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Technical Memorandum No. 33-204

# SPODP-Single Precision Orbit Determination Program

Michael R. Warner Melba W. Nead



JET PROPULSION LABORATORY California Institute of Technology Pasadena, California

February 15, 1965

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## SPODP - Single Precision Orbit Determination Program

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JET PROPULSION LABORATORY
CALIFORNIA INSTITUTE OF TECHNOLOGY
PASADENA, CALIFORNIA
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## JPL TECHNICAL MEMORANDUM NO. 33-204

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#### ABSTRACT

This Technical Memorandum describes the Single Precision Orbit Determination Program which was developed for operation under the data processing system of the Space Flight Operations Facility. Included are sections containing flow diagrams, load maps, the common storage map, input and output descriptions, subroutine writeups, operating instructions, and check cases. The computational methods employed by the program are described in the subroutine documentation.

#### I. INTRODUCTION

The Single Precision Orbit Determination Program (SPODP; more commonly expressed as ODP) for the IBM 7094 computer was written to meet the specifications of the Jet Propulsion Laboratory for a reliable and accurate method of tracking and predicting the motion of lunar and interplanetary spacecraft. The uses of the ODP may conveniently be separated into real-time and nonreal-time applications.

Real-time applications:

- 1. To establish a reliable set of orbital elements for the space-craft.
- 2. To provide an acquisition ephemeris for the world-wide network of tracking stations.
- 3. To assist JPL engineers in evaluating the performance of tracking stations and the quality of tracking data.

Nonreal-time applications:

- 1. To provide a high-speed computing method necessary for orbit determination and tracking data accuracy studies (premission).
- 2. To provide a high-speed computing method necessary for a sophisticated orbital analysis based on large numbers of observations (post-mission).

In addition to solving for the six initial conditions of the spacecraft's motion, the ODP has the capability of solving for 14 physical constants and the Earth radius, latitude, and longitude of 15 tracking stations. From this set of 66 parameters, a subset containing from one to twenty is extracted by the user. The ODP obtains solutions for the parameters in this subset.

The ODP, since it must have initial estimates of the parameters, is an orbit improvement program. It differentially connects the estimates by means of an over-determined system of equations, employing a modified least-squares method. This procedure may be briefly outlined as follows:

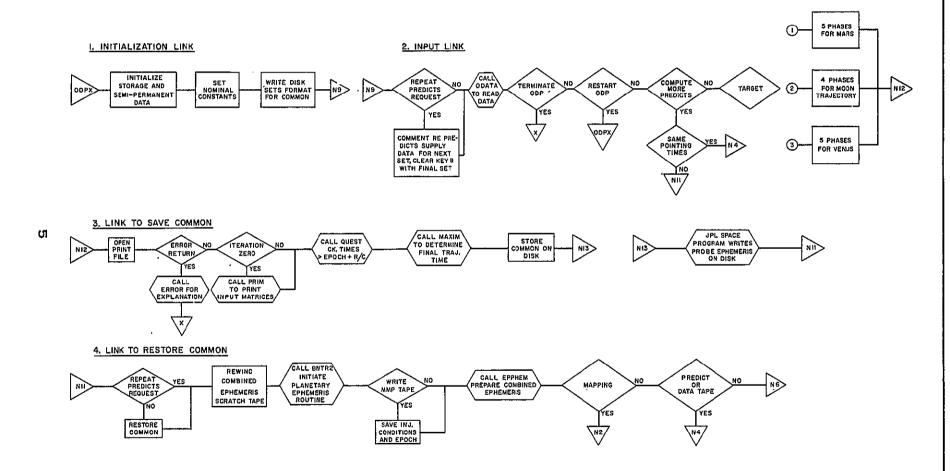
- 1. Input parameter estimates.
- 2. Write probe ephemeris file on disk based on orbit estimate.
- 3. Read i<sup>th</sup> observation from data file, G<sub>i</sub>. This observation may be slant range, range rate, one of four angle types, one of four doppler types, or DSIF ranging.
- Using probe ephemeris, determine the value of the observation based on orbit estimate, F<sub>i</sub>.
- 5. Obtain the residual  $\Delta F_i = G_i F_i$ .
- 6. Calculate the partials of the observations with respect to the n parameters to be estimated,  $\partial F_i/\partial Q_i$ , ...,  $\partial F_i/\partial Q_n$ .
- 7. Multiply the column matrix of partials by itself to form a matrix  $J_i^*$ .
- 8. Add  $J_i^*$  to the accumulated matrix  $J^* = J_1^* + J_2^* + \cdots$ ,  $+ J_{i-1}^*$ .
- 9. Multiply the column matrix of partials by the residual to form a column matrix  $R_{i}$ .
- 10. Add  $R_i$  to the accumulated column matrix  $R = R_1 + R_2 + \cdots + R_{i-1}$ .
- Repeat steps 3 through 10 until all observations are processed.
- 12. Obtain the final J matrix by adding in the <u>a priori</u> uncertainties.  $J = J^* + \tilde{\Gamma}^{-1}$ .
- 13. Solve the normal equations  $J\Delta Q = R$  for the column matrix of changes to the parameter estimates,  $\Delta Q$ .
- 14. Repeat steps 2 through 13 until the procedure converges.

The ODP is written on disk in thirteen links. Each link constitutes a logical section of the overall orbit determination. The linking is accomplished by use of the JPTRAJ source program, which is also used to pass certain information between links.

#### II. FLOW CHARTS

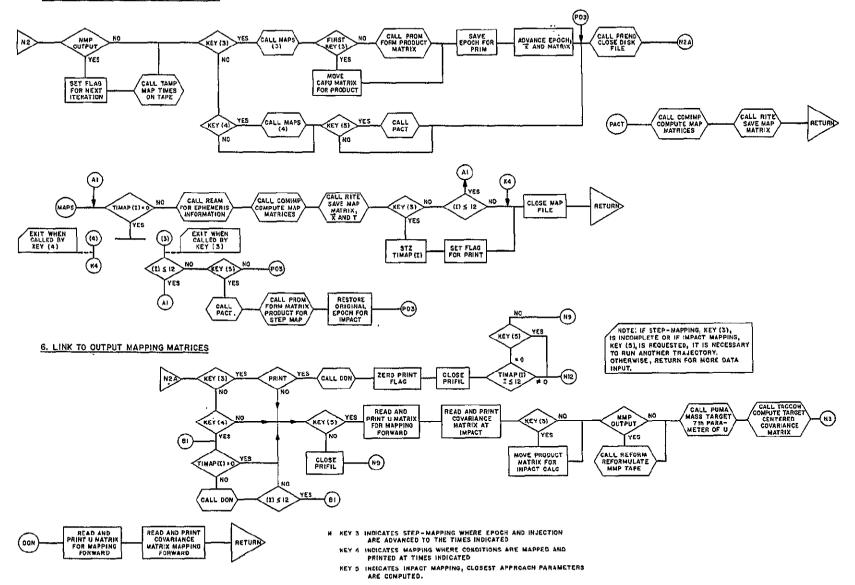
## A. LINKS

- 1. Initialization Link
- 2. Input Link
- 3. Link to Save COMMON
- 4. Link to Restore COMMON
- 5. Link to Compute Mapping Matrices
- 6. Link to Output Mapping Matrices
- 7. Link to Compute Closest Approach Parameters
- 8. Predictions or Data Simulation Link
- 9. Link to Sort Predictions
- 10. Data Fitting Link
- 11. Link to Output Solution
- 12. Link to Output Residuals and Statistics
- 13. Link to Output Auxiliary Quantities

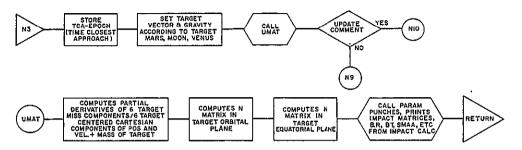


#### 5. LINK TO COMPUTE MAPPING MATRICES

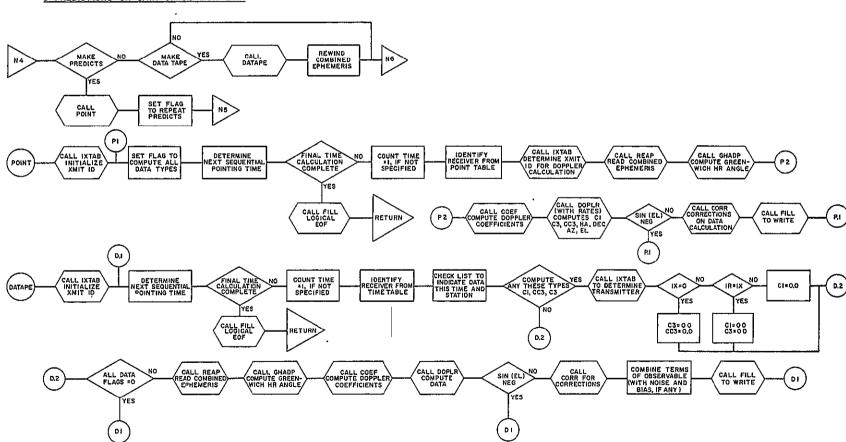
a



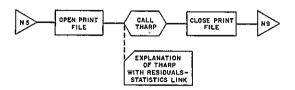
#### 7. LINK TO COMPUTE CLOSEST APPROACH PARAMETERS



#### 8 PREDICTIONS OR DATA SIMULATION LINK

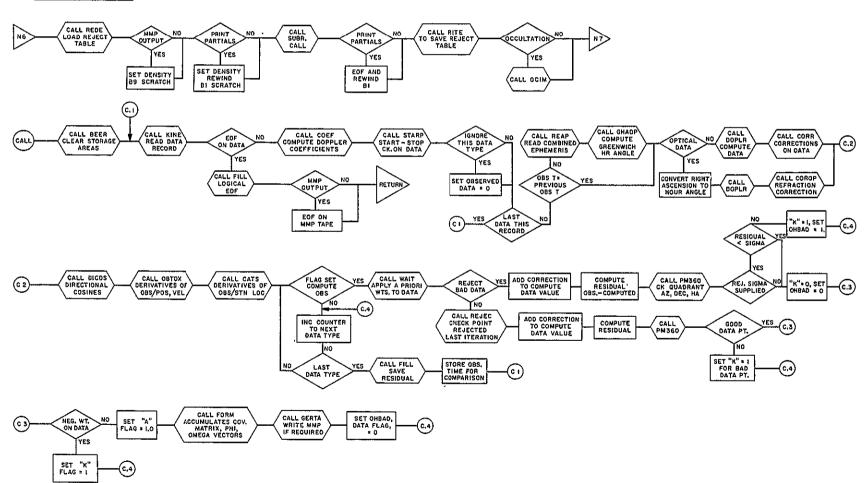


#### 9. LINK TO SORT PREDICTIONS



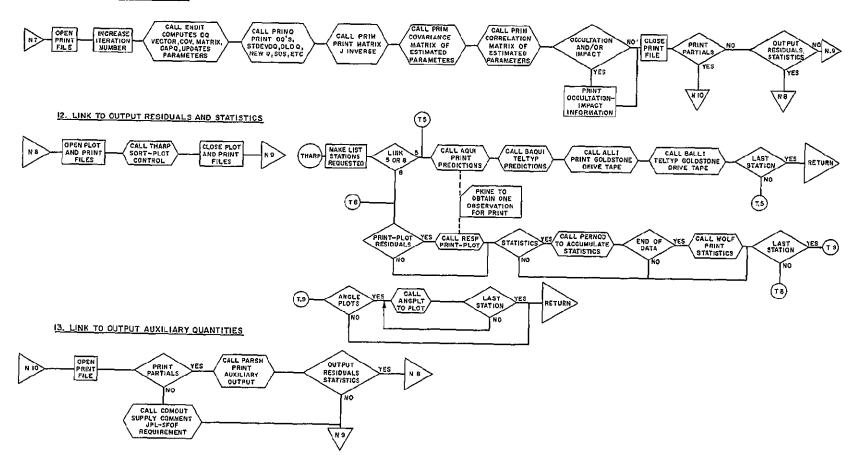
#### 10. DATA FITTING LINK

 $\infty$ 



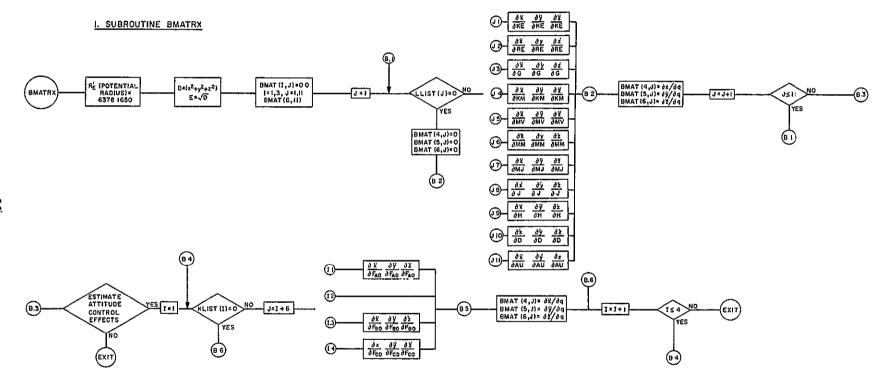
#### II LINK TO OUTPUT SOLUTION

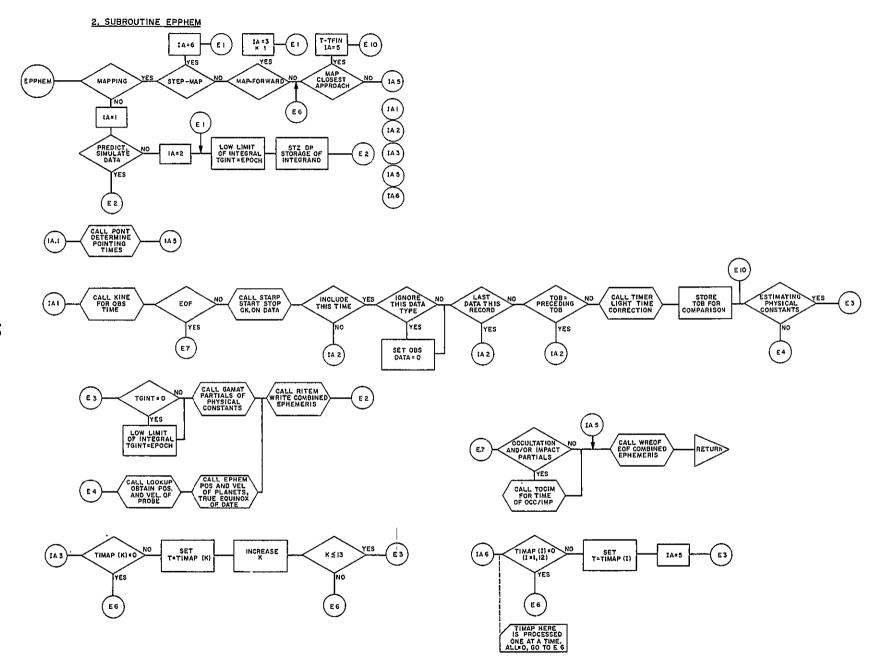
φ



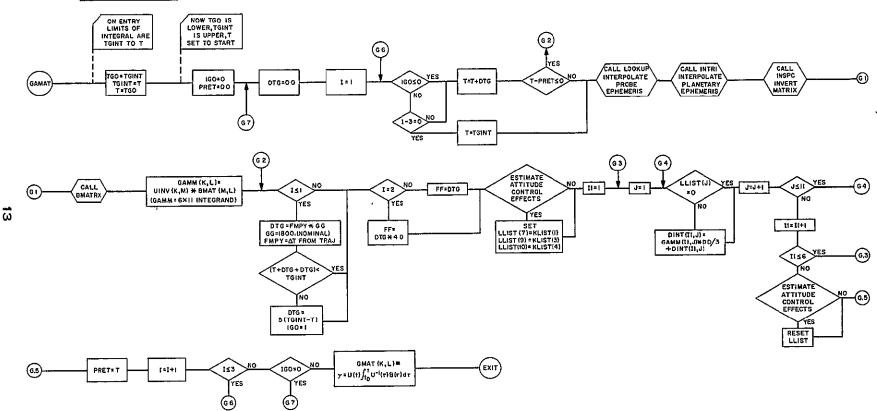
## B. SUBROUTINES

- 1. BMATRX
- 2. EPPHEM
- 3. GAMAT
- 4. LOOKUP
- 5. MAXIM

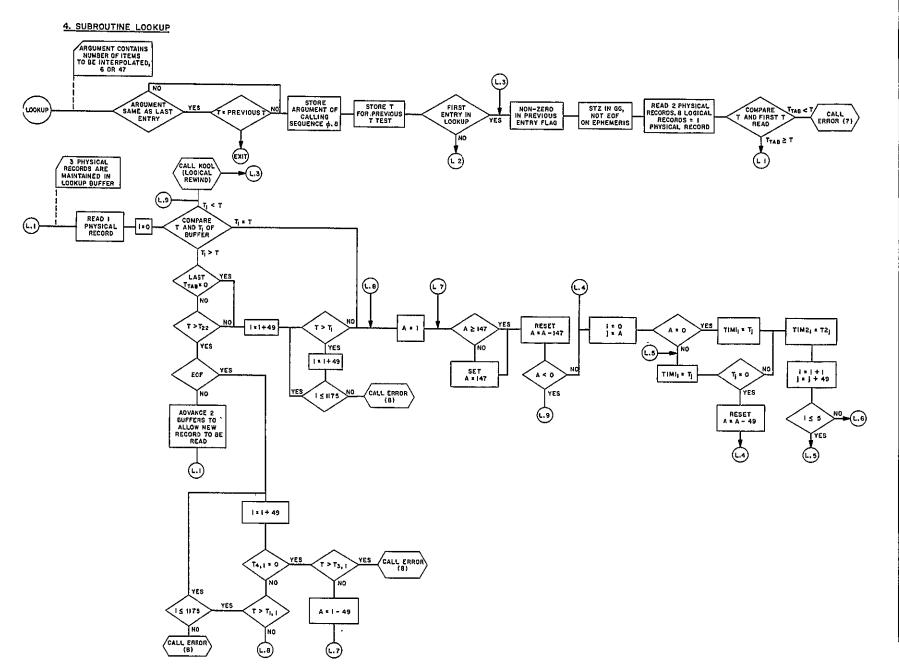




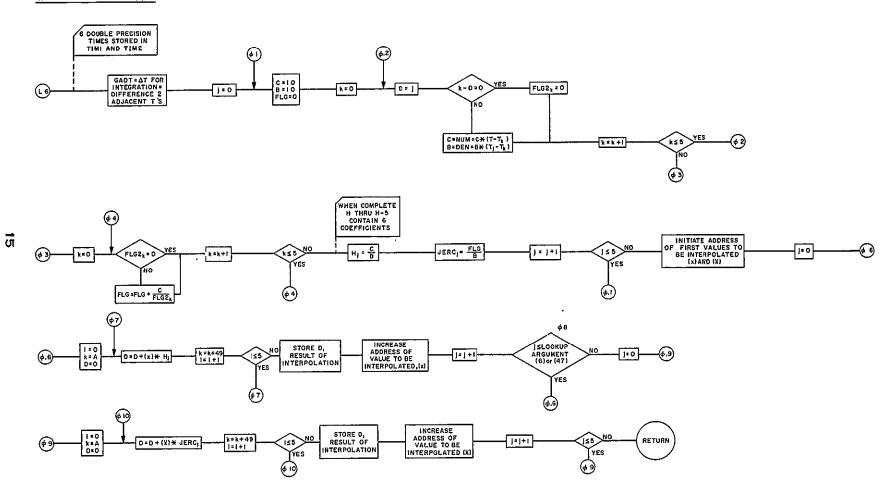


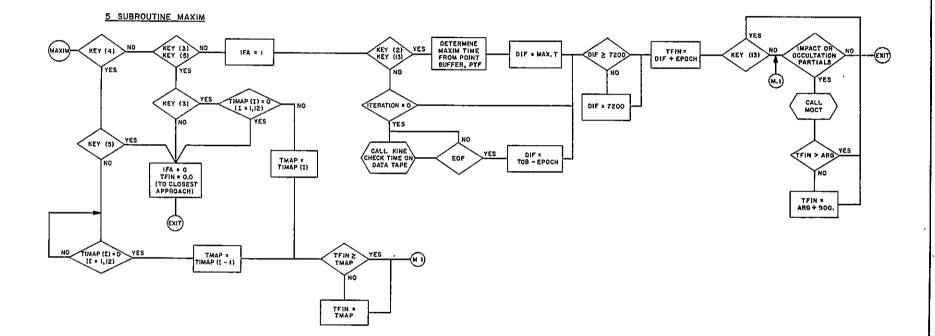






#### 4. SUBROUTINE LOOKUR p 2





## III. PHYSICAL LIMITATIONS

## A. DISK AND TAPE USAGE UNDER SFOF

The ODP uses tape unit B6 for the planetary ephemeris (EPHEM) tape and tape unit A4 for the combined ephemeris scratch tape. Disk allocation is restricted to the 2800 record block named JPLODP.

## B. DEBUGGING FACILITIES

Snaps may be inserted in links ODPX, LA2A, LA3, LA5, LA7, LA8, LA10, and LA12, as these links contain subroutine PROUT.

Octal correctors may be inserted in any link through the JPTRAJ source deck.

## IV. INPUT AND STORAGE ALLOCATION

Primary input to the ODP is a deck of control and data cards, which are described in detail in Section IV. A. These cards contain the parameter estimates, the desired options, and all variables which are not a function of each observation. Subroutine CARDS is used for card input.

Also input to the ODP is a file on disk containing the tracking data, and a planetary ephemeris tape. The data file is written on disk by the Orbit Data Generator Program, ODG, while the ephemeris tape is the standard JPL EPHEM system tape. The system routine DCP is used for disk reading and subroutine TAPIO is used for reading the planetary ephemeris tape.

A. CARD INPUT

```
ЕРОСН
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       (01
                                                                                                                                                              EPOCH ASSOCIATED WITH THE TRAJECTORY, WHERE YY= LAST TWO DIGITS OF THE YEAR MH MONTH OF THE YEAR OPERATED DE DAYS HH MOURS NN= MINUTES SS- SECONDS
                 YYM400DHH.N4SSEFF
                                                                                 GEOCENTRIC POSITION AND VELOCITY AT EPOCH
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       102
              * SYMBOLS ALLOWED X,Y,Z,CG,XG,XG
                                                                                    INPUT DIHER-OPTIONS AND CONSTANTS
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       103
                                                                                                                                                    #ED EXPLANATION AND/BR NOMINAL

K(EARTH) [398603.2KM3/SEC2)
EARTH RADIUS (6378.3149KM)
K(MON9)

MASS RATIO VENUS (2.447118E=6)

MASS RATIO MAND (3.2487420E=7)

MASS RATIO MAND (3.2280420E=7)

MASS RATIO MAND (3.2280420E=7)

MASS RATIO MAND (3.2280420E=7)

MASS RATIO MAND (4.22860E=7)

COSEF. FOURTH HARMONIC (-.975E=5)

COSEF. FOURTH HARMONIC (-.975E=5)

ASTRONOMICAL UNIT (1.49599EKKM)

VELOCITY OF LIGHT (299702-5KM/SEC)

STATION RANGE

STATION LATITUDE (GEOCENTRIC)

STATION LATITUDE (1000CENTRIC)

STATION
            • SYMBOL ALLGAED
                                                                                                                                                                                                                                                                                     EXPLANATION AND/OR NOMINAL
              KÉ
RE
                 C
RI[[]
              13(1)
NSDISP(1) | NEXT | NEXT
                                                                                                                                                                                                                                                                             NORTH-SOUTH DISPLACEMENT, DEG EAST-WEST DISPLACEMENT, DEG THORK OF REFRACTION (140-0) TRANSPONDER DRIFT, CPS/SEC INTITAL TRANSCHURY STEP SIZE, SEC PUMERED FLIGHT UDRAFTION MOTHER COUNT JETON FOR ALC. 19EVITFICATION FOR PREDICTS ORBIT STORED ON DISY IN LINK 6 MATRIX STORED ON DISY IN LINK 6
                                                                                                                                                                                                                                                                        TJ WRITE DISK FOR H/C AND SPACE
L.T.-U.T., SEC (35.)
WEIGHT OF FUEL, IB.
FLOW RATE, LB/SEC
I-RUST, LB
RIAS ANGLE, DEG
IARGET RADIUS, KM (MARS.3400.0)
INDUI FOR RAHUOM NOMBER, GCTAL (1)
MAXIMUM SLAMI RANGE FOR DATA, DEG
MINIMUM LEVATIOL ANDLE FOR DATA, DEG
RESIDUAL PRIMI HIPLACED BY
STANDARD DEVIATIONS
KE IS OITERNINGED BY INC. MELATION
K-M CUBE HOUT IRE. SELTING
RADIATION PRESSURE UPTION (1.031E8)
       DJT
FJELWT
RTFUFL
THRUST
       BRNHU
RS TOP
       RAND
RYAX
       ELMIN
DATA SIGHA BUIPUI
         REM CONSTRAINT
    RADOPT
                                                                                                                                                                                                                                                                                NON-ZERO, RESIDUALS IN SLUATING PL.
  FLRES
                                                                                                                                          PUNCH MAIRIX, 4

1 INPUT J MAIRIX OF ESTIMATO PARAMETERS
2 INPUT COVARIANCE MAIRIX OF CONSIDERED PARAMETERS
3 J INVESTS
4 COVARIANCE MAIRIX OF ESTIMATED PARAMETERS
5 COVARIANCE MAIRIX OF ESTIMATED PARAMETERS
6 COVARIANCE MAIRIX AT IMPACT
7 (ALREADY PUNCHEO)
8 J MAIRIX
9 INPUT COVARIANCE MAIRIX OF ESTIMATED PARAMETERS
10 CORPELATIONS DASED UN J MAIRIX
11 U PRUDUCT MAIRIX
    PANNAS
                                                                                                                                                    TO CHANGE AREA, MASS, GAMMA B OF SPACECRAFT (11.12) (259.00) (.096) (2.789) (340.20)
  ARMARS
HSMARS
GBMARS
ARMININ
HSMUNN
```

```
(3.0)
{3.83)
{198.22)
(.383)
                                    THE FOLLOWING ARE DOPPLER COEFFICIENTS AND FREQUENCIES WHICH ARE A FUNCTION OF THE BAND. NOVINALLY, STATIONS 1,2, AND 3 ARE S BAND. 4 AND 5 ARE L-S BAND.
                                                               L-BAND
                                                                                            L-S BAND
TFREQ TRANSPONDER FREQ
XFREQ TRANSMITTER FREQ
                                    (C1)
(C1)
(CC3)
(CC3)
(C3)
(C3)
                                                               930-15E6 9.375E6 1.0E6

-96875 3125 1.0

-1E6 9.375E6 1.0E6

32.359559561 32.57918552 104.25339367

930.15 9.375E6 1.0E6

31.348314605 32.57918552 104.25339367
D3P1(5)
D3P1(6)
                                    SAME AS ABOVE FOR STATION 2
SAME AS ABOVE FOR STATION 3
SAME AS ABOVE FOR STATION 4
SAME AS ABOVE FOR STATION 5
1115900
INFRQ GROUND STATION INJECTION FREQUENCY, CPS 123.6279142E61
SYNFRQ GROUND STATION SYNFHESIZER FREQUENCY, CPS 122.04209265E61
                ESTIMATE THESE PARAMETERS
                                                                                                                                                                       104
                CONSIDER THESE PARAMETERS
                                                                                                                                                                       105
                THESE CARDS ARE FOLLOWED BY LISTS OF PARAMETER NAMES
WHICH TELL THE OUP TO ESTIMATE (CONSIDER) THE
CORRESPONDING PARAMETER. NO NUMERIC DATA REQUIRED.
LIST OF PARAMETER MANES WHICH CAN BE ESTIMATED
(CONSIDERED).
SYMBBLS ALLOWED
X,Y,Z,DX,DY,DZ,KE,RE,G,KM,MY,MH,MJ,J,H,D,AU,C,RI((),LA((),LO(1)
X:Y:Z:DX.DY.DZ.KE.RE.G.KH.HV.HH.FA.J.FB.FC.AU.C.RI([].LA(]).LD(])
                REJECTION SIGNAS
                THE 480VE-CARO IS FOLLOWED BY DATA INDICATING WHICH OF THE 385EAVABLES ARE TO BE CHECKED FOR POSSIBLE REJECTION OF 8AD POINTS. IF A SYMBOL IS ACCOMPANIED BY A NUMERICAL VALUE THE DBSERVATION WILL BE REJECTED IF THE ABSOLUTE YALVE OF THE RESIDUAL EXCEEDS THE INPUT VALUE.

SYMBOLS ALLOWED
                                     R([],DR([],EL([],AZ([],DEG([],HA([],CI([]),CC3[[]),
C3[[],DI([],RU([])
                                      INVERSE COVARIANCE MATRIX OF ESTIMATED PARAMETERS
COVARIANCE MATRIX OF ESTIMATED PARAMETERS
COVARIANCE MATRIX OF CONSIDERED PARAMETERS
                                                                                                                                                                       (07
(10
(11
                                      PROVISION IS MADE TO ENIER ONE OF THREE MATRICES IN ONE OF THREE WAYS
                                      TO ENTER THE DIAGONAL TERM OF THE MATRIX
 DI AG
                                      TO ENTER THE MAIRIX BY RUN
 R01=
                                    - (CONTINUE ON FOLLOWING CARD, IF NECESSARY)
 R02=
                                      TO ENTER WITH NO SYMBOLS SUPPLY A FORTRAN TYPE ARRAY FOR A 20X20 MATRIX
                  DELETE THESE DATA TYPES
                                                                                                                                                                        (13
                                      SYDBOLS ALLOHED AS IN 106 ABOVE
                                                                                                                                                                         114
                  STATISTICS AND RESIDUALS FOR THESE DATA TYPES
                                    DESIGNATION OF AT LEAST UNE DATA TYPE PER STATION IS NECESSARY FOR THE CALCULATION OF STATISTICS. INCLUDE ALL DATA TYPES FUR WHICH STATISTICS AND RESIDUALS ARE DESIRED. THE VERTICAL PLOTTING SCALE CAN BE DEFINED AS NIGHT. S. THIS LATTER IS NORMALLE. THIS LATTER IS NORMALLE. THIS LATTER IS NORMALLE. THIS LATTER IS NORMALLE. HOUR A MAXIMUM OF 8 DATA TYPES PER STATION MAY BE REQUESTED.

SYMBOLS ALLOWED AS FULLOWS
                                      R(I),DR(I),EL(I),AZ(I),DEC(I),HA(I),CL(L),CC3(I)
C3(I),DL(I),RU(I),NHR
                                                                                                                                                                         115
                   NOMINAL VALUES CORRESPONDING TO COVARIANCE MATRIX
                                     IF A CJVARIANCE MATRIX FOR THE ESTIMATED PARAMETERS OR ITS INVERSE IS ENTERED THE VALUES OF THE ESTIMATED PARAMETERS ESTIMATED PARAMETERS ASSOCIATED WITH MATRIX MAY ALSD BE ENTERED. THESE VALUES FOLLOW THIS CONTROL CARD AND MUST BE IN THE SAME ORDER AS THE PARAMETERS IN THE MATRIX.

IF NJMINAL VALUES FOR THE PARAMETERS ARE NOT ENTERED THE PAGGRAW WILL SET THE NDMINAL VALUES TO THE PATRIAL ESTIMATE OF THE PARAMETERS. NO SYMBOLS ARE ALLOWED.
```

```
WEIGHTS BY DATA TYPE AND STATION
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            (16
                                                                                                                MONIVALLY THE ODP COMPUTES THE WEIGHT FOR EACH DATA TYPE FROM INFORMATION THE ODG PUTS ON THE TRACKING DATA FILE. THE DEFRATOR CAN OVERRIDE THIS FEATURE BY FOLLDWING THE ABOVE CARD MITH WEIGHTS (SIGHA RATHER THAM SIGHA SOJARBE) FOR AMY OR ALL OF THE DATA TYPES-WEIGHTS MUST BE SPECIFIED IN THIS MANNER WHEN FITTING AN DOP-SIMULATED DATA FILE. THE USER MAY 4LSO IMPUT OCTAL WEIGHT COOES, E.G., CC3121-316333/8
SYMBJLS ALLOWED
                                                                                                                    R([],DR([],EL([],AZ([],DEC([),HA([),C1([],CC3([)
C3([),D1([],RU([)
POINTING TIMES, SAMPLE RATE, COUNT TIMES

AS THE NAME SUGGESTS, DATA FOLLOWING THIS CONTROL
CARD DEFINES THE ANOUNT OF DUTPUT OBTAINED WHEN THE
PROCEAM IS REPRENIVE A DATA FILE OR COMPUTING POINT-
INC PREDICTIONS. A SYNDOLIC NAME IS ASSOCIATED MITH
EACH OF THE 15 POSSIBLE TRACKING STATIONS, AND EACH
MAME MAY BE FOLLOWED BY DNE, TWO, OR THREE GROUPS OF
DATA. THE FORMAT OF EACH GROUP IS AS FOLLOWS, FIRST
IME, LAST TIME, SAMPLE MATE, COUNT TIME. POINTING
PREDICTIONS (TAR A DATA FILE WILL BE GENERATED FOR
HIE HYDICATED STATION IF THE PROBE IS ABOVE STATION'S
HORIZON ANYTHE BETWEEN THE FIRST THE AND THE LAST
IME. THE HYTERVAL BETWEEN THE FIRST THE AND THE LAST
IME SAMPLE MATE AND MAY ROUSSIED DOPPLER CAL-
CULATIONS MILL BE BASED ON THE INDICATED COUNT TIME.
HE MID THES SAME GREEWLICH TIME AND ARE COMPOSED OF
HOU JORDS EACH, AS DEFINED ELSEMBER. THE SAMPLE
HATE AND THE TOUST TIME ARE GIVEN AS FLOATING POINT
SECOVOS. AS AN EXAMPLE,
HATE AND THE TOUST TIME ARE GIVEN AS FLOATING POINT
SECOVOS. AS AN EXAMPLE,
HATE AND THE SOUTH THE AND SOUTH OF ACCORDING POINT
SECOVOS. AS AN EXAMPLE,
HATE AND THE SOUTH THE AS FLOATING POINT
SECOVOS. AS AN EXAMPLE,
HATE AND THE SOUTH THE AS SOUTH AND ACCORDING POINT
SECOVOS. AS AN EXAMPLE,
HATE AND THE SOUTH THE AS SOUTH AND ACCORDING POINT
SECOVOS. AS AN EXAMPLE,
POINTING PREDICTIONS WILL ALSO BE GENERATED FOR
HE MIS EVERT TO SEC BETWEEN 0300001 AND 0400002 UN JULY
21, 1962 MITH A DOPPLER AVERAGING TIME UP S SEC.
POINTING PREDICTIONS WILL ALSO BE GENERATED FOR
OTDODOOT TO OGGODOOD ON THE SAME DAY SUT WITH A 30-SEC
SYMBOLS ALLOWED
SY
                                                          POINTING TIMES, SAMPLE RATE, COUNT TIMES
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            (17
                                                                                                                    JEIGLY, JETGLZ, CANJET, DUNJET, JUBSJET, ANIGUA, ASCENS, PUERTO, PRETOR, JOHNEL, TGAPE, SANSAL, HANAIL, TRINID
                                                          DATA FILE SIGNA
                                                                                                                    LF A DATA FILE IS BEING PREPARED THE DATA TYPES FOLLOWING THIS CARD WILL BE ENTERED UN THE DOP DATA FILE. IF NUMERIC FIELD APPEARS WITH A SYMBUL THE PROGRAM MILL PUT A NUMBRIALLY DISTRIBUTED RANDOM NUMBER WITH MEAN ZERGY ON THE CALCULATED VALUE OF THE DATA
                                                                                                                  TYPE. THE UNE-SIGNA VALUE OF THE DISTRIBUTION IS SET TO THE VALUE FOLLOWING THE SYMBOLIC MAME OF THE DATA TYPE. IF NO WUMERIC FIELD FOLLOWS A SYMBOL THE UNE-SIGNA VALUE OF THE DISTRIBUTION IS SET TO ZERO. SYMBOLS ALLOWED
                                                                                                                    R(1),DR(1),EL(1),AZ(1),DEC(1),HA(1),CI(1),CC3(1)
C3(1),DI(1),RU(1)
                                                          DATA FILE BIAS
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            122
                                                                                                                    SAME AS THE ABOVE EXCEPT THAT THE NUMERIC FIELD INDICATES THE CONSTANT THE ODP WILL AND TO THE CALCULATED VALUE OF EACH DATA TYPE.
                                                          PASE HEADING
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             (23
                                                                                                                    UP TO 11 BCD WORDS MAY FOLLOW THIS CONTROL CARD.
THE COMMENT MUST BE INITIATED WITH A LEFT PARENT-
THESIS AND CLOSED WITH RIGHT PARENTHESIS. THE
COMMENT WILL BE PRINTED ABOVE VARIOUS OUTPUT GROUPS.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             124
                                                          MAP COVARIANCE MATRIX TO
                                                                                                                INIS CARO IS FOLLOWED BY UP TO 12 MONOTOVICALLY INCREASING GREEWHICH TIMES. THE COVARIANCE MATRIX FROM THE PREVIOUS ITERATION WILL BE MAPPED FORWARD TO THESE TIMES.
SYMBLIS ALLOWED, NONE
                                                          TRANSMITTER ID TABLE
                                                                                                            THE DATA FOLLOWING THIS CONTROL CARD DEFINES THE TRANSMITTING STATION AS A FUNCTION OF TIME. THIS DATA IS REQUIRED ONLY MHEN GENERATING DATA FILES OR POINTING PREDECTIONS SINCE THE DOG IDENTIFIES THE TRANSMITTER FOR EACH TIME POINT APPEARING ON THE ODD DATA FILE. THE FORMAT OF THE TRANSMITTER TABLE IS, IXII, FREQII, TIME(I), IXI2, FREQIZ)...... IX IS A FIXED NUMBER, FREQ IS A FLOATING POINT NUMBER, TREO IS A FLOATING POINT NUMBER THE PREC IS A STATION TXII) IS TRANSMITTING WITH A FREQUIT UNTIL THE THE THE TABLE IS AS FOLLOWS, STATION TXII) IS TRANSMITTING WITH A FREQUIT UNTIL THE TABLE TO THE TABLE TABLE TO THE TABLE TABLE TO THE TABLE TO THE TABLE TABLE TABLE TABLE TO THE TABLE TABLE
                                                                                                                1. IF ELEVATION AT STATION IX IS NEGATIVE SET IX=0
2. IF IX=0 ONIT CC3 AND C3
3.IF IX=IR+0 ONIT C1 AND CC3 WHERE IR IS*THE RECEIVER
```

```
4. IF IX=IR GAIT CL AND C3
SYMBOLS ALLOWED, NONE
                             RESTRICTIONS
A MAXIMUM OF 134 IX-TIME PAIRS IS ALLOHED
              OFFLINE CONTROL
                                                                                                                                   127
BURN START TIME
                                                                                                                                    (30
                             THIS TIME, IN THE USUAL GREENWICH FORMAT, DENOTES THE BEGINNING OF THE POWERED FLIGHT.
              SPHERICAL INJECTION CONDITIONS
                             IN PLACE OF CARTESIAN INJECTION CONDITIONS, THE USER MAY JAPUT SPHERICAL CONDITIONS UNDER THIS CONTROL CARD. THE SPHERICAL CONDITIONS ARE TRANSFORMED TO CARTESIAN CONDITIONS AT JAPUT TIME.
SYMBOLS ALLOWED
                             RAD, LAT, LONG, VE, ELE, AZE
              OCCULTATION TIMES
                             THIS DATA TYPE IS INPUT BY CARD, RATHER THAN THROUGH THE DOS. TWO OCCULTATION TIMES APPEAR ON EACH CARD. THE FORMAT IS OCCX-FIT,F2_WEIGHT,REJECTION SIGHA, WHERE II AND T2 ARE GREENMICH TIMES IN THE USUAL FORMAT. RESTRICTIONS
ONLY STATIONS I THROUGH 5 PERMITTED. LF DNLY DNE TIME IS TO BE USED THEN T2 MUST BE 0,0.
                             0001-0002-0003-0004-0005
                              SAME AS DECULTATION TIME, EXCEPT THAT UNLY ONE TIME
PER STATION IS INPUT. THE FORMAT IS IMPX=T,WEIGHT,
REJECTION SIGNA.
                              RESIRELTIONS
ONLY STATIONS & THROUGH 5 PERMITTED
SYMBOLS ALLOWED
                              IMP1, IMPZ, IMP3, IMP4, IMP5
               START AND STOP TIMES
                              THIS INPUT PERMITS THE FITTING OF DATA ONLY DURING THE INTERVAL BETWEEN T-START AND T-STOP. UP TO THREE PAIRS MAY BE INPUT IN THE USUAL GHT FORMAT.
                                                                                                                                     too
               END DATA
                              THIS CARD MUST TERMINATE EACH SET OF CONTROL DATA READ BY THE ODP.
```

B. LOAD MAPS AND COMMON STORAGE MAP

THE HAME O	OF THIS PROGR	AH IS 'LA2	2/22/65			
ITRY NAME	ENTRY ADD. 22302	TRANSFER VECTORS	LOAD ADD. 22300	DCTAL LENGTH 00011	DECIMAL LENGTH	COMMON BR
		COMAP				
COKAP	22322	OFFSYS TYPRYT	22311	00273	00187	
		TAMP PROM				
		FLOT				
		REAM COMIMP				
		DCP				
PRANS	22606	FLAX DCP	22604	00043	00035	
=		FLAX		*****		
CONTRO	22660	PRAMS (DEMP)	22647	01357	00751	47055
		(DFAD)				
		HUKAK (B270)				
ADB	24226	**************************************	24226	00050	00040	
ARS EN ARCOS	24306 24302	*NO4E*	24276	00144	00100	77152
QARSIN	24300					
QARCOS	24276					
ERRARS ERRARC	24427 24433					
ARTAN	24442	*ROYE*	24442	00103	00067	77152
DAYS	24550	FIX FLOAT	24545	00041	00033	
		ADD				
ERROR FIXT	24606 24620	*NBYE* FIXII	24606 24617	00011 00007	00009	
FIXTT	24626	NOVE	24626	00312	00202	50031
FLOT	24734 25115					
FLOAT	25121					
DAYE	25131					
FLAT FLAPR	25142 25170	ERROR Typryt	25140	00102	00066	
FLAK	25177					
JUKAN JUKAN	25246 25556	"NOYE"	25242 25552	00310 00130	00200 00088	50031 77461
PNUT	25710	\$1N	25702	00336	00222	50001
PRON	26244	COS *NOVE*	26240	00156	00110	77461
QUIZ	26416	.NGAE.	26416	00003	00003	11401
REAP REAM	26424 26424	RÉADB Flat	26421	02543	01379	
		ERRDR				
ROT	31170	'NDYE'	31164	00266	00182	50031
OFFSYS ENDSYS	31452 31453	*ND4E*	31452	00010	80000	
FINSYS	31454					
SCHAIN PGSTRT	31455 31456					
PGSTOP	31457					
DCP TYPRYT	31460 31461					
SIN	31462	*ND4E*	31462	00242	00162	
QSIN QSIN	31465 31471					
0005	31473					
TANP	31736	WRITEB	31724	00126	00086	
EPOX (DFAD)	32032 32052	FLAT 'NDYE'	32052	00121	00081	77776
(DESB)	32072					
(DFMP) (DFDP)	32112 32140					
READO	32174	(100)	32173	00365	00245	
READS WRITED	32176 32201					
WRITEB	32203					
BSREC	32412					
B\$FILE REWIND	32415 32420					
UNLOAD	32423					
ENDFIL SETLOW	32426 32404					
SETHE	32407					
(TAPETO	32555 32555					
[[00]	32563	'NONE'	32560	00030	00024	

UNUSED CORE LIES FROM 32610 THROUGH 47055, LEAVING 14246 OCTAL OR 06310 DECIMAL LOCATIONS.

ENTRY POI DEFI ENDF	NTS TO SUBROUTINES NE ATTACH IL REWIND	OPEN CLI	LIBRARY, DSE (DFKP) FSB)	(DFAD)	SQRI FEDDUT	PAQUI	WRITES
THE	NAME OF THIS PROC	RAH IS "LAZA "	2/22/65				
ENTRY **P *APO	CB 22302	TRANSFER VECTOR MAPOUT DEFINE ATTACH OPEN TYPAYT OFFSYS CLOSE DCP FLAK PRAM PRIM REFORM TACCOR	SS LDAD ADD. 22301 22310	OCTAL LENGIH 00007 01564	DECIMAL LENGTH 03007 03884	COMMON BREAK	
AAT	24102	(DFMP)	24074	00116	99076	7746I	
8180		-NOVE+	24212	00055	00045 00069	C++>+	
DIAG ERRO	R 24374	SORT "HONE"	24267 24374	00105 00011	60000	50031	
FIXT	T 24414	+H04E+	24405 24414	00007 03312	00007 002 <b>0</b> 2	50031	
FLOT FIX	24703						
FLQA DAYE	24717						
FLAT Ft ap:		ERROR TypryT	24726	00101	00065		
ARGK RRGK	24765 AY 25034	TRDZ	25027	00252	92149	77461	
PRIH PRAH	25321 25310	FGD99F PRUV1 818CD FLAPR PRIME NORMAY PRIT DJAGO	25300	01423	00787	50001	
PR [AI AEFO	E 26730 RM 26751	WRITES FLAT ENDFIL REXIND	26723	00177	00127		
PRIT	27124	READS Fixt	27122	00036	00030		
SIND	27160	, KOAE, B18CD	27160	00002	00002		
OFFST ENDST FINST SCHAT POSTE POSTE DCP TYPRT	rS 27163 rs 27164 lM 27165 RT 27166 pp 27167 27170 rt 27171	*NONE*	27162	00010	40048		
TAPR SCALE TACCI	27200	"NOYE" 'NOYE" 'NOYE" 'CALE (DFMP) (DFAD) AAT (DFS8)	27172 27174 27237	00002 00043 01617	00911 00911	77461 47055	
IDFA	31056	, KDAE,	31056	00120	00000	17776	
10F51 (0FM) 10F01	21 31116						
SORT PAGUI FGDOI PACHI PROVI PROVI FSXA ISXB	31176 7 31266 JT 32532 7 31433 12 31404	VNOYE' OUTUS ACTIND BFLG RESTKA REGIND WRITE PROON RGGSAY	31176 31252	00054 03031	00044 01561	17773	
READE	34304	RĞĞİTR EKIND EKACT EIDU}	34303	00365	00245		
READE WRITE WRITE BSREC BSFIL REMIN UNLOA ENDFI SETVI CUNIT TAPE (1001	D 34311 8 34313 34522 E 34525 D 34530 D 34530 L 34536 M 34514 34517 J 34665	**O <b>YE</b> *	24/20	00030			
OUTUS BFLG ENDBU CKIND CKACT REGIN ACTIN ACTIN AESTK PLICO PLICO PRECON	36722 35556 37557 37151 37151 37150 U 37150 D 37147 A 37145 N 37230 N 37231 N 37232	RGGSAY RGGSYR	34670 34720	02320	0024 01232		
RGUSA RGGST 19CS OEFIN JDIN ATTAC CLOSE DEEN READ WAITE GGPY TREW WEF BSF	V 37240 R 37322 37374 E 37375 37400 H 37403 37406 37411	CLUCK	37240 37373	00133 04335	00191 02269		

'LAZA ' JUST LOADED.

UNUSED CORE LIES FROM 44036 THROUGH 47055, LEAVING 03020 OCTAL OR 01552 DECIMAL LOCATIONS.

ENTRY POINTS TO DEFINE	D SUBROUTINES ATTACH	REQUESTED FROM LIBR OPEN CLOSE	MARY, WRITEB	ENDFIL	FGDOUT SQRT	PROUT	LOG
THE NAME	OF THIS PROGR	* EAJ' 21 HAR	2/22/65				
ENTRY NAME **PCB OZ	ENTRY ADD. 22302 22316	TRANSFER VECTORS OZ DEFINE ATTACH OPEN I YPRYT OFFSYS UMAT CLOSE	LOAD ADD. 22301 22307	OCTAL LENGTH DOODS 01367	DECIMAL LENGT 00006 00759	H COMMON BREAK	
ARTAN	23676	*NOVE *	23676	00103	00067	77152	
FIXT FIXTT FLOT FIX FLOAT DAYE	24002 24010 24116 24277 24303 24313	* 11/17 * 11/17	24001 24010	00007 00312	00007 00202	50031	
ERROR Flat Flapr	24322 24336 24364	*NOYE* ERROR TYPRYT	24322 24334	00012 00101	00010 00065		
FLAK Gerpu	24373 24440	WRITES	24435	00100	00064		
ORBEQ	24544	FLAT ENDFIL ARTAN COS	24535	00167	00119	77461	
PARAM	24732	SIN FGBOUT SORT FIXT PROUT INORM	24724	01320	00720		
QUIZ OFFSYS ENDSYS FINSYS SCHAIN PGSTRT	26244 26246 26247 26250 26251 26252	ORBEO 'NDVE* 'NOVE*	26244 26246	00002 00010	00002 00008		
PGSTOP DCP Typryt Sin	26253 26254 26255 26256	**************************************	26256	00243	00163		
COS QSIN QCOS	26261 26265 26267		24274	002.43	******		
ÎNDRH UHAT	26526 27004	SQRT SQRT LOG Gerpu Artan Param	26521 26773	00252 03221	00170 01681	77461 77 <u>4</u> 61	
L06 L0G10	32222 32214	PACHE!	32214	24000	00054	77774	
SURT PROUT FGODUT PRCAY PROUT2 PROUT3 ISXA ISXA	32302 32372 33636 32537 32510 32523 34545 35335	MOUE* DUTMS ACTIND BFLG RESIKA REGIND WARIE PACON ROGSAY ROGSTR CKIND CKACT	32302 32356	00054 03031	00044 0156 <u>1</u>	77773	
READD READS WRITED WRITED SREC BSFILE REWIND UNLOAD ENDFIL SETLOW SETHI (UNIT) TAPEIO	35410 35412 35415 35417 35526 35631 35634 35637 35642 35620 35623 35771	1100)	35407	00365	00245		
DUTUS BFLC ENDOUT CKIND CKACT REGIND ACTIND RESTKA PLICON PLEGON PROON	35777 36026 36662 36613 40255 40320 40253 40251 40334 40335 40336 40337	* NGUSE* RGGSAY RGGSTR	35774 36024	00030 02320	00024 01232		
AGGSAV AGGSTA IOCS DEFINE JOIN ATTACH	40344 40426 40500 40501 40504 40507	CFOCK .WOME.	40344 40477	00133 04335	02269		

UNUSED	CORE LIES FR	M 45142	THROUGH	50031.	LEAVING	02670	OCTAL OR	01464	DECIMAL LOCATIONS.
*LA3 *	JUST LOADED.								
XHEN	45034								
MINUTE	45034								
CLOCK	45037	•	NONE .	45034		00106		00070	
STASH	40545								
BSF	40542								
BSR	40537								
WEF	40534								
REW	40531								
CDPY	40526								
WRITE	40523								
READ	40520								
OPEN	40515								
CLOSE	40512								

ENTRY POINTS TO	D SUBROUTINES (DFAD)		LIBRARY. DFS8) SQRT	DHOD	(DFOP) EXP(3	READB
THE NAME	OF THIS PROGR	RAH IS "LA4"	2/22/65			
EHTRY NAME **PCS PREDA	ENTRY ADD. 22302 22314	TRANSFER VECT PREDA TYPRYT POINT DATAPE REWIND	70RS LOAD ADD. 22301 22310	OCTAL LENGTH OCOOT GOOSO	DECIMAL LENGTH 00007 00040	CONMON BREAK
ABDD ADD ARSIN ARCDS GARSIN GARCOS ERRARS	22360 22410 22470 22464 22462 22460 22611	**************************************	22360 22410 22460	00030 00050 00144	00024 00040 00100	77152
EARARC ARTAN COEF	22615 22624 22736	'NQYE' (DFAD) (DFAP) (DFSB)	22624 22726	00102 00770	00066 00504	77152 50031
CORR	23724	SQRT	23716	00274	88100	47675
DATAPE	24230	SIN IXTAB OFFSYS OFFSO READ GHADP COEF BUPLA CORR ND2F F11T	24212	00447	00295	50031
DAYS	24564	FIX FLOAT ADD	24661	00041	00033	
DNAME SIPREG ODOFF SPHX	24735 24736 24737 24740	NONE.	24722 24736	00014 00003	00012 00003	
DGPLR	24756	RATES DHOD VEC ABDD (DFMP) (DFAD) SQRT ARTAH ARSIN	24741	01313	00715	50031
ERROR FILT FILT FIXT FIXTI FLOT FLOAT	26254 26270 26552 26566 26574 26702 27063 27067	*NONE* RITE PISA FIXTT *NONE*	26254 26266 26565 26574	00012 00277 00007 00312	00010 00191 00007 00202	50001 50031
DAYE Flat Flapr Flak	27077 27110 27136 27145	ERROR Typryj	27106	00102	88000	
GHADP	27222	(DFOP) (DFSB) (DFMP) (DFAD) DMDD COS	27210	0032 <b>0</b>	09208	50031
IXTAB NO2F PNUI	27530 27616 27674	'NONE' 'NOYE'	27530 27616	00066 00050 00337	00054 00040 00223	50001
POINT	30242	COS IXTAB COEF OFFSYS (DFAD) REAP GHADP DOPLR CORR	27666 30225	00305	00197	50031
QUIZ RATES	30532 30546	FILL 'NOVE' COS SIN SQRT ARSIN EXPL3	30532 30535	00003 00662	00003 00434	50031
REAP " REAM	31422 31422	READB FLAT ERROR	31417	02543	01379	
ROT OFFSYS ENOSYS FINSYS SCHAIM	34166 34450 34451 34452 34453	*NOVE*	34162 34450	00266 00010	00000	50031

PGSTRT	34454					
PGSTOP	34455					
DCP	34456					
TYPRYT	34457					
SIN	34460	"NONE"	34460	00242	00162	
COS	34463					
QS [N	34467					
QCOS	34471					
VEC	34722	'AD4E'	34722	00017	00015	50031
REDE	34742	DCP	34741	00421	00273	
RITE	34745					
PISA	35171					
EXP(3	35362	'AD4E'	35362	00136	00094	77773
(OFAD)	35520	'NOYE'	35520	00120	00080	77776
(OFSB)	35548					
[DFMP]	35560					
(OFOP)	35606					
OHOD	35640	*NONE*	35640	00060	00048	77776
SORT	35720	NOVE	35720	00055	00045	77773
READO	35776	(100)	35775	00365	00245	
READB	36000					
HRITED	36003					
WRITEB	36005					
BSREC	36214					
BSFILE	36217					
REWIND	36222					
UNLOAD	36225					
ENDFIL	36230					
SETLOW	36206					
SETHI	36211					
CUNITI	36357					
TAPEID	36357					
(100)	36365	*ND4E*	36362	00030	00024	
*LA4 * JU	ST LOADED.					

UNUSED CORE LIES FROM 36412 THROUGH 47675, LEAVING 11264 OCTAL OR 04788 DECIMAL LOCATIONS.

DEFINE	ATTACH	OPEN	CLOSE PRO	UT FGDDUT	TELTYP	ENDOUT	
THE NAME	OF THIS PAGE	RAH IS 'LAS	2/22/65				
ENTRY NAME	ENTRY ADD.	TRAVSFER VEC				AL LENGTH	COMMON BREA
* *PCB .	, 22302	SDRT	22301	00006		0006	
SURT	22316	TYPRYT		02055	0	1069	
		DEFINE					
		ATTACH					
		OPEN IHARP					
		CLOSE					
		FLAK					
BIBCO	24611	'ADYE*	24364	00054	o	0044	
DNAME	24453	*370K*	24440	00014	ō	0012	
AQU1	24512	PROUT	24454	00741	ō	0481	50031
ALLI	24472	FGDOUT					
BALLI	24467	PKIYE					
BAQUI	24464	FIXT					
		DAYE					
		TELTYP					
		ENDOUT					
950000	251.21	81850	25415	00007	•	0007	
PERNOD WBLF	25421 25420	KINE	27417	00001	v	0001	
RESP	25420						
ANGPLI	25417						
HONRED	25424	• NONE •	25424	00002	0	0002	
ERROR	25426	'NONE'		00012		0010	
FILL	25442	RITE	25440	00277	Ó	0191	50001
FILT	25724	PISA					
FIXT	25740	FIXTT	25737	00007	a	0007	
FIXIT	25746	*ADVE	25746	00312	0	0202	50031
FLOT	26054						
FIX	26235						
FLOAT	26241						
DAYE	26251					0066	
FLAT Flapr	26262 26310	EARDA TYPRYT	26260	00102	v	nneo	
FLAK	26317	ITPKT					
KINE	26364	REDE	26362	00216	n	0142	50001
MINC	20304	ERROR	20302	00210	٠		,,,,,,,,
QUIZ	26600	NOVE	26600	00002	e	0002	
OFFSYS	26502	'NDNE'		00010		0008	
ENDSYS	26603						
FINSYS	26504						
SCHAIN	26505						
PGSTRT	26506						
PGSTOP	26507						
DCP	597.0						
TYPRYT	26511				_		
THARP	26526	OFFSYS	26612	00573		0379	
PKINE	26733	XINE					
		AQU1 ALL1					
		BAQUI					
		BALLI					
		PERNOS	1				
		RESP					
		WOLF					

		ANGPLT BIBCD			
		TYPRYT			
REDE	27406	DCP	27405	00422	00274
RITE	27411	-	21405	00422	00214
PISA	27635				
TELTYP	30036	PROUT2	30027	01101	00577
SCCTTY	30253	ACTIND			
		PROUT3			
		TSXA			
		RGG\$AV			
		RGGSTR			
_		T5X8			
PROUT	31144	OUTUS	31130	03032	01562
FGDBUT	32410	ACTIND			
PRCMV	31311	8FLG			
PROUT2 Prout3	31262 31275	RESTXA REGIND			
TSXA	33317	WRITE			
TSXB	34107	PROON			
	,	RGGSAV			
		RGGSTR			
		CKIND			
		CKACT			
DUTUS	34164	REGSAV	34162	02320	01232
BFLG	35020	RGGSTR			
ENDOUT	34551				
CKIND	36413				
CKACT REGIND	36456 36412				
ACTIND	36411				
RESTKA	36407				
PLICON	36472				
PL2CON	36473				
PLECON	36474				
PRCON	36475				
RGGSAV	36502	.HDAE.	36502	00133	00091
RGGSTR	36564				
1002	36636	CLOCK	36635	04335	02269
DEFINE JDIN	36637 36642				
ATTACH	36645				
CLOSE	36650				
OPEN	36653				
READ	36656				
WRITE	36661				
COPY	36664				
REW	36667				
WEF	36672				
BSR	36675				
BSF Stash	36700 36703				
CLUCK	36703 43175	"NBYE"	43172	00106	00070
MINUTE	43172	1015	73172	00108	50070
XHIN	43172				
	UST LOADED.				

UNUSED CORE LIES FROM 43300 THROUGH 50001, LEAVING 04502 OCTAL OR 02370 DECIMAL LOCATIONS.

SETHI WRITEB	REWIND READS	ENDFIL (D	FSB) (DFAD)	(DFMP)	SQRT EX	EP (3 0HDD	(DF
THE NAME	OF THIS PROG	RAH IS 'LAG '	2/22/65				
ENTRY NAME **PCB FIT	ENTRY ADD. 22302 22320	TRANSFER VECTOR FIT REDE IYPRYI SETHI REWIND CALL ENDFIL RITE OCIM FLAK	22301 22307	OCTAL LENGTH ODOOG OO735	DECIMAL LEM DOODS 00477	IGTH COMMON BREJ	iK
ABDD ADD ARSIN ARCOS QARSIN QARCOS ERRARS	23244 23274 23354 23350 23346 23344 23475	. WDAE. . WDAE.	23244 23274 23344	00030 00050 00144	00024 00040 00100	77152	
ERRARC ABTAH BEER CALL	23510 23510 23614 23736	HOVE* REJEC BEER KINE COEF STARP IDFSBJ REAP GHADP DOPLR COROP CORR DICOS OBTOX CATS OFFSYS WAII QUIZ REJEC PM360 FORM GERTA FILL	23510 23613 23704	00103 00071 00523	00067 00057 00339	77152 47675 51235	
ÇATS	24436	GREOF VEC ABOD COS	24427	00645	00421	47675	
COEF	25304	(DFAD) (OFAP) (DFSB)	25274	00770	00504	\$0031	
COL	26272	OUIZ COS SQRT	26264	00406	00262	50031	
COROP	26702	SURT	26672	00170	00120	50031	

	CORE LIES FROM	45150 THROUGH	46711.	LEAVING	01542	OCTAL OR	00866	DECIMAL LOCATIONS.
TAPEIO (100)	45115 45123 JUST EDADED.	* 4045	45120		00030		00024	
UNLOAD ENDFIL SETLOW SETHI (UNIT)	44763 44766 44744 44747 45115							
BSREC BSF1LE RENIND	44752 44755 44760							
READO READS WRITED WRITES	44534 44536 44541 44543	(100)	44533		00365		00247	
(DFDP) DMOD SQRT	44344 44376 44456	'NDVE' 'NOVE'	44376 44456		00060 00055		00048 00045 00245	11776 77773
(DFAD) (DFSB) (DFMP)	44256 44276 44316	'NONE'	44256		00120		02080	77776
RITE PISA EXPIS	43503 43727 44120	*NONE *	44120		00136		00094	77773
WOCT REDE	43464 43500	COS ERROR 'NOVE' DCP	43464 43477		00013		00011 00273	
WAIT	43130	COL DECOD SORT WDCT	43116		00346		00230	50031
OCOS STARP VEC	42615 43046 43076	'NOVE'	43046 43076		00030		00024 00016	50031
PGSTRT PGSTOP DCP TYPRYT SIN COS QSIN	42600 42601 42602 42603 42604 42607 42613	*9V6*	42604		00242		00162	
ROT OFFSYS ENDSYS FINSYS SCHAIN	42312 42674 42575 42576 42577	**************************************	42306 42574		00010		00182 00008	50031
REJEC RECO	42230 42302	ERROR "NOVE"	42230		00056		00046	46711
REAP REAM	37470 37470	ARSIN EXPLO REAGB FLAI	37465		02543		01379	
QUIZ RATES	36600 36614	"YONE" SIM SORT	36603 -		00062		00003 00434	50031
PM360 PNUI	361 76 36250	SIN COS	36172 36242		00050		00040	\$0001 \$0001
OCPAR PARLEY	35416 36130	FLAT ERROR FORM WRITED FLAT	35411 36126		00515 00044		00333 00036	50031
OBTOX	34676 35174	ERRÛR NOVE IMPAR OCPAR READB	34672 35167		00275 00222		00189 00146	50031
EMPAR INSPC KINE	33752 34150 34456	OHOD CDS FORM *NUVE* REDE	33745 34144 34454		00177 00310 00216		00127 00200 90142	50031 50031 50001
GREOF GHADP	33436	ENDFIL (OFDP) (DFS8) (DFMP) (DFAD)	33424		00321		00209	50031
GERTA	33212 33332	(OFMP) (OFAD) (DFOP) SURT MRITES FLAT	33207		00215		00141	
FLAT FLAPR FLAK FORM	32022 32050 32057 32132	ERROR IYPRYI PARLEY	32020		01066		00065	50031
FLOT F1X FLOAT DAYE	31614 31775 32001 32011							
FILL FILT FIXT FIXTT	31202 31464 31500 31506	RITE PISA FIXTT "NONE"	31200 31477 31506		00277 00007 00312		00191 00007 00202	50001
ERROR	31166	VEC ABDD (OFMP) (OFAD) SGRT ARTAN ARSIN *NONE*	31166		00012		00010	
DOPLE	27466 27670	SIN CDS RATES DMBD	27460 27653		00173 01313		00123 00715	50031 50031
DECOD.	27420	FLOAT ADD *NOVE* *	27420		00040		00032	50031
CORR	27070 27362	TADZ SIN FIX	27062 27357		00275 00041		00189	47675
		HIZ						

ENTRY POINTS TO DEFINE CLOCK	G SUBROUTINES ATTACH (DFDP)	REQUESTED FROM LIDS OPEN CLOSE	RARY FGDOUT	PROUT	SORT IDESD)	DFMP}	1 OF
THE NAME	OF THES PAGE	RAH 15 *LAT *	2/22/65				
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS APRIER	LDAD ADD. 22300	DCTAL LENGTH 00012	DECIMAL LENGTH DOOLD	COMMON BREAK	
PRINT	22330	PAINT DEFINE ATTACH OPEN ENDIT RITE OFFSYS PAINO PAIN CLOSE REDE FOROUT FIXT PROUT FLAX	22312	02514	D1 356		
BIBCD DIAG	25053 25110	*NONE* SQRT	25026 25103	00055 00106	00045 00070	50031	
DIAGO DHAME Endif	25216 25331 25346	SORT 'NOIME' 'DESB! (DESB! (DESB) 'DESAD) PRIN SORI SIPREG REVRI NOUT	25211 25316 25332	01,05 00014 01474	00069 00012 00828	50031 50031	
ERRUR FIXT	27026 27040	PROME*	27026 21037	00011	00009 00007		
FIXTI FLOT FIX FLOAT DAYE	27046 27154 27335 27341 27351	***************************************	27046	00912	00202	50031	
FLAT FLAPR FLAK	27352 27410 27417	ERROR I YPRYT	21360	00101	00065+		
NGRHAY NOUT	27466 27734	SQRT PROUT	27461 27732	00251	00169 00064	7746L	
BLEW BUKEY OPKEY	27776 30032 30036	TYPRYT "KOYE"	30032	00040	00032		
COKEY AFKEY	30053 30060						
PRIM Pran	30105 30113	FGOOUT PROUT 618CO FLAPR PRIME NORMAY PRIF	30072	01424	40788	50001	
TAMP PRIME	31517 31520	4H04E*	31516	00005	00005		
TAPR DIAGO	31521						
PRING PRIT APRIOR	31532 31763 32154	FGDOUT DIAG CLOCK PADUT FLAPR FIAT BIBCD	31523	00435	00285	50031	
QU1Z REVRT	32160 32170	CDXEY	32160 32162	00002	00002 00134	50031	
OFFSYS ENDSYS FINSYS SCHAIN PGSIRT PGSIOP OCP TYPRYT	32370 32371 32372 32373 32374 32375 32376 32377	RLEW *NONE*	32370	00050	0000ê		
STPREG	32410	(OFAO) (OFAP) OSQRT (OFAP)	32400	02055	01069	71757	
DSQR! REDE RITE	34564 34567	ERROR OCP	34455 34563	00421	0007 <i>0</i> 00273	71776	
PISA (DFAD) (DFSB) (DFMP) (DFMP)	35013 35204 35224 35244 35272	• ND4E*	35204	00150	00000	77776	
SORT GLOCK HINDTE	35324 35403 35400	*#04E*	35324 35400	00054 00106	00044	77773	
XMIN PROUT	35400 35522 36766	อนาบร	35506	03032	01562		
FGDDUT PRCHY PROUTZ PROUT3 TSXA TSX8	35667 35660 35653 37675 40465	ACTINO BFLG RESTXA REQINO MATTE PRCON RGCSAY RGGSTR CKIND CKACT					
OUTUS BFLC ENDOUT CKIND CKACT REGIND ACTIND RESTKA PLICON PLZCON PL3CON PROON	40542 41376 41127 42771 43034 42770 42765 43050 43051 43052 43052 43053 43060	RGSSAV RGGSTR	43060	02320	01232		
RGGSIR IOCS	43142 43214	CLOCK	43213	04335	02269		
DEFINE JOIN ATTACH	43215 43220 43223		22				

CLOSE 43226

DPEN 43231

READ 43234

MRITE 43237

COPY 43245

REM 43245

MEF 43250

BSR 43253

SSF 43256

STASH 43261

\*LA7 \* JUST LOADED.

UNUSED CORE LIES FROM 47550 THROUGH 50001, LEAVING 00232 OCTAL OR 00154 DECIMAL LOCATIONS.

ENTRY POINTS TO DEFINE	SUBROUTINES HIARTA	REQUESTED FROM LIB	RARY, WEF	WRITE	FSDOUT	PROUT	SQRT
THE NAME	OF THIS PROG	RAH IS 'LAB '	2/22/65				
ENTRY NAME **PCB CLOCK	ENTRY ADD. 22302 22310	TRANSFER VECTORS RESID INDITE	LOAD ADD- 22301 22310	DCTAL LENGTH ODDOZ ODDOZ	0000 0000	)7 )2	COMMON BREAK
#ESI0	22320	DEFINE ATTACH OPEN TYPRYT THARP CLOSE	22312	01301	007		
ANGPLT	23520	PKIYE FIXT BISCO CXPLOT DIFD	23613	00577	003		51235
BIBCD CXPLOI	24437 24470	"ND1E" KEE	Z4412 Z4466	00054 03320	000		
XXXFSR	25327	WRITE	24400	0,,,,,	• • • • • • • • • • • • • • • • • • • •		
CDC PLTONE	25675 24521						
8ET	24472 30006	*ROYE*	30006	00026	000	22	
DIFO DNAME	30047	*4846*	30034	00015	000	13	
BALL [ BAQUE	30056 30057	KIHE	30051	00007	000	07	
ALLI	30052						
AQUI YUUY	30053 30054						
ERROR	30060	*NO4E*	30060	11000	000		
FIXT FIXTT	30072 30100	FIXIT *NOVE*	30071 30100	00007 00312	000 002	02	50031
FLOT F1X	30206 30367						
FLOAT	30373						
DAYE Flat	30403 30414	ERROR	30412	00102	000	66	
FLAPR	30442	TYPRYT					
FLAK KINE	30451 30516	REDE	30514	00215	001	41	50001
PERNOD WOLF	30736 31067	ERROR F[X] FGDDUT	30731	00503	003	<b>2</b> 3	51235
		PROUT SQRF					
QUIZ	31434	COLA "NOVE"	31434	00002	000	02	
RESP COLA	31450 32736	CDC FGDOUT	31436	01564	800		Sooni
LULA	32136	CXPLOT PKINE PERNOD FIXT PROUT FLAPR BIBCD FRXJR					
OFFSYS ENDSYS FINSYS SCHAIN PGSIRT PGSIOP DCP TYPRYT	33222 33223 33224 33225 33226 33227 33230 33231	**************************************	33222	00010	600	08	
STATID THARP PKINE	33232 33266 33373	NOVE* GFFSYS KINE AQUI ALLI BAQUI BALLI PENUD RESP WOLF ANGPLT BISCD TYPRYT	33232 33252	00020 00573	000 003		
REDE Rité	34046 34051	DCP	34045	00421	002	73	
PISA SQRT	34275 34466	"NONE"	34466	00054	000	44	77773
PROUT FGDOUT PRCHV PROUT2 PROUT3 TSXA TSXB	34556 36022 34723 34674 34707 36731 37521	OUTUS ACTIND BFLG RESTKA REGIND WRITE PRODN RGGSAV RGGSTR CKIND	34542	03032	015		
OUTUS BFLG ENDOUT CKIND CKACT REGIND ACTIND	37576 40432 40163 42025 42070 42024 42023	CKACT RGGSAV RGGSTR	37574	02320	012	32	

RESTKA	42021				
PL1CON	42104				
PLZCON	42105				
PL3CON	42106				
PRCON	42107				
REGSAY	42114	*NOYE*	42114	00133	00091
		-4045	46114	00133	00031
RGGSTR	42176				
1005	42250	CLOCK	42247	D4335	02269
DEFINE	42251				
MIOL	42254				
ATTACH	42257				
CLOS€	42262				
OPEN	42265				
READ	42270				
WRITE	42273				
COPY	42276				
REW	42301				
WEF	42304				
BSR	42307				
asr	+2312				
STASH	42315				
'LAB ' JU	ST LØADED.				

UNUSED CORE LIES FROM 46604 THROUGH 50001, LEAVING 01176 DCTAL DR 00638 DECIMAL LOCATIONS.

EHTRY POINTS REWIND	TO SUBROUTINES ENDFIL	REQUESTED FAON LI	(BRARY, OP) (DFS8)	(DFMP)	(DFAD) DROD	DĒFINE	ATTACH
OPEN	CLOSE	EXP(3 RGGS		CLOCK	READ		
THE HAM	E OF THIS PROG	RAH IS 'LA9 '	2/22/65				
ENTRY NAME	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	CONNON BREAK	
MODPH1 ODBFF	22460 22414	TYPRYT ODATA	22300	01322	00722		
READS	22306	REWIND					
		OFFSYS					
		€NDF1L					
TELTP1	23524	ERROR *NOVE*	23622	00003	00003		
TELTYP	23623	-1015	23022	00003	00003		
LUNIT)	23522						
FIXT	23526	FIXIT	23625	00010	00008		
DIAG ERROR	23642 23744	SQRT Typryt	23635 23742	00105 00442	00069 00290	50031	
CHION	23744	ONLIN	20112	00772	20270		
FIXTT	24404	.NDKE.	24404	00312	00505	50031	
FLOT F1X	24512 24673						
FLOAT	24677						
DAYE	24707						
GHADP	24730	(DFOP)	24716	00320	00208	50031	
		(DFSB)					
		(DFAD)					
		DMDD					
		COS 'NDYE'			00002		
QUIZ OFFSYS	25236 25240	'ND4E'	25236 25240	00002 00010	00002		
ENDSYS	25240 25241		2,52,10	00010	00000		
FINSYS	25242						
SCHAIN PGSTRT	25243 25244						
PGSTOP	25245						
DCP	25246						
TYPRYT	25247 25250						
SIN COS	25250 25253	'NOVE'	25250	00243	00163		
QSIN	25257						
gcos	25261						
SPHX	25522	COS SIN	25513	00323	00211	50031	
		GHADP					
STPREG	26046	(DFAD)	26036	02055	01069	71757	
		(DFOP) DSQRT					
		{DFMP}					
DSQRT	30114	ERADR	30113	00106	00070	77776	
REDE	30222 30225	DCP	30221	00422	00274		
PISA	30451						
ODATA	30566	TYPRYT	30643	03305	01733	50001	
ONLIN	30770	DEFINE					
*****	31136	ATTACH Open					
		CARDS					
		OFFSYS					
		ERRDR FINSYS					
		Crose					
		ODOFF					
		READS					
		FLOT MODENT					
		SPHX					
		STPREG					
		DIAG REDE					
		DCP					
		EXP(3					
CARDS CRDCON	34154 34430	RGGSAV RGGSTR	34150	99220	01206		
CROCON	34450	CLOCK					
		READ					
EXP(3	36436	"NOVE"	36436	00136	00094	77773	
(DFAD) (DFSB)	36574 36614	"NOVE"	36574	00120	00080	77776	
(DEMP1	36634						
(DFOP)	36662			444	****		
DMDD SQRT	36714 36774	* ND4E *	36714 36774	00060 00054	00048 00044	77776 77773	
CLOCK	37053	*ND4E*	37050	00107	00071	11113	
MINUTE	37050		·				
XMIN READD	37050 37160	(נסנ)	27167	00365	00245		
READB	37162	1 700 1	37157	20500	VUZ93		
WRITED	37165						

WRITEE	37167				
BSREC					
	37376				
BSFILE					
REWIND					
UNLOAD					
ENDFIL					
SETLOX					
SETHI	37373				
(UNIT)	*****				
TAPEIC					
£10N}	37547	• NOVE •	37544	00030	00024
RGGSAV	37574	'NONE'	37574	00133	00091
REGSTR	37656				
LOCS	37730	CLOCK	37727	04335	02269
DEFINE	37731				
MIOL	37734				
ATTACH	37737				
CLOSE	37742				
OPEN	37745				
READ	37750				
WRITE	37753				
COPY	37756				
REH	37761				
WEF	37764				
BSR	37767				
BSF	37172				
STASH	37175				
P4 15	* JUST LOADED.				

UNUSED CORE LIES FROM 44264 THROUGH 50001, LEAVING 03516 OCTAL OR 01870 DECIMAL LOCATIONS.

ENTRY POINTS TO	SUBROUTINES DEFINE	REQUESTED FROM LI	BRARY, CLOSE	COMENT	READB PROUT	
THE NAME	OF THIS PROG	RAM IS 'LAID '	2/22/65			
ENTRY NAME **PCB PARSH	ENTRY ADD. 22302 22316	TRAYSFER VECTORS PARSH FGDDUT DEFINE ATTACH OPEN COMDUT WASH CLOSE	E LOAD ADD- 22301 22307	DCTAL LENGTH 00006 01226	DECIMAL LENGTH DOODS 00662	COMMON BREAK
CBMQUT FIXT FLOT FLOT FIX	23536 23554 23562 23670 24051	FIXIT FIXIT COMENT	23535 23553 23562	00016 00007 00312	00014 00007 00202	S0031
FLOAT DAYE Flat Flapr	24055 24065 24076 24124	ERROR TYPAYT	24074	00102	00066	
FLAK ERROR	24133 24176	*NOVE *	24176	00011	08009	
WASH	24214	READB FLAT FIXT PROUT FLAPR	24207	00623	00403	
OFFSYS ENDSYS FINSYS SCHAIN PGSIRI PGSIGP DCP	25032 25033 25034 25035 25036 25037 25040	· NOVE ·	25032	00018	80000	
TYPAYT PROUT FROUT PROUT PROUT2 PROUT3 TSXA TSXA	25041 25056 26322 25223 25174 25207 27231 30021	OUTUS ACTIAD BELG RESTKA REGIND HRITE PRODN RGGSAV RGGSAV CKIND	25042	03031	01561	
READD READB REITED REITEB BSREC BSFILE REWIND UNLOAD ENDFIL SETLON SETHI (UNII) IAPEIO	30074 30076 30101 30103 30312 30315 30320 30323 30326 30304 30307 30455	(IOU)	30073	00365	00245	
OUTUS BFLG ENDOUT CKIND CKACT REGIND ACTIND RESTKA PLICON PLICON PRECON	30463 30512 313746 31377 32741 33004 32740 32737 32737 32737 33020 33022 33022 33022	*NOVE* RGGSAV RGGSTR	30460 30510	00030 02320	00024 01232	
RGGSAV RGGSTR IDCS DEFINE JOIN ATTACH CLOSE OPEN READ MRITE	33030 33112 33164 33165 33170 33173 33176 33201 33204 33207	CLOCK	33030 33163	00133 04336	0009L 02270	

UNUSED CORE LIES FROM 43026 THROUGH 50031, LEAVING 05004 OCTAL OR 02564 DECIMAL LOCATIONS.

COPY 33212
REM 33215
WEF 33220
BSR 33223
BSR 33223
STASH 33231
CONENT 37522 CLOCK 37521 03177 01663
CLOCK 42723 "NOME" 42720 00106 00070
MINUTE 42720
VLA10 "JUST LOADED-

REWIND ENDFIL		(DFSB) READS	BSREC	(DFAD)	(OFMP) (DFOP)	DKOD	547
THE NAME	OF THIS PROG	RAH IS *LA11 *	2/22/65				
SHAN YATHS	ENTRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL LENGTH	COMMON BREAK	
••PCB	22304	EPOX Trade	22300	00022	00018		
		TAPEX					
1000	22332	LOCO REDE	22322	00165	00117		
TRADE	22455	REWIND	LESCE	00103	00111		
EPOX	22466	KOOL BNTR2					
		Dir.					
		TYPRYT					
		EPPHEM FLAK					
ACE	22514	SQRT	22507	00067	00055	47675	
BMATRX	22610	SQRT	22576	01432	00794	47675	
		G1G2 RCAL					
		RCOM					
		RSIG					
ERROR	24230	ACE •NONE•	24230	00012	00010		
ADO	24242	'NOYE'	24242	00050	00040		
ARSIN ARCOS	24322	"KONE"	24312	00144	00100	77152	
QARSIN	24316 24314						
QARCOS	24312						
ERRARS FRRARC	24443 24447						
ARTAN	24456	'NOYE'	24456	00103	00067	77152	
DAYS	24564	FIX	24561	00042	00034		
		FLOAT ADD					
EPPHEM	24644	QUIZ	24623	00604	00388	47675	
		PONT Kine					
		STARP					
		(DFSB)					
		TIMER LOOKUP					
		INTRI					
		KOOL					
		GAMAT Ritem					
		TOCIM					
FIXT	25430	WREDF	25/22	00007	20007		
FIXIT	25436	FIXTT "NONE"	25427 25436	00312	00202	50031	
FLOT	25544						
FIX Float	25725 25731						
DAYE	25741						
FLAT Flapr	25752 26000	ERROR Typryt	25750	00105	88000		
FLAX	26007						
G162	26056	*HONE!	26052	00266	90182	50001	
EPHEN TAPEX	26344 27524	REWIND READB	26340	01452	00810		
		BSREC					
PHUT	30050	SIN	30012	00336	DD Z 2 2	50001	
ROT	30354	COS "HOME"	30350	00266	00182	+50031	
GAHAT	30652	(DFAD)	30636	00614	00396	47675	
		(DFSB) LOSKUP					
		INTR1		•			
		INSPC					
		BHATRX (DFMP)					
		(DFDP)					
GHADP	31464	(OFDP)	31452	00320	00208	50031	
		(DFS8)					
		(DFAD)					
		DKOD					
INSPC	31776	COS *NOME*	31772	00310	00200	50031	
BNTR2	36365	EPHEN	32302	04214	02188		
INTAL	36406	PHUT Rot					
		ERROR					
KINE	36520	REDE	36516	00200	00128	50001	
LGOXUP	36720	ERROR ERROR	36716	00463	00307	47777	
KOGL	37250	REDE					
PONT	37414	ÜFFSYS	37401	00311	00201	51235	
		(OFAD) TIMER					
		ŁOCKUP					
		INTRI Gamat					

QUIZ	37712	*NONE*	37712	00003	00003	
RCAL	37722	SORT	37715	00249	00160	50031
ACOM	40162	SQRT	40155	00133	00091	50031
WREOF	40431	FLAT	40310	02520	01360	20031
RITEH	40314	WRITEB	40310	02320	01360	
	103.1	ENDFIL				
		RÉWIND				
RSEG	43034					
DFFSYS		'NOVE'	43030	00064	00052	47675
	43114	******	43114	00016	00008	
ENDSYS	43115					
FINSYS	43116					
SCHAIN	43117					
PGSTRT	43120					
PGSTQP	43121					
DCP	43122					
TYPRYI	43123					
SIN	43124	"NOYE"	43124	00242	00162	
COS	43127		,,,,,	30272	00105	
NIZO	43133					
gcos	43135					
STARP	43366	'NONE'	43366	00031	00025	
TIMER	43426	(DESB)				****
TINCS	43420		43417	00142	00098	50031
		LOOKUP				
T		SQRT				
TOCIN	43600	OFFSYS	43561	01126	00598	
		TIMER				
		LOOKUP				
		INTRI				
		SQRT				
		KOOL				
		GAHAT				
		CHADP				
		COS				
		SIN				
		ARSIN				
		ARCDS				
		WRITEB				
		FLAT				
		RITE				
REDE	44710	OCP	44707	00421	00273	
RITE	44713					
PISA	45137					
(DFAD)	45330	MOVE	45330	00120	08000	77776
{DF\$B}	45350					
{DFMP}	45370					
{OFOP1	45416					
D×OD	45450	*4D4E*	45450	00060	00046	77776
SQRT	4553C	*NONE *	45530	00055	00045	77773
READD	45506	(100)	45605	00365	00245	
READB	45610		17007	VV 707	00273	
WRITED	45613					
WRITES	45615					
BSREC	46024					
BSFILE	46027					
REWIND	46032					
UNLOAD	46035					
ENDFIL	46040					
SETLOW ,	46016					
SETHI	46321					
(UNIT)	46167					
TAPELO	46167					
(100)	46175	• NOVE •	46172	00030	00024	
					<del>-</del> -	
'LAII ' J	UST LOADED.					

UNUSED CORE LIES FROM 46222 THROUGH 47675, LEAVING 01454 OCTAL OR 00812 DECIMAL LOCATIONS.

DEFINE	ATTACH	REQUESTED FROM LIB OPEN CLOSE	SQRT	PROUT	(OFS8)	(DFAD)	FGDD91	E
THE NAME	OF THIS PROG	RAH IS "LAIZ "	2/22/65					
ENTRY NAME	EYIRY ADD.	TRANSFER VECTORS	LOAD ADD.	OCTAL LENGTH	DECIMAL	L LENGTH	COMMUN BREAK	
*****	22302	KUSVAZ	22301	00123		083		
SAVCOM	22440	DEFIYE ATTACH OPEN	22424	02154	01	132		
		PRIM QUEST						
		HAXIH OFFSYS						
		TYPRYT RITE CLOSE						
		ERROR FLAC						
BIBCO	24025	'NDYE'	24600	00055	000	045		
DIAGO	24662	SQRT	24655	00105		069	50031	
ERROR	24766	TYPRYT PROJT ENDSYS RECOV	24762	00544	00	356		
FLAT	25530	ERROR	25526	00102	000	240		
FLAPR	25556	TYPRYT	27720	00101	401	000		
FLAX	25565	.,						
KINE	25632	REDS ERROR	25630	00215	00:	141	50001	
HAXIN	26056	QU[Z kine (DFSB) (DF 1D) HOCT	26045	60473	003	315	50001	
HOET	26544	'NOVE'	26540	00107	000			
NORHAY	26654	SORT	26647	00252	001		77461	
APRED	27204	FIXT	27121	00143	000		11401	
UPREJ OPERA	27231 27122			40.43	000	.,,		
DPKEY	27264 27270	*3704*	27284	00040	000	03Z		
COKEY	27305							
AFKEY	27312							
PRIM Pram	27345 27334	PROUT	27324	01424	007	188	50001	
		BISCD FLAPR PRIME NORMAY						
		PRIT DIAGO						
TAMP	30751	'YDYE'	30750	00004	000	104		
PRIKE	30752					•		
TAPR	30753							
DIAGO	*****							

PRIT	30756	FIXT BIBCO	30754	00035	00029	
quest	31020	SORT ERROR	31011	00165	00117	47675
		EXIT			*****	
QUIZ	31176	"NDNE"	31176	00002	00002	
RECOY	31200	**04E*	31200	00402	00002	50031
SETUP	31204	RITE	31202	00105	00069	20031
4444		ENDSYS	24267	00007	00007	
FIXT	31310	FIXTT	31307		00202	\$0031
F1XTT FLDT	31316 31424	•NO4E•	31316	00312	03204	30031
F1X FLOAT	31605 31611					
DAYE	31621					
OFFSYS	31630	"NONE"	31630	00011	00009	
ENDSYS	31631					
FINSYS	31632					
SCHAIN	31633					
PESTRT	31634					
PGSTOP	31635					
DCP	31636					
TYPRYT	31637					
3032	31642	DCP	31641	00421	00273	
RITE	31545					
PISA	32071	anour.	32262	00210	00136	
EXIT	32263	'NOVE'	32202	00210	00130	
EXSEL	3236 <del>6</del> 32274					
EXITEN ERROMP	32262					
(DFAD)	32472	*NONE*	32472	00120	00080	77776
(DFSB)	32512	115 12				
(DEMP)	32532					
(OFOP)	32560					
SQRT	32612	*NB4E*	32612	00054	4+DDD	77773
PROUT	32702	OUTUS .	32666	<b>Q3032</b>	01562	
FGDOUT	34146	ACTIND				
PRCNV	33047	BFLG				
PROUTZ	33020	RESTKA				
PROUT3	33033	REGIND				
TSXA	35055	WRITE				
TSXB	35645	PACON RGGSAV				
		RGGSTR				
		CK1ND				
		CXACT				
OUTUS	35722	RGGSAV	35720	02320	01232	
BFLG	36556	RGGSTR				
ENDOUT	36307					
EKIND	40151					
CKACT	40214					
KEDIND	40150					
ACTIND	40147					
RESTKA	40145					
PL1CON PL2CON	40230 40231					
PLICON	40232					
PACON	40233					
RGGSAV	40240	*NONE *	40240	00133	00091	
RGGSTR	40322					
1005	40374	CLOCK	40373	04335	02269	
DEFINE	40375					
JOIN	40400					
ASTACH	40403					
CLOSE	40406					
DPEN	40412					
READ Write	40414 40417					
COPY	49417 40422					
REW	10122					
NEF	40430					
8SR	40433					
BSF	40436					
STASH	40441					
CLOCK	44733	*HB%E*	44730	<b>GB1G6</b>	00070	
MINUTE	44730					
XMIH	44730					
	WET LOLDED					
-LAIZ	JUST LOADED.					
_						DECEMBE A DESTROY

UNUSED CORE LIES FROM 45036 THROUGH 47675, LEAVING 02640 OCTAL OR 01440 DECIMAL LOCATIONS.

DEFINE SORT	O SUBROUFINES AFFACH	"REQUESTED FROM LIB OPEN PROUT	CLOSE	(DFDP)	(DFSB) (DFMP)	(DFAD)
THE NAME	OF THIS PROG	RAM IS "DDPX "	2/22/65			
ENTRY NAME FIRST	ENTRY ADD- 22312	TRANSFER VECTORS FDATA DEFINE ATTACH OPEN PROUT NOMNL RITE CLOSE FLAX	LOAD ADD. 22301	OCTAL LENGTH 02070	DECIMAL LENGTH	COMMON BREAK
• =PCB	24372	FIRST	24371	50000	00007	
ERROR	24400	"NDNE"	24400	00012	00010	
FDATA	24412	*NOVÉ*	24412	01117	00592	
FIXT	25532	FIXTT	25531	Op007	00007	
FIXTT	25540	"NONE"	25540	00312	00202	50031
FLOT	25546					
FIX	26027					
FLOAT	26033					
DAYE	26043					
GHADP	26064	(DFOP) (DFSB) (DFMP) (DFAD) DMOD COS	26052	00320	00208	50031
NONNL	26376	'NOVE'	26372	00763	00499	47777
HORMAY	27362	SQRT	27355	00251	00169	77461
OUIZ	27626	*NOVE*	27626	00002	00002	

SCHAIN	27533					
PGSTRI	27634					
PGSTOP	27635					
DCP	27636					
TYPRYT	27637					
SIN	27540	'NONE'	27640	00243	00113	
cos	27543	-vore -	27040	00243	00163	
QSIN	27647					
QCQS	27651					
REDE	30104	OCP	30103	00/01		
RITE		ULP	30103	00421	00273	
PISA	30107					
FLAT	30333 30526		30524			
FLAPR	30554	ERROR TYPRYT	30324	00102	00466	
FLAK	30563	ITPRIT				
(DFAD)	30626	INONE	30626			77776
		. 40AE.	30020	00120	08000	11116
(DFS81 LDFMP1	30646 30666					
(OFDP1	30714					
			207//	****		*****
GKBO SQRT	30746 31026	HOVE	30746	00060	99948	77776
PROUT	31116	*37UK* 20100	31026 31102	00054	00044	77773
FGDDUT	32362	DRITON	31102	03032	01562	
PRÉNY	31263	BFLC				
PRDUT2	31234	RESTRA				
PADUT3	31247					
		REGIND				
TSXA TSXB	33271	WRITE				
ISAB	34061	PRCON				
		RGGSAV RGGSTR				
		CKIND				
		CKAGT				
OUTUS	34136	RGGSAV	34134	02320	01 232	
BFLG	34772		24114	02320	01232	
ENDOUT	34523	RGGSTR				
CKIND	36365					
CKACI	36430					
REGIND	36364					
ACTIND	36363					
RESTK4	36361					
PLICON	36444					
PL 2CON	36445					
PL 3CON	36446					
PRCON	36447					
RGGSAV	36454	*NOVE *	36454	00133	00091	
RGGSIR	36536	-4046 -	20434	00133	40032	
2301	36610	CLOCK	36607	04335	<b>02269</b>	
DEFINE	36611		2000	41333	02207	
JOIN	36614					
ATTACH	36517					
CLOSE	36622					
DPEN	36625					
READ	36630					
HRITE	36533					
EOPY	36636					
REW	36641					
HEF	36544					
BSR	36647					
BSF	36552					
STASH						
	36655					
	36655 63167	*NOVE*	43166	00106	00070	
CLOCK	43147	'NUNE'	43144	00106	00070	
		'NONE'	43144	00108	00070	

UNUSED CURE LIES FROM 43252 THROUGH 47777, LEAVING 04526 BCTAL DR 02390 DECLMAL LOCATIONS.

\*DDPX \* JUST LOADER.

#### QUE CHEHON HAP, JANUARY 4,1965.

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| 17401 | T CUMMUN | 2 | RUNNING TIPL | |
| 17457 | EFR CUMMUN | 2 | EPICH |
| 17457 | TER CUMMUN | 2 | EPICH |
| 17451 | TER CUMMUN | 2 | EPICH |
| 17451 | TER CUMMUN | 2 | EPICH |
| 17451 | TER CUMMUN | 2 | EPICH |
| 17451 | TER CUMMUN | 2 | EPICH |
| 17451 | TER CUMMUN | 2 | EPICH |
| 17452 | CHAYUN | 2 | EPICH |
| 17453 | CHA CUMPUN | 1 | EPICH |
| 17454 | CHAYJ CUMBUN | 1 | EPICH |
| 17451 | EPICH | CAMPUN | 1 | EPICH |
| 17452 | EPICH | SCO |
| 17453 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17455 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17455 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17455 | EPICH | SCO |
| 17454 | EPICH | SCO |
| 17455 | EPICH | SCO |
| 17456 | EPICH | SCO |
| 17457 | EPICH | SCO |
| 17457 | EPICH | SCO |
| 17457 | EPICH | SCO |
| 17458 | EPICH | SCO |
| 17459 | EPICH | SCO |
| 17450 | EPICH | SCO |
| 17451 | E
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DEC COMMON
HA CUMHON
C1 CUMHON
C01 CUMHON
CU1 CUMHON
CU3 CUMHON
XXX CUMHON
                                                                                                                                                                       LOCAL DECLINATION ANGLE
LOCAL HOUR ANGLE
1 MAY DUPPLER
COMERENT 3 MAY DOPPLER
3 MAY DUPPLER
DIFFERENCEO 1 MAY DUPPLER
DIFF RANGING
EXTRA DATA TYPE
   77360
77357
77356
77355
77354
77353
77352
77351
                                  DAVEC COHNON
DRUCT COMMON
DEL COMMON
DAZ CHMON
DUEC COMMON
DUEC COMMON
DC1 COMMON
DC2 COMMON
DC3 COMMON
   77350
77347
77346
77345
77344
77343
77342
77341
77340
77337
77336
77335
                                                                                                                                                                       CORRECTIONS
                                  RVECO COMHON
ROOTO LUMMON
ELU CUMMON
AZIU CUMMON
DECU CUMMON
CIU CUMMON
CIU CUMMON
CIU CUMMON
CIU CUMMON
CIIO COMMON
CUIO COMMON
XXXU CUMMON
  /7334
77333
77332
77331
77330
77327
77326
17325
77325
77325
77322
17323
                                                                                            FORTRAN CAND OS
                                                                                                                                                                       GBSERVED VALUES
                                 RYECD CUMON ROOTS COMBON ELU CUMON AZIO CUMON DECO CUMON HAD CUMON CID CUMON CID CUMON CID CUMON CID CUMON COID CUMON XXXD CUMON XXXD CUMON
                                                                                            FURIRAN CARD 05
  77320
77317
77316
77315
77314
77313
77312
77311
77310
77307
77306
77305
                                                                                                                                                                     RESIDUALS TUBSERVED-CALCULATEDS
                                                                                          FORTRAN CARD OF
1
                                RYECH COMHON
ROOTH COMHON
ELW COMHON
AZIM COMHON
DEGN COMHON
CAM COMHON
CAM COMHON
CAM COMHON
COMHON
COMHON
COMHON
COMHON
COMHON
COMHON
COMHON
 77304
77303
77302
77301
77300
77277
77276
77275
                                                                                                                                                                       WEEGHTS
  77274
77273
  77272
77271
                                 FURTRAN CARD UL

AR CUMMON 1

BU CUMMON 1

PASS CUMMON 1

AUP COMMON 6

ARRA CORRON 6

AVEN CUMMON 6

XVEN CUMMON 6

XMED CUMMON 6

XMED CUMMON 6

XMED CUMMON 6

XMED CUMMON 7

COMMON 1
                                                                                          L
FURTRAN CARD US
                                                                                                                                                                   RECEIVER 1.D.
XMITTER 1.D.
CURRENT DATA TYPE
PASS 10ENTIFICATION
JUPITER VECTOR
SATURN VECTOR
HARS VECTOR
VENUS VECTOR
SUN VECTOR
MODA VECTOR
 77270
77267
77266
77265
17246
17240
17232
77224
77216
 77210
                                   DOR CONKON
DOR CONKON
DO CONKON
77202
77201
77200
77177
77176
77175
77174
                                                                                      FURTRAN CARD 09
1
1
1
1
1
1
1
1
56
1
1
FORTRAN CARD 10
                                                                                                                                                                    PROBE VECTOR
                                                                                                                                                                    VARIATIONAL EQUATIONS NUTATION IN LONGITUDE NUTATION IN OBLIQUITY
 77130
77127
                               XAC CUMMON
YAC COMMON
ZAC CUMMON
XJERK CUMMON
YJERK CUMMON
JERK CUMMON
UIAV CUMMON
BHAT CUMMON
 77126
77125
77124
77123
77122
77121
77120
77054
                                                                                                                                                                    PROBE ACCELERATION
                                                                                                                                                                   PROUE JENK (THING CERIVATIVE)
                                                                                       1
36
6+12
6+11
FURTRAN CARD 11
                                                                                                                                                                    INVERSE OF VARIATIONAL EQUATIONS PARTIALS(DBSERVABLES/CARTESIAN) B MATRIX OF PARTIALS
 76744
                                      ELX COMMON
ELY COMMON
ELY COMMON
AX CUMHON
AX CUMHON
AZ COMMON
DY COMMON
DY COMMON
TAY COMMON
TAY CUMHON
TAY CUMHON
TAY CUMHON
TOY CUMHON
TOY CUMHON
TOY CUMHON
TOY CUMHON
TOY CUMHON
76642
76640
76640
76636
76635
76635
76634
76633
76631
76631
76630
76625
76625
                                                                                                                                                                   DIRECTION COSINES
                                                                                                                                                                   DIRECTION CUSINES
                                                                                                                                                                   DIRECTION CUSINES
                                                                                                                                                                  DIRECTION COSINES
                                                                                                                                                                  DIRECTION CUSINES
                                                                                         FURTRAN CARD 12
76623
/5622
76614
76601
                               TAU COMMON 1
ILIST CUMMON 6
LLIST CUMMON 11
MLIST CUMMON 1
                                                                                                                                                                  DOPPLER AYERAGING TIME
POSITION AND VELCCITY FLAGS
PHYSICAL COMSTANTS PARTIALS FLAGS
VELUCITY OF LIGHT FLAG
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NLISI COMBIN.

NEST COMBIN.

NCY CUMBIN

EQU CHARIN

EQU EP-6

EQU EP-7

EQU
                                                                                                                                                                                                                                         STATION LOCATION FLAGS
NU. ESTIMATED PARAMETERS-
NU. CONSIDERLY PARAMETERS
 76600
16523
76522
76521
16521
76523
76512
76511
76471
76471
76471
76466
76457
76421
76421
76421
76421
76421
                                                                                                                           100
EP-1
EP-6
EP-7
EP-4
EP-25
EP-25
EP-34
EP-34
EP-49
EP-49
EP-48
EP-48
EP-40
EP-100
EP-100
EP-100
EP-100
                                                                                                                                                                                                                                         E.T.-U.T.
                                                                                                                                                                                                                                       BANUI-1817=LS 0=L 1817=S)
K1 FREQUENCY
K5 FREQUENCY
BAND DESIGNATION (15 MORDS)
START TIME, ATTITUDE CUNTROL
                                                                                                                                                                                                                                          UP
STOP TIME, ATTITUDE CONTROL
                                                                                                                                                                                                                                       MCIM COMENI BUFFER 130 MORUS)
NOT LERU TU READ CARDS FRUM A2
10 FOR PREDICTS
POWERD FLIGHT PARAMETERS (10 MORUS)
STEP MAP FLAC-U PROJUCT
START TIME, STEP MAP PRINTOULT
 76375
76374
76372
76371
76350
76347
76346
16345
76235
                                                                                                                                                                                                                                          THE STEP HAP PRINTED OP FLAG TO REPCAT TRAJ FOR STEP HAP SPACE COMENT BUFFER (17 NURUS) MAPPLO FORWARD EPULH
                                                                                                                                                                                                                                          FLAG FUR PUNCHING MAIRICES
NOT ZERO TO BYPASS LAIL
                                                                                                                                                                                                                                         REJECTION SIGNAS
                                          SIGX3 CUMHON
                                                               A CUMHON

B COMHON

C CUMHON

D CUMHON

F COMHON

G COMHON

D CUMHON

P COMHON

P COMHON

R COMHON

R COMHON

S CUMHON
                                                                                                                              FORTRAN CARD 11.
                                                                                                                                                                                                                                         CORMON
STORAGE
AVAILABLE
                                                                                                                                                                                                                                         FOR
FAP
OR
FORTRAN
USAGE
 15741
15740
15737
15736
15735
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FURTRAN CARD 14
2
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6
6
Furtran Card 15
 15734
15732
15730
15726
15724
15722
15720
15712
                                          RNJ COMMON
PRINS COMMON
INTRA COMMON
VNJ COMMON
ELWJ COMMON
AZNJ COMMON
XNJ COMMON
                                                                                                                                                                                                                                         SPHERICAL INJECTION CONDITIONS
                                                                                                                                                                                                                                       CARTESIAN INJECTION CONDITIONS
                                  CRAVE CUMMUN
RE COMMUN
RE COMMUN
GRUB LUMMON
RYEN CUMMUN
                                                                                                                                                                                                                                         KLEARTH)
EARTH RADIUS
RADIATION PRESSURE CONSTANT
 75104
75703
75102
75701
15700
75617
75616
75615
15614
75613
75612
75671
                                                                                                                                                                                                                                    RADIATION PRESSURE CONSTANT
-KIFHORNI
MASS RATIO PERS
MASS RATIO PARS
HASS RATIO PUPILER
CUFFICIENT SECOND HARPWHIC
CUFFICIENT FOUNTH HARMONIC
ASTRONOMICAL UNIT
VELCCITY OF LIGHT
                                                                                                                              FURTRAY CALD 16
 75670 RI CUNHON
75651 PHII CUPHON
75632 THETAI CUMMON
                                                                                                                                                                                                                                       STATION RANGE
STATION LATITUDE (GEOGENIRIC)
STATION LONGITUDE
                                          GRAVS CUMHUN
GRAVY CUMHUN
GRAVY CUMHUN
GRAVJ CUMHUN
KI CUMHUN
YI CUMHUN
YI CUMHUN
EE CUMHUN
RAI CUMHUN
75613
75612
75611
75610
75601
75606
75605
75604
75603
                                                                                                                                                                                                                                       KISUN)
KIYENUS)
K(MAKS)
K(AMS)
K(JUPITER)
ANGULAR ROTATIU- RATE OF CAKIH
STATION X(T)
STATION X(T)
STATION X(T)
                                                                                                                                                                                                                                         STATION ZITT
STATION RIGHT ASCENSION
                                                                                                                     15642
15563
15544
15525
                                                       UI CUMMON
VI CUMMON
FNI COMMON
TAB CUMMON
                                                                                                                                                                                                                                       NORTH SUUTH DISPLACEMENT (GECD-GEOC)
EAST WEST DISPLACEMENT
LOCAL INDEX OF REFRACTION
                                                                                                                                                                                                                                      DATA FILE 1-PUT BUFFER (400 WORDS)
DATA FILE LUTPLI BUFFER (400 WORDS)
SCRATLM AREA, LAG (200 WORDS)
PROBE EPHEMFAIS INTERPULATION BUFFER
SCALL FACTORS
PLOT FLANS
PLUT STEP
                                                TAB CUMMUN
EQU
EQU
EQU
YI CUMMUN
RESK CUMMUN
IRES CUMMUN
MHR CUMMUN
 15525
74705
74065
74065
73555
71325
71325
71135
                                         DRUC CUMHUN
DIM3 CUMHUN
CPPH1 CUMHUN
CPPH2 CUMHUN
CPTH1 GUMHUN
XJ CUMHUN
XJ2 CUMHUN
XX CUMHUN
XX CUMHUN
RIGHI CUMHUN
RIGHI CUMHUN
RIGHI CUMHUN
RIGHI CUMHUN
                                                                                                                                                                                                                                      VELECTIY OF LIGHT PARTIALS
STATION PARTIALS BY DATA TYPE
CAP PHI MATRIX OF PARTIALS
 70744
70730
                                                                                                                     12-11-15
20
20
20
20
20-20
20-20
400
20-20
20-20
400
20
20-20
60
                                                                                                                           12-3-15
70730
61674
61650
67624
61600
67554
66114
65274
64454
                                                                                                                                                                                                                                    CAP PHI MATRIX OF PARTIALS UP CAP INCIA MATRIX OF PARTIALS DP JATRIX UP SCHAICH ZONG MATRIX SCRAICH ZONG PATRIX SCRAICH ZONG PATRIX RESIDUAL-CPPHI ACCUMULATEU DP
                                            CPGAM CGMMON
GPGM2 CGMMON
GFLF2 CGMMON
GFLF2 CGMMON
GAMTO CGMMON
DIMI CGMMON
64404
                                                                                                                                                                                                                                       CUVARIANCE MATRIX
                                                                                                                                                                                                                                          UP
INPUT COVARIANCE HAIRIX (1 4/ERSE)
                                                                                                                                                                                                                                       DP
GUV MX OF UHADJUSTED PARAMETERS
V MATRIX THTEGRAND
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DIN2 COMMON 66
DIG COMMON 48
EQU 01C
PLOFIL EQU 01C-23
PUNFIL EQU 01C-35
PRIFIL EQU 01G-47
FORTRAN CARD 21
60362
60260
60260
60231
60215
60201
                                                                                                                                                                                                                                                                               DP
                                                                                                                                                                                                                                                                                IUCS PLUT CONTROL (12 MCROS FORMARD)
IUCS PUNCH CUNTROL (12 MDRUS FURMARD)
IUCS PRINT CONTROL (12 MDROS FURMARD)
                                                 YEAR COMMON HONTH COMMON DAY COMMON HON COMMON HIN COMMON SEC COMMON FRAC COMMON
 60200
60177
60176
60175
60174
60173
                                                                                                                                                                                                                                                                               7-WURD TIME BLOCK
INTEGERS IN ADDRESS PART
FAP USE ONLY
 60172
                                                                                                                                                   1
FORTRAN CARD 22
                                                       IFL COMMON
IF2 CONHON
IF3 CONHON
IF4 CONHON
IF5 COMMON
IF6 CUNHON
IF6 CUNHON
IF6 CONHON
IF6 CONHON
IF6 CONHON
 60171
60170
60167
60165
60165
60164
60163
60162
60161
                                                                                                                                                                                                                                                                                CONTENTS OF KEYS 10-17
                                                                                                                                                                                                                                                                               ZERO TO CLOSE DATA FILE/PREDICT OUTPUT
DATA FILE READ LB17=EGF 2817=NORMAL
                                                                                                                                                                                                                                                                               NOT ZERO TO CHECK RESIDUAL REJECTION
                                                                                                                                                                                                                                                                               NOT ZERO TO PRINT PHI VECTORS
 60160
                                                 KLIST COMMON
FABC COMMON
ELMIN CUMMON
RNAX CUMMON
RUN CUMMON
GA CUMMON
GB CUMMON
TOBL CUMMON
XX CUMMON
XX CUMMON
XX CUMMON
XX CUMMON
                                                                                                                                                   FORTRAN CARD 23
                                                                                                                                                                                                                                                                             FLAGS FOR ATTITUDE CONTROL ESTIMATION
ATTITUDE CONTROL CUEFFICIENTS
MINIMUM ELEVATION ANGLE
MAXIMUM SLANT RANGE
DATA FILE GUIPUT COUNTER
   60157
   60157
60153
60142
60141
60140
60137
60136
60123
60115
                                                                                                                                                                                                                                                                             PARTIALS DF/DQ
PARTIALS DF/DXO
PACTIONS TUB
SCRATCH COLUMN VECTUR
SCRATCH 200-20 HAIRIX
SCRATCH 200-20 HAIRIX
                                                                                                                                                2
20
400
400
FURTRAN CARD 24
12*15
   57247
                                                                                                                                                                                                                                                                               NOT ZERU TU IGNURE DATA POINT
SUM OF SQUARED RESIDUALS
INPUT BUFFER
DP INPUT BUFFER
                                               1GNAZ CORHON
SOUR CORHON
BUZZ5 CURHON
BUZZ CURHON
   56427
56143
56141
55600
                                                                                                                                                  2
15+15
12
FORTRAN CARD 25
                                                 XN COHHON
DXN COMHON
DDXN CUMHON
TODXN CUMHON
IS COMHON
TL COMHON
GSG COMHON
GSG COMHON
   55564
55561
55556
55553
55550
55547
55107
54447
54441
                                                                                                                                                                                                                                                                                CORPECTED PROBE COGRDINATES
                                                                                                                                                                                                                                                                             DATA SAMPLE INTERVAL
S==2 MEIGHTING TABLE
I MEIGHTING TABLE
G MEIGHTING TABLE
NOT ZERO IF DATA TYPE REJECTED LASI
                                                                                                                                                   12*6*4
12*6*4
                                                                                                                                                6
12
FORTRAN CARD 26
20
20
                                                  OHBAD COMMON
OTIC COMMON
   $4425
54401
54355
54303
54303
54300
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                                                                                                                                                                                                                                                                                ESTIMATED CONDITIONS AND CONSTANTS
                                                                                                                                                                                                                                                                                DOPPLER COEFFICIENTS
                                                                                                                                                INITIAL POINTING TIMES POINTING COUNT INTERVALS FINAL POINTING TIMES
                                                                                                                                                                                                                                                                                OCCULTATION-IMPACT TIMES (30 MUROS)
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                                                                                                                                                                                                                                                                                SCRATCH 20.20 MATRIX
SCRATCH 20.20 MATRIX
V HATRIX OF PARTIALS
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                                                                                                                                                   12-15
12-15
                                                                                                                                                                                                                                                                               TRANSHITTER ID TABLE
HAPPING TIMES
S/C AREA
S/C NASS
PHYSICAL CONSTANTS INTEGRATION HATRIX
HAPPING MATRIX TO ENCOUNTER
                                                                                                                                                  6+45
24
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```

### V. OUTPUT

Primary output from the ODP is in the form of a BCD tape which may be listed on the IBM 1401. This listing contains the following items:

- 1. List of card input
- 2. A priori covariance matrix
- 3. Inverse of a priori covariance matrix
- 4. Trajectory information
- 5. J matrix
- 6. J matrix correlations
- 7. Parameter estimates and statistics
- 8. Inverse of J matrix
- 9. Covariance matrix
- 10. Covariance matrix correlations

The following items are output on option:

- Residuals, computed observables, weights, and frequencies
- 2. Residual statistics
- 3. Pointing ephemeris
- 4. Mapping matrix
- 5. Mapped covariance matrix
- 6. Position and velocity of probe at mapping times
- 7. Encounter parameters and statistics
- 8. Supplementary printout at data times--vectors and partials All items, with the exception of the supplementary partials printout, may be printed on the SC-3070 in SFOF Mode II operation.

Any matrix which is printed by the ODP may also be punched on option. In this case the card images are written on the same BCD output tape for punching by the 1401.

The ODP also writes a plotting tape for the SC-4020 plotter, containing plots of residuals versus time, and angle residuals versus angles.

A binary tape for the Matrix Manipulation Program, containing data partials and mapping matrices, is generated on option. This capability does not exist in Mode II operation.

Error comments and status-of-program messages are printed on the administrative printer in Mode II and the on-line printer in Mode IV. Refer to Subroutine 28, ERROR, for a listing of these comments.

### A. DISK AND TAPE FORMATS

The Planetary Ephemeris format is presented in Ref. 4, and the Matrix Manipulation Program tape format in Ref. 11 (Section IX). The other formats are covered in Section VI; the Prediction File is presented in Subroutine 30, FILL; the Data File in Subroutine 43, KINE; the Probe Ephemeris in Subroutine 45, LOOKUP; and the Combined Ephemeris in Subroutine 86, RITEM.

## B. OUTPUT SYMBOLS AND DEFINITIONS

The following symbols appear on the output listings. The corresponding symbols used in this report and the definitions are given:

Output symbol	Report symbol	Definition	
X	* )		
Y	у	position of probe, geocentric equatorial true of date, km	
Z	z )	-	
DX	× )		
DY	ý }	velocity of probe, geocentric equatorial true of date,	
DZ	ż)	km/sec	
KE	GM <sub>e</sub>	gravitational constant of Earth, km <sup>3</sup> /sec <sup>2</sup>	
RE	$R_{ m e}$	equatorial radius of Earth, km	
G	γB	solar pressure constant	
KM	$^{ m GM}_{ m m}$	gravitational constant of Moon, km <sup>3</sup> /sec <sup>2</sup>	
MV	${f M}_{f v}$	mass of Venus, solar masses	
MM	${f M}_{f r}$	mass of Mars, solar masses	

Output symbol	Report symbol	Definition		
MJ	${f M}_{f j}$	mass of Jupiter, solar masses		
J	J	second harmonic coefficient		
Н	Н	third harmonic coefficient		
D	D	fourth harmonic coefficient		
AU	a <sub>e</sub>	astronomical unit, km		
FA	$F_{ao}$	attitude		
FB	$^{ ext{F}}_{ ext{bo}}  angle$	control		
FC	F <sub>co</sub>	coefficients		
С	c	speed of light, km/sec		
RI(I)	$\mathtt{R}_{\mathbf{i}}$	radius of Earth at station i, km		
LA(I)	$\boldsymbol{\phi}_{\mathtt{i}}$	geocentric latitude of station i, deg		
LO(I)	$\lambda_{\mathbf{i}}$	longitude of station i, deg		
R	ρ	slant range, km		
DR	ρ̈́	slant range rate, km/sec		
EL	γ	elevation angle, deg		
AZ	σ	azimuth angle, deg		
DEC	δ	declination, deg		
HA	a	hour angle, deg		
Cl	$^{\mathrm{f}}$ 1	one-way integrated doppler frequency, cps		
CC3	<sup>f</sup> c3	coherent three-way integrated doppler frequency, cps		
C3	f <sub>3</sub>	three-way integrated doppler frequency, cps		
D1	<sup>f</sup> dl	differenced one-way integrated doppler frequency, cps		

Output symbol	Report symbol	Definition
RU	ρDSIF	DSIF ranging
В	B	vector from target center of mass perpendicular to probe asymptote, km
B∙ RO	$\left.\begin{array}{c} \underline{\mathbf{B}} \cdot \underline{\mathbf{R}}_{0} \\ \underline{\mathbf{B}} \cdot \underline{\mathbf{T}}_{0} \end{array}\right\}$	dot products, target orbital plane, km
В∙ТО	B·T <sub>0</sub>	prane, kin
B.RT	$\underline{\mathbf{B}} \cdot \underline{\mathbf{R}}_{\mathbf{T}}$	dot products, target
$\mathtt{B}\boldsymbol{\cdot}\mathtt{TT}$	$\left. \begin{array}{c} \underline{\mathbf{B}} \cdot \underline{\mathbf{R}}_{\mathbf{T}} \\ \underline{\mathbf{B}} \cdot \underline{\mathbf{T}}_{\mathbf{T}} \end{array} \right\}$	equatorial plane, km
${ m TL}$	$^{ m t}_{ m L}$	linearized time of flight, hours (lunar missions) or days
TF	$t_f$	true time of flight, hours (lunar missions) or days
SMAA	a	semi-major axis of dispersion ellipse at target, km
SMIA	b	semi-minor axis, km
THETA	θ	inclination of dispersion ellipse to target orbital plane, deg
DEL T	$\sigma_{t}$	standard deviation of linearized time of flight, sec
DEL B	$^{\sigma}$ B	standard deviation of $\underline{B}$ vector, km
DEL S	${}^{\sigma}{}_{ m S}$	standard deviation of asymptote unit vector, km
DEL BR	$^{\sigma}$ B $_{ullet}$ R	standard deviation of $\underline{B} \cdot \underline{R}$ , km
DEL BT	$\sigma_{\mathbf{B} \cdot \mathbf{T}}$	standard deviation of $\underline{B} \cdot \underline{T}$ , km
C3	c <sub>3</sub>	vis viva energy, $km^2/sec^2$
TC	au	doppler count time, sec
Q	q	transmitter index
FRQ	$\left\{egin{array}{c} \mathbf{f}_{\mathbf{q}} \end{array}\right.$	if q $\neq$ 0, station frequency less 29.66 x 10 <sup>6</sup> cps
<del>-</del> -	(f <sub>T</sub>	if q = 0, probe frequency less 960 x 106 cps
	46	

# VI. SUBROUTINES

Section VI contains the documentation for the 104 Subroutines which were written primarily for use by the SPODP. Documentation is not included for JPL general-purpose routines and SFOF routines.

1

## IDENTIFICATION

AAT

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

This subroutine forms the matrix product of two vectors.

# RESTRICTIONS

Standard Fortran II arrays.

### USE

CALL AAT (X, Y, PRO)

X (1,6) F

Y (6, 1) P<sup>T</sup>

PRO (6, 6) Product matrix

## CODING INFORMATION

Length of subroutine is 77 (10) or 115 (8) words.

ABDD

Charles Coltharp, JPL IBM 7094 Fap

January 4, 1965

#### PURPOSE

ABDD computes the function.

$$S = \frac{Z_{i}(ax + by) - cz}{d},$$

the form of which has frequent application in ODP station location partial expressions.

#### RESTRICTIONS

COMMON break: 47675

USE

CALL ABDD

PZE Ď d

PZE x, y,  $z(1 \times 3)$ R

COMMON input:

0 a .

В b

С

Z - component of geocentric station vector, km.

COMMON output:

s s

# CODING INFORMATION

Length of subroutine is 24 (10) or 30 (8) words.

ACE

Melba Nead, JPL Fortran II, Version 3 January 4, 1965

### PURPOSE

To compute the unit vector as needed in calculation of the attitude control effects partials in BMATRX.

## RESTRICTIONS

- a. COMMON from ODP
- b. Subroutines used: SQRT

\_\_\_\_

USE

CALL ACE (DDX)

Computes R, X/R etc, and stores the result in DDX(I).

# CODING INFORMATION

Length of subroutine is 57 (10) or 71 (8) words.

ANGPLT

Alfred Schoepke, JPL

IBM 7094 Fap

January 4, 1965

## PURPOSE

ANGPLT plots hour angle residuals versus hour angle, and declination residuals versus declination. A tape containing the plotting information is written for the SC-4020.

## RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutine used:

PKINE

FIXT

BIBCD

CXPLOT

### USE

CALL ANGPLT

COMMON input:

IR station identification

RVECD residual array

RVEC data type array

# CODING INFORMATION

Length of subroutine is 383 (10) or 577 (8) words.

5

AQUI/ALLI/BAQUI/BALLI

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

To format, list, and send, via TELTYP, predictions needed by the tracking stations.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

PROUT

**FGDOUT** 

PKINE

FIXT

TELTYP

ENDOUT

BIBCD

In MODE 2, the predictions are transmitted in real time by TELTYP. In MODE 4, a magnetic tape (A9) is written by TELTYP. This tape is then processed to punch paper tape acceptable to the teleprinters.

### USE

CALL	AQUI	to list predictions
CALL	BAQUI	to prepare for transmission
CALL	ALLI	to list drive tape for Goldstone
CALL	BALLI	to prepare drive tape for transmission

## CODING INFORMATION

Length of subroutine is 460 (10) or 714 (8) words.

### REFERENCE

Heller, J., IOM to F. Curl, Nominal Predict Format, June 10, 1963.

BEER

Michael R. Warner, JPL IBM 7094 Fap

January 4, 1965

### PURPOSE

BEER stores zeroes in certain ODP common locations prior to the data fit. It also executes the reset entry of REJEC.

## RESTRICTIONS

- a. COMMON break: 46711
- b. Subroutine called:

REJEC

#### USE

CALL BEER

COMMON output: zeroes in DINT, DIN2, QA, QB, TOBL, TOBL-1, TGINT, TGINT-1, XMU, RIGHT, RITE2, XX, XC, XJ, XJ2, XK, XK2, IF2, IF4. BIGNO in IF1 and IF5.

### CODING INFORMATION

Length of subroutines is 57 (10) or 71 (8) words.

7

BIBCD

Melba W. Nead, JPL IBM 7094 Fap

January 4, 1965

## PURPOSE

To convert the binary equivalent of 2 digit decimal numbers to BCD for print-out.

# RESTRICTION

Range of numbers:  $1 \le n \le 99$ 

# USE

With argument in the accumulator

CALL BIBCD

The BCD equivalent is in the accumulator on return.

## CODING INFORMATION

Length of subroutine is 45 (10) or 55 (8) words.

IDENTIFICATION 8-1 of 4

**BMATRX** 

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

BMATRX is written to compute the analytic partials of the physical constants when these parameters are being estimated or considered in the ODP.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. When computing partials of the attitude control effects, partials of MJ, H, and D are inhibited, thus a maximum of 11 of the possible 14 can be computed at one time.
- c. Subroutines used:

ACE

GIG2

RCAL

RCOM

RSIG

SQRT

#### METHOD

Formulas for the partial derivatives of acceleration with respect to physical constants,  $\partial \hat{r}/\partial q$ , are derived from the equations of motion. These are integrated to obtain  $\partial r/\partial q$ .

The equations for these partials follow:

$$\frac{\partial \dot{x}}{\partial GM_e} = \frac{-xg_1}{r^3}$$

$$\frac{\partial \ddot{y}}{\partial GM_e} = \frac{-yg_1}{r^3}$$

$$\frac{\partial \dot{z}}{\partial GM_e} = \frac{-zg_2}{r^3}$$

$$\frac{\partial \ddot{\mathbf{x}}}{\partial R_{e}} = \frac{-GM_{\mathbf{m}}}{R_{e}} \left\{ 2\mathbf{x}_{\mathbf{m}} \left[ \frac{1}{\left|\underline{\mathbf{r}}_{\mathbf{m}} - \underline{\mathbf{r}}\right|^{3}} - \frac{1}{\mathbf{r}_{\mathbf{m}}^{3}} \right] - 3 \frac{\left(\underline{\mathbf{r}} \cdot \underline{\mathbf{r}}_{\mathbf{m}} - \mathbf{r}^{2}\right) \left(\mathbf{x} - \mathbf{x}_{\mathbf{m}}\right)}{\left|\underline{\mathbf{r}}_{\mathbf{m}} - \underline{\mathbf{r}}\right|^{5}} - \frac{3\dot{\mathbf{x}}}{\left|\underline{\mathbf{r}}_{\mathbf{m}} - \underline{\mathbf{r}}\right|^{3}} \right\} \qquad \mathbf{x} \to \mathbf{y}, \mathbf{z}$$

$$\frac{\partial \ddot{\mathbf{x}}}{\partial \Upsilon \mathbf{B}} = \frac{c_1 \frac{\mathbf{A}_p (\mathbf{x} - \mathbf{x}_s)}{\mathbf{M}_p |\mathbf{r} - \mathbf{r}_s|^3}}{\mathbf{x} \to \mathbf{y}, \mathbf{z}}$$

where

$$c_1 = 1.031 \times 10^8$$
 $A_p = \text{surface area of probe, m}^2$ 
 $M_p = \text{mass of probe, kg}$ 

$$\frac{\partial x}{\partial GM_{m}} = -\left[\frac{x_{m}}{r_{m}} + \frac{(x - x_{m})}{\left|\underline{r_{m}} - \underline{r}\right|^{3}}\right] \qquad x \to y, z$$

$$\frac{\partial \ddot{x}}{\partial M_{v}} = -GM_{s} \left[ \frac{x_{v}}{r_{v}^{3}} + \frac{(x - x_{v})}{\left| \underline{r}_{v} - \underline{r} \right|^{3}} \right] \qquad x \to y, z$$

$$\frac{\partial \ddot{x}}{\partial M_{r}} = -GM_{s} \left[ \frac{x_{r}}{r_{r}^{3}} + \frac{(x - x_{r})}{\left| \underline{r}_{r} - \underline{r} \right|^{3}} \right] \qquad x \to y, z$$

$$\frac{\partial \dot{x}}{\partial M_{j}} = -GM_{s} \left[ \frac{x_{j}}{r_{j}^{3}} + \frac{(x - x_{j})}{\left| \underline{r_{j}} - \underline{r} \right|^{3}} \right] \qquad x \to y, z$$

$$\frac{\partial \dot{x}}{\partial J} = -x \left[ 1 - 5 \left( \frac{z}{r} \right)^2 \right] \frac{GM_e}{r^2} \frac{R_e^2}{r^3}$$
  $x \to y$ 

$$\frac{\partial \mathbf{z}}{\partial \mathbf{J}} = -\mathbf{z} \left[ 3 - 5 \left( \frac{\mathbf{z}}{\mathbf{r}} \right)^2 \right] \frac{GM_e}{r^2} \frac{R_e^2}{r^3}$$

$$\frac{\partial \dot{x}}{\partial H} = x \left[ 7 \left( \frac{z}{r} \right)^2 - 3 \right] \frac{z}{r} \frac{GM_e}{r^3} \frac{R_e^3}{r^3}$$
  $x \to y$ 

$$\frac{\partial z}{\partial H} = \left[1 - 10\left(\frac{z}{r}\right)^2 + \frac{35}{3}\left(\frac{z}{r}\right)^4\right] \frac{3r}{5} \frac{GM_e}{r^3} \frac{R_e^3}{r^3}$$

$$\frac{\partial x}{\partial D} = -x \left[ 3 - 42 \left( \frac{z}{r} \right)^2 + 63 \left( \frac{z}{r} \right)^4 \right] \frac{GM_e}{7r^3} \frac{R_e^4}{r^4} \qquad x \to y$$

$$\frac{\partial \mathbf{z}}{\partial \mathbf{D}} = -\mathbf{z} \left[ 15 - 70 \left( \frac{\mathbf{z}}{\mathbf{r}} \right)^2 + 63 \left( \frac{\mathbf{z}}{\mathbf{r}} \right)^4 \right] \frac{GM_e}{7r^3} \frac{R_e^4}{r^4}$$
8-3 of 4

$$\frac{\partial \ddot{\mathbf{x}}}{\partial a_e} = \frac{1}{a_e} \sum_{k=1}^4 \mathrm{GM}_k \left[ \mathbf{x}_k \left( \frac{1}{\left|\underline{\mathbf{r}}_k - \underline{\mathbf{r}}\right|^3} - \frac{1}{\mathbf{r}_k^3} \right) - \frac{3\left(\mathbf{r}^2 - \underline{\mathbf{r}} \cdot \underline{\mathbf{r}}_k\right) \left(\mathbf{x} - \mathbf{x}_k\right)}{\left|\underline{\mathbf{r}}_k - \underline{\mathbf{r}}\right|^5} \right] \qquad \quad \mathbf{x} \to \mathbf{y}, \, \mathbf{z}$$

where

The partial derivatives of acceleration with respect to the attitude control effects are given by

$$\begin{bmatrix} \frac{\partial \ddot{x}}{\partial F_{AO}} & \frac{\partial \ddot{x}}{\partial F_{BO}} & \frac{\partial \ddot{x}}{\partial F_{CO}} \\ \\ \frac{\partial \ddot{v}}{\partial F_{AO}} & \vdots \end{bmatrix} = \frac{10^{-8}}{m} \begin{bmatrix} A_{x} & B_{x} & C_{x} \\ A_{y} & B_{y} & C_{y} \\ \\ A_{z} & B_{z} & C_{z} \end{bmatrix}$$

where the unit vectors along the three body fixed axes are defined by

$$\overline{A} = \frac{-\overline{H}(\overline{E} \cdot \overline{H}) + \overline{E}}{|\overline{H}(\overline{E} \cdot \overline{H}) - \overline{E}|} = A_x, A_y, A_z$$

$$\overline{B} = \frac{\overline{E} \times \overline{H}}{\left| \overline{E} \times \overline{H} \right|} = B_{x}, B_{y}, B_{z}$$

$$\overline{C} = -\overline{H} = C_x, C_y, C_z$$

and

$$\overline{E} = E_{x}, E_{y}, E_{z}$$
 = Unit vector directed from  
the probe toward Canopus

$$\overline{H} = H_x, H_y, H_z =$$
Unit vector directed from the probe toward the Sun

USE 8-4 of 4

## CALL BMATRX

The subroutine checks lists in the COMMON area to determine what is required.

## CODING INFORMATION

Length of subroutine is 794(10) or 1432(8) words.

## REFERENCES

- a. Anderson, John D., RFP 312-37, August 29, 1961.
- b. Null, George W., RFP 312-179, Addendum 3, December 11, 1963.

IDENTIFICATION 9-1 of 2

CALL

Michael R. Warner, JPL Fortran II, Version 3

January 4, 1965

### PURPOSE

CALL serves as the logical control routine for the data fitting link, LA6. All steps from reading the ODG data file through computation of partials and residuals to writing the residual file are initiated by this routine.

### RESTRICTIONS

a. COMMON break: 46711

b. Subroutines called:

BEER

KINE

COEF

STARP

(DFSB)

REAP

CHADP

DOPLR

COROP

CORR

DICOS

OBTOX

CATS

**OFFSYS** 

WAIT

QUIZ

REJEC

PM360

FORM

GERTA/GREOF

FILL

#### USE

9-2 of 2

CALL CALL

## COMMON input:

IGNAZ data type delete flags (12 × 15)

PTFD-30 data start and stop times

LLIST physical constants solve flags (11)

EPOCH ODP epoch, d.p. sec past 1950.0

THETAI station longitude, deg

SIGX3 rejection sigmas  $(12 \times 15)$ 

DXDR-27 occultation-impact data flag

## COMMON output:

HAO observed hour angle from observed optical right ascension

RVECD residual array

TOBL previous observation time

## CODING INFORMATION

Length of subroutine is 339 (10) or 528 (8) words.

IDENTIFICATION 10-1 of 5

CATS

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

CATS computes the partials of the ODP data types with respect to the station locations. Such a partial may be non-zero only if the tracking datum is from the same station as the specified station location.

### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines called:

VEC

ABDD

COS

### **METHOD**

$$\begin{array}{l} X_{i} = R_{i} \cos(\alpha_{G} + \lambda_{i}) \cos \phi_{i} \\ \\ Y_{i} = R_{i} \sin(\alpha_{G} + \lambda_{i}) \cos \phi_{i} \\ \\ Z_{i} = R_{i} \sin \phi_{i} \\ \\ \\ \frac{\partial \rho_{i}}{\partial R_{i}} = \frac{1}{R_{i}} \left(\underline{R}_{i} \cdot \underline{L}\right) \\ \\ \frac{\partial \rho_{i}}{\partial \phi_{i}} = Z_{i} \left\{ \left[\underline{L}_{x} \cos(\alpha_{G} + \lambda_{i}) + \underline{L}_{y} \sin(\alpha_{G} + \lambda_{i})\right] - \underline{L}_{z} R_{i} \cos \phi_{i} \right\} \\ \\ \frac{\partial \rho_{i}}{\partial \lambda_{i}} = Y_{i} \underline{L}_{x} - X_{i} \underline{L}_{y} \\ \end{array}$$
 (slant range partials)

where

10-2 of 5

$$\underline{\mathbf{L}} = \frac{\underline{\rho} + \underline{\mathbf{r}}_{\mathbf{S}}}{\rho}$$

 $\lambda_i$  = station longitude

 $\phi_{i}$  = station geocentric latitude

R; = station Earth radius

$$\frac{\partial \dot{\hat{\rho}}_{i}}{\partial R_{i}} = \frac{1}{R_{i}} \left( \underline{R}_{i} \cdot \underline{L}' \right)$$

$$\frac{\partial \dot{\hat{\rho}}_{i}}{\partial \phi_{i}} = -Z_{i} \left\{ \left[ \underline{L}_{x}' \cos(\alpha_{G} + \lambda_{i}) + \underline{L}_{y}' \sin(\alpha_{G} + \lambda_{i}) \right] + \underline{L}_{z}' R_{i} \cos \phi_{i} \right\}$$

$$\frac{\partial \dot{\hat{\rho}}_{i}}{\partial \lambda_{i}} = -Y_{i} \underline{L}_{x}' + X_{i} \underline{L}_{y}'$$

$$\left\{ (range rate partials) \right\}$$

where

$$L_{x}' = \frac{1}{\rho_{i}} \left( -\dot{x} - 2\omega Y_{i} + \dot{\rho}_{i} L_{x} + \omega y \right)$$

$$L_{y}^{i} = \frac{1}{\rho_{i}} \left( -\dot{y} + 2\omega X_{i} + \dot{\rho}_{i} L_{y} - \omega x \right)$$

$$L_{z}^{\prime} = \frac{1}{\rho_{i}} \left( -\dot{z} + \rho_{i} L_{z} \right)$$

$$\begin{split} \frac{\partial \gamma_{i}}{\partial R_{i}} &= \frac{-1}{R_{i}^{\rho} \rho_{i}} \left( \underline{R}_{i} \cdot \widetilde{\underline{D}} \right) \\ \frac{\partial \gamma_{i}}{\partial \phi_{i}} &= \frac{1}{\rho_{i}} \left\{ Z_{i} \left[ \widetilde{D}_{x} \cos(\alpha_{G} + \lambda_{i}) + \widetilde{D}_{y} \sin(\alpha_{G} + \lambda_{i}) \right] - \widetilde{D}_{z} R_{i} \cos \phi_{i} \right\} \end{split}$$
 (elevation angle partials) 
$$\frac{\partial \gamma_{i}}{\partial \lambda_{i}} &= \frac{1}{\rho_{i}} \left( Y_{i} \widetilde{D}_{x} - X_{i} \widetilde{D}_{y} \right) \end{split}$$

where

$$D_x = -\sin \delta_i \cos(\alpha_G + \lambda_i)$$

$$D_y = -\sin \delta_i \sin(\alpha_G + \lambda_i)$$

$$D_z = \cos \delta_i$$

$$A_x = -\sin(\alpha_G + \lambda_i)$$

$$A_y = \cos(\alpha_G + \lambda_i)$$

$$A_z = 0$$

$$\widetilde{D}_{x} = \sin \gamma_{i} \left[ \sin \sigma_{i} \sin(\alpha_{G} + \lambda_{i}) + \cos \sigma_{i} \sin \phi_{i} \cos(\alpha_{G} + \lambda_{i}) \right]$$

$$+ \cos \phi_{i} \cos(\alpha_{G} + \lambda_{i}) \cos \gamma_{i}$$

$$\widetilde{D}_{y} = -\sin \gamma_{i} \left[ \sin \sigma_{i} \cos(\alpha_{G} + \lambda_{i}) - \cos \sigma_{i} \sin \phi_{i} \sin(\alpha_{G} + \lambda_{i}) \right]$$

$$+ \cos \phi_{i} \sin(\alpha_{G} + \lambda_{i}) \cos \gamma_{i}$$

$$\widetilde{D}_{\mathbf{z}} = -\cos\sigma_{\mathbf{i}}\cos\phi_{\mathbf{i}}\sin\gamma_{\mathbf{i}} + \sin\phi_{\mathbf{i}}\cos\gamma_{\mathbf{i}}$$

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$$\widetilde{A}_{x} = -\cos \sigma_{i} \sin(\alpha_{G} + \lambda_{i}) + \sin \sigma_{i} \sin \phi_{i} \cos(\alpha_{G} + \lambda_{i})$$

$$\widetilde{A}_{y} = \cos \sigma_{i} \cos(\alpha_{G} + \lambda_{i}) + \sin \sigma_{i} \sin \phi_{i} \sin(\alpha_{G} + \lambda_{i})$$

$$\widetilde{A}_{z} = -\sin \sigma_{i} \cos \phi_{i}$$

The vectors  $\underline{A}$ ,  $\underline{D}$ ,  $\underline{\widetilde{A}}$ , and  $\underline{\widetilde{D}}$ , defined above, are computed by subroutine DICOS.

$$\frac{\partial f_{1i}}{\partial S_i} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_i}{\partial S_i}$$
 (one-way doppler partials)

$$\frac{\partial f_{c3i, q}}{\partial S_i} = \frac{\Omega_4}{c} \frac{\partial \dot{\rho}_i}{\partial S_i}$$
 (two-way doppler receiver partials)

$$\frac{\partial f_{3i, q}}{\partial S_i} = \frac{\Omega_6}{c} \frac{\partial \dot{\rho}_i}{\partial S_i}$$
 (three-way doppler receiver partials)

$$\frac{\partial f_{dli,j}}{\partial S_i} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_i}{\partial S_i}$$
 (differenced one-way doppler receiver i partials)

$$\frac{\partial f_{c3i, q}}{\partial S_{q}} = \frac{\Omega_{4}}{c} \frac{\partial \rho_{q}}{\partial S_{q}}$$
 (two-way doppler transmitter partials)

$$\frac{\partial f_{3i, q}}{\partial S_q} = \frac{\Omega_6}{c} \frac{\partial \dot{\rho}_q}{\partial S_q} \quad \text{(three-way doppler transmitter partials)}$$

$$\frac{\partial f_{dli,j}}{\partial S_j} = \frac{\Omega_2}{c} \frac{\partial \dot{\rho}_j}{\partial S_j}$$
 (differenced one-way doppler receiver j partials)

where

 $\Omega_2$  = one-way multiplier

 $\Omega_4$  = two-way multiplier

 $\Omega_{6}$  = three-way multiplier

USE

10-5 of 5

CALL CATS

# COMMON inputs:

RI 
$$R_i$$
, km

PHII  $\phi_i$ , deg

THETAI  $\lambda_i$ , deg

GHA  $\alpha_G$ , deg

ELX  $L$ 

AX  $A$ 

DX  $D$ 

TAX  $A$ 

TDX  $D$ 

RVEC  $\rho_i$ , km

OMEGA  $\omega$ , deg/sec

X  $I$ , km

XDOT  $\dot{I}$ , km/sec

CZ2  $\Omega_2$ 

CZ4  $\Omega_4$ 

CZ6  $\Omega_6$ 

COMMON output:

DIM3 partials  $\frac{\partial F}{\partial S_i}$  (12 × 3 × 15)

# CODING INFORMATION

Length of subroutine is 421 (10) or 645 (8) words.

# REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM33-168, March 18, 1964.
- b. Anderson, John D., RFP 312-37, August 29, 1961.

IDENTIFICATION 11-1 of 5

COEF

Michael R. Warner, JPL Fortran II, Version 3

January 4, 1965

## PURPOSE

COEF calculates the coefficients used by the ODP in doppler calculations. The subroutine is cognizant of band differences. All intermediate calculations are in double precision.

## RESTRICTIONS

COMMON break: 47675

Subroutines used:

(DFAD)

(DFMP)

(DFSB)

QUIZ

### METHOD

$$f_T = f_{T \text{ base}} + \Delta f_T$$
 (transponder frequency)

$$f_q = f_{q \text{ base}} + \Delta f_{q}$$
 (transmitter frequency)

$$\Omega_2 = \omega_2 [f_T + D(t_{ob} - t_0)]$$

$$\Omega_1 = \omega_1 + \Delta \omega_1 - \Omega_2$$

$$\Omega_4 = \omega_4 f_q$$

$$\Omega_3 = \omega_3 + \Delta \omega_3$$
 for predicts  $\Omega_3 = 0$  for data fitting

$$\Omega_6 = \omega_6 f_q$$

$$\Omega_5$$
 =  $\omega_5$  +  $\Delta \omega_5$  -  $\Omega_6$  for predicts  $\Omega_5$  = 0 for data fitting

where for L to S band,

11-2 of 5

$$f_{T base} = 2290.0 \times 10^6 \text{ cps}$$
 $f_{q base} = 20.00 \times 10^6 \text{ cps}$ 
 $k_1 = f_{q base} + \Delta k_1$ 
 $\omega_2 = 0.3125$ 
 $\omega_1 = 9.375 \times 10^6 \text{ cps}$ 
 $\omega_4 = 32.579185520$ 
 $\omega_3 = 9.375 \times 10^6 \text{ cps}$ 
 $\omega_6 = 32.579185520$ 
 $\omega_5 = 9.375 \times 10^6 \text{ cps}$ 
 $\Delta \omega_1 = 30 \text{ k}_1$ 
 $\Delta \omega_3 = 30 \text{ k}_1$ 
 $\Delta \omega_5 = 30 \text{ k}_1$ 

for L band,

$$f_{\text{T base}} = 960.0 \times 10^6 \text{ cps}$$
 $f_{\text{q base}} = 29.66 \times 10^6 \text{ cps}$ 
 $\omega_2 = 0.96875$ 
 $\omega_1 = 930.15 \times 10^6 \text{ cps}$ 

$$\omega_4 = 32.359550561$$

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$$\omega_3 = 100000$$
, cps

$$\omega_6 = 31.348314605$$

$$\omega_5 = 930.15 \times 10^6 \text{ cps}$$

$$\Delta\omega_1 = \Delta\omega_3 = \Delta\omega_5 = 0$$

for S band,

$$f_{\text{T base}} = 2290.0 \times 10^6 \text{ cps}$$

$$f_{q \text{ base}} = 20.00 \times 10^6 \text{ cps}$$

$$k_s = f_{q \text{ base}} + \Delta k_s$$

$$\omega_2 = 1.0$$

$$\omega_{1} = 1000000.$$

$$\omega_4 = 104.25339367$$

$$\omega_3 = 1000000$$
.

$$\omega_6 = 104.25339367$$

$$\omega_5 = 1000000$$
.

$$\Delta \omega_1 = \omega_6 k_s$$

$$\Delta\omega_3 = 0$$

$$\Delta \omega_5 =$$

```
11-4 of 5
  also
                                  D = transponder (beacon) drift
                                t<sub>ob</sub> = observation time
                                 t_0 = ODP epoch
USE
   CALL
           COEF
   COMMON input:
      EP-6
                   band indication
      EP-7
                    \Delta k_1, cps
      EP-8
                    \Delta k_s, cps
                   6 \times 15 array of \omega_1^{},\;\omega_2^{},\;\ldots,\;\omega_6^{} coefficients for each station, d.p.
      TXCON
                    \Delta f_T, cps
      TFREQ
                    \Delta f_{q}, cps
      XFREQ
      DXDR-4
                   D, cps/sec
                   observation time, d.p. sec past 1950.0
      TOB
                   ODP epoch, d.p. sec past 1950.0
      EPOCH
                   observed one-way doppler (flag)
      C10
                   observed differenced one-way doppler (flag)
      CDIO
      CC3O
                   observed two-way doppler (flag)
      C30
                   observed three-way doppler (flag)
                   observed DSIF ranging (flag)
      CD3O
   COMMON output:
      CZ1
                   \Omega_1
      CZ2
                   \Omega_2
      CZ3
                   \Omega_3
                   \Omega_4
      CZ4
```

 $\Omega_{5}$ 

 $\Omega_5$ , d.p.

 $\Omega_6$ , d. p.

CZ5 CZ6 RNJ

PHINJ

# CODING INFORMATION

11-5 of 5

Length of subroutine is 503 (10) or 767 (8) words.

## REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Wollenhaupt, Wilber R., RFP 312-268, April 16, 1964.
- c. Wollenhaupt, Wilber R., RFP 312-319, September 3, 1964.

IDENTIFICATION 12-1 of 3

COL

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

COL calculates the G<sup>2</sup> terms in the weighting function

$$\sigma_{j}^{2} = \sum_{p=1}^{6} S_{pjk}^{2} G_{pj}^{2} \max \left(\frac{t_{pjk}}{\tau}, 1\right)$$

## RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

COS

SQRT

## **METHOD**

The computation of the g value depends on the value of the weight code group (p) and the data type (j). The following table indicates the g term for each of the p, j combinations:

72

12-2 of 3

12-3 of 3 where

```
P = slant range
  \dot{\rho} = range rate
  \gamma = elevation angle
   \delta = declination
\Delta_{r}\rho = refraction correction to slant range
\Delta_r \dot{\rho} = refraction correction to range rate
\Delta_r \delta = refraction correction to declination
\Delta_r a = refraction correction to hour angle
\Omega_2 = one-way doppler multiplier
\Omega_4 = two-way doppler multiplier
\Omega_6 = three-way doppler multiplier
  \tau = doppler averaging time
  c = velocity of light
```

### USE

```
CALL
      COL
```

RVEC

# COMMON input:

RDOT è, km/sec ELγ, deg DEC δ, deg DRVEC  $\Delta_{r^{\rho}}$ , km  $\Delta_r \hat{\rho}$ , km/sec DRDOT  $\Delta_{\mathbf{r}}\delta$  ,  $\deg$ DRDEC DRHA Δ<sub>a</sub>, deg CZ2  $\Omega_2$ CZ4  $\Omega_4$ CZ6  $\Omega_6$ TAU  $\tau$ , sec VELC c, km/sec COMMON output:  $g^2 (1 \times 6)$ 

 $\rho$ , km

# CODING INFORMATION

GSQ

Length of subroutine is 262 (10) or 406 (8) words

COMAP

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

### PURPOSE

COMAP checks the mapping requests which are made of the ODP. When step-mapping the product of the U matrices for mapping is computed, thus the U matrix which is used in the calculation of the closest approach parameters is:

$$\mathbf{U} = \mathbf{U}_{\mathrm{IMP}} * \mathbf{U}_{\mathrm{T}_{\mathrm{N}}} * \dots * \mathbf{U}_{\mathrm{T2}} * \mathbf{U}_{\mathrm{TI}}$$

where  $U_{TN}$  are the matrices computed for each step-map time and  $U_{IMP}$  is evaluated at time of closest approach. Results of the mapping matrix calculations are stored on disk to be printed in a subsequent link.

# RESTRICTIONS

a. ERROR condition: disk error indicated by DCr

b. COMMON break: 47055

c. Subroutines used:

TAMP

PROM

REAM

COMIMP

FLAK

FLOT

DCP

**OFFSYS** 

TYPRYT

# USE

This link is called under control of the JPTRAJ Source Deck.

### CODING INFORMATION

Length of subroutine is 186(10) or 272(8) words.

IDENTIFICATION 14-1 of 2

COMIMP

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

### PURPOSE

To project the statistical information contained in the covariance matrix  $\Gamma(t_0)$  to any later time.

### RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used:

MAMUL

PRAMS

(DFAD)

(DFSB)

(DFMP)

### METHOD

The procedure employs a mapping matrix

$$Ut_0, t = \frac{\partial Q_t}{\partial Q_{t_0}}$$

which is an extended matrix of variational partials representing all estimated parameters  $Q_{t_0}$ . If the parameter set  $Q_{t_0}$  consists only of the six initial conditions, then this matrix is identical to the familiar U matrix. A similar mapping matrix is employed for including the effect of the considered parameter set  $\widetilde{Q}_{t_0}$ :

$$Vt_0$$
,  $t = \frac{\partial Q_t}{\partial \widetilde{Q}_{t_0}}$ 

The mapping operation is then accomplished by

$$\Gamma_{t} = U\Gamma_{t_{0}}U^{T} - V\Gamma_{\widetilde{Q}}V^{T}J^{-1}U^{T} - UJ^{-1}V\Gamma_{\widetilde{Q}}V^{T} + V\Gamma_{\widetilde{Q}}V^{T}$$

If no parameters are being considered, this reduces to

14-2 of 2

$$\Gamma_{t} = U\Gamma_{t_0}U^{T}$$

The covariance matrix of estimated parameters is defined as

$$\Gamma = J^{-1} + J^{-1}KJ^{-1}$$

where

$$J^{-1} = (J^* + \widetilde{\Gamma}^{-1})^{-1}$$

and

$$K = \nu \Gamma_{\widetilde{Q}}^{\nu}$$

where

 $\Gamma_{\widetilde{Q}}$  = a priori covariance matrix of the m considered parameters

$$v = \sum_{i=1}^{N} \phi_i w^{-1} \theta_i^{T}$$
 ( $\theta_i$  is the analogous vector of partials of considered parameters)

USE

# CALL COMIMP

The matrix thus generated is used in the calculation of the target centered covariance matrix (TACCOM) and the closest approach parameters.

# CODING INFORMATION

Length of subroutine is 750(10) or 1356(8) words.

### REFERENCE

Anderson, John D., RFP 312-37, August 29, 1961.

COMOUT

Michael R. Warner, JPL

IMB 7094 Fap

January 4, 1965

# PURPOSE

COMOUT writes the midcourse and trajectory information on the COMENT region of disk.

## RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

COMENT

USE

CALL COMOUT

COMMON input:

EP-63 midcourse COMENT buffer (29 words forward)

EP-104 trajectory COMENT buffer (17 words forward)

## CODING INFORMATION

Length of subroutine is 14(10) or 16(8) words.

IDENTIFICATION 16-1 of 2

COROP

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

COROP computes the refraction corrections to optically obtained hour angle and declination.

## RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SQRT

EXP(3)

ARTAN

SIN

### METHOD

$$\Delta_{r} \gamma = \tan^{-1} \left( \frac{A}{\rho - B} \right)$$
 (correction to elevation angle)

where

$$A = \frac{0.00211}{(\gamma/57.2957795 + 0.0598)^2.42}$$

$$B = \sqrt{C^2 - R_e^2 + R_e^2 \sin^2 \gamma} - R_e \sin \gamma$$

$$C = R_e + 51.2064$$

ρ = slant range

R<sub>e</sub> = earth equatorial radius

The program then uses  $\Delta_{r}\gamma$  to obtain  $\Delta_{r}a$  and  $\Delta_{r}\delta$  in the same manner as subroutine CORR (q.v.).

USE

16-2 of 2

CALL COROP

COMMON input:

RVEC P, km

EL γ, deg

RE R<sub>e</sub>, km

COMMON output:

DHA  $\Delta_{r}a$ , deg

DDEC  $\Delta_{r}\delta$ , deg

# CODING INFORMATION

Length of subroutine is 120(10) or 170(8) words.

# REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Cain, Dan L., TM 312-275, February 14, 1963.

IDENTIFICATION 17-1 of 3

CORR

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

CORR calculates the refraction corrections and the vertical displacements for angular tracking data.

# RESTRICTIONS

a. COMMON break: 47675

b. Subroutines called:

SIN

SQRT

**METHOD** 

Δγ	=	57. 2957795 $\frac{n}{340.0}$ b <sub>1</sub> b <sub>2</sub>	$\gamma < 0.3$ radians
٧١		31.6/31./3 340 0 5132	. < 0.5 Tadians

$$\Delta_{r}\gamma = 57.2957795 \text{ n} \times 10^{-6} \frac{\cos \gamma}{\sin \gamma}$$
  $\gamma \ge 0.3 \text{ radians}$ 

$$\Delta_{\mathbf{r}}^{\alpha} = \frac{\Delta_{\mathbf{r}} \gamma \cos \phi \sin^{2} \alpha}{\cos^{2} \gamma \sin \sigma} \qquad \delta < 87^{\circ}$$

$$\Delta_{\mathbf{r}} \delta = \frac{(\sin \phi \cos \gamma - \sin \gamma \cos \phi \cos \sigma) \Delta_{\mathbf{r}} \gamma}{\cos \delta} \qquad \delta < 87^{\circ}$$

$$\Delta_{\mathbf{r}}^{a} = \Delta_{\mathbf{r}}^{\delta} = 0$$
  $\delta \ge 87^{\circ}$ 

$$b_1 = 1.0 - (1.216 \times 10^5 b_3 \gamma_{rad}) - (51.0 - 300.0 \gamma_{rad}) \sqrt{b_3}$$

$$b_2 = \left[7.0 \times 10^{-4}/(0.0589 + \gamma_{rad})\right] - 1.26 \times 10^{-3}$$

$$b_3 = \frac{1}{10^3 (r - R_i)}$$

 $\Delta_{\mathbf{v}} \alpha = \frac{\mathbf{v}}{\cos \phi}$   $\Delta_{\mathbf{v}} \gamma = \mathbf{u} \cos \sigma$   $\Delta_{\mathbf{v}} \sigma = \mathbf{u} \sin \sigma \frac{\sin \gamma}{\cos \gamma}$   $\Delta_{\mathbf{v}} \delta = 0$ 

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where

 $\gamma$  = elevation angle ( $\gamma_{rad}$  =  $\gamma$  in radians)

 $\sigma = azimuth angle$ 

a = hour angle

δ = declination

 $\Delta_r$  = refraction correction

 $\Delta_{v}^{-}$  = vertical correction

n = index of refraction (nominally 340.0)

 $\phi$  = geocentric latitude of station

r = geocentric probe distance

R; = Earth radius at station i

u = north-south displacement of vertical

v = east-west displacement of vertical

### USE

## CALL CORR

# COMMON input:

SINEL sin y

IFA-9  $\neq$  0 to bypass  $\Delta_r$  and  $\Delta_v$  calculation

EL

ELZ sin δ

X, Y, Z geocentric probe vector

RI(i) Earth radius at station i

FNI(i) index of refraction at station i

COSEL cos y

HA a

COSPHI  $\cos \phi$ 

SINAZ sin σ

# JPL TECHNICAL MEMORANDUM NO. 33-204

COSAZ  $\cos \sigma$  17-3 of 3

UI(i) north-south displacement, station i

VI(i) east-west displacement, station i

# COMMON output:

DEL  $\Delta_{r}\gamma + \Delta_{v}\gamma$ 

DHA  $\Delta_r a + \Delta_v a$ 

DAZ  $\Delta_{v}\sigma$ 

DDEC  $\Delta_r \delta$ 

# CODING INFORMATION

Length of subroutine is 188(10) or 274(8) words.

# REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Cain, Dan L., IOM to W. Hoover, July 6, 1960.
- c. Cain, Dan L., IOM to W. Hoover, October 4, 1960.

IDENTIFICATION 18-1 of 6

CXPLOT/XXXFSR/CDC John R. Schoeni, JPL IBM 7094 Fap January 4, 1965

### PURPOSE

To generate on the computer plotting tapes to be displayed on the SC-4020 microfilm recorder. Controlled by a set of pseudo-operations, the routine will scale data, generate and label grid lines, and annotate the plotted information.

### RESTRICTIONS

- a. ERROR return when incorrect calling sequence is given.
- b. Subroutine used:

**IOCS** 

### USE

CXPLOT uses IOCS to control I/O buffering and the user before calling CXPLOT must open a reserve file for the subroutine's use. In defining the buffer pool to be attached to the plotting file, it is recommended that at least two buffers of up to 200 words in length be assigned. The entry into CXPLOT is from a FAP calling sequence and consists of a call CXPLOT followed by pseudo-operations which specify the operations to be performed. The calling sequence is terminated by an error exit of the form, PZE A, where A is the location of the users error routine. Normal return is to the location following the error return. The following pseudo-operations are recognized.

FVE N, T, D FVE O, O, O	}	Selection of tape unit and indication of film advance
SIX SX, L, SY SIX F, SIZE, D	}	Printing horizontal titles
SVN SX, L, Sy SVN F, SIZE, D	}	Printing vertical titles
PON Rx <sub>0</sub> , Rx <sub>n</sub> PON Ry <sub>0</sub> , Ry <sub>n</sub> PON n, e, p PON m, e, q PON Fx, Fy PON X <sub>0</sub> , X <sub>n</sub> PON Y <sub>0</sub> , Y <sub>n</sub>		Generation and labelling of grid

PTW Lx<sub>0</sub>, EX, Ly<sub>0</sub>
PTW N, A, CHAR

Generation of graph

PTH X, t, Y

Generation of special grid line or form flash

MZE O, t, N

Termination of plot

PZE A,

Error return

# CONTROL PSEUDO-OPERATIONS

## FVE N, T, D

An FVE pseudo-operation must be the first pseudo-operation encountered in the first calling sequence to CXPLOT in a program because it causes various addresses throughout the subroutine to be initialized.

- $\underline{N}$  In the FVE pseudo-operation, N indicates that every Nth point will be plotted. N = 0 = 1
- $\underline{T}$  T = 1 will cause a film advance command to be generated. T = 0 indicates no frame advance.
- D is the symbolic location of the File Control Block that has been opened for this file.

# FVE O, O, O

A second FVE pseudo-operation is used when the output is to be written on the IBM 7094 using a packed format and high density tape mode. The plot tape generated by the use of this option requires processing on the 1401 before being plotted on the SC-4020. The Standard SFOF mode of operation is high density and packed.

# MZE O, T, N

The MZE pseudo-operation gives information required to terminate a particular plot or set of plots.

If  $\underline{N}$  is zero, the instruction is treated as a skip, and control will pass to the next pseudo-operation.

If  $\underline{N}$  is non-zero, STOP PRINT and ADVANCE FRAME commands will be generated and written on the tape previously specified.

T not used.

# PZE A

The PZE pseudo-operation indicates the end of the calling sequence. Normal exit is to the instruction following this pseudo-operation. "A" is the address of a user supplied error routine.

### HORIZONTAL TITLES

18-3 of 6

# SIX SX, L, SY SIX F, SIZE, D

The SIX pseudo-operations give information required for printing horizontal titles. The first operation in the pair describes where the title is to be located. The second operation tells the location of the format statement and the letter size. Six pseudo-operations must always occur in pairs.

- L = 1 SX and SY designate the X and Y coordinates locating the center of the title.
- L = 0 SX and SY designate the X and Y coordinates locating the position at which the first character of the title is to be centered.
- F designates the location of the format statement to be used for the title. The format statement starting at location F must be of the form

- x an integer indicating the number of hollerith characters which follow the H.
- y if blank the current frame will be used. This is the normal case.
- y if equal to 1, a film advance command is generated before printing this title.
- z is always a blank character.
- SIZE designates the letter size and may have the values 0, 1, 2, 3 specifying that letters of 1, 2, 4, or 8 times normal size are to be used. (Caution: a character 8 times normal size is 64 by 128 units in size.)
- D When printing normal size letters, SIZE is zero, D indicates whether the letters will be generated by the vector generator or by the plotting matrix.
  - D = 0: plotting matrix.
  - D = 1: vector generator.

## VERTICAL TITLES

# SVNSX, L, SY

# SVN F, SIZE, D

The SVN pseudo-operations give information required for printing titles vertically. They are analogous to the SIX pseudo-operations in format.

### GRIDS & LABELS

### PON

The PON pseudo-operations specify the grid lines to be drawn, the exposure to be used in drawing them, the formats to control their labelling and the scaling to be used in the plot.

The first five PON pseudo-operations must be used as a group.

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PON 
$$Rx_0$$
,  $Rx_n$   
PON  $Ry_0$ ,  $Ry_n$   
PON n, e, p  
PON m, e, q  
PON FX,, Fy

# PON Rx<sub>0</sub>,, Rx<sub>n</sub>

 $Rx_0$  is the location of the lower end of the range of x values.

 $\overline{Rx}_n$  is the location of the upper end of the range of x values.

The limits for the range will be in floating point form.

# PON Ry<sub>0</sub>,, Ry<sub>n</sub>

is analogous to the PON  $Rx_0$ ,  $Rx_n$  for the range of y.

# PON n, e, p

contains an integer n specifying the number of uniform intervals in the x-direction (vertical grids) to be bounded by grid lines, and an integer p indicating that the first grid line and every p<sup>th</sup> grid line following will be labelled. If e = 1, the labelled grid lines will be drawn heavy; otherwise they will be light (vertical grids).

## PON m, e, q

This is analogous to the third PON but refers to the y-direction (horizontal grids).

# PON Fx,, Fy

contains the locations of format statements to be used in labelling the horizontal and vertical grid lines, respectively. These should be of the form:

where

<u>F</u> indicates that a conversion from a floating point number to fixed point number should be performed, and that the fixed point number should be printed.

 $\underline{c}$  designates the "column width." ( $c \leq 9$ )

 $\underline{d}$  designates the number of digits to be retained to the right of the decimal point. d < (c-1)

$$\frac{\text{PON } X_0, X_n}{\text{PON } Y_0, Y_n}$$

The sixth and seventh PON pseudo-operations specify the area of the character on surface ( $1023 \times 1023$ ) to be used for plotting. These two PON pseudo-operations may be omitted, and if so the standard case of

PON 96,,992 PON 0.,896 18-5 of 6

will be assumed by the subroutine.

# PON $X_0$ ,, $X_n$

is the range of the horizontal positions to be used

$$(0 \leqslant X_0 < X_n \leqslant 1023).$$

PON  $Y_0, Y_n$ 

contains the range of vertical positions to be used for the graph

$$(0 \le Y_0 < Y_n \le 1023).$$

## GENERATION OF GRAPH

PTW Lx<sub>0</sub>, EX, Ly<sub>0</sub>

# PTW N, A, CHAR

The PTW pseudo-operations specify the character and the exposure to be used in plotting, the memory area containing the sets of (x, y) values to be plotted, and the scaling factors to be used.

 $\frac{Lx_0}{}$  designates the starting memory location for the sequence of values of x to be plotted.

 $\frac{Ly_0}{N}$  designates the starting location for the corresponding values of y. designates the number of pairs of values of x and y to be plotted.

Thus, locations  $\text{Lx}_0$  to  $\text{Lx}_0^+\text{N}$  - 1 contain the values of x corresponding to the values of y found in locations  $\text{Ly}_0^+\text{N}$  - 1.

Note the special case where the data are in x, y pairs. N is the number of c and y values, and the address of the FVE pseudo-operation is set to 2. The same type of logic applies to triplets, etc.

- <u>EX</u> designates the exposure to be used in plotting. If EX = 0, the exposure will be heavy; otherwise, the light mode will be used.
- A=0 The scaling factors computed from the previous set of PON pseudo-operations will be used.
- A#0 The four locations following the pair of PTW instructions should contain the location of floating point values of a, b, c, and d in that order.

For each pair of values, x, y, a corresponding pair of coordinates (X, Y) is computed by the subroutine as follows:

$$X = ax + b$$

$$Y = c_Y + d$$

where 18-6 of 6

a, b, c and d are the scaling parameters. They depend upon the range of coordinates to be used when plotting the range of values specified.

If, in scaling, a point is either 0 > X > 1024 or 0 > Y > 1024, the point is discarded.

CHAR The Hollerith character to be used for the plot is given in the six bit positions 30 to 35 of location CHAR.

### JOINING POINTS BY VECTORS

If the character specified is the space the vector generator will be used for the plot, and successive points will be connected using straight line segments.

# PTH X, t, Y

The PTH pseudo-operation allows the programmer two options: one to draw grid lines directly in either the x or y direction, and one to control usage of the form flash.

- t = 0 A horizontal grid line will be drawn beginning at X, Y and extending to the right edge of the frame.
- t = 1 A vertical grid line will be drawn from X, Y to the upper edge of the frame.
- t = 4 A form flash command will be generated. Care must be taken not to generate more than one form flash on the same plot.

### CODING INFORMATION

Length of subroutine is 1744(10) or 3320(8) words.

IDENTIFICATION 19-1 of 2

DATAPE

Melba W. Nead, JPL, Fortran II, Version 3 January 4, 1965

## PURPOSE

To generate a simulated data file for checking and for study programs.

### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

IXTAB

OFFSYS

(DFAD)

TIMER

LOOKUP

INTR1

GHADP

COEF

DOPLR

CORR

ND2F

FILT

Types of doppler are limited as follows:

```
If IX (transmitter) = 0, C3 and CC3 are zero.
```

If IX  $\neq$  0, IR (receiver)  $\neq$  IX, C1 and CC3 are zero.

If  $IX \neq 0$ , IR = IX, Cl and C3 are zero.

### METHOD

Data identified under the control card DATA TAPE SIGMA will be included on the file. Weights for each data type must be included under WEIGHTS BY DATA TYPE AND STATION. The equation for the computation of the data value in the file is:

$$R(I) = Rc(I) + \Delta R(I) + F(I, IR) * A + B(I, IR)$$

where

R(I) = the array of data where I references the data type

Rc(I): = calculated value of the data point

# JPL TECHNICAL MEMORANDUM NO. 33-204

 $\Delta R(I)$  = correction on calculated value

19-2 of 2

F(I, IR) = noise factor on data by type and station. This provided by setting the data type under DATA TAPE SIGMA to a floating point number.

A = random number

B(I, IR) = bias. Obtained by value under DATA TAPE BIAS according type.

Data is calculated by the same routines employed in the fitting of data.

USE

CALL DATAPE

## CODING INFORMATION

Length of subroutine is 301(10) or 455(8) words.

DECOD

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

## PURPOSE

DECOD determines the group value for the given group number of the weight code word. This word consists of six 3-bit groups from bit S through bit 17. Each group may have a value from 0 through 7. DECOD also floats the data sample rate, which occupies the address portion of the weight code word.

### RESTRICTION

COMMON break: 47675

## USE

CALL DECOD

PZE J group number PZE K group value

COMMON input:

RVECW weight code array  $(1 \times 11)$ 

ID data type identification

COMMON output:

TS sample rate, sec.

## CODING INFORMATION

Length of subroutine is 32 (10) or 40 (8) words.

DIAG/DIAGO

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

To compute standard deviation of the mapped forward covariance matrix or standard deviation of input covariance matrix.

## RESTRICTIONS

- a. COMMON break: 47675
- b. Each subroutine computes the square root of the diagonal of a specific matrix.
- c. Subroutines used:

SQRT

## USE

CALL DIAG Standard deviation of input covariance matrix

CALL DIAGO Standard deviation of mapped forward covariance matrix

## CODING INFORMATION

Length of each subroutine is 105 (10) or 151 (8) words.

DICOS

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

DICOS computes the direction cosine vectors  $\underline{\underline{A}}$ ,  $\underline{\underline{D}}$ ,  $\underline{\underline{\widetilde{A}}}$ , and  $\underline{\underline{\widetilde{D}}}$ , which are defined in the documentation of subroutine CATS.

## RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

SIN/COS

## USE

## CALL DICOS

# COMMON input:

DEC

GHA  $a_G$ , deg THETAI  $\lambda_i$ , deg EL  $\gamma$ , deg

 $\delta$ , deg

AZI  $\sigma$ , deg

PHII  $\phi_i$ , deg

# COMMON output:

 $\begin{array}{ccc} \text{AX} & \underline{\textbf{A}} \\ \text{DX} & \underline{\textbf{D}} \\ \text{TAX} & \underline{\widetilde{\textbf{A}}} \\ \text{TDX} & \underline{\widetilde{\textbf{D}}} \end{array}$ 

# CODING INFORMATION

Length of subroutine is 123 (10) or 173 (8) words.

IDENTIFICATION 23-1 of 4

(DISCBU) REDE/RITE/PISA Charles Coltharp, JPL IBM 7094 Fap January 4, 1965

## PURPOSE

DISCBU provides elementary buffering for reading and writing scratch information on disk.

## RESTRICTIONS

Subroutines used:

DCP

## METHOD

DISCBU uses DCP for reading and writing the disk. All physical records are 200 words in length. Each logical record is preceded on the disk by a control word,

PZE O,,L

where L is the number of words in the logical record. L may be zero, but L will never be greater than 200.

A control word of

MZE O,,O

denotes a logical end of file.

USE

To write CALL RITE

PZE FILE,, ERREOF

IOXY A,, N

To read CALL REDE 23-2 of 4

PZE FILE.. ERREOF YXOI Α,, Ν

A is the location into which information is read, or from which it is written.

TCH, transfer in channel, will be recognized by both read and write operations.

The tag portion of all IO commands is ignored.

The address and decrement portions of IOCD commands are ignored.

Closed or unopened files are opened automatically by the first read or write operation requested.

# Read Operation

- Rewind file; close file. When IOCD is used it must be the only command IOCD in the command list.
- Read N words, ignoring logical record marks, and proceed to next command. IOCP If this command precisely finishes a record, the logical end of record may be recognized by the next command.
- Read N words, ignoring logical record marks, and terminate IO. Unread IOCT words in a partially read record will be lost.
- Read to logical record mark (read one whole record, or finish a record IORP already started), and proceed to next command. Put number of words read into the decrement portion of the IORP command. N is not interpreted.
- Same as IORP, but terminate IO. IORT
- Same as IORP if the number of words in the record is less than or equal to N. IOSP Otherwise, transmission is stopped after N words have been read. Unread words in this record will be lost. Control proceeds to the next command.
- IOST Same as IOSP, but terminate IO.

## Write Operation

- Write logical EOF, rewind file; close file. When IOCD is used it must be the IOCD only command in the command list.
- Write N words, and proceed to next command. IOCP
- Write N words followed by a logical record mark, and terminate IO IOCT
- Write N words followed by a logical record mark, and proceed to next IORP command.

IORT Same as IOCT.

23-3 of 4

IOSP Same as IORP.

IOST Same as IORT.

If at any time in writing, a record reaches a length of 200 words before a logical record mark is indicated by the user, DISCBU will insert a logical record mark before continuing the transmission.

### File Block

FILE BCI 1, XXXXXX

+1 PZE J (decrement = P)

+2 PZE K (decrement = Q)

+3 PZE BUFFER (decrement = R)

+4 PZE 200

XXXXXX an entry in the disk directory, defined as scratch

- J that record in XXXXXX with which this file begins
- K that record in XXXXXX with which this file ends

BUFFER the location of a 400 word block of core storage

The decrements of FILE +1 through FILE +3 (the items in parentheses, above) will be used by DISCBU for bookkeeping. These items will be zero when the file is not open, and they should be set zero initially by the user.

- P last disk record transmitted
- Q buffer designation
  - 0 = next disk transmission will be from (or into) BUFFER + 0
  - 1 = next disk transmission will be from (or into) BUFFER + 200
- R location of next word in BUFFER to be read or written.

FILE +4 is the location given to DCP for the flag word. Initially this must be a positive, non-zero number.

# Error - End of File Return

Control is returned to ERREOF under the conditions described below.

If an error is indicated by DCP, that indication will be in the AC on return. The error codes are described in the design specifications for DCP.

If the user tries to exceed the self-imposed file bounds (K is the largest usable disk record in XXXXXX), the AC on return will be MZE 0,0,-1.

No provision is made for continuing if the condition occurs on a read operation.

Before the file may be used again, the user must issue the sequence

CALL REDE

PZE FILE,, ERREOF

IOCD

to close and rewind the file.

23-4 of 4

If the condition occurs on a write operation, the user may continue, provided all seven index registers are preserved, by executing

TRA \$PISA.

However, K should be increased if the user wishes to read the material back at a later time.

If a logical EOF is encountered by a read operation, the AC will be plus zero.

# CODING INFORMATION

Length of subroutine is 273 (10) or 421 (8) words.

DNAME

Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

## PURPOSE

DNAME contains a list of the BCD names of the eleven ODP data types.

# USE

The BCD names, one per word, are stored backward. The entry point, DNAME, is assigned to the first logical location (last physical location).

# CODING INFORMATION

Length of subroutine is 12 (10) or 14 (8) words.

IDENTIFICATION 25-1 of 5

DOPLR

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

DOPLR obtains the calculated values of the ODP data types.

## RESTRICTIONS

COMMON break: 47675

Subroutines called:

ABDD

ARSIN

ARTAN

DMOD

RATES

SQRT

VEC

(DFAD)

(DFMP)

# METHOD

DOPLR first obtains the fine light-time correction in a two-iteration procedure:

$$\epsilon = 0$$

$$\underline{\mathbf{r}}_t = \underline{\mathbf{r}}_t^t + \epsilon \dot{\underline{\mathbf{r}}}_t^t$$

$$\underline{\dot{\mathbf{r}}}_{\mathsf{t}} = \underline{\dot{\mathbf{r}}}_{\mathsf{t}}^{\mathsf{t}} + \epsilon \underline{\ddot{\mathbf{r}}}_{\mathsf{t}}^{\mathsf{t}}$$

$$\frac{\ddot{\mathbf{r}}_{t}}{\mathbf{r}_{t}} = \frac{\ddot{\mathbf{r}}_{t}^{\dagger}}{\mathbf{t}} + \epsilon \frac{\ddot{\mathbf{r}}_{t}^{\dagger}}{\mathbf{t}}$$

$$\ddot{\underline{r}}_t = \ddot{\underline{r}}_t^{\dagger}$$

$$\Delta t = \frac{\rho + \Delta t^{\dagger} \dot{\rho}}{C + \dot{\rho}}$$

 $\epsilon = \Delta t' - \Delta t$ ; the procedure is then repeated.

where

25-2 of 5

Δt' = coarse correction from subroutine TIMER

 $t^1 = t_{ob} - \Delta t^1$ 

 $\rho$  = slant range at  $t^1$ 

 $\dot{p}$  = range rate at t'

c = speed of light

The eleven ODP data types are then obtained. DOPLR calls subroutine RATES to obtain slant range and its derivatives and all other leg-dependent quantities:

$$\rho = \left| \frac{\mathbf{r}_{t}}{\mathbf{t}} - \frac{\mathbf{R}_{it}}{\mathbf{ob}} \right| \qquad \text{(slant range)}$$

where

 $\underline{r}$  = geocentric position vector of probe

 $\underline{\mathbf{R}}_{\mathbf{i}}$  = geocentric position vector of station i

$$\dot{\rho} = \frac{\rho \cdot \dot{\rho}}{\rho} \qquad \text{(range rate)}$$

where

 $\underline{\rho}$  = topocentric position vector of probe

 $\frac{\dot{\rho}}{2}$  = topocentric velocity vector of probe

$$\gamma = \sin^{-1} \frac{R_i \cdot L}{R_i}$$
 (elevation angle)

where

$$\underline{\mathbf{L}} = \frac{\underline{\rho} + \underline{\mathbf{r}}_{s}}{\underline{\rho}}$$

R; = Earth radius at station i

 $\frac{\mathbf{r}}{-\mathbf{s}}$  = geocentric position vector of Sun

$$\sigma = \tan^{-1} \frac{\sin \sigma}{\cos \sigma}$$
 (azimuth angle).

where

25-3 of 5

$$\begin{split} \sin \sigma &= \frac{L_{y} \cos(\alpha_{G} + \lambda_{i}) - L_{x} \sin(\alpha_{G} + \lambda_{i})}{\cos \lambda_{i}} \\ \cos \sigma &= \frac{-L_{x} \sin \phi_{i} \cos(\alpha_{G} + \lambda_{i}) - L_{y} \sin(\alpha_{G} + \lambda_{i}) \sin \phi_{i} + L_{z} \cos \phi_{i}}{\cos \gamma_{i}} \end{split}$$

a<sub>G</sub> = Greenwich hour angle

 $\lambda_i$  = longitude of station i

 $\phi_i$  = geocentric latitude of station i

$$\alpha = \alpha_G + \lambda_i - \tan^{-1} \left( \frac{\rho_y}{\rho_x} \right)$$
 (hour angle)

$$\delta = \sin^{-1} L_{z}$$
 (declination)

$$f_1 = \Omega_1 + \Omega_2 \phi_1$$
 (one-way integrated doppler frequency)

$$f_{c3} = \Omega_3 + \Omega_4 \phi_3$$
 (coherent three-way integrated doppler frequency)

$$f_3 = \Omega_5 + \Omega_6 \phi_3$$
 (three-way integrated doppler frequency)

where

 $\Omega_1$ ,  $\Omega_2$ ,  $\cdots$   $\Omega_6$  are defined in COEF writeup.

$$\phi_1 = \frac{\dot{\rho}_i}{c} - \frac{1}{c^2} h_1 - \frac{\ddot{\rho}_i}{c} \frac{\tau^2}{24}$$

$$\phi_3 = \frac{1}{c} (\dot{\rho}_i + \dot{\rho}_q) - \frac{1}{c^2} h_3 - \frac{1}{c} (\ddot{\rho}_i + \ddot{\rho}_q) \frac{\tau^2}{24}$$

r = doppler counting interval

q = transmitter index

i = receiver index

$$\begin{aligned} \mathbf{h}_1 &= \frac{\dot{\rho}}{\rho} \ \underline{\rho} \cdot (\underline{\mathbf{r}} - \underline{\mathbf{r}}_s) + \frac{1}{2} \left( \dot{\mathbf{R}}_{it_{ob}} - \dot{\mathbf{r}}_t^2 \right) \\ \\ \mathbf{h}_3 &= \dot{\rho}_i^2 + \dot{\rho}_q \dot{\rho}_i + \frac{\dot{\rho}_i}{\rho_i} \underline{\rho}_i \cdot \left( \underline{\dot{\mathbf{R}}}_{it_{ob}} - \dot{\underline{\mathbf{r}}}_{st} \right) \\ \\ &- \frac{\dot{\rho}_q}{\rho_q} \underline{\rho}_q \cdot \left( \underline{\dot{\mathbf{R}}}_{qt_{tr}} - \dot{\underline{\mathbf{r}}}_{st} \right) + \frac{1}{2} \left( \dot{\mathbf{R}}_{it_{ob}}^2 - \dot{\mathbf{R}}_{qt_{tr}}^2 \right) \\ \\ \mathbf{f}_{dl} &= \mathbf{f}_{li} - \mathbf{f}_{lj} \qquad \text{(differenced one-way doppler from stations i and j)} \end{aligned}$$

where  $f_{li}$  and  $f_{lj}$  are one-way doppler values which must be taken simultaneously at stations i and j

$$\rho_{\text{DSIF}} = \frac{\Omega_6(\rho_{\text{q}} + \rho_{\text{i}})}{16 \text{ c}} + \epsilon$$
 (DSIF ranging)

where

 $\epsilon$  = ranging system bias

USE

```
CALL DOPLR
```

COMMON input:

DELT  $\Delta t^{\dagger}$ , d.p. sec past 1950.0 X  $\underline{r}(1 \times 3)$ , km XDOT  $\underline{\dot{r}}(1 \times 3)$ , km/sec

XAC  $\underline{\ddot{r}}(1 \times 3)$ , km/sec<sup>2</sup> XJERK  $\underline{\ddot{r}}(1 \times 3)$ , km/sec<sup>3</sup>

VELC c, km/sec CZ1,  $\cdots$  CZ6  $\Omega_1$ ,  $\cdots$   $\Omega_6$ 

RVECO observed values of ODP data types  $(1 \times 11)$ 

OMEGA Earth rotation rate, deg/sec

COMMON output:

RVEC calculated values of ODP data types  $(1 \times 11)$ 

DELT2 Δt, d.p. sec past 1950.0

# CODING INFORMATION

25-5 of 5

Length of subroutine is 715 (10) or 1313 (8) words.

# REFERENCE

Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.

IDENTIFICATION 26-1 of 2

ENDIT

Michael R. Warner, JPL Fortran II, Version 3

January 4, 1965

## PURPOSE

ENDIT sets up and solves the normal equations. The covariance matrix

$$\Gamma = J^{-1} + J^{-1} KJ^{-1}$$

is obtained from the matrices accumulated by subroutine FORM. The  $\Delta Q$  vector is retained for subsequent output.

# RESTRICTIONS

COMMON break: 47675

Subroutines used:

(DFSB)/(DFMP)/(DFAD)

PRIM

SQRT

STPREG

REVRT

NOUT

# METHOD

The matrix inversion is accomplished by subroutine STPREG (q.v.).

USE

CALL ENDIT

COMMON input:

KLIST attitude control estimate flags

FABC attitude control coefficients

3	ILIST	initial conditions estimate flags	26-2 of 2
•	XNJ	initial conditions	
	LLIST	physical constants estimate flags	
	GRAVE	physical constants	
	MLIST	velocity of light estimate flag	
	VELC	velocity of light	
	NLIST	station locations estimate flags	
	RI	station locations	
	GFLIP	a priori $\Gamma^{-1}$	
	QTILD	nominal values corresponding to QTILD	
	NEST	number of estimated parameters	
	NCON	number of considered parameters	
	RIGHT	accumulated right-hand side	
	XJ	accumulated normal matrix	
	SOUR	sum of squares	
	NTOT	number of data points	
	XX	accumulated consider matrix	
	XMU	accumulated consider vector	
COMMON output: .		out: .	
	QNU	updated parameter list	
	RIGHT	$\Delta Q$ vector	
	ХJ	J inverse	
	CPGAM	covariance matrix	
	XNJ )		
	GRAVE	undeted segmentage	
	RI	updated parameters	
	FABC		
	CAPQ	Q sum of squares	

# CODING INFORMATION

Length of subroutine is 828 (10) or 1474 (8) words.

**EPPHEM** 

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

## PURPOSE

To prepare an ephemeris tape with complete information required from the planetary ephemeris and the probe ephemeris for each time point of the ODP. This program checks the requirements of each iteration and interpolates and if necessary integrates the partials of the physical constants. The information is written on tape A-4.

## RESTRICTIONS

- a. COMMON break: 47055
- b. Subroutines used:

GAMAT

LOOKUP

RITEM

TOCIM

INTR 1

PONT

STARP

WREOF

KINE

QUIZ

TIMER

(DFSB)

### METHOD

EPPHEM is primarily a logical decision box, which accomplishes its task through the use of subroutines. A flow chart is given to supply further clarification.

#### USE

CALL EPPHEM

### CODING INFORMATION

Length of subroutine is 361 (10) or 551 (8) words.

28-1 of 2

ERROR

Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

## PURPOSE

ERROR prints ODP error condition messages. There are three versions of this subroutine in the ODP; the version in link LA9 prints recoverable error condition messages, the version in link LA12 prints unrecoverable error messages, and the version in the other links stores the calling sequence for subsequent action by LA12.

### RESTRICTIONS

COMMON break: 47675

Subroutines used: (LA9)

TYPRYT

ONLIN

Subroutines used: (LA12)

TYPRYT

PROUT

**ENDS YS** 

RECOV

# USE

CALL	ERR	OR
PZE	ΙÌ	
PZE	J	Decrement integers
₽7₽	ĸ	

<u>+</u>	Error message	Version
1	INCORRECT RECORD NUMBER IN DISK CALLING SEQUENCE	LA12
2	WEIGHT (J = ID) FOR STATION (K = IR) = $0$	LA12
3	DISK STORAGE NOT ALLOCATED FOR THIS JOB	LA9

		28-2 of 2
4	ESTIMATE OR CONSIDER LIST EXCEEDS 20 PARAMETERS	LA9
5	NO INPUT RECEIVED WITHIN TIME LIMIT	LA9
6	PROGRAM UNABLE TO READ/WRITE PROBE EPHEMERIS	LA12
7	FIRST PROBE EPHEMERIS TIME GREATER THAN	
	LOOKUP ARGUMENT	LA12
8	LAST PROBE EPHEMERIS TIME LESS THAN LOOKUP	
	ARGUMENT	LA12
9	INJECTION CONDITION = 0	LA9
10	INPUT J MATRIX NOT N*N	LA9
11	INADMISSIBLE SYMBOL OR DATA FIELD	LA9
12	INPUT CONSIDER MATRIX NOT M*M	LA9
13	POINTING TIME LESS THAN EPOCH + R/C	LA9
14	MAPPING TIME LESS THAN EPOCH	LA9
15	DISK TRANSMISSION ERROR	LA12, LA9
16	DISK BUFFER ALLOCATION INSUFFICIENT	LA12
17	PLANETARY EPHEMERIS ERROR	LA12
18	DATA POINT OUT OF TIME SORT	LA12
19	ERROR IN CXPLOT CALLING SEQUENCE	LA12
20	ERROR IN DP SQRT SUBROUTINE	LA12, LA9
21	LOOKUP TIME FAILS TO MATCH TIME ON EPHEMERIS	
	TAPE	LA12

The second line of messages 1, 2, 3, 6, 7, 8, 15, 16, 17, 18, 19, 20, 21, is RECOVERY IMPOSSIBLE. JOB ABORTED.

The second line of messages 4, 5, 9, 11, 13, 14, is
USER MUST CORRECT AND RELOAD INPUT DECK WITHIN 3 MINUTES.

The second line of messages 10, 12, is CONDITION IGNORED. JOB WILL CONTINUE.

# CODING INFORMATION

Length of the LA9 version is 286 (10) or 436 (8) words.

Length of the LA12 version is 326 (10) or 506 (8) words.

Length of the dummy version is 9 (10) or 11 (8) words.

FDATA

Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

# PURPOSE

FDATA stores the nominal values of the S and T weighting tables in COMMON arrays SSQ and TL, respectively. The S values are squared prior to their storage in SSQ.

## RESTRICTIONS

COMMON break: 47675

USE

CALL FDATA

## CODING INFORMATION

Length of subroutine is 591 (10) or 1117 (8) words.

## REFERENCE

Hamilton, Thomas W., Inter-Office Memorandum to M. Warner, June 1962.

30-1 of 2

FILL/FILT Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

#### PURPOSE

FILL writes the ODP residual file on disk in the same physical format as the ODGgenerated data file. It is also used to write the tracking predictions on disk for subsequent output. FILT writes a simulated ODG data file on disk. Both entries use the RITE entry of the buffered disk routine DISCBU.

## RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

RITE

PISA

## USE

CALL FILL

PZE

CALL FILT

PZE

# COMMON input:

EP-6

TOB time of observation, d.p. sec past 1950.0 RVEC array of computed data types ( $I \times 11$ )

RVECD array of residuals  $(1 \times 11)$ RVECW array of weights  $(1 \times 11)$ 

IF5 # 0 normal entry, = 0 end-of-file entry

band designation XFREQ transmitter frequency . TFREQ transponder frequency IR receiver ID, B17

1Xtransmitter ID, B17

TAU doppler averaging time, sec

```
COMMON output:
```

30-2 of 2

```
TAB-799
            400 word output buffer for DISCBU
```

The disk logical record format is as follows:

```
Word
            word count (2N + 5, B17) or - 0 for EOF
        1
            TOB
        2 TOB-1
        3 XFREQ if IX # 0
            TFREQ if IX = 0
        4
            0
        5
            bit S
                     1 if L - S band
                      0 if LorS band
                      1 if RVEC # 0
               1
               2
                      1 if RVEC - 1 \neq 0
               3
                      1 if RVEC - 2 ≠ 0
                      1 if RVEC - 10 \neq 0
               11
               12-15 not used
                     l if L - S or S band
               16
                      0 if L band
               17
                      not used
               18-21 IX
               22-25 IR
               26-35 TAU
        6
            first member of RVEC which is non-zero
            bits S-17 RVECD corresponding to word 6
               18-35 RVECW corresponding to word 6
        8
            second member of RVEC which is non-zero
            second RVECD - RVECW word
            Nth member of RVEC which is non-zero
    2N + 4
```

2N + 5Nth RVECD - RVECW word

## CODING INFORMATION

Length of subroutine is 191 (10) or 277 (8) words.

FIRST

Melba W. Nead, JPL IBM 7094 Fap

January 4, 1965

## PURPOSE

Main program for link 1 of SFOF-JPTRAJ version of the JPL Orbit Determination Program. FIRST clears COMMON and formats the rejection table on the disk.

# RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 46711
- c. Subroutines used:

**FDATA** 

NOMNL

FLAK

DISCBU

#### USE

Initiates ODP by clearing COMMON and calling subroutines to store the permanent or semi-permanent information. This link is called only once per "run" of the ODP.

# CODING INFORMATION

Length of subroutine is 427 (10) or 653 (8) words.

32

FIT

Melba W. Nead, JPL IBM 7094 Fap

January 4, 1965

## PURPOSE

Main program for link 6. The purpose of the routine is to control calculation of the observables and the linkage.

#### RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 46711
- c. Subroutines used:

CALL

OCIM

FLAK

DISCBU

TAPIO

## USE

When fitting data, this link is called by the JPTRAJ monitor.

# CODING INFORMATION

Length of subroutine is 476 (10) or 734 (8) words.

33

FLAT/FLAPR/FLAK

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

# PURPOSE

Subroutine to check the I/O activity flags.

# RESTRICTIONS

- a. ERROR exit to indicate disk I/O error
- b. Subroutines used:

ERROR

TYPRYT

## USE

CALL	FLAT	To check TAPIO flag
CALL	FLAPR	To check PROUT flag
CALL	FLAK	To identify disk error return

## CODING INFORMATION

Length of subroutine is 64 (10) or 100 (8) words.

## REFERENCE

EPD - 125, Rev. 1, April 1, 1964

IDENTIFICATION 34-1 of 2

FORM

Michael R. Warner, JPL Fortran II, Version 3 January 4, 1965

## PURPOSE

FORM sets up the column matrix of partials  $\partial F/\partial Q$  and accumulates the normal matrix J.

## RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

PARLEY

(DFMP)/(DFAD)/(DFSB)

SQRT

## METHOD

The partials  $\partial F/\partial Q$  are obtained by the chain rule when Q is an initial condition or physical constant:

$$\frac{\partial \mathbf{F}}{\partial Q} = \frac{\partial \mathbf{F}}{\partial \underline{\mathbf{r}}} \frac{\partial \underline{\mathbf{r}}}{\partial Q}$$

If Q is the velocity of light or a station parameter,  $\partial F/\partial Q$  are computed analytically and input to FORM.

## USE

A = 1.0 to execute entire routine

CALL FORM

A = -1.0 to bypass chain rule section (occultation or impact time partials)

CALL FORM

# COMMON input:

RVECW	weight for data type ID
ID	data type identification
POBX	partials $\partial F/\partial \underline{r}$ (6 × 12)
GMAT	partials $\frac{\partial \mathbf{r}}{\partial q}$ (6 × 11)
ប	partials $\frac{\partial \mathbf{r}}{\partial \mathbf{r}_0}$ (6 × 6)
ILIST	initial conditions flags (6)
LLIST	physical constants flags (11)

	MLIST	velocity of light flag (1)	34-2 of 2
	NLIST	station locations flags $(3 \times 15)$	
	QA	partials $\partial F/\partial q$ from occultation-impact	
	QB	partials $\partial F/\partial r_0$ from occultation-impact	
	DRDC	partials $\partial F/\partial \overline{c}$	
	DIM3	partials $\partial F/\partial S_{\mathbf{i}}$	
	NEST	number of estimated parameters	
	NCON	number of considered parameters	
	RVECD	residual, data type ID	
	IF10	phi vector output flag	
	DXDR-13	MMP output flag	
C	OMMON output:		
	СРРНІ	ordered partials $\partial F/\partial Q$	
	CPTHT	ordered partials $\partial F/\partial \widetilde{Q}$ (considered)	
	XJ	J matrix	
	RIGHT	right-hand side of normal equations	
	XX	J matrix for considered parameters	
	NTOT	number of data points	
	SOUR	sum of squares	

# CODING INFORMATION

Length of subroutine is 566 (10) or 1066 (8) words.

G1G2

Melba W. Nead, JPL Fortran II, Version 3

January 4, 1965

#### PURPOSE

G1G2 computes the gravity equations  $g_1$  and  $g_2$ .

#### RESTRICTION

COMMON break: 47675

#### METHOD

$$g_{1} = 1 + \left[3 - 42\left(\frac{z}{r}\right)^{2} + 63\left(\frac{z}{r}\right)^{4}\right] \left[\frac{D}{7}\left(\frac{R_{e}}{r}\right)^{4}\right] + \left[H\left(\frac{R_{e}}{r}\right)^{3}\right] \left[3 - 7\left(\frac{z}{r}\right)^{2}\right] \frac{z}{r}$$

$$+ \left[J\left(\frac{R_{e}}{r}\right)^{2}\right] \left[1 - 5\left(\frac{z}{r}\right)^{2}\right]$$

$$g_{2} = 1 + \left[15 - 70\left(\frac{z}{r}\right)^{2} + 63\left(\frac{z}{r}\right)^{4}\right] \left[\frac{D}{7}\left(\frac{R_{e}}{r}\right)^{4}\right] - \left[H\left(\frac{R_{e}^{3}}{r}\right)\right] \left[1 - 10\left(\frac{z}{r}\right)^{2}\right]$$

$$+ \frac{35}{3}\left(\frac{z}{r}\right)^{4} \frac{3}{5}\frac{r}{z} + \left[J\left(\frac{R_{e}}{r}\right)^{2}\right] \left[3 - 5\left(\frac{z}{r}\right)^{2}\right]$$

USE

CALL G1G2 (G1, G2) Results are stored in G1 and G2.

## CODING INFORMATION

Length of subroutine is 182 (10) or 266 (8) words.

## REFERENCE

Anderson, John D., TM 312-131, August 23, 1961.

IDENTIFICATION 36-1 of 4

GAMAT

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

To obtain the partials  $\partial r/\partial q$  from the following integral using Simpson's rule:

$$\Upsilon(t) = U(t) \int_{t_0}^{t_U-1} (t^*)B(t^*) dt^*$$

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

BMATRX

INSPC

INTR1

LOOKUP

(DFAD)

(DFDP)

(DFMP)

(DFSB)

# METHOD

The partials of acceleration with respect to physical constants  $\partial \ddot{r}/\partial q$  are evaluated in BMATRX. These partials are then numerically integrated to obtain  $\partial r/\partial q$ .

Using the acceleration partials as evaluated in BMATRX, if

36-2 of 4

$$B = \begin{cases} 0 & 0 & \cdots & 0 \\ 0 & 0 & \cdots & 0 \\ 0 & 0 & \cdots & 0 \end{cases}$$

$$\frac{\partial \ddot{x}}{\partial q_1} \frac{\partial \ddot{x}}{\partial q_2} \cdots \frac{\partial \ddot{x}}{\partial q_n}$$

$$\frac{\partial \ddot{y}}{\partial q_1} \frac{\partial \ddot{y}}{\partial q_2} \cdots \frac{\partial \ddot{y}}{\partial q_n}$$

$$\frac{\partial \ddot{z}}{\partial q_1} \frac{\partial \ddot{z}}{\partial q_2} \cdots \frac{\partial \ddot{z}}{\partial q_n}$$

$$\frac{\partial \ddot{x}}{\partial x_0} \cdots \frac{\partial \ddot{x}}{\partial x_0}$$

$$U = \begin{cases} \frac{\partial x}{\partial x_0} \cdots \frac{\partial x}{\partial x_0} \\ \frac{\partial x}{\partial x_0} \cdots$$

and

36-3 of 4

$$\Upsilon = \left\{ \begin{array}{cccc} \frac{\partial x}{\partial q_1} & & & \frac{\partial x}{\partial q_n} \\ & & & \frac{\partial y}{\partial q_1} & & & \frac{\partial y}{\partial q_n} \\ & & & & \frac{\partial z}{\partial q_1} & & & \frac{\partial z}{\partial q_n} \\ & & & & \frac{\partial \dot{x}}{\partial q_1} & & & \frac{\partial \dot{x}}{\partial q_n} \\ & & & & \frac{\partial \dot{y}}{\partial q_1} & & & \frac{\partial \dot{y}}{\partial q_n} \\ & & & & \frac{\partial \dot{z}}{\partial q_1} & & & & \frac{\partial \dot{z}}{\partial q_n} \end{array} \right.$$

then the partials  $\partial r/\partial q$  are obtained from the following integral using Simpson's method:

$$\Upsilon(t) = U(t) \int_{t_0}^{t_U-1} (t*)B(t*) dt*$$

The Simpson procedure uses an integration step based on the data times; if no data times exist in the interval, the time steps employed in the trajectory integration are used. The inverse of the U matrix is obtained not by the usual numerical methods but by an inspection method as defined in the subroutine INSPC.

USE

CALL GAMAT Result of the integration is placed on the combined ephemeris tape for use as needed.

# CODING INFORMATION

36-4 of 4

Length of subroutine is 369 (10) or 561 (8) words.

# REFERENCE

Anderson, John D., RFP 312-37, August 29, 1961.

GERTA/GREOF

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

## PURPOSE

GERTA writes the partials  $\partial F/\partial Q$  (the "phi vector") on the MMP output tape, B7-SYSUT9. Each set of partials is tagged by time, receiver, transmitter, and data type.

## RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

WRITEB/ENDFIL

FLAT

## USE

CALL GERTA

CALL GREOF for end-of-file

## COMMON input:

CPPHI partials  $\partial F/\partial Q$  (1 × 20) TOB time, d.p. sec past 1950.0

IR receiver ID

IX transmitter ID

ID data type ID

## CODING INFORMATION

Length of subroutine is 141 (10) or 215 (8) words.

IDENTIFICATION 38-1 of 2

GHADP

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

GHADP calculates the Greenwich hour angle for the time given in COMMON location TOB. Intermediate calculations are in double precision. Output is in COMMON location GHA.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

(DFDP)

(DFAD)

(DFMP)

(DFSB)

DMOD

COS

#### METHOD

$$a_G = 100^{\circ}.0755426 + 0^{\circ}.985647346d + 2^{\circ}.9015$$
  
  $\times 10^{-13} d^2 + \omega s + \Delta \lambda \cos \overline{\epsilon} \pmod{360^{\circ}}$ 

where  $^aG$  = Greenwich hour angle

d = integer days past 1950.0

s = sec past 0<sup>h</sup> of d th day

 $\omega$  = Earth rotation rate

 $\bar{\epsilon}$  = mean obliquity

 $\Delta \lambda$  = nutation in longitude

The mean obliquity is given by

$$\overline{\epsilon}$$
 = 23°.445759 - 0°.1309404T - 0°.88 x 10<sup>-6</sup> T<sup>2</sup> + 0°.5 × 10<sup>-6</sup> T<sup>3</sup>

where T = Julian centuries past 1950.0

USE

38-2 of 2

CALL GHADP

COMMON input:

TOB observation time, d.p. sec past 1950.0

OMEGA Earth rotation rate, deg/sec

DLO nutation in longitude, deg

COMMON output:

GHA Greenwich hour angle, deg

CODING INFORMATION

Length of subroutine is 208 (10) or 320 (8) words.

REFERENCE

Holdridge, D. B., TR 32-223, March 2, 1962.

**IMPAR** 

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

IMPAR computes the partials of impact time with respect to the estimated and considered parameters.

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

FORM

METHOD

$$\frac{\partial T_{\underline{I}}}{\partial Q} = \frac{-1}{\underline{\rho} \cdot \dot{\underline{\rho}}} \left( \underline{\rho} \cdot \frac{\partial \underline{\rho}}{\partial Q} \right)$$

where

T<sub>I</sub> = impact time

Q = parameter

 $\underline{\rho}$  = target centered probe position

 $\frac{\dot{p}}{p}$  = target centered probe velocity

USE

CALL IMPAR

PZE XTAR geocentric target position and velocity

COMMON input:

X geocentric probe position and velocity (1 × 6)

U partials  $\partial \underline{\rho}/\partial r_0$  (6×6)

GMAT partials  $\partial \rho / \partial q$  (6×11)

LLIST physical constants flags (11)

COMMON output:

QA partials  $\partial T_I/\partial q$  (1×11) QB partials  $\partial T_I/\partial r_0$  (1×6)

## CODING INFORMATION

Length of subroutine is 127 (10) or 177 (8) words.

INSPC

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

## PURPOSE

INSPC inverts the U matrix of variational partials by inspection.

# RESTRICTIONS

COMMON break: 47675

#### METHOD

If the U matrix is partitioned

$$\mathbf{U} = \begin{bmatrix} \mathbf{U}_{11} & \mathbf{U}_{12} \\ \mathbf{U}_{21} & \mathbf{U}_{22} \end{bmatrix}$$

then its inverse may be written as

$$\mathbf{U}^{-1} = \begin{bmatrix} \mathbf{U}_{22}^{\mathrm{T}} & -\mathbf{U}_{12}^{\mathrm{T}} \\ -\mathbf{U}_{21}^{\mathrm{T}} & \mathbf{U}_{11}^{\mathrm{T}} \end{bmatrix}$$

USE

CALL INSPC

COMMON input:

U matrix of variational partials (6 x 6)

COMMON output:

UINV U<sup>-1</sup>

# CODING INFORMATION

Length of subroutine is 200 (10) or 310 (8) words.

#### REFERENCE

Anderson, John D., TM 312-409, March 24, 1964.

INTR1/BNTR2

Alan D. Rosenberg and Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

## PURPOSE

This version of INTRl uses the latest version of the JPL Ephemeris routines, EPHEM, to obtain the necessary planetary information. INTRl adjusts this output so it matches output from the original INTRl rotated to true of date for the ODP.

#### RESTRICTIONS

- a. Portions of the ODP COMMON storage must be made available to the routine. ERROR return for planetary ephemeris tape error.
- b. Subroutines used:

**EPHEM** 

PNUT

ROT

ERROR

#### USE

CALL BNTR2 Enter one time to set up blocks of storage necessary

to EPHEM

CALL INTR1 Time is used from COMMON

# CODING INFORMATION

Length of subroutine is 2188 (10) or 4214 (8) words.

#### REFERENCES

- a. Holdridge, D. B., TR 32-223, Space Trajectories Program for the IBM 7090 Computer, March 2, 1962.
- b. Peabody, P. R., Scott, J. F., Orozco, E. G., TR 32-580, User's Description of JPL Ephemeris Tapes, March 2, 1964.

IXTAB

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

## PURPOSE

IXTAB does a lookup in the table of transmitter ID's and frequencies by means of a time argument.

#### RESTRICTION

COMMON break: 47675

#### USE

CALL IXTAB

PZE N

N is the location of a flag which is zero for the initialization entry and non-zero for the normal entry. The initialization entry returns the lookup pointer to the first point in the table.

# COMMON input:

TOB lookup time, d.p. sec past 1950.0

EPOCH ODP epoch, d.p. sec past 1950.0

YYl table of times, sec past epoch (1 × 90)

YY1-90 table of transmitter ID's (1 × 90)

YY1-180 table of frequencies (1 × 90)

# COMMON output:

IX transmitter ID

XFREQ transmitter frequency if IX \neq 0
TFREQ transponder frequency if IX \neq 0

# CODING INFORMATION

Length of subroutine is 54 (10) or 66 (8) words.

IDENTIFICATION 43-1 of 2

KINE

Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

# PURPOSE

KINE reads the ODG data file or the ODP residual file from disk. Each entry loads COMMON with the contents of the next sequential logical record. KINE uses the REDE entry of the buffered disk routine DISCBU.

## RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

REDE

ERROR

c. Error conditions:

Disk parity error

Attempted read outside of file limits

### USE

CALL	KINE				
PZE	RVECO	for ODG	PZE	RVEC	for ODP
PZE	RVECW	data	PZE	RVECD (	residual
PZE	= 1001	reading	PZE	= 1	reading
PZE	= 2000	) Touching	PZE	= 1000	
PZE	IF6	1B17 = EOF, 2B17 =	normal		

### COMMON output:

TOB	observation time, d.p. sec past 1950.0
RVEC	array of computed data types
RVECO	array of observed data types
RVECW	array of weight code-words
RVECD	array of residual-weight words
PASS	pass identification
XFREQ	transmitter frequency, cps
TFREQ	transponder frequency, cps
IR	receiver identification
IX	transmitter identification
TAU	doppler averaging time, sec

The ODG logical record format is as follows:

```
43-2 of 2
```

```
0 word count (2N + 5, B17) or -0 for EOF
Word
       1 TOB
       2 TOB - 1
       3 XFREQ if IX ≠ 0
         TFREQ if IX # 0
       4 PASS
                     1 if L - S band
       5 bit S
                     0 if L or S band
                     1 if RVECO ≠ 0
             1
                     1 if RVECO - 1 \neq 0
             2
                     1 if RVECO - 2 \neq 0
              3
             11
             12-15
                     not used
                     1 if L - S or S band
             16
                     0 if L band
                     l if optical data
             17
             18-21
            22-25
                     IR
            26-35
                    TAU.
       6 first member of RVECO which is non-zero
       7 bits S-17 weight codeword corresponding to word 6
             18-35
                     data sample rate, sec
       8 second member of RVECO which is non-zero
```

```
2N + 4 Nth member of RVECO which is non-zero
```

9 second weight-sample rate word

2N + 5 Nth weight-sample rate word

The corresponding residual record format is found in the documentation of subroutine FILL.

## CODING INFORMATION

Length of subroutine is 142 (10) or 216 (8) words.

LOCO

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

## PURPOSE

This is the control routine for the link which restores the ODP COMMON from disk following computation of the trajectory. In addition it controls preparation of the combined ephemeris on tape A4.

#### RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. COMMON break: 47675
- c. Subroutines used:

KOOL

TAPIO

INTR1

**EPPHEM** 

FLAK

TYPRYT

# USE

This is one of the basic links of the ODP in that it restores COMMON and prepares the ephemeris. It is called under control of the JPTRAJ monitor.

#### CODING INFORMATION

Length of subroutine is 116 (10) or 164 (8) words.

IDENTIFICATION 45-1 of 2

LOOKUP/KOOL Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

## PURPOSE

To perform look-up on the probe ephemeris generated by SPACE.

## RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

ERROR Called when time incompatible with the ephemeris which has been generated.

REDE

## METHOD

The subroutine computes the coefficients for a 5th degree Lagrangian interpolation and provides values for the position, velocity, acceleration and jerk (3rd derivative) of the probe, variational equations, nutation in longitude and nutation in obliquity. (Flow chart is included)

$$y(x) = \sum_{k=0}^{w} 1_k(x)f(x_k)$$

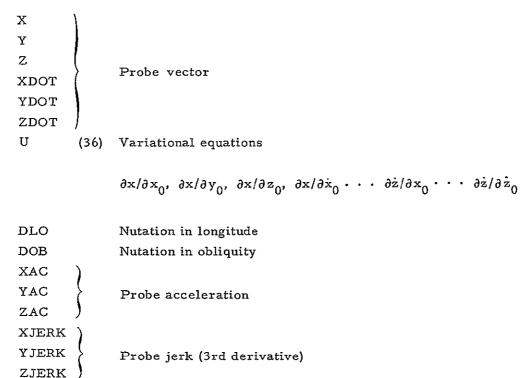
where

$$l_{i}(x) = \frac{\pi(x)}{(x - x_{i})\pi'(x_{i})}$$

$$= \frac{(x - x_{0}) \cdot \cdot \cdot (x - x_{i-1})(x - x_{i+1}) \cdot \cdot \cdot (x - x_{n})}{(x_{i} - x_{0}) \cdot \cdot \cdot (x_{i} - x_{i-1})(x_{i} - x_{i+1}) \cdot \cdot \cdot (x_{i} - x_{n})}$$

45-2 of 2

For each time that LOOKUP is called the following information is stored in COMMON



# USE

CALL LOOKUP (n) n = number of items to be interpolated, 6 or 49. When 6, the probe vector only is given.

CALL KOOL to provide a logical reset on the probe ephemeris file and to set flag for LOOKUP to read 3 physical records needed on first entry to the subroutine.

# CODING INFORMATION

Length of subroutine is 314 (10) or 472 (8) words.

## REFERENCE

Hildebrand, F. B., Introduction to Numerical Analysis, McGraw-Hill, New York, 1956.

MAMUL

Melba W. Nead, JPL Fortran II, Version 3

January 4, 1965 ·

# PURPOSE

To multiply two matrices and store in a third array.

# RESTRICTION

Maximum array:  $20 \times 20$ 

USE

CALL MAMUL (A, B, C, M, N, L)  

$$A(M, L) * B(L, N) = C (M, N)$$

## CODING INFORMATION

Length of subroutine is 87 (10) or 127 (8) words.

MAPOUT

Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

## PURPOSE

MAPOUT reads all matrices which were computed and stored on disk in Link 2. Here they are printed and in some cases punched. Also, the mapping information is saved for the MMP (MATRIX MANIPULATOR PROGRAM) input tape.

#### RESTRICTIONS

a. ERROR condition: disk error indicated by DCP

b. COMMON break: 47055

c. Subroutines used:

PRIM

FLAK

REFORM

PUMA

PRAM

DCP

IOCS

TYPRYT

OFFSYS

TACCOM

## USE

This link is called under control of the JPTRAJ Source Deck.

# CODING INFORMATION

Length of subroutine is 866 (10) or 1542 (8) words.

MIXAM

Melba Nead, JPL Fortran II, Version 3

7 4 10/5

January 4, 1965

## PURPOSE

Checks data tape, mapping times, predictions, pointing times, etc. so that the trajectory can be run a minimum length of time and still satisfy all demands on the probe ephemeris.

## RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

KINE

MOCT

QUIZ

(DFAD)

(DFSB)

## METHOD

Flow chart is included.

USE

CALL MAXIM

# CODING INFORMATION

Length of subroutine is 315 (10) or 473 (8) words.

MOCT

Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

## PURPOSE

MOCT obtains the latest occultation or impact time for determining the trajectory link time stop.

# RESTRICTION

COMMON break: 47675

## USE

CALL MOCT

PZE LAST latest occultation/impact time, d.p. sec past 1950.0

COMMON input:

PTFD occultation/impact input area

DXDR-27 occultation/impact flag

# CODING INFORMATION

Length of subroutine is 71 (10) or 107 (8) words.

ND2F

Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

#### PURPOSE

This is a SHARE routine which has been modified so that it is possible to start at a predetermined portion of the random number generator.

# RESTRICTION

COMMON break: 47675

# METHOD

RANO, an octal number, can be input into the ODP data. This number is chosen from a pre-calculated set.

#### USE

To provide noise on a simulated data tape.

# CODING INFORMATION

Length of subroutine is 39 (10) or 47 (8) words.

NOMNL

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

NOMNL stores the nominal values of all constants, other than the weighting tables and the target-dependent solar pressure constants.

# RESTRICTION

COMMON break: 47675

#### USE

CALL NOMNL

This subroutine outputs into approximately 150 COMMON locations.

### CODING INFORMATION

Length of subroutine is 499 (10) or 1051 (8) words.

# REFERENCES

- a. Clarke, Victor C., TR 32-604, March 6, 1964.
- b. Scott, James F., IOM 317.21/318, September 1, 1964.

NORMAY

Melba Nead, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

To normalize a matrix on its diagonal terms.

# RESTRICTIONS

a. Maximum matrix:  $20 \times 20$ 

b. Subroutines used:

SQRT

# USE

CALL NORMAY (XMAT, YMAT, N)

XMAT = matrix A be normalized

YMAT = location to store normalized matrix

N = order of matrix

#### CODING INFORMATION

Length of subroutine is 169 (10) or 251 (8) words.

NOUT

Michael R. Warner, JPL, IBM 7094 Fap
January 4, 1965

#### PURPOSE

NOUT prints comments indicating the rows and columns of the normal matrix which were deleted by subroutine STPREG during inversion:

VARIABLE NO. XX REJECTED BY DIAGONAL TEST.

# RESTRICTION

Subroutines used:

PROUT

USE

CALL NOUT

PZE INOUT

INOUT is a 1  $\times$  20 array in which STPREG has indicated the status of each variable. COMMON input:

PRIFIL PROUT file control block for printing.

# CODING INFORMATION

Length of subroutine is 64 (10) or 100 (8) words.

IDENTIFICATION 54-1 of 3

OBTOX

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

OBTOX computes the partials of the ODP data types with respect to the probe position and velocity at the observation time.

#### RESTRICTION

COMMON break: 47675

# METHOD

$$\frac{\partial \rho_{\mathbf{i}}}{\partial \underline{x}} = \underline{L}$$

$$\frac{\partial \rho_{\mathbf{i}}}{\partial \dot{x}} = 0$$

$$\frac{\partial \dot{\rho}_{\mathbf{i}}}{\partial x} = \frac{1}{\rho_{\mathbf{i}}} (\dot{x} + \omega y_{1} - \dot{\rho}_{\mathbf{i}} L_{x})$$

$$\frac{\partial \dot{\rho}_{\mathbf{i}}}{\partial y} = \frac{1}{\rho_{\mathbf{i}}} (\dot{y} - \omega x_{1} - \dot{\rho}_{\mathbf{i}} L_{y})$$

$$\frac{\partial \dot{\rho}_{\mathbf{i}}}{\partial z} = \frac{1}{\rho_{\mathbf{i}}} (\dot{z} - \dot{\rho}_{\mathbf{i}} L_{z})$$

$$\frac{\partial \dot{\rho}_{\mathbf{i}}}{\partial z} = \underline{L}$$

$$\frac{\partial \gamma_{\mathbf{i}}}{\partial \underline{r}} = \underline{D}$$

$$\frac{\partial \gamma_{\mathbf{i}}}{\partial z} = 0$$

$$(elevation angle partials)$$

$$\frac{\partial \sigma_{i}}{\partial \underline{x}} = \frac{\widetilde{\underline{A}}}{\rho_{i}}$$
(azimuth angle partials)
$$\frac{\partial \sigma_{i}}{\partial \underline{x}} = 0$$

$$\frac{\partial \delta_{i}}{\partial x} = \frac{-\cos \alpha_{xi} \sin \delta_{i}}{\rho_{i}}$$

$$\frac{\partial \delta_{i}}{\partial y} = \frac{-\sin \alpha_{xi} \sin \delta_{i}}{\rho_{i}}$$
(declination partials)
$$\frac{\partial \delta_{i}}{\partial z} = 0$$

$$\frac{\partial \delta_{i}}{\partial z} = 0$$

$$\frac{\partial \delta_{i}}{\partial z} = \frac{\cos \delta_{i}}{\rho_{i}}$$

$$\frac{\partial \delta_{i}}{\partial z} = 0$$

$$\frac{\partial \alpha_{i}}{\partial x} = \frac{\sin \alpha_{xi}}{\rho_{i} \cos \delta_{i}}$$
(hour angle partials)
$$\frac{\partial \alpha_{i}}{\partial z} = 0$$

$$\frac{\partial \alpha_{i}}{\partial z} = \frac{\alpha_{2}}{c} \frac{\partial \dot{\rho}_{i}}{\partial z} \quad \underline{r} \to \dot{\underline{r}} \quad \text{(one-way doppler partials)}$$

$$\frac{\partial^{f}_{C3i, q}}{\partial \underline{x}} = \frac{\Omega_{4}}{c} \left(\frac{\partial \dot{\rho}_{i}}{\partial \underline{r}} + \frac{\partial \dot{\rho}_{q}}{\partial \underline{x}}\right) \quad \underline{r} \to \dot{\underline{r}} \quad \text{(two-way doppler partials)}$$

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$$\frac{\partial f_{3i,q}}{\partial \underline{r}} = \frac{\Omega_6}{c} \left( \frac{\partial \dot{\rho}_i}{\partial \underline{r}} + \frac{\partial \dot{\rho}_q}{\partial \underline{r}} \right) \quad \underline{\underline{r}} \to \underline{\dot{r}} \quad \text{(three-way doppler partials)}$$

$$\frac{\partial f_{\text{dli,j}}}{\partial \underline{r}} = \frac{\Omega_2}{c} \left( \frac{\partial \dot{\rho}_i}{\partial \underline{r}} - \frac{\partial \dot{\rho}_j}{\partial \underline{r}} \right) \quad \underline{r} \to \underline{\dot{r}} \quad \text{(differenced one-way partials)}$$

(all symbols used here are defined in the subroutine CATS documentation)

#### USE

CALL OBTOX

COMMON input:

ELX 1

TDX D

RVEC ODP computed data type array

 $x \underline{r}, \dot{r}$ 

OMEGA  $\omega$ , deg/sec

RAI a, deg

CZ2  $\Omega_2$ 

CZ4  $\Omega_4$ 

CZ6  $\Omega_6$ 

VELC c, km/sec

COMMON output:

POBX ∂F/∂r, ∂F/∂r

#### CODING INFORMATION

Length of subroutine is 189 (10) or 275 (8) words.

# REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Anderson, John D., TM 312-409, March 24, 1964.

OCIM

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

# PURPOSE

OCIM reads the ODP combined ephemeris tape, extracting occultation and impact time parameters necessary for partials calculation.

# RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines called:

**IMPAR** 

OCPAR

READB

FLAT

ERROR

c. Error conditions:

Ephemeris tape permanent redundancy

#### USE

### CALL OCIM

### COMMON input:

DXDR-6	target (1 = Mars 2 = Moon 3 = Venus)
DXDR-27	1B17 = occultation time 2B17 = impact time
OOMX	geocentric Moon vector
XVEN	geocentric Venus vector
XMAR	geocentric Mars vector
PTFD	occultation-impact times input array
X	geocentric probe vector
XAC	geocentric probe accelerations
XJERK	geocentric probe jerks
RVECD	residual
RVECW	weight
DINT	physical constants partials
T	corrected occultation/impact time

IDENTIFICATION 56-1 of 3

OCPAR

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

OCPAR computes the partials of occultation time with respect to the estimated and considered parameters.

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

FORM

**METHOD** 

$$\frac{\partial T_{0}}{\partial Q} = \frac{\underline{\rho} \cdot \underline{\rho}_{m} \left( \frac{\underline{\rho}_{m} \cdot \frac{\partial \underline{\rho}_{m}}{\partial Q}}{\underline{\rho}_{m}^{2}} + \frac{\underline{\rho} \cdot \frac{\partial \underline{\rho}}{\partial Q}}{\underline{\rho}^{2}} \right) - \underline{\rho} \cdot \frac{\partial \underline{\rho}_{m}}{\partial Q} - \underline{\rho}_{m} \cdot \frac{\partial \underline{\rho}}{\partial Q}} + \frac{|\underline{\rho}| |\underline{\rho}_{m}| \sin \phi_{p} R_{m}}{\underline{\rho}_{m}^{2} \sqrt{\underline{\rho}_{m}^{2} - R_{m}^{2}}} \underline{\rho}_{m} \cdot \frac{\partial \underline{\rho}_{m}}{\partial Q}} {\underline{\rho}_{m}^{2} \sqrt{\underline{\rho}_{m}^{2} - R_{m}^{2}}} \underline{\rho}_{m} \cdot \frac{\partial \underline{\rho}_{m}}{\partial Q}}$$

$$= \frac{\underline{\rho} \cdot \underline{\rho}_{m} \left( \underline{\rho}_{m} \cdot \underline{\dot{\rho}}_{m} + \frac{\underline{\rho} \cdot \dot{\underline{\rho}}}{\underline{\rho}^{2}} \right) - \underline{\rho} \cdot \underline{\rho}_{m} - \underline{\rho}_{m} \cdot \underline{\rho}_{m}} + \underline{\rho}_{m} \cdot \underline{\rho}_{m} + \frac{\underline{\rho} \cdot \dot{\underline{\rho}}}{\underline{\rho}^{2}} - \underline{\rho}_{m} \cdot \underline{\rho}_{m}}{\underline{\rho}_{m}^{2} \sqrt{\underline{\rho}_{m}^{2} - R_{m}^{2}}} \underline{\rho}_{m} \cdot \underline{\rho}_{m} \cdot \underline{\rho}_{m}}$$

where

 $\underline{\rho}_{m} = \underline{r}_{m} - \underline{R}_{i}$ 

R<sub>m</sub> = radius of target

 $\underline{\underline{r}}_{m}$  = geocentric target vector

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$$\phi_{\rm p} = \cos^{-1} \frac{\frac{\rho}{\rho} \cdot \frac{\rho}{m}}{\left|\frac{\rho}{\rho}\right| \left|\frac{\rho}{m}\right|}$$

 $\rho$  = topocentric probe vector

 $\underline{R}_i$  = geocentric station vector

# USE

CALL OCPAR

PZE XTAR 
$$\underline{r}_{m}$$
, km

PZE AA  $A = \underline{\Gamma} \cdot \dot{\underline{\Gamma}}$ 

PZE AB  $B = \underline{\beta} \cdot \dot{\underline{\beta}}$ 

PZE AC  $C = \sqrt{\Gamma^{2}\beta^{2}} - (\underline{\Gamma} \cdot \underline{\beta})^{2}$ 

PZE AK  $K = \underline{\Gamma} \cdot \dot{\underline{\beta}} + \underline{\beta} \cdot \dot{\underline{\Gamma}}$ 

PZE AM  $M = R_{m}C/\beta^{2}\sqrt{\beta^{2} - R_{m}^{2}}$ 

PZE GDB  $\underline{\Gamma} \cdot \underline{\beta}$ 

PZE BETA  $|\underline{\beta}| (\underline{\beta} = \underline{\rho}_{m})$ 

PZE GAMMA  $|\underline{\Gamma}| (\underline{\Gamma} = \underline{r} - \underline{R}_{i})$ 

PZE XXX  $\underline{\rho}_{m} \cdot \underline{r}$ 

PZE XXI  $\underline{r} \cdot \underline{R}_{i}$ 

PZE SI  $\underline{R}_{i}$ 

# COMMON input:

COSPHI  $\sin \phi_i$ 

U	partials $\partial P/\partial \underline{r}_0$ (6 × 6)
GMAT	partials $\partial \underline{\rho}/\partial q$ (6 × 11)
LLIST	physical constants flags (II)
RI	$\left  \frac{\mathbb{R}_{\mathbf{i}}}{\mathbb{R}_{\mathbf{i}}} \right $ , km
NLIST	station locations flags (3 $\times$ 15)
COSRAI	cos a
SINRAI	sin. a <sub>ri</sub>

# JPL TECHNICAL MEMORANDUM NO. 33-204

# COMMON output:

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QA partials  $\partial T_0/\partial q$ QB partials  $\partial T_0/\partial \underline{r}_0$ DIM3 partials  $\partial T_0/\partial s_i$ 

# CODING INFORMATION

Length of subroutine is 333 (10) or 515 (8) words.

# REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Liu, Anthony, RFP 312-136, May 20, 1963.

IDENTIFICATION 57-1 of 2

ODATA/ONLIN/.....
Michael R. Warner, JPL
IBM 7094 Fap
January 4, 1965

#### PURPOSE

ODATA performs the following functions:

- a. In conjunction with the conversion subroutine CARDS, reads ODP control cards and the symbolic and/or numeric data cards. The cards may be read from the on-line reader or input tape A2 (mode 4) or the user area reader (mode 2).
- b. Manipulates the symbolic and numeric data and stores them in the appropriate COMMON locations for subsequent ODP use.
- c. Under sense switch (or console key) control, terminates ODP execution.
- d. Prints the trajectory target on the administrative (or on-line) printer.
- e. Stores the target-dependent solar pressure constants.
- f. On option, reads the previously computed injection conditions, constants, and covariance matrix from disk.
- g. Applies Gauss' constraint to the planetary masses:

$$GM_s = 3.9640160 \times 10^{-14} a_e^3$$

$$GM_v = \frac{M_v}{M_s} GM_s$$

$$GM_r = \frac{M_r}{M_s} GM_s$$

$$GM_j = \frac{M_j}{M_s} GM_s$$

where

a = astronomical unit

s, v, r, j = subscripts denoting the Sun, Venus, Mars, Jupiter.

h. On option, applies the scaling constraint to the Earth radius:

$$R_{em} = 86.315745 (GM_e + GM_m)^{1/3}$$

where

57-2 of 2

R<sub>em</sub> = earth radius for scaling the lunar ephemeris.

#### RESTRICTIONS

- a. If any of the following error conditions are detected, ODATA calls subroutine ERROR for the appropriate comment printout and subsequent action:
  - 1. Error in symbolic or numeric input.
  - 2. Checksum or redundancy error.
  - 3. Cards not received within time limit (mode 2).
  - 4. Illegal card in deck (mode 2)
  - 5. End-of-file indication (mode 4)
  - 6. Data on input card inconsistent with ODP requirements.
  - 7. Error in reading disk.
- b. COMMON break: 47675
- c. Subroutines called:

TYPRYT

IOCS - DEFINE/ATTACH/OPEN/CLOSE

CARDS

**OFFSYS** 

ERROR

FINSYS

READS/MOOPH1/ODOFF

FLOT

SPHX

STPREG

DIAG

DISCBU - REDE

EXP(3

# USE

#### CALL ODATA

ODATA uses most of the ODP COMMON map. Section IVB has the COMMON listing. The input descriptions are in Section IVA.

ONLIN is used for returning to ODATA after a corrected card input error (TTR \$ONLIN).

..... represents the entry point for the JPTRAJ program control block of LA9.

#### CODING INFORMATION

Length of subroutine is 1723 (10) or 3273 (8) words.

ORBEQ

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

ORBEQ rotates the encounter noise moment matrix from the ODP  $\underline{R}$ ,  $\underline{T}$  plane to the midcourse program  $\underline{R}$ ,  $\underline{T}$  plane.

# RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used

ARTAN

COS

SIN

METHOD

$$\theta = \tan^{-1} \frac{\underline{B} \cdot \underline{R}_{m/c}}{\underline{B} \cdot \underline{T}_{m/c}} - \tan^{-1} \frac{\underline{B} \cdot \underline{R}_{odp}}{\underline{B} \cdot \underline{T}_{odp}}$$

$$R = \begin{bmatrix} \cos \theta & -\sin \theta & 0 \\ \sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

USE

CALL	ORBEQ	
PZE	BRO	ODP $\underline{B} \cdot \underline{R}$ , km
PZE	BTO	ODP $\underline{B} \cdot \underline{T}$ , km
PZE	BRM	$M/C \underline{B} \cdot \underline{T}$ , km
PZE	BTM	$M/C B \cdot T$ , km
PZE	CNO	ODP noise moment matrix (6 $\times$ 6)
PZE	CNM	M/C noise moment matrix $(3 \times 3)$ .

#### CODING INFORMATION

Length of subroutine is 119 (10) or 167 (8) words.

ΟZ

Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

### PURPOSE

OZ is the control program for the calculation of the closest approach parameters.

# RESTRICTIONS

- a. COMMON break: 47055
- b. Subroutines used:

UMAT

TYPRYT

IOCS

**OFFSYS** 

# USE

When impact or closest approach parameters are requested, this link is called under control of the JPTRAJ Source Deck.

# CODING INFORMATION

Length of subroutine is 766 (10) or 1376 (8) words.

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PARAM

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

# PURPOSE

PARAM outputs the encounter parameters page of the ODP printout. It also loads the buffers for COMENT output.

# RESTRICTIONS

- a. COMMON break: 47055
- b. Subroutines called:

SQRT

FIXT

PROUT

TNORM

ORBEQ

# USE

CALL	PARAM		
PZE	CNO	noise moment matrix of encounter parameters (6 $\times$ 6)	
PZE	BRO	$\underline{B} \cdot \underline{R}$ , km (ODP plane)	
PZE	BTO	$\underline{B} \cdot \underline{T}$ , km (ODP plane)	
PZE	ELPS	miss ellipse configuration (1 $\times$ 3 - semi-major axis,	
		semi-minor axis, orientation angle)	
PZE	Α	semi-major axis of encounter conic	
PZE	UNJ	$\partial M/\partial Q_0$ matrix of encounter partials	
PZE	TL	linearized time of flight, days	
PZE	ŪΤ	$\partial M/\partial Q$ matrix of encounter partials	
COMMON input:			

DXDR-6	target identification
DXDR-21	$\underline{B} \cdot \underline{R}$ (midcourse plane)
DXDR-20	$\underline{B} \cdot \underline{T}$ (midcourse plane)
DXDR-2	COMENT update flag
GRAVE	$GM_{Earth}$ , $km^3/sec^2$
GRAVR	$^{ m GM}_{ m Mars}$
GRAVV	$GM_{Venus}$

#### JPL TECHNICAL MEMORANDUM NO. 33-204

GRAVM GM<sub>Moon</sub> 60-2 of 2

GRAVS GM<sub>Sun</sub>

TFIN time of impact/closest approach, d.p. seconds past 1950.0

COMTS ODP page heading, BCD PRIFIL PROUT file control block

XNJ ODP initial conditions (1 × 6), d.p. EPOCH ODP epoch, d.p. sec past 1950.0

XDOT probe velocity at encounter,  $km/sec (1 \times 3)$ 

COMMON output:

COMBU midcourse program COMENT parameters
COMBW trajectory program COMENT parameters

#### CODING INFORMATION

Length of subroutine is 720 (10) or 1320 (8) words.

61-1 of 2

PARLEY

Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

#### PURPOSE

To write time, partials, position and velocity of probe and planets, etc., on. tape for each data point while the data is being processed.

# RESTRICTIONS

- a. COMMON break: 47675
- b. Cannot be used in SFOF MODE 2 (real-time mode).
- c. Subroutines used:

FLAT TAPIO

#### USE

#### CALL PARLEY

The following items are written on a scratch tape to be read and printed in a subsequent link:

TOB			Observation time
GHA			Greenwich Hour Angle
X		1	
Y			
Z		(	Probe vector
XDOT			Trobe vector
YDOT		)	
ZDOT		1	
U	(36)		Variational equations
DLO			Nutation in longitude
DOB			Nutation in obliquity
XAC		)	
YAC		}	Probe Acceleration
ZAC		)	
XJERK		)	
YJERK		}	Probe jerk (3rd derivative)
ZJERK		)	
XSUN	(6)		Sun vector

XVEN	(6)	Venus vector	61-2 of 2
XMOO	(6)	Moon vector XTARG vector only	
XMAR	(6)	Mars vector	
CPPHI	(20)	CAP PHI matrix of partials	
CPTHT	(20)	CAP THETA matrix of partials	
DELT		Light time correction	
DELT2		Second light time correction	
DHA	1		
DDEC			
DAZ	(	N. C	
DEL	Refraction corrections		
DRDOT			
DRVEC		1	

# CODING INFORMATION

Length of subroutine is 30 (10) or 36 (8) words.

# REFERENCE

Null, George, W., RFP 312-179, August 15, 1963.

62

PARSH

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

This link is called only when the COMENT region is to be updated, or when in MODE 4, the printing of the partials is requested. The program is called under control of the JPTRAJ monitor.

# RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutine used:

COMOUT

WASH

FGDOUT

DISCBU

#### USE

Control program for writing COMENT on disk or printing partials.

# CODING INFORMATION

Length of subroutine is 662 (10) or 1226 (8) words.

PERNOD/WOLF

Charles Coltharp/Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

#### **PURPOSE**

PERNOD accumulates statistics on data residuals. It is called during the processing of each residual. WOLF is called at the end-of-file for each station's residuals. It obtains the standard deviation, root-mean-square, mean, and second moment for each data type and each pass identification.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

FIXT

PROUT/FGDOUT

SQRT

COLA

# USE

CALL PERNOD after each residual is read

CALL WOLF after each station is completed

### COMMON input:

TOB observation time, d.p. seconds past 1950.0

RVECD residual array  $(1 \times 11)$ 

ITNO iteration number
PASS pass identification

IR receives identification

PRIFIL IOCS printing file control block

# CODING INFORMATION

Length of subroutine is 323 (10) or 503 (8) words.

PM360

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

PM360 adjusts angle residuals by 360 deg if they are greater than 180 deg or less than -180 deg.

#### RESTRICTIONS

COMMON break: 47675

#### USE

CALL PM360

COMMON input:

DECD declination residual

HAD hour angle residual

AZID azimuth residual

COMMON output:

DECD ]

HAD residuals adjusted by 360 deg if necessary

AZID

# CODING INFORMATION

Length of subroutine is 40 (10) or 50 (8) words.

IDENTIFICATION 65-1 of 2

PNUT

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

PNUT calculates the rotation matrix for nutation-precession.

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SIN

COS

#### METHOD

$$N = \begin{bmatrix} 1 & -\Delta\lambda \cos \overline{\epsilon} & -\Delta\lambda \sin \overline{\epsilon} \\ \Delta\lambda \cos \overline{\epsilon} & 1 & -\Delta\epsilon \\ \Delta\lambda \sin \overline{\epsilon} & \Delta\epsilon & 1 \end{bmatrix}$$

$$a_{11} = 1 - 0.29697 \times 10^{-3} T^2 - 0.13 \times 10^{-6} T^3$$

$$a_{12} = -0.02234988T - 0.676 \times 10^{-5} T^2 + 0.221 \times 10^{-5} T^3$$

$$a_{13} = -0.00971711T + 0.207 \times 10^{-5} T^2 + 0.96 \times 10^{-6} T^3$$

$$a_{21} = -a_{12}$$

$$a_{22} = 1 - 0.24976 \times 10^{-3} \text{ T}^2 - 0.15 \times 10^{-6} \text{ T}^3$$

$$a_{23} = -0.10859 \times 10^{-3} \text{ T}^2 - 0.3 \times 10^{-7} \text{ T}^3$$

$$a_{31} = -a_{13}$$

$$a_{32} = a_{23}$$

65-2 of 2

$$a_{33} = 1 - 0.4721 \times 10^{-4} T^2 + 0.2 \times 10^{-7} T^3$$

#### where

 $\Delta \lambda$  = nutation in longitude

 $\Delta \epsilon$  = nutation in obliquity

 $\bar{\epsilon}$  = mean obliquity (see writeup for GHADP)

T = Julian centuries past 1950.0

#### USE

# CALL PNUT

# COMMON input:

T time, d.p. sec past 1950.0

DOB nutation in obliquity, deg

DLO nutation in longitude, deg

# COMMON output:

ROTMX nutation-precession matrix (NA)

### CODING INFORMATION

Length of subroutine is 222 (10) or 336 (9) words.

# REFERENCE

Holdridge, D. B., TR 32-223, March 2, 1962.

POINT

Melba W. Nead, JPL Fortran II, Version 3

January 4, 1965

#### PURPOSE

To generate the pointing predictions for the tracking stations.

# RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

COEF

FILL

**OFFSYS** 

CORR

GHADP

REAP

DOPLR

IXTAB

(DFAD)

USE

CALL POINT

### CODING INFORMATION

Length of subroutine is 196 (10) or 304 (8) words.

# REFERENCE

Trask, D.W., RFP 312-37, Addendum 6, April 4, 1962.

PONT

Melba W. Nead, JPL Fortran II, Version 3

January 4, 1965

# PURPOSE

To compute the times at which the combined ephemeris shall be written for a simulated data tape or predictions.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

GAMAT

INTR1

LOOKUP

OFFSYS

RITEM

TIMER

(DFAD)

Flow chart included

USE

CALL PONT

# CODING INFORMATION

Length of subroutine is 201 (10) or 311 (8) words.

PRAMS

Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

#### PURPOSE

To save the U matrix when it is calculated by COMIMP. Core space forces the calculation and the print-out of these matrices to be done in separate links.

#### RESTRICTIONS

- a. ERROR condition: disk error indicated by DCP
- b. Subroutines used:

FLAK DCP

USE

CALL PRAMS

### CODING INFORMATION

Length of subroutine is 18 (10) or 22 (8) words.

PREDA

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

This is the main program of the link which computes predictions for the DSIF stations or prepares a simulated data tape.

# RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

POINT

DATAPE

TYPRYT

TAPIO

# USE

This link is called under control of the JPTRAJ Source Deck when predictions or a data tape are requested.

# CODING INFORMATION

Length of subroutine is 39 (10) or 47 (8) words.

IDENTIFICATION 70-1 of 2

PRIM/PRAM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

To print any matrix up to  $20 \times 20$  from a possible list of 63. Included is the capability of punching the array in a format acceptable as input back into the ODP. When stepmapping, probe position and velocity and the mapping matrix are automatically punched.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

BIBCD

FLAPR

PRIME

NORMAY

PRIT

DIAGO

FGDOUT

PROUT

#### USE

CALL PRIM

PZE ,,N

### where N is as follows:

- 1. Input J matrix of estimated parameters
- 2. Input covariance matrix of considered parameters
- 3. Jinverse
- 4. Covariance matrix of estimated parameters
- 5. Correlation matrix of estimated parameters
- 6. Covariance matrix at impact
- 7. Covariance matrix mapped forward
- 8. J matrix
- 9. Input covariance matrix of estimated parameters
- 10. Correlations based on J matrix
- 11. U product matrix in step-mapping

CALL PRAM

70-2 of 2

TSX CAPU

Prints U matrix for mapping forward

To punch a matrix PUNCH=n under OTHER PARAMETER VALUES in the data input. This will punch the matrix only one time.

# CODING INFORMATION

Length of subroutine is 787 (10) or 1423 (8) words.

# REFERENCE

Trask, D.W., REP 312-37, Addendum 6, April 4, 1962.

PRIME/REFORM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

To write impact times, probe and target vectors and the mapping matrices on the MATRIX MANIPULATOR PROGRAM input tape.

#### RESTRICTIONS `

- a. COMMON break: 47675
- b. Subroutines used:

TAPIO

FLAT

#### USE

CALL REFORM To read mapping times from preliminary storage tape. TFIN is

added to those originally saved and all are written on the final

MMP tape.

CALL PRIME To write the mapping matrices on the MMP tape.

# CODING INFORMATION

Length of subroutine is 98 (10) or 142 (8) words.

#### REFERENCE

Peterson, G. E., ED 218, May 15, 1964.

PRINQ/PRIT/APRIOR

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

To print DQ's, new Q's, old Q's, standard deviation of covariance matrix of estimated parameters, etc.

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

**FGDOUT** 

DIAG

CLOCK

PROUT

FLAPR

FIXT

BIBCD

USE

CALL PRINQ Prints DQ's etc. of estimated parameter.

CALL PRIT Entered with double precision time in seconds in ACC and MQ.

These are converted to year, month, day, hour, minute, second,

and stored in COMMON.

APRIOR Buffer in which the standard deviation of the aprior covariance

matrix is stored through the "WANT" capability of the JPTRAJ

monitor.

#### CODING INFORMATION

Length of subroutine is 259 (10) or 403 (8) words.

### REFERENCE

Trask, D.W., RFP 312-37, Addendum 6, April 4, 1962.

PRINT

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

PRINT controls the printing of the DQ's and the matrices which have been computed in the preceding link. Also, when occultation calculations are requested it reads this information from disk and prepares it for printing.

#### RESTRICTIONS

a. ERROR condition: disk error indicated by DCP

b. COMMON break: 47675

c. Subroutines used:

PRINQ

PRIM

FIXT

PROUT

FGDOUT

FLAK

DISCBU

IOCS

ENDIT

OFFSYS

# USE

Upon completion of the calculation of the data fitting link, the print link is called under control of the JPTRAJ MONITOR.

#### CODING INFORMATION

Length of subroutine is 1352 (10) or 2510 (8) words.

PROM

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

# PURPOSE

To multiply two square matrices and store in a third array.

# RESTRICTION

Maximum array:  $20 \times 20$ 

USE

CALL PROM (A, B, C, N)  

$$A(N, N) * B(N, N) = C(N, N)$$

# CODING INFORMATION

Length of subroutine is 110 (10) or 156 (8) words.

QUEST

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

#### PURPOSE

Checks to ascertain that the times specified are greater than EPOCH plus light time correction. Exits with an error return if time is not compatible.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

ERROR

SQRT

EXIT

USE

CALL QUEST

# CODING INFORMATION

Length of subroutine is 117 (10) or 165 (8) words.

QUIZ

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

### PURPOSE

QUIZ allows a Fortran program to interrogate the COMMON location KEYS, which contains the "key settings" specified by the input deck.

# RESTRICTION

COMMON break: 47675

USE

In Fortran,

B A = (octal representation of desired key setting)
IF (QUIZF (A)) S<sub>1</sub>, S<sub>2</sub>, S<sub>1</sub>

where

 $\mathbf{S}_1$  is the normal return indicating that the key is set  $\mathbf{S}_2$  is the normal return indicating that the key is not set

COMMON input:

KEYS key setting word

### CODING INFORMATION

Length of subroutine is 2 (10) or 2 (8) words.

IDENTIFICATION 77-1 of 4

RATES

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

### PURPOSE

RATES obtains the topocentric slant range and its first three derivatives. Both the up-leg and down-leg are calculated using the light-time correction supplied by subroutine DOPLR. Four other range-dependent quantities are also obtained.

### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines called:

COS

SIN

SQRT

ARSIN

EXP(3

METHOD

$$a_{r_i} = \lambda_i + a_G - 2\omega\Delta t$$
 (station right ascension)

$$X_i = R_i \cos \phi_i \cos \alpha_{ri}$$

$$Y_i = R_i \cos \phi_i \sin \alpha_{ri}$$

$$Z_i = R_i \sin \phi_i$$

$$\dot{X}_{i} = -\omega Y_{i}$$

$$\ddot{X}_{i} = -\omega Y_{i}$$

$$\ddot{Y}_{i} = \omega X_{i}$$

(station coordinates and derivatives)

$$\ddot{X}_{i} = -\omega Y_{i}$$

$$\ddot{Y}_{i} = \omega X_{i}$$

$$\frac{\rho_{i}}{\dot{\rho}_{i}} = \frac{\dot{x}}{\dot{\gamma}} - \frac{\dot{R}_{i}}{\dot{\alpha}_{i}}$$

$$\frac{\dot{\rho}_{i}}{\dot{\rho}_{i}} = \frac{\dot{x}}{\dot{\gamma}} - \frac{\dot{R}_{i}}{\dot{\alpha}_{i}}$$

$$\frac{\ddot{\rho}_{i}}{\ddot{\rho}_{i}} = \frac{\ddot{x}}{\ddot{\gamma}} - \frac{\ddot{R}_{i}}{\dot{\alpha}_{i}}$$

$$\frac{\ddot{\rho}_{i}}{\ddot{\rho}_{i}} = \frac{\ddot{x}}{\ddot{\gamma}} - \frac{\ddot{R}_{i}}{\dot{\alpha}_{i}}$$

$$A = \frac{\rho_{i}}{\dot{\gamma}} \cdot \frac{\dot{x}}{\dot{\gamma}}$$

$$B = \frac{\dot{R}_{i}}{\dot{\gamma}} \cdot \frac{\dot{x}}{\dot{\gamma}}$$

$$C = \dot{X}_{i}^{2} + \dot{Y}_{i}^{2}$$

$$D = \frac{\rho_{i}}{\dot{\gamma}} \cdot (\frac{\dot{R}_{i}}{\dot{\gamma}} - \frac{\dot{x}}{\dot{\gamma}})$$

$$a = |\rho|$$

$$b = \frac{\rho}{\dot{\gamma}} \cdot \frac{\dot{\rho}}{\dot{\rho}}$$

$$c = \frac{\rho}{\dot{\gamma}} \cdot \frac{\dot{\rho}}{\dot{\rho}} + \frac{\dot{\rho}}{\dot{\gamma}} \cdot \frac{\dot{\rho}}{\dot{\rho}}$$

$$\rho = a$$

$$\dot{\rho} = \frac{\dot{b}}{\dot{a}}$$

$$\ddot{\rho} = \frac{d - 3\dot{\rho}\dot{\rho}}{\dot{a}}$$

$$(slant range)$$

$$\Delta_{\mathbf{r}} \rho = \frac{0.0018958}{(\sin \gamma + 0.06483)^{1.4}} \frac{n_{\mathbf{i}}}{340.0}$$

$$\Delta_{\mathbf{r}} \rho = \frac{0.0018958}{\tau} \left[ \frac{1}{(\sin F + 0.06483)^{1.4}} - \frac{1}{(\sin G + 0.06483)^{1.4}} \right] \frac{n}{340.0}$$
(slant range)

where

 $\lambda_i$  = longitude station i

 $^{a}G$  = Greenwich hour angle

 $\omega$  = Earth rotation rate

 $\Delta t = light time correction$ 

 $\phi_i$  = geocentric latitude, station i

R; = Earth radius at station i

 $\underline{r}$  = geocentric probe coordinates

 $\underline{r}_{s}$  = geocentric sun coordinates

 $\gamma$  = elevation angle

n; = index of refraction, station i

 $\tau$  = doppler averaging time

$$F = \gamma + \frac{\tau \dot{\gamma}}{2}$$

$$G = \gamma - \frac{\tau \dot{\gamma}}{2}$$

USE

CALL RATES PZERHO P, km  $\hat{\rho}$ , km/sec DRHO PZEρ, km/sec<sup>2</sup> DDRHO PZEö, km/sec<sup>3</sup> PZE DDDRHO PZE A PZEВ В PZE C PZE D D PZEDT PZE Ι station index (IX for up-leg, IR for down-leg)

```
COMMON input:
                                                                                                                77-4 of 4
                    \theta_i, deg
    THETAI
                    \dot{a}_G^{}, deg
    GHA
    OMEGA
                    ω, deg/sec
    PHII
                    \phi_i, deg
    RI
                    R, km
    XN
                    <u>r</u>, km
                    \dot{\underline{r}}, km/sec
    DXN
                                                light time corrected
                   <u>r</u>, km/sec<sup>2</sup>
    DDXN
                   r, km/sec<sup>3</sup>
    DDDXN
    XSUN
                    \frac{\mathbf{r}}{\mathbf{s}}, \frac{\dot{\mathbf{r}}}{\mathbf{s}}
    EL
                    γ, deg
    FNI
                    n,
    TAU
                    τ, sec
COMMON output:
    RVEC
                    \rho, km
    RDOT
                    ρ, km/sec
    DRVEC
                    \Delta_r \rho, km
                    \Delta_{r}\rho, km/sec
    DRDOT
```

### CODING INFORMATION

Length of subroutine is 435 (10) or 663 (8) words.

### REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Anderson, John D., TM 312-409, March 24, 1964.

RCAL

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

### PURPOSE

RCAL is to be used to compute the coefficients as required in BMATRX:

$$A = \left(\frac{1}{||\overrightarrow{RV} - \overrightarrow{R}||^3} - \frac{1}{RV^3}\right)$$

$$B = \left(\frac{3(\overrightarrow{RV} \cdot \overrightarrow{R} - R^2)}{||\overrightarrow{RV} - \overrightarrow{R}||^5}\right)$$

$$C = \left(\frac{3}{||\overrightarrow{RV} - \overrightarrow{R}||^3}\right)$$

### RESTRICTIONS

a. COMMON break: 47675

b. Subroutine used:

SQRT

USE

CALL RCAL(RV) RV, location of the body vector. Results are stored in COMMON locations A, B, and C.

# CODING INFORMATION

Length of subroutine is 159 (10) or 237 (8) words.

**RCOM** 

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

#### PURPOSE

RCOM computes the coefficients as defined in BMATRX:

$$A = \left(\frac{1}{||\vec{R}\vec{V} - \vec{R}||^3}\right)$$

$$B = \left(\frac{1}{RV^3}\right)$$

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SQRT

USE

CALL RCOM(RV) RV, location of the body vector. Results are stored in COMMON locations A and B.

# CODING INFORMATION

Length of subroutine is 91 (10) or 133 (8) words.

80

READS/ODOFF/MOOPH1

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

READS is the main program of the link of the ODP which reads data. The routine also includes phasing for all three possible targets, Moon, Mars, and Venus. The target is checked and the phasing is moved so as to be available to the trajectory through the WANT capability of the JPTRAJ monitor.

### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

ODATA

TAPIO

**OFFSYS** 

ERROR

#### USE

Upon the completion of each iteration, the logic of the program causes it to cycle back through READS, all under control of the Source Deck and the JPTRAJ monitor.

CALL ODOFF

MOOPH1

To end ODP and return to system as called from ODATA. Entry to allow JPTRAJ "WANT" feature to save information from this area of READS, namely, the phasing for SPACE.

#### CODING INFORMATION

Length of subroutine is 704 (10) or 1300 (8) words.

REAP/REAM Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

### PURPOSE

To read the combined ephemeris tape as it is written by the subroutine RITEM.

### RESTRICTIONS

- a. ERROR indication received by encountering EOF on tape
- b. Subroutines used:

TAPIO

FLAT

ERROR

#### METHOD

REAP-REAM read the combined ephemeris tape which has been written by RITEM. The subroutine uses two buffers for efficiency of operation.

# USE

CALL REAP Compares TOB with that saved on tape
CALL REAM Compares T with that saved on tape

# CODING INFORMATION

Length of subroutine is 1375(10) or 2537(8) words.

REJEC

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

REJEC keeps a record of those residuals which were rejected by the 3-sigma test. Each bit in the rejection table represents a residual. The table is ordered sequentially.

### RESTRICTION

COMMON break: 46711

#### USE

CALL REJEC

PZE :

PZE J

PZE K

#### where

I = 0 to initialize (set pointer at first bit)

= l for normal entry

J = 0 to determine rejection status of current point

= 1 to update rejection status of current point

COMMON input and output:

CAPU 500 word rejection table, with recording capability for 18,000 residuals

# CODING INFORMATION

Length of subroutine is 46 (10) or 56 (8) words.

RESID

Melba W. Nead, JPL IBM 7094 Fap

January 4, 1965

# PURPOSE

Control program for the link which prints and plots residuals and prints statistics.

# RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

THARP

TYPRYT

IOCS

### USE

Upon request for residuals and statistics, this link is called by the JPTRAJ monitor.

# CODING INFORMATION

Length of subroutine is 704 (10) or 1300 (8) words.

RESP/COLA

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

This subroutine is designed to provide both listings and plots of information from the receiving stations. The plots are the residuals, observed minus calculated values, for one, two, or three types versus time. Date, station number, time at start of frame, and pass number are included on each frame. In addition to the above, the listings contain count time, sample time, computed values for the data, and weights. Rejected data points are starred. It is possible to list the residual in fixed (nominal) or floating point (FLRES = 1.0).

### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

**FGDOUT** 

CXPLOT

PKINE

PERNOD

STATID

FIXT

PROUT

FLAPR

BIBCD

ERROR

CDC

A maximum of 8 data residuals can be plotted or listed on one iteration.

### USE

CALL RESP With each call to RESP a maximum of 3 residuals is plotted or listed.

The routine must be called again for more; the maximum number is 8.

#### CODING INFORMATION

Length of subroutine is 883 (10) or 1563 (8) words.

# REFERENCE

Trask, D.W., RFP 312-37, Addendum 6, April 4, 1962.

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RITEM/WREOF Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

# PURPOSE

To write the combined ephemeris tape.

### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

FLAT

TAPIO

# METHOD

RITEM uses TAPIO to write the combined ephemeris tape. The subroutine uses two buffers alternately to provide efficient timing. Each buffer contains four logical records of 156 words each.

### USE

CALL RITEM Writes logical record on tape A4. Each record contains the following information:

T (2)		Time of data point
TOB (2)		Observation time
XJUP (6)		Jupiter vector
XERM (6)		
XMAR (6)		Mars vector
XVEN (6)		Venus vector
XSUN (6)		Sun vector
XMOO (6)		Moon vector
X.	1	
Y	)	
Z	(	
XDOT		Probe vector
YDOT	)	
ZĐOT	1	
U (36)		Variational equations
DLO		Nutation in longitude
DOB	185	Nutation in obliquity

XAC )	85-2 of 2
YAC	Probe acceleration
zac )	
XJERK )	
yjerk }	Probe jerk (3rd derivative)
zjerk )	
GMAT (66)	Gamma matrix

CALL WREOF To write end of file and rewind tape A4 when ephemeris is complete.

# CODING INFORMATION

Length of subroutine is 1350(10) or 2506(8) words.

ROT

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

ROT rotates the planetary ephemerides from the mean equator and equinox of 1950.0 to the true equator of equinox of date.

### RESTRICTION

COMMON break: 47675

### METHOD

$$\begin{pmatrix} x \\ y \\ z \end{pmatrix}$$
 of date = NA  $\begin{pmatrix} x \\ y \\ z \end{pmatrix}$  1950.0

where

NA = nutation-precession matrix obtained by subroutine PNUT

USE

CALL ROT

COMMON input:

ROTMX NA

XSUN geocentric position and velocity of Sun, 1950.0

XVEN geocentric position and velocity of Venus

XMAR geocentric position and velocity of Mars

XJUP geocentric position and velocity of Jupiter

XMOO geocentric position and velocity of Moon

COMMON output:

XSUN, XVEN, XMAR, XJUP, XMOO rotated to true-of-date

# CODING INFORMATION

Length of subroutine is 182 (10) or 266 (8) words.

RSIG

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

### PURPOSE

The partial of AU is expressed as a sum of terms evaluated for SUN, VENUS, MARS, and JUPITER. RSIG evaluates and sums these terms as they are needed in BMATRX.

$$\sum_{k=1}^{4} GM_{k} \left[ x_{k} \left( \frac{1}{\left| \underline{r}_{k} - \underline{r} \right|^{3}} - \frac{1}{r_{k}^{3}} \right) - \frac{3 \left( \underline{r}^{2} - \underline{r} \cdot \underline{r}_{k} \right) (\underline{x} - \underline{x}_{k})}{\left| \underline{r}_{k} - \underline{r} \right|^{5}} \right]$$

where

k = 1 = Sun

k = 2 = Venus

k = 3 = Mars

k = 4 = Jupiter

### RESTRICTION

COMMON break: 47675

USE

CALL RSIG(GRA, XPL, SIGX) where GRA is gravity of the body, XPL position of body, and SIGX is the sum. The routine is called for the four bodies and SIGX is accumulated.

### CODING INFORMATION

Length of subroutine is 51 (10) or 63 (8) words.

SAVCOM

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

#### PURPOSE

Control for the program to save COMMON. This routine also prints the input matrices when the ODP is initiated and checks the times supplied so the trajectory can be run only for the required period.

#### RESTRICTIONS

- a. ERROR condition; disk error indicated by DCP
- b. COMMON break: 47675
- c. Subroutines used:

PRIM

TYPRYT

QUEST

MAXIM

ERROR

FLAK

DISCBU

OFFSYS

#### USE

To save COMMON during the trajectory link, this routine is under control of the JPTRAJ monitor.

### CODING INFORMATION

Length of subroutine is 1118 (10) or 2136 (8) words.

SCALE

Melba W. Nead, JPL Fortran II, Version 3 January 4, 1965

### PURPOSE

When the target is Mars or Venus, the target vector is divided by AU as required in calculation of the target centered covariance matrix.

### USE

CALL SCALE (XPOS, P) XPOS is the target vector, the quotient is stored in P.

### CODING INFORMATION

Length of subroutine is 34 (10) or 42 (8) words.

SORT

Melba W. Nead, JPL IBM 7094 Fap January 4, 1965

#### PURPOSE

Time sorted predictions are prepared in Link 4. This is the main program in which these predicts are station sorted and prepared for transmission to the tracking stations.

# RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

THARP

FLAK

TYPRYT

**IOCS** 

# USE

Link 5 is called under control of the JPTRAJ Source Deck.

# CODING INFORMATION

Length of subroutine is 1069 (10) or 2055 (8) words.

IDENTIFICATION 91-1 of 2

SPHX

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

### PURPOSE

SPHX converts Earth-fixed spherical coordinates to geocentric equatorial cartesian coordinates. No provision is made for nutations; thus, the output is in the mean equator and equinox of date.

### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

SIN

COS

**GHADP** 

#### **METHOD**

$$x_{0} = r_{0} \cos \phi_{0} \cos(\alpha_{G} + \lambda_{0})$$

$$y_{0} = r_{0} \cos \phi_{0} \sin(\alpha_{G} + \lambda_{0})$$

$$z_{0} = W$$

$$\dot{x}_{0} = (\dot{U} - \omega V)\cos \alpha_{G} - (\dot{V} + \omega U)\sin \alpha_{G}$$

$$\dot{y}_{0} = (\dot{U} - \omega V)\sin \alpha_{G} + (\dot{V} + \omega U)\cos \alpha_{G}$$

$$\dot{z}_{0} = \dot{W}$$

$$U = r_{0} \cos \phi_{0} \cos \lambda_{0}$$

$$V = r_{0} \cos \phi_{0} \sin \lambda_{0}$$

 $W = r_0 \sin \phi_0$ 

$$\dot{\mathbf{U}} = \mathbf{v}_0(\sin \gamma_0 \cos \phi_0 \cos \lambda_0 - \cos \gamma_0 \sin \gamma_0 \sin \sigma_0 \qquad 91-2 \text{ of } 2$$

$$-\cos \gamma_0 \sin \phi_0 \cos \lambda_0 \cos \sigma_0)$$

$$\dot{\mathbf{V}} = \mathbf{v}_0(\sin \gamma_0 \cos \phi_0 \sin \lambda_0 + \cos \gamma_0 \cos \gamma_0 \sin \sigma_0$$

$$-\cos \gamma_0 \sin \phi_0 \sin \lambda_0 \cos \sigma_0)$$

$$\dot{\mathbf{W}} = \mathbf{v}_0(\sin \gamma_0 \sin \phi_0 + \cos \gamma_0 \cos \phi_0 \cos \sigma_0)$$

where

$$\mathbf{x}_0$$
,  $\mathbf{y}_0$ ,  $\mathbf{z}_0$ ,  $\dot{\mathbf{x}}_0$ ,  $\dot{\mathbf{y}}_0$ ,  $\dot{\mathbf{z}}_0$  = geocentric equatorial position and velocity  $\mathbf{r}_0$ ,  $\phi_0$ ,  $\lambda_0$ ,  $\mathbf{v}_0$ ,  $\gamma_0$ ,  $\sigma_0$  = Earth-fixed spherical coordinates 
$$\mathbf{c}_G = \text{Greenwich hour angle}$$
  $\omega$  = Earth rotation rate

USE

# CALL SPHX

# COMMON input:

RNJ

ro, km PHINJ  $\phi_0$ , deg  $\lambda_0$ , deg THTNJ v<sub>0</sub>, km/sec VNJ ELNJ  $\gamma_0$ , deg  $\sigma_0$ , deg AZNJ

EPOCH epoch of state vector, d.p. sec past 1950.0

OMEGA ω, deg/sec

# COMMON output:

XNJ  $x_0, y_0, z_0, \dot{x}_0, \dot{y}_0, \dot{z}_0 (1 \times 6)$ 

# CODING INFORMATION

Length of subroutine is 211 (10) or 323 (8) words.

STARP

Michael R. Warner, JPL Fortran II, Version 3 January 4, 1965

# PURPOSE

STARP determines whether an observed time point should be deleted or retained on the basis of input data start and stop times.

### RESTRICTION

COMMON break: 47675

USE

CALL STARP

PZE I I = 0 for point outside allowed range = 1 for point within allowed range

COMMON input:

TOB observation time, d.p. sec past 1950.0

PTFD-30 start-stop times

### CODING INFORMATION

Length of subroutine is 24 (10) or 30 (8) words.

IDENTIFICATION 93-1 of 3

STPREG

Charles L. Lawson/Terry Kinney, JPL Fortran II, Version 3 January 4, 1965

#### PURPOSE

Double precision solution of the normal equations of a linear least squares regression problem using the stepwise procedure of M. A. Efroymson.

This subroutine is particularly intended for use in least-squares problems in which the user is aware of a large number of basic functions which might bear a linear relationship to the object function but expects that only some of these basic functions make a significant contribution to the determination of the object function.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:
   (DFAD)/(DFMP)/(DFDP)
   DSORT

#### METHOD

In an n-parameter problem this subroutine provides successively a 1-parameter fit, a 2-parameter fit, ..., a k-parameter fit, where the new parameter introduced at each step is the one which will make the greatest reduction in the sum of squares of residuals. This criterion is equivalent to the requirement that the new parameter be selected as the one which will have the highest statistical significance in the sense that the ratio of its magnitude to its standard deviation will be largest or equivalently its "F" value will be largest (see Ref. c). The termination level k, which will be  $\leq$  n, is determined by the subroutine on the basis of tolerance parameters SDIN, SDOUT, and CD provided by the user.

For an exposition of this procedure, see Ref. a of this subroutine. For a complete single precision regression package which incorporates this procedure in its system solving subroutine, see Ref. b. For a discussion of this procedure including details of its implementation in this program see Ref. c.

#### USE

Let the overdetermined system of linear equations which expresses the linear regression problem be denoted by

93-2 of 3

where G is an m by n matrix (m > n), c is an n-vector, f is an m-vector. The elements of G and f are known and the coefficient vector c is to be computed as a weighted least squares solution.

The normal equations Ac = b are formed from the overdetermined system by pre-multiplication by G\*W, (\*denotes transposition) where W is an m by m non-negative definite weighting matrix (W may be the identity matrix).

This computation, as well as the computation of the scalar quantity s = f\*Wf, is to be done by the user's program. STPREG may then be called as follows:

D DIMENSION A (20, 20), B (20), S (1)
DIMENSION INOUT (20)
CALL STPREG (A, B, S, N, M, SDIN, SDOUT,
CD, NIN, INOUT)

The parameters are defined as follows:

$$A(I, J), I = 1, N; J = 1, N$$

On entry: The matrix A, computed by the user as A = G\*WG.

A is normalized by STPREG before inversion.

On return:  $A^{-1}$  if NIN = N. If NIN < N then only an NIN by NIN submatrix of A will have been inverted and the elements of this inverse will be in the locations A (I, J) corresponding to those values of I and J for which both INOUT (I) = +1 and INOUT (J) = +1. The other A (I, J) locations for  $I \le N$ ,  $J \le N$  will contain zero.  $A^{-1}$  is unnormalized before the return.

$$B(I), I = 1, N$$

On entry: The vector b, computed by the user as p = G\*Wf.

On return: The solution vector c if NIN = N. If NIN < N then only NIN components of c will have been computed and they will be those in the locations B (I) for which INOUT (I) =  $\pm$  1. The other B (I) locations for I  $\leq$  N will contain zero.

S

On entry: The scalar s, computed by the user as s = f\*Wf.

On return: The weighted sum of squares of residuals:

$$(f - Gc)*W(f - Gc) = (f - Gc)*Wf$$
  
=  $f*Wf - (Gc)*Wf = f*Wf - c*b$ 

- N The dimension, n, of the normal system.
- M The dimension, m, of the f vector.

#### SDIN, SDOUT

If DSIN > 0., these two numbers will be used by the "significance" logic of the program. A coefficient will be permitted to enter the solution if the ratio of its

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magnitude to its standard deviation will exceed SDIN. If this ratio subsequently drops below SDOUT this coefficient will be removed from the solution. To avoid looping, SDIN should exceed SDOUT.

Typical values for SDIN and SDOUT might be SDIN = 1.96, SDOUT = 1.64, which could be interpreted as meaning that a coefficient will be permitted to enter the solution if there is a 95% probability that it is different from zero and will later be removed from the solution if this probability falls below 90%.

If  $SDIN \le 0$  all significance tests will be skipped. This permits use of the program with meaningless data in B and S, as may be the case when only matrix inversion is desired.

Tolerance for relative reduction in magnitude of diagonal elements. The i<sup>th</sup> coefficient will not be entered into the solution if the i<sup>th</sup> diagonal element becomes smaller than CD times the original i<sup>th</sup> diagonal element of the matrix A. For example if CD = 10. \*\*(-12) then a potential pivot element will not be used if it has lost more than its first 12 significant decimal places.

NIN On return: The number of coefficients in the final solution. INOUT (I), I = 1, N On return: Flags to indicate final status of each coefficient.

- +l Included in solution.
- -1 Rejected by diagonal test.
- -2 Rejected by significance tests.

### CODING INFORMATION

Length of subroutine is 1069 (10) or 2055 (8) words.

### REFERENCES

- a. Efroymson, M. A., Multiple Regression Analysis, Mathematical Methods for Digital Computers, ed. Ralston and Wilf, Wiley, 1960, pp. 191-203.
- b. Efroymson, M.A., Stepwise Multiple Regression with Variable Transformations, SHARE Distribution No. 1194, October, 1961.
- c. Lawson, C.L., Computation of the Most Significant Coefficients in Least-Squares Estimation, JPL Section Report No. 372-5, August, 1962.

IDENTIFICATION 94-1 of 2

TACCOM

Melba W. Nead, JPL

Fortran II, Version 3

January 4, 1965

#### PURPOSE

To evaluate the target-centered cartesian Covariance Matrix.

### RESTRICTIONS

a. COMMON break: 47055

b. Subroutines used:

AAT

SCALE

(DFAD)

(DFMP)

(DFSB)

### METHOD

In the following, A will represent the ephemeris scaling factor. Thus, A = REM when target = MOON, and A = AU when target = MARS or VENUS.

a. When A is among the estimated parameters, compute the target centered covariance matrix as

$$\Gamma_{E} = \Gamma_{X} + PP^{T}\sigma_{A}^{2} - \Gamma_{XA}P^{T} - P\Gamma_{XA}^{T}$$

where

 $\Gamma_{\rm X}$  = upper left hand 6 × 6 sub-matrix of the geocentric covariance matrix  $\Gamma$ 

P = 6-vector giving the geocentric position and velocity of the target in units of A and A/sec, referred to the Earth equatorial system

 $\sigma_{\Lambda}^2$  = diagonal element of  $\Gamma$  corresponding to A

 $\Gamma_{\rm XA}$  = first 6 elements of the column of  $\Gamma$  corresponding to A

b. When A is not among the estimated parameters

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$$\Gamma_{\rm E} = \Gamma_{\rm X}$$

USE

CALL TACCOM (GAME, ITAR)

GAME contains  $\Gamma_{\hbox{\scriptsize E}}$  as calculated in TACCOM

ITAR contains the target reference

ITAR = 1 MARS

ITAR = 2 MOON

ITAR = 3 VENUS

# CODING INFORMATION

Length of subroutine is 938 (10) or 1652 (8) words.

#### REFERENCE

Carey, C., RFP 312-252, March 5, 1964.

TAMP/EPOX

Melba W. Nead, JPL

IBM 7094 Fap

January 4, 1965

# PURPOSE

Writes mapping times on scratch tape A7 to be saved for matrix manipulation program input tape.

### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

TAPIO

FLAT

USE

CALL TAMP

EPOX

provides space for storage of injection conditions which are saved through the WANT cards feature of JPTRAJ

# CODING INFORMATION

Length of subroutine is 87 (10) or 127 (8) words.

THARP/PKINE Michael R. Warner, JPL IBM 7094 Fap January 4, 1965

#### PURPOSE

THARP serves as the logical control routine for the predictions and residuals output links, LA5 and LA8. It sets up lists of desired stations and calls the output subroutines AQUI, PERNOD, RESP, and ANGPLT. PKINE is called by these output routines in order to obtain a meaningful logical record from the disk read routine KINE.

#### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

**OFFSYS** 

KINE

AQUI/ALLI/BAQUI/BALLI

PERNOD/WOLF

RESP

ANGPLT

BIBCD

TYPRYT

### USE

CALL THARP

CALL PKINE

PZE SUBR = 1 if called from ANGPLT

= 2 if called from AQUI

= 3 if called from PERNOD

= 4 if called from RESP

### COMMON input:

IF3 ODP option flags

IRES residual output flags (8 × 15)

RESR plotting scale factors (8 x 15)

ITNO iteration number

PT pointing times (used as flags)

#### CODING INFORMATION

Length of subroutine is 379 (10) or 573 (8) words.

IDENTIFICATION 97-1 of 2

TIMER

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

TIMER obtains the primary light time correction which adjusts the observation time for ephemeris lookup purposes.

# RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

(DFSB)

LOOKUP

SQRT

METHOD

$$\Delta t = 0$$

$$t = t_{ob} - \Delta t$$

$$r = \sqrt{x^2 + y^2 + z^2}$$

$$\Delta t = \frac{\mathbf{r} - \mathbf{R}_{e}}{C}$$

The procedure is then repeated with the new  $\Delta t$ .

USE

CALL TIMER

COMMON input:

TOB t<sub>ob</sub>, d. p. sec past 1950.0

RE R<sub>e</sub>, km

VELC c, km/sec

COMMON output:

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T t, d.p. sec past 1950.0

DELT Δt, d.p. sec

# CODING INFORMATION

Length of subroutine is 98 (10) or 142 (8) words.

TNORM

Melba W. Nead/Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

# PURPOSE

TNORM normalizes an N x N matrix on its diagonal terms. N may be 2, 3, 4, 5, or 6 but the matrix must be stored in a  $6 \times 6$  array.

# RESTRICTION

Subroutine called:

SQRT

# USE

CALL	TNORM	
PZE	YMAT	input matrix

PZE XMAT output normalized matrix

PZE N N

# CODING INFORMATION

Length of subroutine is 170 (10) or 252 (8) words.

IDENTIFICATION 99-1 of 3

TOCIM

Michael R. Warner, JPL

IBM 7094, Fap

January 4, 1965

#### PURPOSE

TOCIM computes occultation time and impact time, and obtains the residual. It writes a record on the combined ephemeris tape for each observable, and outputs on disk for subsequent printing.

### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

**OFFSYS** 

TIMER

LOOKUP/KOOL

INTRI

SQRT

COS/SIN

GAMAT

GHADP

ARCOS/ARSIN

FLAT

RITE

WRITEB

# METHOD

TOCIM uses a Newton-Raphson formula for both data types:

For occultation time,

$$\left[T_{0}^{(r)} - T_{0}^{(r-1)}\right]^{\frac{d(\phi_{p} - \phi_{m})}{dt}} \Big|_{T_{0}^{(r-1)}} = \phi_{m} - \phi_{p}$$

where

$$\phi_{\rm m} = \sin^{-1} \frac{R_{\rm m}}{\rho_{\rm m}}$$

$$\phi_{p} = \cos^{-1} \frac{\underline{\rho} \cdot \underline{\rho}_{m}}{|\underline{\rho}| |\underline{\rho}_{m}|}$$

$$\frac{d (\phi_{p} - \phi_{m})}{dt} = \frac{\underline{\rho} \cdot \underline{\rho}_{m}}{a_{1}} \left[ \left( \frac{a_{2}}{\rho^{2}} + \frac{a_{3}}{\rho_{m}^{2}} \right) - a_{4} \right] + \frac{a_{3}R_{m}}{\rho_{m}^{2} \sqrt{\rho_{m}^{2} - R_{m}^{2}}}$$

$$a_{1} = |\underline{\rho}| |\underline{\rho}_{m}| \sin \phi_{p}$$

$$a_{2} = \underline{\rho} \cdot \underline{\dot{\rho}}_{m}$$

$$a_{3} = \underline{\rho}_{m} \cdot \underline{\dot{\rho}}_{m}$$

$$a_{4} = \underline{\rho} \cdot \underline{\dot{\rho}}_{m} + \underline{\rho}_{m} \cdot \underline{\dot{\rho}}$$

For impact time,

$$\begin{bmatrix} T_{I}^{(r)} - T_{I}^{(r-1)} \end{bmatrix} \begin{vmatrix} \dot{\underline{r}} - \dot{\underline{r}}_{m} \end{vmatrix} = \begin{vmatrix} \underline{r} - \underline{r}_{m} \end{vmatrix} - R_{m}$$

(all quantities are defined in the subroutine OCPAR documentation)

USE

#### CALL TOCIM

### COMMON input:

DXDK-6 target identification

IFA-4 R<sub>m</sub>, km

PTFD occultation-impact data buffer

DXDR-27 occultation-impact buffer

LLIST physical constants estimate flags

RI station Earth radius, km

PHII station geocentric latitude, deg

THETAI station longitude, deg

EPOCH ODP epoch, d.p. sec past 1950.0

OMEGA Earth rotation rate, deg/sec

# CODING INFORMATION

99-3 of 3

Length of subroutine is 998 (10) or 1746 (8) words.

# REFERENCE

Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.

100-1 of 9

UMAT

Kenneth Oslund/Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

### PURPOSE

UMAT computes the encounter conic, rotating from the geocentric equatorial system to the target-centered  $\underline{R}$ ,  $\underline{T}$  system. All encounter parameters and statistics are obtained by this routine.

# RESTRICTIONS

Subroutines called:

SQRT

LOG

GERPU

ARTAN

PARAM

#### **METHOD**

The following inputs are in the UMAT calling sequence:

r<sub>tp</sub>, i target centered probe position and velocity at time of encounter

<u>r</u>ct, <u>r</u>ct position and velocity of target, centered at that body's primary, at time of encounter

 $\Gamma_{\rm e}$  covariance matrix at encounter (7 × 7)

 $U_{e}$  mapping matrix to encounter  $(7 \times 7)$ 

 $\mu$   $GM_{target}$ 

t<sub>f</sub> time of flight

First, the rotation matrix is obtained:

$$r'^2 = \underline{r}_{ct} \cdot \underline{r}_{ct}$$

$$\dot{s}^{12} = \dot{\underline{r}}_{ct} \cdot \dot{\underline{r}}_{ct}$$

100-2 of 9

$$\mathbf{r'\dot{r'}} = \underline{\mathbf{r}_{ct}} \cdot \underline{\dot{\mathbf{r}}_{ct}}$$

$$\underline{\mathbf{u}_{p}} = \frac{\mathbf{r'\dot{\mathbf{r}}_{ct}}}{\mathbf{r'}}$$

$$\underline{\mathbf{v}_{p}} = \frac{\mathbf{r'\dot{\mathbf{r}}_{ct}} - \dot{\mathbf{r'}_{ct}}}{\mathbf{r'\sqrt{\dot{s'}^{2} - \dot{\mathbf{r'}^{2}}}}}$$

$$\mathbf{a}_{33} = \sqrt{\left(\mathbf{u}_{pr} + \mathbf{v}_{py}\right)^{2} + \left(\mathbf{u}_{py} - \mathbf{v}_{px}\right)^{2} - 1}$$

$$\mathbf{a}_{23} = \sqrt{\mathbf{u}_{pz}^{2} + \mathbf{v}_{pz}^{2}}$$

$$\mathbf{a}_{11} = \frac{1 + \mathbf{a}_{33} \left[\mathbf{v}_{pz}(\mathbf{u}_{px} + \mathbf{v}_{py}) + \mathbf{u}_{pz}(\mathbf{u}_{py} - \mathbf{v}_{px})\right]}{\mathbf{a}_{23}}$$

$$\mathbf{a}_{12} = \frac{1 + \mathbf{a}_{33} \left[\mathbf{v}_{pz}(\mathbf{u}_{py} - \mathbf{v}_{px}) - \mathbf{u}_{pz}(\mathbf{u}_{px} + \mathbf{v}_{py})\right]}{\mathbf{a}_{23}}$$

$$\mathbf{a}_{21} = -\mathbf{a}_{12}\mathbf{a}_{33}$$

$$\mathbf{a}_{22} = \mathbf{a}_{11}\mathbf{a}_{33}$$

$$\mathbf{a}_{31} = \mathbf{a}_{12}\mathbf{a}_{23}$$

$$\mathbf{a}_{32} = -\mathbf{a}_{11}\mathbf{a}_{23}$$

The target-centered probe vector is then rotated:

$$\underline{r}_{tp} = A\underline{r}_{tp}'$$

$$\underline{\dot{r}}_{tp} = A\underline{\dot{r}}_{tp}'$$

and the following column vectors are formed:

100-3 of 9

$$\underline{X}_{p} = \begin{pmatrix} \underline{x}_{tp} \\ 0 \end{pmatrix}$$
  $\underline{\dot{X}}_{p} = \begin{pmatrix} \dot{x}_{tp} \\ 0 \end{pmatrix}$ 

$$\underline{\mathbf{x}}_{\mathbf{v}} = \begin{pmatrix} \mathbf{0} \\ \underline{\mathbf{r}}_{\mathbf{tp}} \end{pmatrix} \qquad \underline{\dot{\mathbf{x}}}_{\mathbf{v}} = \begin{pmatrix} \mathbf{0} \\ \dot{\underline{\mathbf{r}}}_{\mathbf{tp}} \end{pmatrix}$$

(the seventh member of these vectors is in each case zero)

The encounter conic is then obtained:

$$\mathbf{r}^2 = \underline{\mathbf{x}}_{\mathbf{p}} \cdot \underline{\mathbf{x}}_{\mathbf{p}}$$

$$\underline{\dot{s}}^2 = \underline{\dot{x}}_p \cdot \underline{\dot{x}}_p$$

$$r\dot{r} = \underline{X}_p \cdot \dot{\underline{X}}_p$$

$$a = \frac{1}{\frac{2}{r} - \frac{\dot{s}^2}{\mu}}$$
 (semi-major axis)

$$e^2 = 1 - \frac{r^2 \dot{s}^2 - (r \dot{r})^2}{\mu/a}$$
 (eccentricity)<sup>2</sup>

e sinh 
$$F = \frac{r\dot{r}}{\sqrt{-\mu/a}}$$

$$e \cosh F = 1 - \frac{r}{a}$$

$$e \exp F = e \sinh F + e \cos F$$

$$ep_z = X_{pz} \frac{e \cosh F}{r} - \dot{X}_{pz} e \sinh F \sqrt{-\frac{a}{\mu}}$$

$$e \sin \phi_s = \left(\frac{X_{pz}}{r} - X_{pz}\sqrt{-\frac{a}{\mu}}\right) \exp F + \dot{X}_p e \sqrt{-\frac{a}{\mu}}$$

$$\underline{B} \cdot \underline{T} \cos \Phi_{S} = \sqrt{-\frac{a}{\mu}} (X_{pz} \dot{X}_{py} - X_{py} \dot{X}_{px})$$

$$\underline{B} \cdot \underline{R} \cos \Phi_{S} = a(\sin \Phi_{S} - e p_{z})$$

$$T_{a} = \frac{3}{2} \sqrt{-\frac{a}{\mu}} (e \sinh F - F + \log e)$$

$$T_{e} = \left(\frac{e \sinh F + 1}{e}\right) a \sqrt{-\frac{a}{\mu}}$$

$$T_{F} = -r \sqrt{-\frac{a}{\mu}}$$

$$v_{\omega} = \frac{1}{\sqrt{-\frac{a}{\mu}}}$$

$$n = \frac{-v_{\omega}}{a}$$

 $t_L = t_f + \frac{(F - e \sinh F - \log e)}{n}$  (linearized time of flight)

The next step is to obtain the encounter partials. The encounter parameter set (M) is

 $\underline{B} \cdot \underline{R}$  as defined above

 $\underline{\mathbf{B}} \cdot \underline{\mathbf{T}}$  as defined above

t, linearized time of flight

 $\begin{array}{c} \underline{S} \cdot \underline{R}_{S} \\ \underline{S} \cdot \underline{T}_{S} \end{array} \begin{array}{c} \text{dot products of incoming asymptote unit vector } \underline{S} \\ \text{with standard } \underline{R} \text{ and } \underline{T} \text{ vectors} \end{array}$ 

C<sub>3</sub> vis-viva energy

These parameters are related to the position and velocity of the probe at encounter and the target gravitational constant. This set is called Q.

$$F_{az} = \frac{\exp F}{r}$$

$$F_{a\dot{z}} = -\sqrt{-\frac{a}{\mu}} (\exp F - e)$$

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$$F_{bz} = \frac{e \cosh F}{r}$$

$$F_{b\dot{z}} = -\sqrt{-\frac{a}{\mu}} e \sinh F$$

$$F_{cx} = -\dot{X}_{py}$$

$$F_{cy} = \dot{X}_{px}$$

$$F_{c\dot{x}} = X_{py}$$

$$F_{c\dot{y}} = -X_{px}$$

$$S_{x} = \left[ \left( \frac{X_{px}}{r} - \sqrt{-\frac{a}{\mu}} \dot{X}_{px} \right) \exp F + e \sqrt{-\frac{a}{\mu}} \dot{X}_{px} \right] \frac{1}{e} x \rightarrow y$$

$$S_z = \sin \phi_s$$

$$T_{sx} = \frac{S_y}{\cos \phi_s}$$

$$T_{sy} = -\frac{S_x}{\cos \phi_s}$$

$$R_{sx} = T_{sy}S_{z}$$

$$R_{sy} = -T_{sx}S_{z}$$

$$R_{sz} = \cos \Phi_{s}$$

$$\underline{\delta a} = 2a^2 \left( \frac{\underline{X}_p}{r^3} + \frac{\underline{\dot{X}}_n}{\mu} \right)$$

$$\delta a_{\mu} = -\frac{a^{2}(1 + e \cosh F)}{r\mu}$$

$$[\underline{\delta}\underline{e}; \delta' e_{\mu}] = \frac{1}{e} \left[ -\frac{\frac{e}{a} \cosh F X_{p}}{r} - \frac{e \sinh F(\underline{X}_{p} + \underline{X}_{v})}{\sqrt{-\frac{\mu}{a}}} + \frac{\delta \underline{a}}{e^{2}} \left( \frac{\frac{1}{2}e^{2} \sinh^{2}F}{a} + \frac{r e \cosh F}{a^{2}} \right) \right]$$

$$\delta e_{\mu} = \delta' e_{\mu} + \frac{1}{e} \left[ -\frac{1}{\mu} \left( \frac{1}{2}e^{2} \sinh^{2}F + \frac{r}{a} e \cosh F \right) + \frac{r^{2}}{2a^{2}} + \frac{r^{2}\delta a_{\mu}}{2a^{3}} \right]$$

$$[\underline{\delta}\underline{F}; \delta' F_{\mu}] = \frac{1}{e^{2}} \left[ \frac{\frac{e}{a} \sinh F X_{p}}{r} + \frac{e \cosh F(\underline{X}_{p} + \underline{X}_{v})}{\sqrt{-\frac{\mu}{a}}} - \frac{\delta \underline{a}}{a} \left( \frac{\frac{1}{2}e^{2} \cosh^{2}F}{a} + \frac{r e \sinh F}{a^{2}} \right) \right]$$

$$\delta F_{\mu} = \delta' F_{\mu} + \frac{1}{e^{2}} \left( -\frac{e \sinh F}{2\mu e \cosh F} - \frac{r^{2} e \sinh F \delta a_{\mu}}{2e \cosh F a^{3}} \right)$$

$$[\delta \Phi_{g}; \delta' \Phi_{g\mu}] = \left[ -\frac{X_{px}X_{p} \exp F}{r^{3}} - \frac{\frac{1}{2}X_{px}(e - \exp F)_{\underline{a}\underline{a}}}{\sqrt{-\frac{\mu}{a}}} \left( -\frac{X_{px}a}{\sqrt{-\frac{\mu}{a}}} + \sin \Phi_{g} \right) \delta_{\underline{e}} - \left( X_{px}X_{p} - X_{px}X_{p} \right) \delta_{\underline{e}} + \frac{1}{e \cos \Phi_{g}}$$

$$[\delta S_{\underline{X}}; \delta' S_{x\mu}] = \frac{1}{e} \cdot \left[ -\frac{X_{px} \exp F X_{p}}{r^{3}} + \frac{X_{px}(\exp F - e)}{2\sqrt{-\frac{\mu}{a}}} \right] \frac{1}{e \cos \Phi_{g}}$$

$$+ \left( \sqrt{-\frac{a}{\mu}} X_{px} - S_{x} \right) \delta_{\underline{e}} - \left( \sqrt{-\frac{a}{\mu}} X_{px} - X_{px} \right) \exp F \delta_{\underline{e}} + F_{\underline{e}\underline{a}}$$

$$+ \left( \sqrt{-\frac{a}{\mu}} X_{px} - S_{x} \right) \delta_{\underline{e}} - \left( \sqrt{-\frac{a}{\mu}} X_{px} - X_{px} \right) \exp F \delta_{\underline{e}} + F_{\underline{e}\underline{a}}$$

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$$\begin{split} \left[\delta S_{\underline{y}} : \delta' S_{\underline{y}\mu}\right] &= \left[\delta S_{\underline{x}} : \delta' S_{\underline{x}\mu}\right] \times \neg y \\ \left[\delta S_{\underline{z}} : \delta' S_{\underline{z}\mu}\right] &= \cos \varphi_{\underline{s}} \delta \underline{\varphi}_{\underline{s}} \\ &= \underbrace{E_{\underline{p}}} = -\frac{e \sinh F}{2\mu} \underbrace{\sqrt{-\frac{a}{\mu}} \dot{X}_{\underline{p}}}_{2\mu} \\ &= \underbrace{\delta' S_{\underline{\mu}}}_{\underline{p}} + \left[\underline{E}_{\underline{p}} - (e \cosh F - e^2) \underbrace{\frac{\dot{X}_{\underline{p}}}{2\mu}}_{\underline{p}\mu} \underbrace{-\frac{1}{e \cos \varphi_{\underline{s}}}}_{\underline{p}\mu} \right] \frac{1}{e \cos \varphi_{\underline{s}}} \\ \delta \varphi_{\underline{s}\mu} &= \delta' \varphi_{\underline{s}\mu} + \delta' S_{\underline{z}\mu} \\ &= \underbrace{\delta' \Phi_{\underline{s}\mu}}_{\underline{r}} + \delta' S_{\underline{z}\mu} \\ &= \underbrace{\left(-\dot{X}_{\underline{p}\underline{z}} e \cosh F \underbrace{X}_{\underline{p}} + \frac{X_{\underline{p}\underline{z}} e \sinh F \underline{\delta}\underline{a}}{2\sqrt{-\frac{\mu}{a}}} \right)}_{2\sqrt{-\frac{\mu}{a}}} \\ &= \underbrace{\left(-\dot{X}_{\underline{p}\underline{z}} e \cosh F \sqrt{-\frac{a}{\mu}} - \frac{e \sinh F}{r}\right)}_{\underline{\delta}\underline{F}} + \underline{F}_{\underline{b}} \\ &= \delta P_{\underline{z}\mu} = (e \, \delta P_{\underline{z}\mu}) + E_{\underline{p}\underline{z}} \\ &= \underbrace{\left(-\dot{X}_{\underline{p}\underline{z}} e \cosh F \sqrt{-\frac{a}{\mu}} - \frac{e \sinh F}{r}\right)}_{\underline{\delta}\underline{F}} + \underline{F}_{\underline{b}} \\ &= \underbrace{\left(-\dot{X}_{\underline{p}\underline{z}} e \cosh F \sqrt{-\frac{a}{\mu}} - \frac{e \sinh F}{r}\right)}_{\underline{\delta}\underline{A}} - a \, e \, P_{\underline{z}} \underline{\delta}\underline{e}} \\ &= \underbrace{\left(-\dot{Y}_{\underline{i}} : v_{\underline{i}\mu}'\right]}_{\underline{i}} = \underbrace{\left(-\dot{E} \sin \varphi_{\underline{p}} - e^2 P_{\underline{z}}\right)}_{\underline{\delta}\underline{a}} - a \, e \, P_{\underline{z}} \underline{\delta}\underline{e}} \\ &- \underbrace{\left(\underline{B} \cdot \underline{R} \, e \sin \varphi_{\underline{p}} a \, e \cos \varphi_{\underline{p}}\right)}_{\underline{\delta}\underline{A}} - a \, e \, P_{\underline{z}} \underline{\delta}\underline{e}} \\ &= \underbrace{\left(-\dot{\underline{B}} \cdot \underline{R} \, e \sin \varphi_{\underline{p}} a \, e \cos \varphi_{\underline{p}}\right)}_{\underline{\delta}\underline{A}} - a \, e^2 \, \underbrace{\delta P_{\underline{z}}'}_{\underline{z}} = \underbrace{\left(-\dot{\underline{A}} \, \underline{F}_{\underline{c}}\right)}_{\underline{e} \, \cos \varphi_{\underline{p}}} \\ \underbrace{\left(-\dot{\underline{A}} \, \underline{B} \, e \sin \varphi_{\underline{p}} a \, e \cos \varphi_{\underline{p}}\right)}_{\underline{\delta}\underline{A}} + \underline{\underline{B}} \cdot \underline{\underline{T}} \, e \sin \varphi_{\underline{p}} \underline{\delta}\underline{\Phi}_{\underline{s}} - e \, \sqrt{-\frac{a}{\mu}} \, \underline{\underline{F}}_{\underline{c}}\right)}_{\underline{e} \, \cos \varphi_{\underline{p}}} \\ \underbrace{\left(-\dot{\underline{A}} \, \underline{B} \, \underline{A} \, e \, e \sin \varphi_{\underline{p}} a \, e \cos \varphi_{\underline{p}}\right)}_{\underline{\delta}\underline{A}} + \underline{\underline{B}} \cdot \underline{\underline{T}}_{\underline{e}} \sin \varphi_{\underline{p}} \underline{\delta}\underline{\Phi}_{\underline{s}} - e \, \sqrt{-\frac{a}{\mu}} \, \underline{\underline{F}_{\underline{c}}}\right)}_{\underline{e} \, \cos \varphi_{\underline{p}}} \\ \underbrace{\left(-\dot{\underline{A}} \, \underline{A} \, \underline{A} \, \underline{A} \, e \, e \, e \, \underline{A} \, \underline{A} \, \underline{A} \, \underline{A} \, \underline{A} + \underline{\underline{A}} \, \underline{A} \, \underline{A} \, \underline{A} + \underline{\underline{A}} \, \underline{A} \, \underline{A} \, \underline{A} \, \underline{A} \, \underline{A} + \underline{\underline{A}} \, \underline{\underline{A}} \, \underline{A} \, \underline{A$$

 $\begin{bmatrix} \mathbf{v}_4 \\ \mathbf{v}_{4\mu} \end{bmatrix} = \frac{\mu}{2} \quad \underline{\delta}\mathbf{a}$ 

 $\left[\mathbf{v}_{5} : \mathbf{v}_{5\mu}^{\prime}\right] = \mathbf{T}_{sx} \frac{\delta \mathbf{S}_{x}}{\delta \mathbf{S}_{x}} + \mathbf{T}_{sy} \frac{\delta \mathbf{S}_{y}}{\delta \mathbf{S}_{y}}$ 

$$\begin{bmatrix} v_{6} & v_{6}^{'} \mu \end{bmatrix} = R_{sx} \frac{\delta S_{x}}{\delta S_{x}} + R_{sy} \frac{\delta S_{y}}{\delta S_{y}} + R_{sz} \frac{\delta S_{z}}{\delta S_{z}}$$

$$v_{1\mu} = v_{1\mu}^{'}$$

$$v_{2\mu} = v_{2\mu}^{'} + \frac{B \cdot T}{2\mu}$$

$$v_{3\mu} = v_{3\mu}^{'} - \frac{\sqrt{\frac{a^{3}}{\mu} (F - e \sinh F - \log e)}}{2\mu}$$

$$v_{4\mu} = v_{4\mu}^{'} - \frac{1}{a}$$

$$v_{5\mu} = v_{5\mu}^{'}$$

$$v_{6\mu} = v_{6\mu}^{'}$$

The following matrix manipulations are then performed:

$$N = V^{T}A\Gamma_{e}A^{T}V$$
 (noise moment matrix)  
 $V_{0} = VAU_{e}$   $\left(V_{0} = \frac{\partial M}{\partial Q_{0}} \text{ matrix}\right)$ 

Finally, the configuration of the miss ellipse is obtained:

$$r = \sqrt{\left(\frac{n_{11} - n_{22}}{2}\right)^2 + n_{12}^2}$$

$$s = \frac{n_{11} + n_{22}}{2}$$

$$a_{m} = \sqrt{s + r} \qquad \text{(semi-major axis)}$$

$$b_{m} = \sqrt{s - r} \qquad \text{(semi-minor axis)}$$

$$\theta = \frac{1}{2} \tan^{-1} \left(\frac{2n_{12}}{n_{22} - n_{11}}\right) \qquad \text{(inclination)}$$

USE

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UMAT CALL  $\underline{\underline{r}}_{tp}^{'}, \ \underline{\underline{\dot{r}}}_{tp}^{'}$  (1 × 6) PZEXT $\underline{\mathbf{r}}_{ct}$ ,  $\dot{\underline{\mathbf{r}}}_{ct}$  (1 × 6) PZE XHPZE  $(7 \times 7)$ ປຼ PZE  $(7 \times 7)$  in  $(20 \times 20)$  array PZE TMU $\mu$  of target  $t_f$ , sec PZETF

# CODING INFORMATION

Length of subroutine is 1681 (10) or 3221 (8) words.

# REFERENCES

- a. Warner, M. R., Nead, M. W., Hudson, R. H., TM 33-168, March 18, 1964.
- b. Anderson, John D., RFP 312-83, October 1962.
- c. Null, George W., RFP 312-179, Addendum 2, September 19, 1963.

**IDENTIFICATION** 

101

VEC

Charles Coltharp, JPL IBM 7094 Fap

Tanana 4 1061

January 4, 1965

# PURPOSE

VEC computes the standard dot product of two 1  $\times$  3 vectors, A =  $\underline{X} \cdot \underline{Y}$ .

### RESTRICTION

COMMON break: 47675

USE

CALL VEC

PZE X

PZE Y

COMMON output:

Α

dot product

#### CODING INFORMATION

Length of subroutine is 15 (10) or 17 (8) words.

IDENTIFICATION

102-1 of Z

WAIT

Michael R. Warner, JPL

Fortran II, Version 3

January 4, 1965

### PURPOSE

WAIT obtains the a priori weight for each data point used in the normal equations.

#### RESTRICTIONS

a. COMMON break: 47675

b. Subroutines used:

COL

DECOD

SQRT

WOCT

COS

ERROR

Error condition:

If the weight obtained is less than  $10^{-20}$ , the program will abort.

### METHOD

Method al.

If a priori coefficients  $\sqrt{\widetilde{\omega}_j}$  are input to the program, the calculated weight is

$$\sqrt{\omega_{\rm j}} = \sqrt{\omega_{\rm j}} \ \beta_1 \ \beta_2 \sqrt{\frac{60}{r_{\rm s}}}$$

where

 $\beta_1 = 1$  when not weighting azimuth or hour angle

=  $\frac{1}{\cos \gamma}$  when weighting azimuth angle

 $= \frac{1}{\cos \delta}$  when weighting hour angle

 $\beta_2$  = 1 when weighting azimuth angle

= 1 +  $\frac{18}{(\gamma + 1)^2}$  when not weighting azimuth angle

 $\tau_{s}$  = data sample rate

102-2 of 2

 $\delta$  = declination

 $\gamma$  = elevation angle

#### Method a2.

If octal weight codes are input to the ODP a priori coefficient table, the program uses Method b.

#### Method b.

If octal weight codes are input by the ODG data file, and if no input was made to the ODP coefficient array, the calculated sigma is

$$\omega_{j} = \sum_{i=1}^{6} S_{pjk}^{2} g_{pj}^{2} \max \left(\frac{T_{pjk}}{\tau}, 1\right)$$

(see documentation for subroutines DECOD and COL).

#### USE

#### CALL WAIT

### COMMON input:

BALNC 
$$\sqrt{\omega_j}$$
 (12 × 15)

ID data type identification

IR receiver identification

SSQ  $S_{pjk}^2$  (12 × 6 × 4)

TL  $T_{pjk}$  (12 × 6 × 4)

GSQ  $g_{pjk}^2$  (6)

TAU  $\tau$  (doppler averaging time, sec)

IF 10 phi vector output flag

DEC  $\delta$ , deg

EL  $\gamma$ , deg

#### COMMON output:

RVECW array of weights

 $\tau_{\rm s}$ , sec

# CODING INFORMATION

Length of subroutine is 230 (10) or 346 (8) words.

# REFERENCE

TS

Hamilton, Thomas W., TM 312-182, April 12, 1962.

IDENTIFICATION 103

WASH

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

### PURPOSE

WASH prints certain time-dependent quantities which are calculated by the ODP (see list below). The subroutine is operative only in Mode-IV or the JPTRAJ production mode.

### RESTRICTIONS

- a. COMMON break: 47675
- b. Subroutines used:

READB

FLAT/FLAPR

FIXT

PROUT

#### USE

# CALL WASH

# COMMON input:

TOB	observation time, d.p. sec past 1950.0
GHA	Greenwich hour angle, deg
X	geocentric probe vector
XSUN	geocentric Sun vector
XMOO	geocentric Moon vector
XVEN	geocentric Venus vector
XMAR	geocentric Mars vector
СРРНІ	partials $\partial F/\partial Q$
CPTHT	partials $\partial \mathbf{F}/\partial\widetilde{\mathbf{Q}}$
บ	partials $\partial \underline{r}/\partial \underline{r}_0$
GMAT	partials <u>dr/</u> dq
DRVEC	refraction corrections

# CODING INFORMATION

Length of subroutine is 401 (10) or 621 (8) words.

IDENTIFICATION 104

WOCT

Michael R. Warner, JPL

IBM 7094 Fap

January 4, 1965

# PURPOSE

WOCT determines whether an input a priori weight represents a floating point quantity or an octal weight code.

# USE

CALL	WOCT	
PZE	WT	a priori weight word
PZE	CODE	output if octal code
PZE	TEST	= 0 if floating point
		# 0 if octal code

### CODING INFORMATION

Length of subroutine is 11 (10) or 13 (8) words.

#### VII. OPERATING INSTRUCTIONS

The ODP may be operated in three modes: SFOF Mode II, SFOF Mode IV, and the JPTRAJ Production Mode. Mode II requires the full SFOF hardware configuration, namely, the 7094-disk-7040, with associated user-area equipment. Mode IV requires the 7094 and disk, properly initialized for SFOF operation. The JPTRAJ Production Mode is independent of the SFOF system, requiring only the 7094 and disk.

# A. MODE II OPERATION

- 1. After the system has been initialized and the user programs loaded on disk, ask the operator to mount the following tapes:
  - A3 scratch for 1401 output
  - A4 scratch for combined ephemerides
  - B6 EPHEM system tape
  - B9 scratch for 4020 plot output (optional)
- Type PS1X\$ at the user area 7 console.
- 3. Depress the TURN ON button, then the TRANSMIT MESSAGE button.
- 4. When requested by the administrative printer, enter the ODPX source deck via the user area card reader.
- 5. At the conclusion of PS1X, type ODPX\$\( \square\$ at the user-area 5 console.
- 6. Depress the TURN ON button, then the TRANSMIT MESSAGE button.
- 7. At the beginning of ODPX execution, the message ENTER SWITCHES.
- 22 = TERMINATE, 23 = DELAY, NONE = PROCEED will be printed by the administrative printer. To proceed, clear the switches and type ODPX\$ at the user area console.
- 8. Depress the OPTION SWITCH ENTRY button, then the TRANSMIT MESSAGE button.
- 9. When requested by the administrative printer, enter the input deck, as described in Section IVA, via the user area card reader. The program will then execute the options requested by the input deck.

- 10. To obtain the full SC-3070 printout, type 1\$\% at the console, depress the AUTO ON button, then the TRANSMIT MESSAGE button.
- 11. The message described in step 7 will be printed between iterations. The user has the opportunity to read a new deck for the next iteration, or terminate ODP execution.

### B. MODE IV OPERATION

- 1. After the system has been initialized and the user programs loaded on disk, mount the following tapes:
  - A3 scratch for 1401 output
  - A4 scratch for combined ephemeris
  - B6 EPHEM system tape
  - B7 scratch for MMP output (optional)
  - Bl scratch for supplementary partials output (optional)
  - B9 scratch for 4020 plot output (optional)
- 2. While the system is at the idle stop, depress sense switch 1 and depress START at the 7094 console.
- 3. At the card reader select, enter the card TURN ON, PS1X\$.
- 4. Raise sense switch 1.
- 5. At the card reader select, enter the ODPX source deck.
- 6. At the conclusion of PSIX (idle stop), depress sense switch 1 and depress START.
- 7. At the card reader select, enter the card TURN ON, ODPX\$.
- 8. Raise sense switch 1.
- 9. At the card reader select, enter the input deck, as described. The program will then execute the options requested by the input deck.
- 10. To obtain on-line printout, depress sense switch 6.
- 11. The card reader will select between iterations for additional input. If console key 22 is depressed at the end of an iteration, the execution will be terminated.

# C. JPTRAJ PRODUCTION MODE OPERATION

- 1. Mount the following tapes:
  - Al JPTRAJ system tape
  - A2 input tape (see below)
  - A3 scratch for 1401 output
  - A4 scratch for combined ephemeris
  - A5 JPTRAJ program tape
  - Bl scratch for system operation
  - B6 EPHEM system tape
  - B7 scratch for MMP output (optional)
  - B9 scratch for 4020 plot output (optional)
- 2. The input tape should contain the following card images:

```
$JOB MRW, 2116000, 542-70285-1-3170, FC (example of JPL job card)
```

\* JPTRAJ

\$RESTORE

\* DATA
ODPX source deck
ODPX input deck
for iteration 1

ODPX input deck

EOF card

\$PAUSE

- 3. To start the system, depress the LOAD TAPE button at the 7094 console.
- 4. Once the program has been restored on disk, the \$RESTORE card and the A5 tape need not be used.

D. SOURCE DECK LISTING

AMALYSIS OF THE JPTRAJ SOURCE PROGRAM FOR THE ODP PROVICES THE BASIC LOSTE OF THE PROGRAM. IN THE FULLDWING THE STARRED ITEMS ARE DIRECTLY FROM THE SOURCE PROGRAM WHICH IS GIVEN IN COMPLETE FORM IN THE APPENDIX. NON-STANDARD EXITS FOR EACH LINK ARE EXPLATIVED. THE STANDARD EXIT IS IN SEQUENCE.

```
•41
              BDPX
N9
                              X,NII,NI,NA
TERNINATE JOB
USE SAME PROBE EPHEMERIS,SKIP TRAJ THIS PASS
USE SAME PROBE EPHEMERIS,SKIP TRAJ THIS PASS
USE SAME COMBINED EPHEMERIS, SKIP TO LINK 4
USE SAME COMBINED EPHEMERIS, SKIP TO LINK 4
THE CARD SMS-1 WILL CAUSE THE ODP TO READ TME INPUT
DECK FROM TAPE A2. NORMAL INPUT IS BY ONLINE READER.
              LA9
X
NII
NI
V4
+19
•412
             LALZ
                               X
TERMINATE JOB
•113
              SPACE
X
                               X
TERMINATE JOB
                               N4.N5.YIZ
EXIT TO PREDICT-DATA TAPE LINK
EXIT TO DATA FITTING LINK
EXIT TO LINK 12 FOR ERROR, THEN TO X
.411
             LA11
              N6
N12
                               NIZ, X EXIT TO LINK 12 FOR ERROR COMMENT, THEN TO X TERMINATE JOB
+42
                               N9,N12
STEP MAP COMPLETED, CHECK FOR MORE INPUT
EXIT TO LINK 12 FOR ERROR, THEN TO X
•¥2A
              LAZA
Ng
               NI2
                               N9,H10
(NO EXIT TO LINK 4)
IMPACI HAPPING COMPLETE, CHECK FOR HORE INPUT
COMMENT DUTPUT REQUESTED
•¥3
              LA3
              49
N10
              LA4
N6
N1Z
                               N6.H12
NO SORYING NECESSARY, GO TO DATA FITTING LINK
EXIT TO LINK 12 FOR ERROR
 -44
                               N9,N12
(NO EXIT TO LINK 6)
PREDICTS SORTED CHECK FOR MORE INPUT
EXIT TO LINK 12 FOR EAROR
              LAS
•¥5
              N9
N12
 -46
               LAG
N12
                               NIZ
EXII TO LINK 12 FOR ERROR
 .47
              LA7
                                N9,N10,N12
                               RETURN FOR MORE INPUT, NO RESIDUALS, STATISTICS REQUESTED EXIT MERE TO PRINT PARTIALS ERROR EXIT
                                N9.N12
(4D EXIT TO LINK LOI
RETURN FOR MORE INPUT
ERROR EXIT
 •48
              LAS
               N9
N12
                               N8,49
(NO EXIT TO LINK X)
AFTER PARTIALS PRINT RETURN TO LINK 8 FOR RESIDUALS
AND STATISTICS
NO OTHER REQUESTS, RETURN TO INPUT LINK
  910
              LAIO
                45
               N9
               END
```

VIII. CHECK CASES

```
SOURCE PROGRAM LISTING
                                                                                            *N1 DDPX
*N9 LA9
55164/*1.DE16
                                                                                                                                                                                                                                                                                                                                                                                  X,N11,N1,N4
| LAIZ | A | SPACE | K | SPACE
                                                                                                 *N12
                                                                                                                                                                                             LA12
SPACE
                                                                                                                                                                                                                                                                                                                                                                    X
SCFORF=2
Y9,1HOGPHL1,200,(MQIGR,BRROPT),18
Y11-(17APEX),6
Y12,(XN),1H3X1,1,(XN)-1,IH3Y),1
(XN)-2,1H3Z1),1(XN)-3,INJBX),1
(XN)-4,1H3Z1),1(XN)-5,INJBX),1
(EPOCH,1H3JT),1,(EPOCH-1,INJT+1),1
(DUT,GRAY-2),1,(AN,SCALE1+1),1
(RE,SCALE1),1,(FORE,HARRHN-5),1
(QD,HARRH+4)-1,(QH,HARRHN-5),1
(QD,HARRH+2)-1,(GRAYE,GRAY),1
(GRAYY,GRAY+3),1,(GRAYE,GRAY),1
(GRAYY,GRAY+3),1,(GRAYE,GRAY),1
(GRAYY,GRAY+3),1,(GRAYE,GRAY),1
(GRAYY,GRAY+3),1,(GRAYE,GRAY),1
(GRAYY,GRAY+3),1,(GRAYE,GRAY),1
(GRAY,GRAY+6),1,(FRAYS,GRAY+4),1
(FRAY,GRAY+6),1,(FRAYS,GRAP)F+3),1
(FRAY,GRAY+6),1,(FRO,GRAGH+1)
(FRO,GRAGP+45),1,(FRO,GRAGH+1)
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGP),1,(FRO,GRAGH+1),1
(FRO,GRAGHH-1),1,(FRO,GRAGH+1),1
(FRO,GRAGHH-1),1,(FRO,GRAGH+1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1,(FRO,GRAGHH-1),1
(FRO,GRAGHH-1),1
(FRO,GRAG
                                                                                                                                                                                                                                                                                                                                                                                  N4, N5, N12

N9, (C137);

N1), (CLPT, TRADE+1), 1, (CLPT+1, TRADE+), 1

(CLPBT, TRADE+2), 1, (CLPBT, TRADE+3), 1

(CLPBT, TRADE+4), 1, (CLPBT, TRADE+5), 1

TTAPK), 6
                                                                                                                                                                                                                                                                                                                                                                                       TTAPEX),6
N12,X
N11,1EPOX),14
N9,N12
N9,N10
N6,N12
N9,N12
V12
N9,N10,N12
N9,N10,N12
N9,N10,N12
N9,N10,N12
N9,N10,N12
N9,N10,N12
N9,N10,N12
N8,N9
                                                                                                                                                                                                                                                                                                                                                                                  THERE WERE NO GLARING SOURCE DECK ERRORS.
```

THE OBJECT STRING HAS 00516 DCTAL OR 334 DECIMAL WURDS.

PAGE MEADING
IMAS CHECK CASE - FIT SINULATED DATA, THEN HAP)
FPOCH
641102815,0757000
GEDLENTRIC POSTITON AND VELOCITY AT EPOCH
x=35062228E4 Y=214067726E4 z=-32403746E4
0X=-1R839477E1 DY-1.097865-8E2 02=-12280548E1
0THER DPTIONS AND CONSTANTS
IARCE=!MAS)
INFQ=3624200ESTIMATE IMESE PARAMETERS
X Y Z DX DY DZ HM AU
COVARIANCE HATRIX OF ESTIMATED PARAMTERS
DIAG=10.10.10.-.01.01.01.-01.1E-17.25.
PONITING ITHES, SAMPLE RATE, COUNT TIME
HOME-F-64.102916.04.103016.0, 3600.-60.
0ATA FILE SIGMA
HA(5)=-001 DEC(5)=-.001 CE3(5)=-.001
HA(5)=-001 DEC(6)=-.01 CE3(5)=-.01
HA(5)=-01 DEC(5)=-.01 CE3(5)=-.01
HA(5)=-.01 DEC(5)=-.01 DE3(5)=-.01
HA(5)=-.01 DE(5)=-.01 DE3(5)=-.01
HA(5)=-.01 DE3(5)=-.01
HA( 0300 134501 0300 13450 (23 101 (02 (03 104 121 {14 (16 126 127 10

```
INPUT COVARIANCE MATRIX OF ESTIMATED PARAMETERS
                                                                                                                                                                                                                ITERATION NUMBER
                   .00000000 00
.0000000 00
.0000000 00
.0000000 00
.0000000 00
.0000000 00
                                                                                                                                                                                                                                                                                                                                          .00000000 00
.00000000 00
                                                                                                                                                                                                                                                                                               .00000000 00
.0000000 00
.0000000 00
DX
DY
DZ
HM
UA
                                   INPUT J MATRIX OF ESTIMATED PARAMETERS
                                                                                                                                                                                                                   ITERATION NUMBER
                                                                                                                                                                    ÐΧ
                    CASE 1
                                                                                                                                                  IBSYS-JPIRAJ-SPACE 022265
DOUBLE PRECISION EPHEMERIS TAPE - EPHEMI
GHE .39860063 05 J .16234500-02 H -.57499999-05 D .78749999-05 RE .63781650 04 REM .63783112 04 G .66709998-19 A .88781796 29 B .88800194 29 C .88835976 29 OHE .41780741-02 AU .14959850 09 GM .49026293 04 GM .3211411 12 GMV .32470627 04 GM .409020000 0 DA .0000000 0 RA .3211410 04 GM .4902629 04 GM .3791870 08 GM .497870 08 GM .497870 08 GM .497870 08 GM .3791870 08 GM .37980320 06 MG .3791870 08 GM .3791870
                                                                                                                                               235700551423202200000000 J.D.= 2438728.13052093 NOV. 28,1964 15 07 57.000
 INJECTION CONDITIONS MARS
GEDCENTRIC X0 .5658222 04 Y0 .21466726 04 20-.32403748 04 0X0-.18839477 01 DV0 .10978654 02 DZ0-.12280548 01 CARTESIAN TO .54476999 05 GHA .29452970 03 GHO .66920765 02 DATE OF RUN 0222658 13460 EARTH IS THE CENTRAL BOOY FOR INTEGRATION COMEL EQUATIONS OF MOTION
                      PROBE IS IN EARTH'S SHADOW
                                                                                                                                              235700551423202200000000 J-D-= 2438728.13052083 NDV. 28.1964 15 D7 57.000
           D DAYS O HRS. O MIN. 0.000 SEC.
                                                                                                                                                                                                                                                                                                                  EQUATORIAL COORDINATES
                         GEOCENTRIC
                                                                                                                                      Z - 32403747 04 DX - 18839476 01 DY - 10078653 02 DZ - 12280547 01
RA - 20747711 02 V - 11206614 02 PTH - 12650650 02 AZ - 90421379 02
LCN - 862712015 02 V - 10777808 02 PTE - 13165718 02 AZ - 90421379 02
LS - 53810146 08 DXS - 27702202 02 DVS - 10340364 02 DZS - 47011071 01
LS - 13864926 08 DXX - 3705146-01 DVH - 9203371 00 DZF - 447917853 00
LT - 33854926 08 DXX - 87298950 01 DVT - 21590059 02 DZF - 491209471 01
LS - 31004216 03 RAS - 24457186 03 RAM - 18662395 03 LDM - 25209425 03
DR - 24549145 01 SHA - 62358861 04 DES - 21387223 02 DEM - 27525167 01
LCL - 21574659 03
       X .56682219 04 Y .21466724 04
R .68729157 04 DEC -28129771 02
XS -58995549 08 YS -12408705 09
XM -38523466 06 YM -44736395 05
XI -19750066 09 YI .58811766 08
RS .14755881 09 YS .30199675 02
GED -28291871 02 ALI .49950940 03
DUI .35000000 02 DI .37500000 01
L11901304 03 MCL .21144471 03
                                                                                                                                                                         GEDCENTRIC CONIC
                                                                                                                                           235700551326202354642400 J.D.= 2438728.12770659 NDV. 28,1964 15 03 53.850 B .24259727 05 SLR .14169237 05 APD .00000000 00 RCA .65656922 04 C1 .75152290 05 TFP .24314986 03 TF -.28142345-02 LTF -.25588951-01 EA .64722201 01 MA .10390285 01 TFJ .00000000 00
EPUCH OF PERICENTER PASSAGE

SMA --41536075 05 ECC .11580720 01

VH .30978206 01 C3 .95964924 01

TA .23551647 02 HTA .14971233 03
                                                                                                                                       -21466724 04
-11163641 03
-17385475 00
-98093365 00
-80323618 00
-41972693 00
-56972462 00
                   .56682219 04
.28132666 02
.43829295 00
.1904006-02
.45496172 00
.77710383 00
-.77518326 00
                                                                       Y
LAN
HY
QY
BY
                                                                                                                                             8 .24259727 05 THA .31712828 01 T VECTOR IN EARTH EQUATOR PLANE
       81Q .24222576 05 8RQ .13420740 04
                                                                                                                                                                                                         ALL VECTORS REFERENCED TO ECLIPTIC PLANE

DX -.18839476 01 DY -95837711 01 DZ -.54945680 01

PX -.35740509 00 MY -.6542501 00 MZ -.37751507 00

PX .89883011 00 PY -.25060393 00 PZ -.35958584 00

RX -.18420546-01 RY -15001224-01 RZ -.999717170 02

TX .63146777 00 TY -.77540212 00 TZ .00000000 00
                                                                                                                                       Z -.38269391 04
APF .21231137 03
HZ .73989141 00
0Z -.56855042 00
BZ -.67230674 00
```

```
CASE 1
                                                                                          IBSYS-JPTRAJ-SPACE 022265
   SXI .77710380 00 SYI .19849447 00 SZI -.59725159 00 DAI -.36673309 02 RAI .14328600 02 SXO -.77518323 00 SYO .63128952 00 SZO .23756120-01 DAO .13612515 01 RAO .14084150 03
    BTC .22379497 05 BRC -.93644235 04 B .24259727 05 THA .33729373 03 T VECTOR IN ECLIPTIC PLANE
            KELIDCENTRIC
                                                                                                                                                                                                ECLIPTIC COORDINATES
                                         Y .13525276 09 Z -.41570000 04 DX -.29676149 02
LAT -.16140947-02 LDN .66431758 02 V .36997370 02
YE .13525209 09 ZE -.33000000 03 DXE -.27792202 02
YT .20267817 09 ZT .76614050 07 DXT -.19062307 02
LDE .66433669 02 LTT .17675855 01 LDT .12434771 03
ESP .13998227-01 SEP .1186336 03 EPM .26249721 02
MSP .13507880 00 SMP .11829815 03 SEM .61571418 02
ETP .27453512-18 TEP .14621349 03 TPS .65283946 02
STE .36781888 02 EST .57931519 02 RPM .39431215 06
                                                                                                                                                             DY .21399603 02
PTH .22053898 02
DYE .11815832 02
DYT -11620746 02
RST .24560294 09
EMP .48000350 00
EMS .11829583 03
ISP .57933486 02
RPT .20883893 09
            O DAYS O HRS. 9 MIN. 49.244 SEC.
PROBE IS LEAVING EARTH'S SHADOW
                                                                                  235700551646202437164374 J.D.= 2438728.13734078 NOV. 28,1964 15 17 46.244
      2 DAYS O HRS. 52 MIN. 3.000 SEC.
                                                                                       23570067734020200000000 J.D.= 2438730.16666666 NDV. 30,1964 16 00 00.000
     2 DAYS O HRS. 52 HEN. 18.000 SEC.
                                                                                    235700677343202600000000 J.D.= 2438730.16684028 NDV. 30,1964 16 00 15.000
    215542322602 214412016141 614626031437 601731456722 204537405524 601471557132 2149808915 17481859072 204537405524 601471557132
     623733473151 223571732535 222507660544 602504407717 201741457741 200715435157 235700677343 20260000000
                                                                                                                                                                                                                              EARTH
END
                                                                                                                                  ITERATION NUMBER
                      J MATREX
                                                                            7
                                                                                                       DX
                                                                                                                                  DY
                                                                                                                                                             ĐΖ
          .86586481 07 .34045452 07 -.50491355 07 -.21792881 10 .13770335 11 -.15816493 10 .23125263 07 -.77185333-04 .34045452 07 .13346596 07 -.1992949 07 -.85587537 09 .54135596 10 -.62224241 09 .91247442 06 -.39473400-04 .50491359 07 -.19929949 07 .25589691 09 -.92586991 09 -.13497956 07 .45940778-04 .7.21792881 10 -.85537537 09 .12758964 10 .540452930 10 .39774509 12 -.55092013 09 .19371962-01 .13770335 11 .54135596 10 -.80622930 10 -.34665872 13 .29774509 12 -.55092013 09 .19371962-01 .13770335 10 .45224241 09 .92586914 09 .39774509 12 -.25151332 13 .39774509 12 -.2515132 13 .39774509 12 -.2515132 13 .39774509 12 -.42327558 09 .14149069-01 .23125263 07 .12747442 06 -.1349765 07 .58092013 09 .36776555 10 -.2227580 09 .297939999 18 .29793198-04 .23125263 07 .12745400-04 .45940778-06 .19371962-01 .12253334 09 .42147580 09 .39999999 18 .29793198-04 .39799999-01
                       CORRELATIONS BASED ON J MATRIX
                                                                                                                                                                                                                     ΑU
           MARS CHECK CASE - FIT SIMULATED DATA, THEN HAP
      ITERATION NUMBER 1 EPOCH 64/11/28 150757.000 CLOCK 135150 SGS .20501 00 QSGS .20501 00
                                                                                                                                                                                 NOMENAL O
                                                                                                                 OLD Q
                                                                                                                                                   NEW Q
                            no
                                                 SIDEVDO
                                                                                                                                           .54682105 04
.21464863 04
-.32403805 04
-.18839629 01
.10978653 02
-.12280560 01
.32383419-06
.14959850 09
                                                                                                                                                                            .56692222 04
.21466726 04
-.32403748 04
-.18839477 01
.10978654 02
-.12280548 01
                                                                                                          .56682222 04
-21465726 04
-32403748 04
-18839477 01
-10978654 02
-12280548 01
                                              .75099736 00
.25575302 00
.77465767 00
.57259488-03
.38780906-03
.75434680-03
                                                                          .31622776 01
.31622776 01
.31622776 01
.99999999-01
                                                                            99999997-09
                                                                                                             -32383419-05
-14959850 09
```

	J	INVERSE			1TERATION	NUK8ER	1	
	×	¥	z	ОX	DY	DZ	ни	AU
DX DY	59524516 -1155615 -1148700 -47341746 -1002857	8-01 .65409607- 3 0067366365- 3-0388619571- 5-0422914837-	04 .42117713-03 04 .29749136-03 04 .57407415-03 1627951063-15	88619571-04 .42117713-03 .32786490-06 .20192139-06 .39447369-06	27914837-04 - .29749136-03 - .20192139-06 - .15039588-06 - .28756925-06	46559043-04 .57407415-03 .39447369-06 .28756925-06 .56903910-06	28395455-16 27951063-15 14904260-18 14411435-18 28010333-18	-16363926-07 -18269573-07 -26668629-06 -14775477-09 -13669769-09 -25951601-09 -21195486-22 -25000000 02
	ÇOV	ARIANCE HATRIX O	F ESTINATED'PARA	IETERS	ITERATION	NUMBER	1	
	x	Y	z	ОX	DY	D2	ня	AU
DX DY	5952451 1155615 1148700 4734174 1002857	8-01	04 .42117713-03 04 .29749136-03 04 .57407415-03 1627951063-15	.42117713-03 .32786490-06 .20192139-06 .39447369-06 14904260-18	.29749136-03 .20192139-06 .15039588-06	.57407415-03 .39447369-06 .28756925-06 .56903910-06	27951063-15 14904260-18 14411435-18 28010333-18 _99999998-18	-16363926-07 -18269573-07 -26668629-06 -14775471-09 -13669769-09 -25951601-09 -21195486-22 -25000000 02
	COR	RELATION MATRIX	OF ESTIMATED PARA	AMETERS	ITERATION	NUMBER	1	
	x	Y	z	рх	DY	DZ	нн	AU
οx	9272694 -5943391 -7992637 -4863592 -5296623	8 00 -1000000 0 0034002631 1 0060514773 4 0023103514 4 0024133051 4-0711102686-	00 .59433910 00 01 ~34002631 00 00 .1000000 01 00 .94952738 00 00 .99025392 00 00 .9923972 00 00 .59023972 00 00 .59023972 00 00 .58852682-07	.10000000 01 .90932032 00 .91327070 00 26029327-06	23103514 0099025392 0090932032 001000000 019829978 0037161160-06	24133051 00 -98239720 00 -91327070 00 -95299978 00 -10000000 01 37131904-06	26029327-06 37161160-06 37131904-06	.13039122-07 .1428688-07 .68852682-07 .51608834-07 .70497417-07 .6805492-07 .42390971-14 .99999999 00
	STATIO	N NUKBER 41 FREQUENCY2038	64/11/28 516.0	ITERATION	NUMBER 1	PASS	NUHBER 000000	PAGE 1
TIME	TC Q	DEC			на		сез	
16000 17000 18000 19000 20000 21000 22000 23000	0 60 41 0 60 41 0 60 41 0 60 41 0 60 41 0 60 41	.78431416 01 .6 .10831269 02 .6 .12099763 02 .6 .12798196 02 .6 .13718797 02 .6 .13479367 02 .6 .13639556 02 .6 .13732021 02 .6	52-010010 52-01 -0010 52-01 -0010 52-01 -0010 54-01 -0000 57-01 -0010	6 .33477567 6 .34646760 2 .35971234 0 .13594732 7 .27772787 3 .42085743	03 .664-01 03 .667-01 03 .667-01 02 .669-01 02 .671-01 02 .676-01	.0007 .23 .0015 .21 0025 .20 .0000 .20 0011 .20	1456650 05 .130- 1700862 05 .130- 835252 05 .130- 1946827 05 .130- 1512531 05 .130- 1302005 05 .131- 1195638 05 .132- 1116875 05 .135-	010010 01 -0005 01 -0000 01 -0002 01 -0007
					64/11/29			
00000 15000 16000	0 60 41	.13762374 02 .7 .14855866 02 .7 .14977514 02 .6	84-01 .000	5 .28953924	02 -796-01 03 -813-01 03 -696-01	-0010 -20 0007 -14 -0001 -14	016154 05 -155- 764306 05 -157- 926591 05 -135-	01 .0005 01 .0005 01 .0005
17000 18000 19000 20000 21000	0 60 0 0 60 0	.15058219 02 .6 .15117309 02 .6 .15157664 02 .6 .15180486 02 .6 .15187077 02 .6	54-01080 52-01080 52-01 .080 52-01 .082	1 .33475542 9 .34990900 9 .50660299	03 -676-01 03 -676-01 01 -674-01	0019 0007 0006 0013 0010		
22000	0 60 51	.15178905 02 .6 .15156356 02 .6			02 -679-01	0004		
23000	0 60 51	.15156356 02 .6	67-01 -001:		02 <b>-</b> 691-01 64/11/30	<b>.</b> 0006		
01000	0 60 51 0 60 51	.15119074 02 .1 .14819789 02 .1	15-01 .000 11 01 .001	0 .65427979	02 .740-01 02 .115 01	0003 0003		
15000 16000		.15275558 0Z .7 .15356074 0Z .6	98-01001		03 -828-01 03 -698-01	0008 0008		

# JPL TECHNICAL MEMORANDUM NO. 33-204

MARINER STATISTICS	STATION 41	TTERATION 1
PASS DATA TYPE BEGINNING TIME	END NUMBER OF STO DEV	RMS FIRST MOMENT SECOND MOMENT
000000 CC3 11/28-160000 HA 11/28-160000 DEC 11/28-160000	11/29-160000 11 -574-03 11/30-160000 22 -101-02 11/30-160000 22 -158-02	.589-03 .133-03 .347-06 .101-02661-04 .103-05 .165-02 .459-03 .271-05
STATION NUMBER 51 64/1 FREQUENCY2038516.0	1/28 ITERATION NUMBER 1	PASS NUMBER 000000 PAGE 2
TIME IC Q DEC	HA	CC3
220000 60 41 .13129207 02 .735-01 230000 60 41 .13532427 02 .667-01	-0017 -28861228 03 -754-01 -00 -0005 -30329097 03 -686-01 -00	
000000 60 41 -13838208 02 -654-01 FREQUENCY -0	0009 _31818084 03 _674-0100	44
010000 60 0 .14068252 02 .652-01 020000 60 0 .14235436 02 .649-01 030000 60 0 .14349870 02 .649-01 040000 60 0 .14420789 02 .652-01 050000 60 0 .14458847 02 .652-01 060000 60 0 .14458844 02 .652-01 060000 60 0 .14465484 02 .659-01 070000 60 0 .14465484 02 .659-01 080000 60 0 .14359550 02 .664-01	-0010 .33319524 03 .671-01 .00 -0008 .34627475 03 .671-01 .00 .0002 .33738108 01 .671-01 .00 .0005 .18457121 02 .671-0100 .0005 .33498143 02 .674-01 .00 -0010 .48477208 02 .681-01 .00 .0015 .63393088 02 .706-01 .00 -0015 .63393088 02 .706-01 .00	05 35 48 05 05 07
FREQUENCY2038516.0  220000 6D 51 .14953242 02 .857-01 230000 60 51 .15068582 02 .676-01	0002 .28509123 03 .986-0100 .0000 .30005508 03 .698-01 .00	
230000 00 31 123000302 02 1010-01	64/11/30	13 - 114339947 D3 - 1153-01 - 20001
000000 60 51 .15137918 02 .657-01 010000 60 51 .15189322 02 .652-01 020000 60 51 .15225421 02 .649-01 040000 60 51 .15246821 02 .649-01 040000 60 51 .15246821 02 .649-01 050000 60 51 .15248872 02 .652-01 050000 60 51 .15248872 02 .652-01 070000 60 51 .15208640 02 .679-01 080000 60 51 .15200640 02 .679-01	-0019 .31513791 03 .681-01 -00 -0002 .3326529 03 .676-0100 .0009 .33561624 03 .674-01 .00 .0002 .57495905 00 .674-01 .00 .0004 .15727202 02 .674-01 .00 .0009 .3085974 02 .676-0100 .0003 .45980994 02 .681-01 .00 -0001 .45101895 02 .703-01000006 .75933573 02 .959-0100	31 .1510839 05 .131-01 .0021 17 .1556-990 05 .130-010020 12 .16022491 05 .130-01 .0007 122 .16497762 05 .130-010017 03 .16937468 05 .131-01 .0029 05 .17399181 05 .132-01 .0002 45 .17385710 05 .135-01 .0012
HARINER STATISTICS	STATION 51	ITERATION 1
PASS DATA TYPE BEGINNING TIME	END NUMBER OF STO DEV TIME POINTS	RMS FIRST HOHENT SECOND HOMENT
000000 CC3 11/29-220000 HA 11/28-220000 0000 135231 OFFLINE CONTROL 0000 13523 END DATA	11/30-080000 11 .144-02 11/30-080000 22 .267-02 11/30-080000 22 .794-03	.146-02 .255-03 .213-05 .269-02 .334-03 .722-05 .850-03 .303-03 .723-06

```
IBSYS-JPIRAJ-SPACE 022265
CASE 1
 DOUBLE PRECISION EPHEMERIS TAPE - EPHEMI
    235700551423202200000000 J.D.= 2438728.13052083 NOV. 28,1964 15 07 57.000
 INJECTION CONDITIONS HARS
 GEOCENIAIC XD -56682105 04 YO -21466863 04 ZO--32403805 04 DXO--188339629 01 DYO -10978653 02 DZO--12280560 01

CARTESIAN TO -54476999 05 GHA -29452970 03 GHO -66920165 02

DATE UF RUN 022265A 13525 EARTH IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION
            PROBE IS IN EARTH'S SHADON
         O DAYS O HRS. O MIN. 0.000 SEC. 235700551423202200000000 J.D.* 2438728.13052083 NOV. 28,1964 15 07 57.000
                                                                                                                                                                                                                                                      EQUATORIAL COORDINATES
                     GEOCENTRIC
                                                                                                                                                              DX -.18839629 01 DY .10978653 02 DZ -.12280559 01 V .11206616 02 PTH .12650726 02 AZ .90421329 02 VE .10175811 02 PTE .13165397 02 AZ .90421329 02 DX .27792202 02 DYS -.1080364 02 DZ .-47011071 01 DYH .92038371 00 DXH .47497853 00 DYH .2729950 01 DYH .2729059 02 DZ .-91209471 01 VH .10102904 01 RT .20883364 09 VT .25010665 02 RAS .24457186 03 RAM .18662395 03 LDM .25209425 03 SHA .62358811 04 DES -.21387223 02 DEM .27525167 01
                                                    Y -21466862 04 Z -32403804 04

DEC -28129837 02 RM .20742872 02

LAT -28129838 02 LON .86213175 02

YS -12408705 09 ZS -55810146 08

YM -44736399 05 ZM .18645579 05

YT .58811766 08 ZT .33854926 08

ALT .49950689 03 LDS .31004216 03

DT .37500000 01 DR .26543294 01

MCL .21144487 03 TCL .21574675 03
           X .56682103 04
R .68729131 04
R .68729129 04
XS -.58995549 08
       XX -.38523466 06

XX -.19750066 09

RS .14755881 09

GED -.28291938 02
                                                                                                                                GEDCENTRIC CUNIC
                                                                                                               235700551326202354532500 J.D.= 2438728-12770658 NDV. 28,1964 15 03 53.849
B .24259711 05 SLR .14169223 05 APO .00000000 00 RGA .65658862 OF
C1 .75152253 05 IFP .24315096 03 IF -28142471-02 LIF -25588937-01
EA .64722590 01 MA .10390338 01 TF -128142471-02 LIF -25580937-01
  EPOCH OF PERICENTER PASSAGE

SMA --41536060 05 ECC .11580719 01

VM .30978211 01 C3 .95964959 01

TA .23551788 02 MTA .14971234 03
                                                                                                                                                              X .56682103 04

1NC .28132733 02

WX .43829371 00

QX .19036494-02

BX .45496110 00

SXI .77710339 00

SXO -.77518320 00
                                                                                                                  8 .24259711 05
                                                                                                                                                                 THA .31712632 01
                                                                                                                                                                                                                    T VECTOR IN EARTH EQUATOR PLANE
       BIO -24222561 05
                                                           88D .13420647 04
                                                                                                                                                                    ALL VECTORS REFERENCED TO ECLIPTIC PLANE

DX -.18837629 D1 DY -.95837699 D1 DZ -.55

HX -.35740735 DD HY ..5542493 DD HZ -.35

PX ..89862780 DD PY -.25660437 DD PZ -.35

RX -.18420855-D1 RY ..15001476-D1 RZ -.97

X ..63146777 DD TY ..77540213 DD TZ ..07
                                                                                                                                                                                                                                                                         DZ --54945689 01

HZ --54945689 01

HZ --37751494 00

PZ --35958650 00

RZ --99971775 00

IZ .00000000 00
                                                                                                             Z -.38269499 04

APF .21231139 03

WZ .73989060 00

QZ -.56855910 00

BZ -.67230762 00
       X -56682103 04
1NC -42277902 02
WX -43829370 00
                                                       Y .68028036 03
LAN .13934386 03
NY .51035333 00
QY .52264020 00
BY .58376315 00
                                                                                                                        IBSYS-JPTRAJ-SPACE 022265
                                                                                                                                                                                                                                                                                                                  2
 CASE 1
       SXI .77710346 00 SYI .19849367 00 SZI -.59725243 00 DAI -.36673369 02 RAI .14328550 02
SXO -.77518328 00 SYO .63128954 00 SZO .23756519-01 DAO .13612746 01 RAD .14084150 03
       BIC .22379481 05 BRC -.93644400 04 8 .24259718 05 INA .33729368 03 I VECTOR IN ECLIPTIC PLANE
                                                                                                                                                                                                                                                              ECLIPTIC COORDINATES
                 HELTOCENTRIC
                                                                                                                                                                                                                 OY .21399602 02 DZ -.56944132 01
PTH .12089884 02 AZ .98735355 02
DYE .11815832 02 DZT .22182012 00
RSI .24560294 09 YSI .22326273 DZ
EMP .4790158 00 MEP .15127010 00
EMS .11829583 03 ESM .13270400 00
ISP .57933486 02 STP .36782550 DZ
RPI .20883893 09 SPN -.29895868 01
                                                                                                                                                                DX -.29616165 02
V .36997382 02
DXE -.27792202 02
DXT -.19062307 02
L0T .12634771 03
EPH .20249846 02
SEM .61571418 02
TPS .85283946 02
RPM .39431214 06
       X .50001217 08 Y .13525276 09 L -41570000 04 DX -.29676165 02 OY .21399602 02 R .415750169 09 L41 -.16140947-02 LDN .66631758 02 V .36997382 02 PTH .12089886 02 XT -.3795549 08 Y .3252209 09 ZE -.3300000 03 DXE -.27792202 02 DYE .1815832 02 XT -.13850512 09 YT .20267817 09 ZT .76614060 07 DXT -.19062307 02 DYT -.11620746 09 EPS .65134058-03 LDE .66633669 02 LTT .1787885 01 LDT .1264771 03 RST .24560294 09 EPS .65134059 02 ESP .13988227-01 SEP .11486342 03 EPH .28249846 02 EMP .47990158 03 EPT .39785765 02 EFP .27453512-18 TEP .14021333 03 EPT .85283946 02 TSP .57933486 02 SET .3586588 02 SET .35861888 02 EST .57931519 02 EPM .3947124 08 EPT .20883893 09 SAC .22280601-09 CE .20098677 03 CCT .27673355 03 STP .85283019 02 CPT .84329569 02 STN .864328642 02 REP .68729131 04 VEP .11206616 02 CPE .12085585 03 CPS .10111911 03
                                                                                                          2 --41570000 04

LOM -66631758 02

ZE --33000000 03

ZT -76614060 07

LTI :17875885 01

SEP -11486342 03

SMP :11829816 03

TEP -16021333 03

EST :57931519 02
        O DAYS O HRS. 9 MIN. 49.241 SEC.
PROBE IS LEAVING EARTH'S SHADOW
                                                                                                                 235700551646202436644705 J.D.= 2438728.13734075 NOV- 28,1964 15 17 46.241
          8 DAYS 19 HRS. 24 MIN. 2.979 SEC.
                                                                                                                  235701335107202775352514 J.D.= 2438736.93888865 DEC. 7,1964 10 31 59.980
          8 DAYS 19 HRS. 24 MIN. 2.979 SEC. 235701335107202775352514 J.D. = 2438736.93888865 DEC. 7,1964 10 31 59.980
CHANGE OF PMASE OCCURS AT THIS POINT SUN IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION
      225 DAYS Z HRS. 44 MIN. 31.060 SEC.
                                                                                                                   235723202427202007527522 J.D.= 2438953.24476920 JULY 11,1965 17 52 28.060
                                                                                                                                                                                                                                                        EQUATORIAL COORDINATES
                       GEOCENTRIC
                                                                                                                                                                DX -.13367881 02 DY -.23273616 02 DZ -.10141585 02 V .28691692 02 PTM .34373873 02 AZ .11345088 03 VE .15343772 05 PTE .60490716-01 AZE .26996480 03 DXS -.27634650 02 DYS -.88887630 01 GZ -.3855931 01 DXM -.97207475 00 DYM -.20357370-01 DXM -.10283720 00 DYM -.92348482 01 DYM -.20357370-01 DXM -.10283720 00 VM .97771122 00 RT .21291549 09 VT .26110227 02 ASS .11078189 03 RAM .26147890 03 UM .69935081 02 SMA -.20585269 09 DES .22069798 02 DEM -.24698652 02
                 -20952635 09 Y -22337826 08
-21092624 09 DEC -25722458 01
-21092624 09 LET -25722258 01
-50004426 08 YS -13176278 09
-16125000 05 YM -36611200 06
-21150188 09 YT -22632994 08
-15207519 09 YS -29283987 02
-25897513 01 ALT -21091986 09
-35000000 02 DT -86399999 05
-10205468 03 MCL -18000385 03
                                                                                                           Z -,94661789 07
RA -18608537 03
LON -34694236 03
ZS -57140097 08
ZM -,16894550 06
ZT -,93652949 07
RM -40336781 06
LOS -27323891 03
DR -16199062 02
ICL -92811081 01
```

3

CASE 1

```
ECLIPTIC COORDINATES
                 KELIOCENTRIC
    X -15952193 09
R -23158289 09
X -50004426 08
XI -16149745 09
LYE -10737647-04
EPS -39457595 02
EPT -17402787 03
SET -7748314 02
SAC -90460970-10
GCE -25793531 03
REP -21092624 09
                                                               Y -.16787898 09
LAT .50131026-01
YE -.14361901 09
YT -.16810964 09
LDE .28919679 03
ESP .62734623 02
MSP .62679107 02
ETP .59161249 01
SIE .39544090 02
                                                                                                                              Z .20262100 06 DX .14266770 02 DY -.15698153 02
LON .22646218 03 V .21212605 02 PTH .41969292 01
ZE .28500000 02 OXE .27634650 02 DYE .96890649 01
ZI .41261150 06 DXI .18411165 02 DYI -.1725897 02
LIT .10141465 00 LOI .22614924 03 RSI .23311467 09
SEP .77407180 02 EPH .10382236 00 EPH .98184596 02
SHP .77355039 02 SEM .15847546 03 EMS .21468911 02
IEP .54625775-01 FPS .13982710 03 TSP .31713545 00
ESI .63047593 02 RPH .21086426 09 RPI .19999992 07
                                                                                                                                                                                                                                                                                                                                  0Z -43642342-01

AZ .90121874 02

DZE .11010766-02

0ZT -76102304 00

VST .23568180 02

MEP .81107110 02

ESM .55952909-01

STP .3985781 02

SPY .39855862 02
                                                                  GCT -87216419 02
VEP -28691692 02
                                                                                                                                    SIP .13973033 03 CPT .93937702 02 SIN .93840928 02 CPE .91869014 02 CPS .82500247 02
                                                                                                                                                                   HELIOCENTRIC CONIC
                                                                                                                                       235700240660202547527522 J.D.= 2438723.37336585 NUV. 23.1964 20 57 38.810 8 .18568449 09 SLR .18086375 09 APO .23379039 09 RCA .1474714 09 C1 .48993033 10 TFP .19860889 08 TF -.47571538 01 PER .52542087 03 RA .16159514 03 MA .15749984 03 TF .22511424 03
EPRICH OF PERICENTER PASSAGE
SMA .19063377 09 ECC .22638500 00
YH .20955964 02 C3 -.69617315 03
TA .16533578 03 HTA .18000000 03
                                                                                                                                                                                                 X -.15952193 09 Y -.15410060 09
1NC .23492319 02 LAN .30826009 00
0X -.24467139-02 WY -.39862036 00
0X -.87558466 00 QY .44210871 00
0AP .20372204 02 RAP .58995966 02
                                                                                                                                           1 --66606276 08
                                                                                                                                  APF -60843575 02
WZ -91711352 00
QZ -19420888 00
8Z --19420890 00
                                                                     BRQ _38467694 08
                                                                                                                                          8 .18568449 09 THA .11956381 02
                                                                                                                                                                                                                                                                     T VECTOR IN EARTH EQUATOR PLANE
       BTQ .18165617 09
                                                                                                                                                                                                      ALL VECTORS REFERENCED TO ECLIPTIC PLANE
DX -14266770 02 DY --15698153 02 DZ --4;
MX -72491693 00 HY --68883299 00 HZ --2;
PX -48287949 00 PY -87568647 00 PZ --3;
RX --14870399-03 RY -26966991-03 RZ --9;
TX -87568673 00 TY --48287963 00 TZ --9;
                                                                                                                                                                                                                                                                                                                                      DZ -43542342-01

MZ -21271140-02

MZ -30795250-03

MZ -99999969

MZ -00000000000000
                  -.15952193 09 Y -.16787898 09
.13177922 00 LAN .68820811 02
.21446737-02 W -.83096782-03
-.87568491 00 QY .48287887 00
.87568413 00 BY -.48287899 00
.17645793-01 RAP .61126355 02
         X --15952193 09
INC -13177922 00
HX -21446737-02
QX --87568391 00
                                                                                                                                    Z -20262100 06

APF -35230551 03

WZ -99999736 00

QZ -22793191-02

BZ --22793196-02
                                                                                                                                           8 .18568446 09 THA .13059492 00 T VECTOR IN ECLIPTIC PLANE
       BTC .18568398 09 BRC -42323416 06
                                                                                                                                                                                                       ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET DX .15289946 02 DY .14689711 02 DZ .63605690 HX .66815475 00 HY .74341242 00 HZ .30118211-PX -88868259 00 PY .45847028 00 PZ -.69293238-RX .61581179-02 RY -31769557-02 RZ -.99997587 X .45648135 00 TY .88870403 00 TZ .0000000
     X .17224305 09 Y -.15479968 09
INC .17264500 01 LAN .31941926 03
WX -.19598038-01 WY -.22880996-01
QX -.45810362 00 QY -.88841510 00
DX .45810368 00 RY .88841512 00
DAP -.39702438 00 RAP .15271084 03
                                                                                                                                    Z --16642050 06
APF -19329738 03
WZ -99954606 00
QZ --29319061-01
BZ -29319065-01
                                                                                                                                                                                                                                                                                                                                       DZ .63605690 00
MZ .30118211-01
PZ -.69293238-02
RZ -.99997587 00
TZ .00000000 00
                                                                                                                                        B .18568448 09 THA .35831986 03 T VECTOR IN ORBIT PLANE OF TARGET
      BTO _18560465 09 BRO --54442254 07
                                                                                                                                       235723202427202007527522 J-D-= 2438953.24476920 JULY 11,1965 17 52 28.060 MARS IS THE CENTRAL BODY FOR INTEGRATION COMELL EQUATIONS OF MOTION
   225 BAYS 2 HRS. 44 MIN. 31-060 SEC.
CHANGE OF PHASE OCCURS AT THIS POINT
CASE 1
                                                                                                                                                18SYS-JPTRAJ-SPACE 022265
   225 DAYS 2 HRS. 44 MIN. 31.060 SEC.
                                                                                                                                             235723202427202007527522 J.D. = 2438953.24476920 JULY 11,1965 17 52 28.060
                                                                                                                                                                                                                                                                                                               EQUATORIAL COORDINATES
                       GEOCENTRIC
                                                                                                                                                                                                                                                                 DY --23273616 02
PTH -35373873 02
PTE -60490716-01
DYS --88887629 01
UYH -20357373-01
DYT -22095207 02
RT -21291549 09
RAM .25747793 03
DES -22069798 02
                                                                                                                                                                                                                                                                                                                               DZ -10141585 OZ
AZ .11345038 O3
AZE .26996480 O3
DZS -.38556930 O1
DZM -.10283720 O0
DZI -.10412922 OZ
YI .26110227 OZ
LDM .69934925 OZ
DEM -.24698632 OZ
       X -20952635 09 Y -22337827 08
R -21092624 09 DEC -25722458 01
X5 -50004427 08 Y S -13176278 09
XM -16126000 05 YM -36611200 06
XT -21150188 09 YM -26641200 06
XE -15207519 09 YS -229283947 02
GED -25597513 01 ALT -71091986 09
DUT -35000000 02 DT -38400000 04
CCL -10206468 03 MCL -18000385 03
                                                                                                                                                                                                 DX -.13367881 02
V .28691692 02
VE .15343772 05
DXS -.27634650 02
DXH .97207451 00
DXT -92234842 01
VM .97771098 00
RAS .11078193 03
SHA -.20585269 09
                                                                                                                                Z --94661791 07
RA .18608537 03
LON .34854236 03
LS .57140097 08
ZH --16854537 06
ZI --93652949 07
RM .40336780 06
LOS .27323891 03
OR .16199062 02
ICL .92811038 01
                    HET LOCENTRIC
                                                                                                                                                                                                                                                                                                                                DZ -.43642342-01
AZ .90121874 02
DZE .11010766-02
DZT -.76102304 00
VST .23588180 02
MEP .81107116 02
ESM .55952909-01
STP .39855791 02
SPN .39855901 02
                                                                                                                                                                                                                                                                     DY -.15698153 02
PIH .41969292 01
DYE .96890667 01
DYT -.14725897 02
RST .23311467 09
EHP .987845597 02
EHS .21468911 02
EHS .21468911 02
EFS .21468910 02
RFT .20000000 07
                                                                                                                                                                                                     DX .14266770 02

V .21212605 02

DXE .27653650 02

DXT .18411166 02

LOT .22614924 03

FPM .10812639 00

SEM .15847546 03

TPS .13902710 03

RPM .21086426 09
          X -.15952193 09
R .23150209 09
XE .50004427 08
XI -.16149745 09
LTE .1036088T-04
EPS .39857594 02
MPS .39965851 02
EFI .17402788 03
SET .77408312 02
SAC .30460971-10
                                                                     Y -.16787898 09 Z .20262100 06
LAT .50131026-01 LDN .22666218 03
YE -.14361901 09 ZE .27500000 02
YT -.16810964 09 ZT .41251150 06
LOE .28919679 03 LTT .10141465 00
ESP .62734623 02 ESP .77407780 02
MSP .62679107 02 MSP .77355038 02
ETP .59161291 01 TEP .5625775-01
STE .39544090 02 EST .63047594 02
                                                                                                                                                                                                                                                                   SIN .93840926 02
                                                                       GCT .87216416 0Z SIP .13973032 03
VEP .28691692 0Z GPE .91869014 02
                       -25793531 03
-21092624 09
                                                                                                                                                                                                                                                                                                                          ECLIPTIC COORDINATES
                        ARECCENTRIC
                                                                                                                                                                                                                                                                      DY -.97225593 00 DZ .71738064 00
PTH -.82615601 02 AZ .29904489 03
DP .15894875-04 ASO .94773534-01
        X .19755248 07 Y .23065303 06

R .20000000 07 DEC -.60269085 01

ALT .19966220 07 SNA -.12817149 07

HDG .32014240 03 SVL .59718631 01

SAC .90460971-10
                                                                                                                                       Z -.20999099 06 DX -.41443964 01
RA .66597225 01 V .43169362 01
ALP .17289476 03 DR -.42811324 01
NNG .14019883 03 SIA .17393110 03
                                                                                                                                                                                                                                                                             AREDCENTRIC EQUATORIAL COORDINATES
                                                                                                                                                                                                                                                                 DY -.40344328 01 Dt -.10989137 01
PTH -.82615601 02 AZ .30389252 03
PTP -.18227299 01 AZP .27013177 03
LOE .2152833 03 LOS .25651603 03
                                                                                                                                   Z .65006682 06 DX .10731629 01
RA .98561966 02 V .43169362 01
LQN .21477832 03 VP .13459592 03
RAS .14029968 03 DES .15866048 02
              X -.28159035 05 Y .18703261 07
R .19999999 07 DEC .18967598 02
R .20000000 07 LAT .18967598 02
AE .99052002 02 DEE .24866261 02
                                                                                                                                                                           AREDCENTRIC CONIC
  EPOCH OF PERICENTER PASSAGE

SMA -23114860 04 ECC .11133878 03 8 .25734764 06 SLR .28651611 08 APO .0000000 00 RCA .255504654 06 VH .43119555 01 C3 .18592940 02 C1 .11098716 07 IFP -4.5904197 00 IF .23034723 03 LIF .23034799 03 ZAE _168645501 03 ZAP .14546764 03 ZAC _9.080230 02 DEF _10293611 01 IR .5198771 04 CP .64745222 01
```

CASE 1

ALL VECTORS REFERENCED TO EARTH EQUATOR PLAME
DX -41443964 01 DY -11774092 01 D2 -27133758 00
RX -15201859 00 RY -98383238 00 RZ -99828441-01
PX -26914620 00 PY -95859092 00 PZ -93073576-01
RX --60460938-01 RX -17183313-01 RZ -97802227 00
DX -35080731 01 RXI -19586392 03
DAU -35080731 01 RAU -19586392 03
DAU -35080731 01 RAU -19585320 03 X .19755248 07 Y .29516781 06
INC .17357811 03 1A4 .34183582 03
MX .34867790-01 W7 .10627491 00
QX .96246738 00 QY .26621408 00
BX .27778184 00 BY .95617919 00
SXI .9601118 00 SYI .27281309 03
SXO .96465594 00 SY0 .25559374 00
E1E .32002128 02 ETS .14294957 03 Z -.10088414 06
APF .56319054 02
WZ -.993872528 00
QZ .62027644-01
BZ .92512715-01
SZU .62861092-01
SZU .61189193-01
ETC .23281654 03 810 --25623962 06 BRO -\_23855108 05 8 .25734764 06 THA -18531672 03 T VECTOR IN EARTH EQUATOR PLANE ALL VECTORS REFERENCED TO ECLIPTIC PLANE

DX -41443964 01 DY -97225593 00 DY -71738064 00

RX -15201645 00 HY -86242615 00 HZ -48261090 00

PX -26914808 00 PY -86242503 00 PZ -446677369 00

RX -16181709 00 PY -37727299-01 RZ -9860989 00

TX -22845947 00 TY -97355343 00 TZ -00000000 00

DAI -95676999 01 RAI -19320618 03

DAI -90808550 01 RAD -19220714 03 X .19755248 07 Y .23066303 06 Z -20999099 06
INC .15038975 03 LAN .35595333 03 APF .70857026 00
QX -946246739 00 QY -221772405 00 QZ .16202702 00
QX -96246739 00 QY -221772405 00 QZ .16202702 00
QX -796246739 00 SYI -22528158 00 SZI .16621285 00
CXI -946048544 00 SYI -22528158 00 SZI .16621285 00
CTE .54838743 02 ETS .16578618 03 ETC .25565315 03 RTC --22689608 06 BRC -.12143301 06 B -25734764 06 THA .20815534 03 T VECTOR IN ECLIPTIC PLANE ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET DX .10731639 01 OY -.41327579 01 DZ .63605730 00 PX .86781782 00 PX .16753827 00 PZ .63605730 00 PX .82467204 00 PY .28043826 00 PZ .49085507 00 PX .3705490-01 RY -14264919 00 RZ -.99808047 00 TX -.96788319 00 TY -.25140034 00 TZ .00000000 00 DAI .84749191 01 RAU .28365039 03 DAO .79644786 01 RAU .28365738 03 Y .19730736 07 Z -.16639353 06
LAW .90054571 02 APF .73760504 02
MY .48695213-03
0Y -.95987187 00 QZ .14297356 00
BY .29904839 00 QZ .4297356 00
SYI -.95731436 00 SZI .14737645 00
SYO -.96235193 00 SZO .13065914 00
EYS .16729771 03 ETC .25716468 03 BTO -.22361396 06 BRO -.12737582 06 B .25734764 06 THA .20966686 03 T VECTOR IN ORBIT PLANE OF TARGET ALL VECTORS REFERENCED TO AREDCENTRIC EQUATOR PLANE
DX .107316399 01 DY -.40344325 01 DZ -.10989137 01
MX .854781858 00 AY -.55548232-01 MZ .52735849 00
PX .62487278 00 PY .55678651-01 PZ .56246999 00
RX -.65441985-01 RY .24555948 00 RZ -.96708837 00
TX -.96637878 00 TY -.25712262 D0 TZ .00000000 00
DAI -.14744945 02 RAI .28489339 03
DAO -.15344397 02 RAD .28403305 03 -.28159081 05 .14172193 03 .51125371 00 .24125622 00 .82267267 00 .24865516 00 .23383781 00 .62416562 02 8TT -. 20890083 06 8RT -. 15029387 05 8 . 25734764 06 THA . 21573316 03 T VECTOR IN TRUE TARGET EQU. PLANE 230 DAYS 10 HRS. 21 MIY. 2.327 SEC. 235723542666202651147325 J.D.= 2438958.56179771 JULY 17.1965 01 28 59.322 CASE 1 18SYS-JPTRAJ-SPACE 022265 GEOCENTRIC EQUATORIAL COORDINATES X -.21527995 09 R .21831805 09 R .21831805 09 X5 -.62469540 08 XM .35175000 06 XT -.21534786 09 RS .15203851 09 GED -.37740790 01 DUT .35000000 02 CCL .10175299 03 Y --33370439 08
DEC --37486067 01
LAT --37486067 01
YS -12716259 09
YM --16575950 06
YT --33127602 08
VS -29292839 02
ALT -21831167 09
DT -9599999 03
MCL -18001457 03 I -.14273386 08
RA .18081126 03
LON .23189716 03
ZS .55144967 08
ZM -.11054600 06
LT -.14298221 08
RM -40425820 06
LOS .15925455 03
DR .1597919 02
ICL .32801724 03 DX --11678219 02 V -29382266 02 VE -15863321 05 0XS --26664940 02 0XM -45401406 00 DXT --74922993 01 VM -97316541 00 RAS -11616865 03 SMA --21117487 09 HELIOCENTRIC ECLIPTIC COURDINATES X -15279459 09
R -23224114 09
xE -62469360 08
XI -15286250 09
LTE -19224623-03
EPS -39289430 02
MPS -39355420 02
EPT -97133889 02
SET -75244898 02
SAC -89948905-10
GGC -25824700 03
REP -21831805 09 DY -.14835227 02 DZ -.57264804-01
PTH .35678491 01 AZ .90158692 02
DYE .12126476 02 DZE .25510788-03
DYI -.13895480 02 DZI -.76217288 00
RST .23212549 09 VSI .23690830 02
EMP .34905585 02 EMP .14102778 03
EMS .36804360 02 ESM .91192092-01
TSP .55071707-01 STP .11712955 03
RPT .25337335 06 SPM .39287756 02 Y --17489919 09
LAT -44918719-01
TE --13060480 09
YI --17468028 09
LUE -2942664 03
ESP -65407611 02
HSP -65497911 02
ESP -852800138 02
STE -39300298 02 Z .18206950 06 LUN .22885904 03 ZE .36950000 05 LTT .15469524-01 SEP .75302954 02 SMP .7514666 02 SMP .7514666 02 SMP .65410602-01 EST .65454801 02 DX -14986720 02 y -21087651 02 0XE -26664940 02 0XT -19172641 02 LUT -22881184 03 FPM -66719594-01 5EM -14310437 03 TPS -62814820 02 RPM -21863250 09 GCT -46264244 02 SIP -62050924 02 VFP -29302266 02 CPE -91154633 02 CPT \_47739595 02 CPS \_82009783 02 SIN .46975699 D2 AREOCENTRIC ECLIPTIC CODRDINATES .67905585 05 .25337335 06 .24999535 06 .32071056 03 .89948905-10 DY -.93974701 00 DZ .70490804 00 PTH .00000000 00 AZ .28059274 03 DP .98313903-03 ASD .76389518 00 Y -.21290933 06 Z .11940061 06 DEC .28115064 02 RA .28768963 03 SHA .22549665 06 ALP .16413254 03 SVL -.28109262 02 HNG .58804424 02 DX -.41859201 01 V .43476369 01 DR .46250916-07 SIA .96369993 02 AREOCENTRIC EDUATORIAL CODRDINATES DY -.40671214 01 BZ -.11289647 01 PTH .29456472-06 AZ .25164103 03 PTP -.44976981-07 AZP .26586315 03 LOE .15278519 03 LOS .19343714 03 Z .14339664 05 DX .10420695 01 RA .37338876 01 V .43476368 01 LON .54251868 02 VP .18982457 02 RAS .14291916 03 DES .15017378 02 Y -13603512 05 DEC -34468284 02 LAT -34468285,02 DEE -24671811 02 AREOGENTRIC CONIC 235723542666202651074600 J.D.= 2438958.56179770 JULY 17,1965 01 28 59.321 8 .25567811 06 SLR .26235053 08 APD .00000000 00 RCA .25337335 06 CI .11015753 07 FP .65103445-03 FF .23043127 03 LTF .23040201 03 FA .00000000 0D HA .69414259-04 FF .23043127 03 LTF .23043127 03 ZAC .98950581 02 OFF .10376917 01 FR .52012245 04 GP .84832144 01 

CASE 1

```
ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE

DX -41859201 01 DY -11476220 01 DZ .27282742 00

PX -96280351 00 AY -26281448 00 PZ .27282742 00

PX .26800603 00 PY -95841794 00 PZ .98013490-01

FX -61238004-01 RY -117311678-01 RZ -97977304 00

IX -22203356 00 TY .96228777 00 IZ .00000000 00

DAI .364635594 01 RAI .19578031 03

DAO .35467473 01 RAO .19475065 03
       X .67905585 05
INC .17331640 03
MX -34382693-01
QX -96280351 00
8X .27671320 00
SXI -96053725 00
SXI -96519082 00
ETE .46388492 02
                                                                            Y --24283757 06
LAM .34281734 03
WY --11119170 00
QY --26281449 00
8Y --95599888 00
SYI --27140216 00
SYO --25412526 00
ETS .14413840 03
                                                                                                                                                           Z -24835528 05
APF -57372230 02
WZ --99320403 00
QZ -62753035-01
BZ -97447246-01
SZI -6036026-01
SZO -61862808-01
ETC -23280735 03
         BTQ --25445631 06
                                                                                   BRQ --24965733 05
                                                                                                                                                                    8 -25567811 06
                                                                                                                                                                                                                                           THA .18560358 03
                                                                                                                                                                                                                                                                                                                      T VECTOR IN EARTH EQUATOR PLANE
                                                                                                                                                                                                                                          ALL VECTORS REFERENCED TO ECLIPTIC PLANE

DX -41859201 01 DY -93974701 00 DZ -70498804 00

PX -30620351 00 HY -21615121 00 MZ -16213590 00

PX -26800602 00 PY -8402980 00 PZ -47124377 00

EX -16205544 00 RY -37157772-01 RZ -93605894 00

IX -22691466 00 IY -97391465 00 IZ -00000000 00

DAI -95783617 01 RAI 19311549 03

DAO -90828306 01 RAD -19219159 03
                                                                            Y -.21290933 06
LAN .35604388 03
br -.49716855 00
QY -.21615121 00
8Y -.83830711 00
SYI -.22375123 00
SYO -.20853347 00
ETS .16698522 03
       X .6790585 05
INC .15010870 03
WX -34382493-01
QX -96280351 00
8X .27671319 00
SXI -96033725 00
SXI -9651902 00
ETE .69235305 02
                                                                                                                                                            Z -11940061 06
APF -71013729 02
MZ -86697247 00
02 -16213590 00
BZ -46975632 00
SZI -16639635 00
SZI -15786215 00
ETC -25565417 03
        810 -.22479983 06
                                                                                   BRC --12180449 06
                                                                                                                                                                    8 .25567811 06 THA .20845039 03 T VECTOR IN ECLIPTIC PLANE
                                                                                                                                                                                                                                                                    ALL VECTORS REFERENCED TO ORBIT PLANE OF TARCET 10420693 01 DY --41749402 01 DZ .62189034 23958636 00 MY --96025503 00 MZ .41304100 82268951 00 DY .27912224 00 PZ .49925014 63659198-01 RY -1.12826059 00 RZ -99905911 90828207 00 TY --24985965 00 MZ -99905910 90828207 01 RAI .28446920 03 TZ .00000000 379639885 01 RAO .28356148 03
       X .20844761 06 Y .70722138 05
INC .14896956 03 LAN .90114076 02
WX .51549237 00 HY .10263785-02
QX .23968636 00 QY .94625502 00
BX .82048545 00 BY .28726589 00
SXI .24712597 00 SYI .95765820 00
SXO .23222710 00 SYO .96274309 00
ETE .70745116 02 ETS .16849503 03
                                                                                                                                                              2 .12548319 06
APF .73889917 02
MZ -85689337 00
QZ .14504100 00
BZ .45993461 00
SZ .14751961 00
SZ0 .13855065 00
ETC .25716398 03
                                                                                                                                                                                                                                          ALL VECTORS:
DX .10420693 01
HX .23968636 00
PX .82268951 00
RX .36859198-01
TX -9682207 00
DAI .84932127 01
DAO .79639885 01
                                                                                                                                                                                                                                                                                                                                                                                                     LANE OF TARGET

DZ .62189034 OD

HZ .14304100 OD

PZ .49925014 OD

RZ -.98905911 OD

TZ .00000000 OD
        8TO -.22151246 06 BRO -.12768525 06
                                                                                                                                                                   8 .25567811 06 THA .20996021 03 T VECTOR IN ORBIT PLANE OF TARGET
                                                                                                                                                                                                                                         ALL VECTORS REFERENCED TO AREOCENTRIC EQUATOR PLANE
DX .104/20493 01 DV --406/11/214 01 D2 --11/28/96
MX .239/806356 00 MY --935/78/84 00 MZ -2596/731
PX .82268953 00 PY .53689554-01 PZ .5559499
RX --650/53/27-01 RY .24608663 00 RZ --95/76/20
TX --946/79/74 00 TY --25554/284 00 TZ .0000000
DAI --14746/201 02 RAI .2846/0575 03
DAO --15354/307 02 RAO .28393511 03
       X -20844761 06 Y -13603502 05

INC -14148794 03 LAY -12412029 03

WX -51549236 00 WY -3492090 00

QX -23948036 00 QY --93547863 00

BX -82048545 00 BY -62158088-01

SXI -24712597 00 SYI --93495411 00

SXO -232227110 00 SYI --93495412 00

ETE -76774132 02 ETS -17452404 03
                                                                                                                                                            2 .14339664 06

APF .11464694 03

WZ -.78247714 00

QZ -.25967317 00

BZ .56827810 00

SZ1 -.2545787 00

SZ0 -.26478719 00

ETC .26319299 03
                                                                                                                                                                                                                                                                                                                                                                                                     DZ -.11289647 01

MZ -.25967318 00

PZ .56594997 00

RZ -.96706279 00

TZ .00000000 00
       81T -. 20687620 06 BRT -. 15024492 06 8 . 25567811 06 THA . 21598922 03 T VECTOR IN TRUE TARGET EQU. PLANE
                                                                                                                                                                          235723543047202007527522 J.D.= 2438958.56699143 JULY 17.1965 01 36 28.060
CASE 1
                                                                                                                                                                            IBSYS-JPTRAJ-SPACE 022265
                              GEOCENTRIC
                                                                                                                                                                                                                                                                                                                                                                          EQUATORIAL COORDINATES
       X -.21528519 09 Y -.33381530 08
R .21832523 09 DEC -.37497550 01
X -.21532523 09 LAT -.37497550 01
XX -.62497325 08 YS .12715759 09
XX -.35195200 06 YN -.16540250 05
XY -.21535122 09 YY -.33138180 08
XX -.21535122 09 YY -.33138180 08
XX -.21535120 09 YS .29292856 02
GED -.37752351 01 ALT .21831086 09
DUT .35000000 02 07 .9599999 03
CCL .10175268 03 MCL .18001657 03
                                                                                                                                                           Z -.14278223 08
RA .18881394 03
LON .23002497 03
ZS .55142802 08
ZK -.1103987 06
ZT -.14303181 08
RM .40424757 06
LOS .15738492 03
DR .15997706 02
ICL .32779482 03
                                                                                                                                                                                                                                       DX --11676525 02

Y -29382748 02

Y -15863822 05

DXS --26663893 02

DXM -45305043 00

DXT --74905242 01

YM -97317924 00

RAS -11017388 03

SHA --21117983 09
                                                                                                                                                                                                                                                                                                                     DY -.24714124 02
PTH .32987712 02
PTE .57779984-01
DYS -11127437 02
UYH .7955750 00
DYT -.23571790 02
RT .21835491 09
RAM .33482840 03
DES .21265476 02
                                                                                                                                                                                                                                                                                                                                                                                           0Z -,10780385 02
AZ .11331638 03
AZE .26996476 03
DZS -,40258048 01
DZM .32755005 00
DZI -,11053182 02
VT .27090775 02
LGM .16039501 02
DGH -,15848649 02
                       HEL1DCENTRIC
                                                                                                                                                                                                                                                                                                                                                                                    ECLIPTIC COURDINATES
                                                                          DY -.14834143 02 DI -.57407438-01
PTH .35668485 01 AZ .90159081 02
DYE .12128819 02 DIE .25791578-03
DYT -.13974648 02 DIT -.762177318 00
RST .23212452 09 VST .23690931 02
EMP .38874725 02 ESP .1100616 03
EMS .36860782 02 ESP .91192092-01
TSP .155514057-01 STP .11752649 03
RPT .25338079 06 SPN .39287185 02
                                                                                                                                                                              .18204450 06
.22886137 03
.36950000 03
.62328000 05
.15345855-01
.75300904 02
.7514586 02
.66351990-01
.65457144 02
                                                                                                                                                                                                                                         OX -14987368 02

V -21087349 02

DXE -26663893 02

DXT -19173369 02

LOT -27881445 03

EPM -66351990-01

SEM -14304783 03

1PS -62418076 02

RPM -21863992 09
            x -.15278787 09
R .23224173 09
XE .62497325 08
xf -.15285389 09
                                                                                                                                                              LON
ZE
ZT
LTT
SEP
SHP
TEP
EST
                     -.15285389 09
.13924627-03
.39288659 02
.39354758 02
.96693188 02
.75242820 02
.89748448-10
.25824731 03
.21032523 09
                                                                                    GCT .46042143 02
VEP .29382748 02
                                                                                                                                                               SIP .61654202 02 CPT .47651963 02
CPE .91153933 02 CPS .82009314 02
                           AREOCENTRIC
                                                                                                                                                                                                                                                                                                                                                                                ECLIPTIC CODRDINATES
                    .66027186 05
.25338079 06
.25000279 06
.32071114 03
.89948448-10
                                                                                                                                                                                                                                                                                                                     OY --93949461 OO DZ .70476573 OO
PTH .43719955 OO AZ .28036057 O3
DP .98308158-O3 ASD .76387384 OO
                                                                                                                                                                                                                                     DX --41860009 01
V .43476370 01
DR .33174608-01
SIA .95929314 02
                                                                                   Y -.21333097 06
DFC .28195282 02
SHA .22469740 06
SVL -.28189059 02
                                                                                                                                                           2 .11971690 06
RA .28719760 03
ALP .16379697 03
HNG .58310008 02
                                                                                                                                                                                                                                                                                                                             AREOCENTRIC EQUATORIAL COURDINATES
                                                                              Y -11778436 05
DEC -34328284 02
LAT -34328285 02
DEE -24671581 02
                                                                                                                                                              2 .14289000 06
RA -32268656 01
LON -51903004 02
RAS .14292172 03
                                                                                                                                                                                                                                    DX -10418224 01
V -43476371 01
VP -19002142 02
DES -15016530 02
                                                                                                                                                                                                                                                                                                                 DY -.40671375 01 DZ -.11291355 01
PIH .43719955 00 AZ .25135458 03
PIP .10002983 00 AZP .26580524 03
LOE .15092650 03 LOS .19159786 03
                                                                                                                                                                                                         AREDCENTRIC CONIC
                                                                                                                                                            235723542666202651705722 J.D.= 2438958.56179777 JULY 17.1965 01 28 59.327 B .25567811 06 SLR .28235069 08 APD .00000000 00 ACA .25337335 06 C1 .11015757 07 TFP .44673269 03 TF .23043127 03 LTF .23040201 03 ACA .43718589 00 MA .4764590 02 LAC .98950570 02 DEF .10378332 01 R .52012236 04 GP .84832033 01
EPOCH OF PERICENTER PASSAGE

SMA =-23152520 04 ECC .11043662 03

VM .45084471 01 C3 .18562716 02

TA .44115593 00 MTA .90518817 02

ZAE .16978823 03 ZAP .14320822 03
```

CASE 1

ALL VECTORS REFERENCED TO EARTH EQUATOR PLAME
DX -41880009 01 DY -11423338 01 D2 .27279727 00
PX -90480854 00 MY -25562721 00 WZ .61996301-01
PX -26800593 00 PY -95861796 00 PZ .98019493-01
TX -61237934-01 RY -17311628-01 RZ -99797302 00
TX -227203348 00 TY .96228779 00 TZ .00000000 00
DAI .3646460 01 RAD .19475064 03 2 .24957948 05
APF .57372299 02
WZ -9932040 00
QZ .62752869-01
BZ .97447249-01
SZI .63637859-01
SZO .61862732-01
ETC .23280735 03 X .66027186 05 INC .17331641 03 WX -34382542-01 QX -94280352 00 8X .27671310 00 SXI -96033726 00 SXO -96519083 00 EIE .46403853 02 Y -.24335024 06
LAN .34281740 03
HY -.11119166 00
QY -.26281441 00
BY -.95599890 00
SYI -.27148208 00
SYU -.25412519 00
ETS .14413949 03 6 .25567811 06 THA .18560358 03 T VECTOR IN EARTH EQUATOR PLANE BRD --24965732 05 RTD --25445630 06 ALL VECTORS REFERENCED TO ECLIPTIC PLANE
DX --41860009 01 DY --93949461 00 DZ -70476573 00
RX --94648853 00 RY --20967486 00 MZ -15850254 00
PX -26800593 00 PY -84029881 00 PZ -41124378 00
RX --16205567 00 RY --31757728-01 RZ --9805985 00
TX --22691465 00 TY -97391465 00 TZ -00000000 00
DAI -95783515 01 RAI 1.9311549 03
DAD -90828213 01 RAO .19219159 03 I -11971690 06
APF -71013749 02
WZ -86697250 00
GZ -16213572 00
BZ -46975634 00
SZI -16639617 00
SZI -15786198 00
EIC -25565417 03 X .66027186 05 INC .15010870 03 MX -34382543-01 QX -96280353 00 BX .27671310 00 SXI -96033727 00 SXO -96519084 00 ETE .69250666 02 Y -.21333097 06
LAN .35604390 03
WY -.49716853 00
OY -.21615121 00
BY -.83830713 00
SY0 -.22375123.00
SY0 -.22853347 08
EIS .16698631 03 THA .20845039 03 T VECTOR IN ECLIPTIC PLANE BRC --12180449 06 BTC -\_22479982 06 ALL VECTORS REFERENCED 10 ORBIT PLANE OF TARGET DX .10418234 01 DY -41749240 01 DZ .62174067 00 PM . 23334517 00 PW -96237565 00 PW .1392233 00 PZ .82268954 00 PY .27912237 UU PZ .49825006 00 PX .38695183-01 RY -14284037 00 RZ .98895915 00 TX -9.982800 00 TY -24985992 00 TZ .00000000 00 DA1 .84831998 01 RAI .2848692Z 03 DA0 .79639766 01 RAU .28356149 03 1 .12576220 06
APF .73889938 02
WZ -85689364 00
QZ .14304079 00
BZ .49393454 00
SZI .14751939 00
SZO .13855045 00
ETC .25716397 03 Y .68848766 05
LAN .90114108 02
WY .1026641-02
QY .94025498 00
AY .28780600 00
SYI -95768817 00
SYO -96274305 00
ETS .16849611 03 X .20891516 06 INC .14896956 03 WX .51549223 00 QX .23968664 00 BX .82048546 00 SXI .24712624 00 5X0 -23222739 00 ETE -70760472 02 810 --22151248 06 840 -- 12768523 06 B .25567811 06 THA .20996020 03 T VECTOR IN ORBIT PLANE OF TARGET ALL VECTORS REFERENCED TO AREDCENTRIC EQUATOR PLANE
DX .10418234 01 DY -40671372 01 DZ -11271355 01
RX .2334517 00 MY -93586417 00 MZ -26602324 00
PX .82268953 00 PY .53689705-01 PZ .56594993 00
RX -4505446-01 RY .24508678 00 RZ -96679266 00
RX -96679766 00 RY -22555313 00
RX -153746214 02 RAD .28393513 03 Z .14289000 06
APF .11464695 03
WZ -78247710 00
02 -25967335 00
BZ .56827807 00
SZ0 -26478736 00
ETC .26478736 00 Y .11778488 05
LAN .12412032 03
AY .34928120 00
QY -93547851 00
BY .62158233-01
SYI -93495399 00
SYO -93592631 00
ETS .17452513 03 -20891516 06 -14148793 03 -51549222 00 -23968664 00 -82048546 00 -24712624 00 8 \_25567811 06 THA \_21598922 03 T VECTOR IN TRUE TARGET EQU. PLANE 220775513340 622733565354 217604377240 603414100141 U MATRIX FOR MAPPING FORWARD ITERATION NUMBER -1620028 06 .32833775 05 .12743440 05 -.32201795-01 .18749275-01 -.67970214-02 .0000000 00 .00000000 00 .55073909 05 .15447655 05 .5574600-04 -.12770882-01 .73212989-02 -.26537464-02 .0000000 00 .00000000 00 .82201993 05 -.21762466 05 -.91111089 04 .1887793-10 1.101989078-01 .34946406-02 .0000000 00 .00000000 00 .37202730 08 -.51210246 07 -.23455938 07 .80190345 01 -.47951245 01 .16743353 01 .0000000 00 .00000000 00 .222529723 09 .54473221 08 .21533065 08 -.51212658 02 .29795080 02 .11093359 02 .0000000 00 .00000000 00 .225683808 08 -.66539341 07 -.19942635 07 .59071261 01 -.33589278 01 .10381906 01 .0000000 00 .20000000 00 .21474836 01 .10468359 12 .74625055 11 -.14336000 05 .82944000 05 .71536000 05 .10000000 01 .00000000 00 .304104500 00 .746875000 00 .71875000 00 -.278687500 00 .13811045-06 .29429793-06 -.58207661-08 .00000000 00 .10000000 01 COVARIANCE MATRIX AT IMPACT ITERATION NUMBER .10637650 05 .22420325 05 .12501384 05 -.29614010-02 .15642850-02 -.28561890-02 .21474894-07 -.11718800 02 .22420325 05 .88347057 05 .47340614 05 -.28874972-01 .16219363-01 -.11518833-01 .18468361-06 -.17668771 02 .12501384 05 .47346614 05 .34772597 05 -.11713870-01 .75178238-02 .769898615-02 .74625093-07 -.74219149 01 .2971400-02 -.28379472-01 -.11713876-01 .20777808-07 -.12004298-07 .27655606-08 -.1533599-12 .33527526-05 .15642850-02 .16219363-01 .75178238-02 .20777808-07 .1262182-08 .1962602-08 .22439494-13 .73574456-05 .24549494-07 .18648361-06 .74625093-07 .1245999912 .27655606-08 .19335091-08 -.17536011-13 -.14550667-06 .21474894-07 .18648361-06 .74625093-07 .143599912 .82943994-13 .7357456-05 .21474894-07 .18648361-06 .74625093-07 .143599912 .82943994-13 .73574560-06 .21474894-07 .18648361-06 .74625093-07 .143599912 .82943994-13 .7357456-05 .21474894-07 .18648361-06 .74625093-07 .143599912 .82943994-13 .4755667-06 .21195486-22 .25000000 02

ENCOUNTER PARAMETERS AND STATISTICS 65/07/17 012859-322

•		ICK PANAMETERS A	no statistics (	3,01,71	,		
ε	3	.25568130 06	SHAA .321	74637 03			
	3.RU	12768892 06	SHIA -835	507688 02			
	3. TO	22151403 06	DEL T .372	26349 02			
E	3.RC	12180449 06	DEL 8 -332	40677 03			
E	3.TC	22479983 06	DEL S -160	38767 03			
1	rt.	.23040207 03	OEL 8R -101	49567 03			
1	TRETA	.10699438 02	DEL BT .316	53260 03			
,	N MATR	ıx					
		8.R0	B.TQ	TL	C3	5-15	S.RS
	B.RO	-10301371 05	17613031 05	17042226 04	-14831570-01	-10897963-02	.74349552-03
1	B. TO	17613032 05	.10019289 06	-10369679 05	-95422841-01	69411049-02	30035637-02
1	TL '	17042227 04	-10369679 05	-13858011 04	-51374437-02	62122480-03	24740746-03
	C3	.14831570-01	.95422841-01	.51374439-02	-66149723-06	11622786-07	69983611-08
	S.TS	-10897963-02	69411049-02	62122480-03	11622786-07	-53718767-09	.25609643-09
	S.RS	.74349549-03	30035638-02	24740745-03	69983607-08	.25609644-09	-15636082-09
	NORMAI	IZED N HATRIX					
		B.RO	8.10	ΤL	C3	\$ <b>.</b> TS	S.RS
	8.RD	.99999999 DO	54823672 00	45105381 00	-17967000 00	.46327063 00	<b>.</b> 58582360 00
	B. ID	54823674 00	.10000000 01	.88002785 00	-37065500 00	94612197 00	75884733 00
	TL	45105381 00	_88002785 00	-10000000 01	.16968076 00	72000467 00	53149401 00
	C3	.17967001 00	.37065500 00	-16968077 00	-99999998 OD	61657117 00	68812661 00
	<b>S</b> _ TS	.46327065 00	94612197 00	72000467 QD	61657117 00	.99999999 00	-88364239 00
	s.RS	.58582357 OO	75884738 00	53149399 00	68812657 00	.85364241 OD	.99999999 00
	DH 4000	HATREX					
	BA7040	B.RD	8.10	ŦL.	сз	5.15	S.RS
	x	.24330516 05	.64995540 05	29677574 05	-13145920 00	11157073-01	-56606676-03
	Y	.98666986 <b>0</b> 4	.27841637 05	11310281 05	-49878439-01	44800132-02	-25911632-03
	Z	13503961 05	40737256 05	-16973777 05	74298253-01	.66583250-02	28822134-03
	DX	59392066 07	13507410 08	<b>₽79610852 07</b>	35645702 02	.26993601 0 <b>1</b>	10983936 00
	DY	_38196933 08	-10563821 09	46808109 08	-20666396 03	17838378 02	.87565422 00
	DŽ	49480323 07	12244892 08	<b>-53021018 07</b>	23715798 02	.20346128 DI	14765713 00
	MU	-24638750 01	72774732 01	-14250550 00	.43780724-03	.10143746-04	60035930-05
	DN/UQ	MATRIX					
		B-RG	8.10	1L	C3	S.TS	S.RS
	x	14297395 00	16182701 00	-22580654 00	-28314225-06	.43595943-07	25130324-07
	Y	28136354-01	-98867121 00	-40707545-01	11281551-05	47139432-07	-27172928-07
	Z	99584158 00	32862216-02	35173765-01	.66306711-06	.55314040-07	-26337118-07
	DX	63716973 04	16846420 04	-91028766 D4	84913467 OI	36826590-03	-21228334-05
	DY	85946945 04	-40625272 04	_1336B759 05	13992037 01	-15366673-02	88579189-03
	DZ	.10419841 05	11039756 03	84511326 04	-12437810 01	-18654026 00	.26722055 00
	RU	.24778832 01	87136302 01	.23160733-01	.43191793-03	-10281612-04	<b>59267003-0</b> 5
	н нат	RIX FOR MAMEUVER	CALCULATIONS				
		, B.RC	B.TC	TL			
	B.RC	-11291969 05	19957052 05	22881360-01			
	8.10	19957053 05	.99202292 05	.11945784 00			
	ŢL.	22881360-01 OFFLINE CONTROL	-11945784 00	.18564079-06	1	(27	
1403 1403 1403	1	FERMINATE JOB END DATA	•	-		10	
03	-						

```
PAGE HEADING

(YEYUS CHECK CASE - STEP MAP1

EPOCH

620900500,2332000

GEOCEMTRIC POSITION AND VELOCITY AT EPOCH

X=-12456379E7 Y=-19998640E7 Z=-10071723E6

DX=-17446099E1 DY=-24232B7TE1 DZ=-11048412E0

DIHER OPTIONS AND CONSTANTS

TARGET=(VENUS)

ESTIMATE THESE PARAMETERS

X Y Z DX DY DZ NV AU

COVARIANCE MAIRIX DE ESTIMATED PARAMETERS

DIAG=1,-1,-1,-0001,-0001,-0001,-1E-17,25,-

MAP COVARIANCE MAIRIX TO

GETLOMOTOGO

DEFLINE CONTROL

KEY13) KEYT5)

END DATA
 0000 141001
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
0000 14100
                                                                                                                                                                                                                                                                                                                 123
                                                                                                                                                                                                                                                                                                                 (01
                                                                                                                                                                                                                                                                                                                 (02
                                                                                                                                                                                                                                                                                                               110
                                                                                                                                                                                                                                                                                                               (24
                                                                                                                                                                                                                                                                                                                 127
                                                                                                                                                                                                                                                                                                                    10
                                      INPUT COVARIANCE MATRIX OF ESTIMATED PARAMETERS
                                                                                                                                                                                                                                         ITERATION NUMBER
                                                                                                                                                                                                                                         DY
                                                                                                                                                                                                                                                                                            ΟZ
                                                                                                                                                                                                                                                                                                                                             NV
                                                                                                                                                                                                                                                                                                                                                                                               ΑU
                    .00000000 00
.00000000 00
.00000000 00
.0000000 00
.99999999-04
                                                                                                                                                                                                                                                                                                                               .00000000 00
.00000000 00
.00000000 00
.0000000 00
                                                                                                                                                                                                                                                                                                                                                                                 -00000000
                                                                                                                                                                                                                                                                             -00000000
                                                                                                                                                                                                                          ITERATION NUMBER
                                       INPUT J MATRIX OF ESTIMATED PARAMETERS
                                                                                                                                                                                                                                         DY
                     CASE 1
                                                                                                                                                                   TASYS-JPTRAJ-SPACE 022265
DOUBLE PRECISION EPHEMERIS TAPE - EPHEMI
GRE .3986D063 05 J .16234500-0Z H -.57499999-05 D .78749999-05 RE .63781650 04 REH .63783112 04 C .687609998-19 A .88761736 29 A .88800194 29 C .88836476 29 UNE .41780741-02 AU .1495850 09 GMM .49072629 04 C .45718170 08 GMJ .7126740735 09 EGM .39860320 06 KGM .490727190 4 JA .2920000-02 HA .00000000 0 DA .0000000 0 RA .31417000 04 RADIATION PRESSURE INPUT ARA .38300000 01 GB .3830000 00 VAS .19822000 03 GB1 .00000000 00 GB2 .00000000 00 SC .10310000 09
                                                                                                     VENUS 235575400641202000000000 J.D.= 2437912-51634260 SEPT. 5,1962 00 23 32.000
 INJECTION CONDITIONS
GEDCENTRIC X0-.14245297 07 Y0-.19398640 07 Z0-.10071723 06 DX0-.17446099 UL UY0-.24232877 01 UZ0-.11048412 00 CARTESIAN ID .14120000 04 GHA .34951873 03 GHD .34361929 03 DATE OF RUN DZZZ652 14103 EARTH IS THE CENTRAL BODY FUR INTEGRATION COMELL EQUATIONS OF MOTION
           PADRE IS OUT OF EARTH'S SHADOW
          0 DAYS 0 HRS. 0 HIN. 0.000 SEC. 235575400641202000000000 J.D.= 2437912.51634260 SEP1. 5,1962 00 23 32.000
                       GEOCENTRIC
                                                                                                                                                                                                                      DX -.1/446098 01 DY -.24232875 01 DZ -.11048412 00 V .29880081 01 PH .89393198 02 AZ .62816414 02 V .17549891 03 PH .97549786 00 AZ .62816414 02 DX .72901603 00 DX -.25999198 02 DZ .711230747 02 DX .72901603 00 DX -.2599198 02 DX .72901603 00 DX .721205738 02 DX .72901603 00 DX .7212057300 00 DX .72120575900 00 DX .72
       X -.14245296 07 Y -.19398639 07
R .2408380 07 DEC -.23963223 01
XS -.14343227 09 YS .42810504 08
XM -.21632510 06 YM -.21794325 06
XI -.88139502 00 YI -.41412272 08
RS .15083166 09 YS .29566049 02
GED -.24126335 01 ALT .24024598 07
DUT .35000000 02 OT .19200000 04
CCL .60702505 02 HCL .18217293 03
                                                                                                                                                  Z -.10071723 06
RA .23370847 03
LUN .24418974 03
ZS .18564279 08
ZM -.81923071 05
ZI -.22861356 08
RM .40143479 06
LOS .17386238 03
DR .29878460 01
ICL .33869367 03
                                                                                                                                                                                   GEOCENTRIC CONIC
 EPOCH OF PERICENTER PASSAGE
SHA --46363768 05 ECC .11464198 01
VH .29321057 01 C3 .85972439 01
TA .15011236 03 HTA .15072477 03
                                                                                                                                                        23557461277320233000000 J.D.= 2437903.65534361 AUC. 27.1967 03 43 41.688 8 .25992350 05 SLR .14571132 05 APD .00000000 00 RCA .6788751 04 C1 .76210645 05 IFP .76559031 06 IF -.88600988 01 IFF .888600978 04 A .27740861 04 IFF .80000000 00
                                                                                                                                                                                                                           Z -.10071723 06

APF .20465420 03

MZ .88884194 00

QZ -.41660539 00

BZ -.45686411 00

SZI -.37051435 00

SZO -.36930149-01
       X --14245296 07
INC -27271914 02
WX --39026894 00
0X --90656441 00
BX --7119641 00
SXI --30283033 00
SXU --58379930 00
                                                                             Y -.19398639 07
LAN -23836288 03
HY -24044364 00
QY -68592011-01
hY -53324496 00
SYI -87814137 00
SYD -881105764 00
                                                                                                                                                                                                                                                                                                   T VECTOR IN EARTH EQUATOR PLANE
        8TO _24871152 05 8RQ -.75521459 04
                                                                                                                                                          8 .25992350 05
                                                                                                                                                                                                                             THA .34310904 03
                                                                                                                                                                                                                              ALL VECTORS REFERENCED TO ECLIPTIC PLANE

DX -1.7446098 0.1 DY -.22672196 0.1 DZ -86269591.00

MX .70574097 0.0 HY -3.1562129 0.0 MZ -6.3441003 0.0

PX -16105420 0.0 PY -8.12272U6 0.0 PZ -.56063743 0.0

RX -.17009355 0.0 RY -.22808077 0.0 RZ -.95739486 0.0

TX -.79257140 0.0 TY -.60977912 0.0 TZ -.00000000 0.0
       X -.14245296 07 Y -.18198130 07

INC _43960422 02 LAN .21470263 03

MX _39028905 00 MY .57420661 00

QX _-9065441 00 QY -.10280938 00

BX _.71196439 00 8Y .30750167 00
                                                                                                                                              Z .67933540 06
APF .23385448 03
HZ .71981946 00
QZ -40950625 00
BZ -.63130772 00
```

CASE 1

```
8TC -25821250 05 BRC -- 29777056 04
                                                                                                    B _25992357 05 THA _35342171 03 | VECTOR IN ECLIPTIC PLANE
                                                                                                   X -.21047606 07 Y .49300253 06
INC .44529416 02 LAN .14007295 03
NX .44573257 UV NY .55647804 00
DX .13066194 00 BY .72572865 00
SXI .56802335 00 SYI .4084229 00
SXO -.87521258 00 SYO .40449869 00
     BTG .25569498 OS BRG -.46834718 O4
                                                                                                    8 _25994920 05 THA _34962039 03 I VECTUR IN DRBIT PLANE OF TARGET
       0 DAYS 8 HRS. 28 HIN. 56.404 SEC.
                                                                                                          235575417563202063552055 J.D.= 2437912.86977319 SEPT. 5,1962 08 52 28.404
       O DAYS & HRS. 28 MIN. 56-404 SEC. 235575417563202063552055 J.D. = 2437912.86977319 SEPT. 5,1962 OB 52 28-404 CHANGE OF PHASE OCCURS AT THIS POINT IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION
     95 DAYS 14 HRS. 14 MIN. 19.059 SEC.
                                                                                               235605301517202607404611 J.D.= 2438008-10961873 DEC. 9.1952 14 37 51.059
             HELIGCENTRIC
                                                                                                                                                                                                                                    EQUATORIAL COORDINATES
                                                x -.51022567 07

R .10968494 09

E .33012548 08

T --57926326 07
    XT -57926326 07

LTE -22612463 02

EPS .12889057 03

HPS .12884821 03

EPF .13652545 03

SEI .34151903 02

SAC .22900140-09

GCE .87219501 02

REP .51206375 08
                                                GCT .32905271 03
VEP .14973609 02
                                                                                                   SIP .34179292 02 CPT .72719283 02 SIN .72577188 02 CPE -80195375 02 CPS .10226691 03
                                                                                                                      HELIOCENTRIC CONIC
                                                                                                      235607605023202427494611 J.D.= 2438037.01349749 JAN. 7,1963 12 19 26.184
8 -12487168 09 SLR .12256196 09 APO .15158156 09 RCA .10260829 09
CI .40330759 10 IFP -,24972951 07 IF .12449715 03 PER .28446125 03
EA -44934634 02 MA -,36123921 02 IFF .95593274
EPOCH OF PERICENTER PASSAGE
SMA .12722493 09 ECC .19144548 00
VN .26606638 02 C3 -.10431455 04
TA -.52176583 02 MTA .18000000 03
                                                                                                                                                ALL VECTORS REFERENCED TD EARTH EQUATOR PLANE
DX -.346494257 0Z DY -.58201960 01 DZ -.33222712'01
MX -.99880593 00 MY -.35811769-01 MZ -.33229952-01
PX -.81748708 0D PY .52867570 00 PZ .23308281 00
MX -.19593895 00 RY .12623598 00 RZ -.92245689 00
TX .54159284 00 TY .8046094 00 TZ .00000000 D
    X -.51022567 07 Y .99263157 08
INC .25099883 02 LAN .35798326 03
WX -.14928005-01 W7 -.42934343 00
QX -.57575346 00 Q7 -.73681204 00
DAP .13478637 02 RAP .14720786 03
                                                                                                   1 .46385104 08

APF .14666950 03

WI .90556966 00

02 -.35442347 00

BI .35442346 00
    81Q -11628279 09 8RQ --45510967 08
                                                                                                   8 -12487168 09 THA -33862552 03 T VECTOR IN EARTH EQUATOR PLANE
CASE 1
                                                                                                             IBSYS-JPTRAJ-SPACE 022265
                                                                                                                                                   ALL VECTORS REFERENCED TO ECLIPTIC PLANE
UX -.36494297 02 UY -.66613023 01 02 -.73627064 00
RX -.99840594 00 HY -.66875705-01 X- -16240136-01
PX -.81748705 00 PY -.57593085 00 PZ .43765742-02
RX -.35288584-02 RY .24861283-02 RZ .99990509 00
TX .57593806 00 TY .81749474 00 TZ .00000000 00
    X -.51022567 07 Y .10952328 09 Z .30665980 07 INC .18528344 01 (AH .33250316 03 APF .17232764 03 AV -.28848005-01 W -.288693037-01 W .9994717 00 QX -.57575334 00 QY -.81699520 00 QZ -.32043274-01 BX .57575339 00 BY .81699527 00 BZ .32043277-01 DAP .24732883 00 RAP .14483478 03
                                                                                                      B _12487166 09 THA _35816372 03 T VECTOR IN ECLIPTIC PLANE
                                                                                                   ALL VECTORS REFERENCED TO ORBIT PLANE OF TARGET [ 1.12362223 07 DX -.15093336 02 DY ..33785281 02 DZ -.27414238 01 APF .22340356 03 MX -.28065725 00 MY ..95702555 00 MZ -.73031397-01 MZ ..99726595 00 PX ..34657532 00 PY ..92401173 00 PZ -.50176334-01 QZ -.53887788-01 RX -.186171393-01 RY -.47232232-01 RZ -.9987101 00 DZ ..53887780-01 TX ..9987101 00 DY -.50104881 00 TZ ..90801000 DD
    X .10522015 09 Y .30951192 08
INC .42377940 01 LAN .20514100 03
NX -31394514-01 WY .64695422-01
QX -42965851 00 QY .3639790 00
DAP -29105203 01 RAP .68465271 02
      BTO -12469110 09 BRD --67127396 07
                                                                                                    8 .12487166 09 THA .35691845 03 T VECTOR IN ORBIT PLANE (F TAPGET
      95 DAYS 14 HRS. 14 MIN. 19.059 SEC. 235605301517202607404611 J.D.= 2438008.10961873 DEC. 9,1962 14 37 51.059
CHANGE OF PHASE OCCURS AT THIS POINT VENUS IS THE CENTRAL BODY FOR INTEGRATION CONELL EQUATIONS OF MOTION
      95 DAYS 14 HRS. 14 MIN. 19.059 SEC.
                                                                                                        235405301517202607404611 J.D.= 2438008.10961673 DEC. 9.1962 14 37 SL.059
                  GEOCENTRIC
                                                                                                                                                                                                                                    EQUATORIAL COORDINATES
                                                                                                Z --10736197 DB
RA -22042683 D3
LDN -28310597 D3
ZS --57121301 D8
ZM -87959625 D5
ZT --12684154 DB
RM -36544041 DB
LDS -31861212 D3
DR -13951461 D2
TCL -61833203 D2
                                                                                                                                                  DX -.69869830 01 DY -.11838899 02 DZ -.59356140 01 V .14973609 02 PIH .68707168 02 AZ .12448867 03 VE .34645765 04 PTE .21920480 00 AZE .25995161 03 DX .25957273 02 DX -.58507273 02 DY -.5921705 00 DZ -.26939151 03 DX -.55921105 00 DY -.59271705 00 DZ -.26939153 00 DY -.59271705 00 DZ -.26939153 00 DY -.876218830 01 DZ -.16046215 01 VH .10777012 01 RT .53048469 08 VI .10472501 02 DX AZE .2593048 03 RM .51840267 02 DM .11522140 03 SMA -.29672879 08 DES -.22812463 02 DEM .13927554 02
    X -.38114804 08 Y -.32466730 08 R .51206374 08 LAT -.12102736 02 X5 -.33012548 08 Y -.33813495 09 XM .21915150 06 XT -.38805180 08 YT -.33873445 09 XS -.13172784 09 YM .2789500 06 XT -.358005180 08 YT -.33873445 09 CEO -.12182757 02 ALT -51199996 08 001 -.35000000 02 DT .38600000 04 CCL .27278050 03 MCC .17997891 03
              HELTOCENTRIC
                                                                                              Z -46385104 08

LON -92942489 02

ZE -57121301 08

ZT -44437148 08

LTT -24385356 02

SEP -35413715 02

SMP -35277547 02

TEP -18580018 01

EST -16063212
                                                                                                                                                                                                                                     EQUATORIAL COORDINATES
                                                                                                                                                DX - 36494256 02 DY - 58201940 01
V ,3710484 02 PTH - 77073143 01
DXE - 225507273 02 DYE .60187031 01
DXT - 35001950 02 DYT - 226941798 01
LOT .93387681 02 RST .10762945 09
EPM .78819662-01 EPM .1190073 02
SEM .15551562 03 EMS .24425006 02
TPS .44421389 02 TSP .75039911 00
RPM .51564818 08 RPT .25000000 07
       X -.51022566 07 Y -99263156 08

R .10968493 09 LAT .25017435 02

XE .33012548 08 YE .13172890 08

XT -.57920326 07 YI -97856442 08

LIE .22812463 02 LOE .75930981 02

EPS .12890927 03 ESP .15695704 02

MPS .12896021 03 HSP .1575495 02

LOE .7575704 02

MPS .12896021 03 HSP .1575495 02
                                                                                                                                                                                                                                                   DZ -.33262711 01
                                                                                                                                                                                                                                                   DT -,33262711 01
Az .92101531 02
DZE .26093425 01
DZT .10045215 01
VST .35209096 02
MEP .16673057 03
ESM .58516955-01
STP .14492820 03
SPN .12888343 03
               -57926326 07

-22812463 02

-12889057 03

-12890621 03

-13652545 03

-14151904 02

-22900140-09

-87219499 02

-51206374 08
                                                 11 -9/35642 03

LOE -75930981 02

ESP -15695704 02

MSP -15756195 02

ETP -41616422 02

STE -12978488 03
```

CASE 1

APHRODIOCENTRIC EQUATORIAL COORDINATES Y .14067148 07 Z .19479566 07 0X -.14027647 01 DY -.31260163 01 DZ -.43307926 01 DEC .51185796 02 RA .63859499 02 V .55222722 01 PTH -.88684816 02 AZ .24853740 03 SHA .14365062 07 ALP .50235699 02 DR -.55208174 01 DP .29047796-05 ASD .14209440 00 SYL .29805915 02 HMG .17362165 02 SIA .13638335 03 x .69037591 06 R .25000000 07 LT .24938001 07 GE .23110942 03 APHRODISCENTRIC CONIC 235605634232202757404611 J.D.= 2438013.29235961 DEC. 14,1962 19 00 59.871 B .57624974 05 SLR .30915540 06 APD .00000000 00 RCA .47676753 05 CA .3168472 06 IFP -444774881 06 IF .100777602 03 IFF .10073765 03 EA -.25498362 03 MA -13134227 05 IFF .10073765 03 FFF .95593274 02 CAC .72681574 02 DEF .21171787 02 IR .13100778 05 GP -.29764390 02 EPOCH OF PERICENTER PASSAGE
SMA --10741161 05 ECC -54573169 01
VH -54986978 01 C3 -30235677 02
TA --99240591 02 MTA -10055855 03
ZAE -13728729 03 ZAP -34431163 03 Z -19479566 07 APF -20564144 03 WZ --58332922 00 02 --73224367 00 8Z --21135770 00 SZI --7842141 00 SZI --65543945 00 ETC -25151733 03 X .69037591 05 INC .12568504 03 WX -.8992326-02 OX -.42689480 00 8X .96716951 00 SXI -.25397198 00 SXI -.25397198 00 SXI -.3536138 00 ETE .30676850 03 Y .14067148 07
LAN .18063363 03
WY .81217684 00
QY -59064020 00
8Y -14110654 00
SYI -56608119 00
SYO -47722978 00
ETS .41834750 02 DZ -.43307926 01 MZ -.22933716 00 PZ -.35148387 00 RZ -.62044313 00 TZ .00000000 00 8 .57624974 05 BRG -19630194 05 THA .16008325 03 T VECTOR IN EARTH EQUATOR PLANE RIO --54178344 05 ALL VECTORS REFERENCED TO ECLIPTIC PLANE

DX -14027647 01 DY -45909108 01 DZ -27297031 01

MX .96107515 00 HY -2326084 00 HZ -14910819 00

PX .90425774 00 PY -36226754 00 PZ -22602173 00

RX .14442028 00 RY -47274711 00 PZ -80928300 00

TX -19553685 00 TY .29216200 00 TZ .00000000 00

DAI -29624573 02 RAI .25301252 03

DAG -24297925 02 RAD .23003940 03 2 -12275376 07
APF -20613286 03
M2 -85829785 00
Q2 -46069866 00
82 -113777493 00
S21 -49431472 00
S20 -41148134 00
E1C -26231466 03 X .69037591 06
INC .14912599 03
WX -.89822391-02
QX -.42689838 00
BX .96716970 00
SXI -.25397199 00
SXI -.25336812 00
ETE .31756582 03 Y .20655587 07
LAT .18100294 03
MY .51307865 00
QY -.77814811 00
BY -.21354349 00
SY1 -.89135503 00
SY0 -.69858971 00
ETS .52632071 02 B .57624970 05 THA .17088060 03 T VECTOR IN ECLIPTIC PLANE BIE -- SA896602 DS 88C -91331112 O5 X .21701731 07 Y -11002964 06 Z .12362209 07 DX -447923321 01 DY .11675437 00 DX -247923321 01 DY .11675437 00 DX .178791010 01 DX .17879101 DX .178791010 01 DX .178791010 01 DX .178791010 01 DX .17879101 810 - 57386433 05 880 -52377827 04 8 -57624970 05 THA -17478494 03 T VECTOR IN ORBIT PLAYE OF TARGET 100 DAYS 19 HRS. 23 MIN. 26.074 SEC. 235605635514202411434156 J.D.= 2438013.32428326 DEC. 14,1962 19 46 58.075 5 CASE 1 IBSYS-JPIRAJ-SPACE 022265 GEUCENTRIC **EQUATORIAL COORDINATES** X -41051080 08 Y -38291325 08
R .57778180 08 DEC -1368687 02
X -57778180 08 LAT -1368687 02
X -19597573 08 Y -13388624 09
X -22533850 06 Y -38295325 08
X -41081809 08 Y -38295325 08
X -41081809 08 Y -38295325 08
X -41081809 08 Y -38295325 08
DUI .35000000 02 9T .2400000 03
CEC .27336213 03 ACL .1798610 03 L -.13671191 08
RA .22300789 03
LON .2032e984 03
LS -.58055864 08
LM .12794737 06
LT -.13647311 08
RM .36631403 06
LOS .24193441 03
UR .16896584 02
UR .32553668 03 OX --74626395 01 V -18177398 02 VE -40878328 04 DXS -29999814 02 DXM --85653022 00 DXI --44356764 01 VX -10184389 01 VX -10184389 03 SHA --35388548 08 HEL LOCENTRIC EQUATORIAL COORDINATES OY -.11365090 02 DZ -.57040765 01
PTH -.72353166 01 AZ .95856804 02
DYE .35449458 01 OZE .15359250 01
RST .10757004 09 VST .35228403 02
EMP .80678812 02 EMP .908943153 02
EMS .43232903 02 EMP .908943153 02
EMS .43232903 02 EMP .10231507 00
ISP .18654685-01 STP .11464353 02
RPT .39109093 05 SPN .12301465 03 X --21453506 08 R .10755761 09 XE .19597573 08 XT --21484235 08 LTE .23221698 02 EPS .12302097 03 APS .12339864 03 EPI .11679526 03 SEI .37796579 02 SAC .23814957-09 CFF .86617970 02 Z .44384672 08 DX -37462473 02
LDN .10264880 03 V .39561833 02
IE .58055864 08 DX -.29999834 02
LT .44408532 08 DXT -.34435511 02
LTT .22382966 02 LDT .10266685 03
SEP .37769753 02 EPM .37600648 00
SMP .37494848 02 EPM .37600648 03
IEP .33542591-01 IPS .10851589 03
EST .19225541 02 RPM .57839493 08 Y .95544916 08
LAT .24372012 02
YE .13388624 09
YT .75590916 08
LGE .81672475 02
LSP .19209271 02
HSP .19106508 02
ETP .61170761 02 STE .12297788 03 UCT .23205454 03 VEP .18177398 02 SIP .99394255 02 CPT .11992423 03 CPE .80314971 02 CPS .10271893 03 SIN -11080259 03 EQUATORIAL COURDINATES X \_30728787 05 Y \_40000217 04 R \_39109093 05 DEC =-37594455 02 ALT \_32209093 05 SHA =-37080350 05 HGE \_23697902 03 SVL =-32544186 02 SAC \_23814957-09 Z -\_23859226 05 DX -\_30269631 01 DY -\_40196605 01 DZ -\_45723879 01 RA \_\_74165933 01 V -\_67990372 01 PTH \_\_75343634-06 AZ \_\_21192412 03 ALP \_\_12618129 02 DR \_\_74910647-07 DP \_\_99607556-02 ASD \_\_91216356 01 HNG \_\_11213077 03 S1A \_\_10967362 03 APHRODIOCENTRIC CONIC 235605635514202411425570 J.D.= 2438013.32428326 DEC. 14.1962 19 46 58.075
B .48858741 05 SLR .21771052 06 APD .00000000 00 RCA .39109093 05
C1 .26590418 06 IFP .98913489-04 IF .10080794 03 LIF .10077252 03
EA .00000000 00 AM .28129103-05
TF .10080794 03
ZAG .71965344 02 DEF .25297576 02 IR .13206245 05 GP -30444610 02 ALL VECTORS REFERENCED TO EARTH COULSTOR PLANE
OX = 30259631 01 OY = 4019605 01 DL = 45723879 01
MX = 44520467 00 MY = 59121025 00 ML = 62725520 00
PX = 70571975 00 PY = 10227857 00 PL = 61006849 00
PX = 33778184 00 PX = 171389431 00 PL = 61339898 00
TX = 33778184 00 PX = 271389431 00 PL = 200000000 00
DX = 50192323 00 TY = 42769475 00 TL = 00000000 00
DA1 = 52164318 02 RA1 = 24467864 03
DA0 = 31506402 02 RA1 = 24467864 03 X -30728787 05 1°C .11477088 03 MX -42946162 00 0X -44520466 00 BX .86413916 00 SXI -26234753 00 SXI -26045214 00 ETE -30229105 03 Z --23859226 05
APF -22221299 03
WZ --41899068 00
QZ --67250520 00
BZ --44800129 00
SZ1 --3897317 00
SZ0 --52259473 00
ETC -25076851 03 Y .40000217 04
L4N .20822792 03
HY .80000596 00
QY -59121024 02
BY .22925609 00
SYI -55546560 00
SYO -59925832 00
ETS .29230400 02

CASE 1

```
BTQ --39373639 05 BRQ -35684405 05 B -48858741 05 THA -13308353 03 T VECTOR IN EARTH EQUATOR PLANE
                                                                                                                                                                                                                                             ALL VECTORS REFERENCED TO ECLIPTIC PLANE
     X .30728787 05 Y -.58220504 04 Z -.23481196 05 INC .13464185 03 LAN .21712739 03 APF .23754776 03 INC .44520457 00 QY -.80995385 00 QZ -.38179519 00 BX .86413918 00 QY -.80995385 00 QZ -.38179519 00 BX .86413918 00 AY .32104914-01 BZ -.50227281 00 SXI -.26234753 00 SYI -.R2289482 00 SZI -.50400189 00 CX -.30465215 00 SXI -.8289482 00 SZI -.50400189 00 EEE .331365265 03 EIS .40592002 02 EIC .26213010 03
                                                                                                                                                                                                                   ALL VECTORS REFERENCED ID ELLIP
DX -.30259631 01 UY -.55069064 01
NX -.44520466 00 NY -.80995365 00
PX -.78571915 00 PY -.1488692 00
RX ..15308934 00 RY .46018908 00
IX -.93275255 00 IY .30374755 00
DAI -.30265119 02 RAI .25231716 03
DAU -.13948913 02 RAI .25231716 03
                                                                                                                                                                                                                                                                                                                                                                 DZ -.25958396 01

#Z -.38179517 00

PZ -.60040245 00

RZ -.86370255 00

TZ .00000000 00
     BTC -.39749469 05 BRC .28410497 05
                                                                                                                                                 8 .48858742 05
                                                                                                                                                                                                                    THA .14444512 03 I VECTOR IN ECLIPTIC PLANE
                                                                                                                                                                                                                  ALL VECTORS REFERENCED TO ORAIT PLANE OF TARGET DX -.60659888 01 DV -14842515 01 DZ -.26884349 01 PX -.89218352 00 MY -.21830320 00 MZ -.39541406 00 PX -40900621-01 PY -.832279864 00 PZ -.55206295 00 RX -.50638494 00 RY -.38010652-01 RZ -.86211938 00 DAI -.336446810 02 RAI -.17795301 03 DAO -.1536278 02 RAO .155797407 03
      X .L5995859 04 Y -.32570000 05 Z -.21590682 05 INC .13722957 03 LAN .13851610 03 APF .23438793 03 LAN .44991472 00 WY .50871417 00 MZ -.7346052 00 OX -89218351 00 QY .21830318 00 QZ -.39541406 00 OX .23527313 00 BY -.86038096 00 AZ -.45207910 00 SXI -.86157462 00 SXI .30641724-01 SZI -.50670514 00 CTE .30926123 03 ETS .61084647 02 ETC .27339349 03
      BTO --41602414 05 BRD .25620609 05 B .48858741 05 THA .14837340 03 T VECTOR IN ORBIT PLANE OF TARGET
   100 DAYS 19 HRS. 26 HIM. 19.059 SEC.
                                                                                                                                                  235605635567202607404611 J.D.= 2438013.32628540 DEC. 14,1962 19 49 51.059
                                                                                                                                                                                                                                                                                                                                          EQUATORIAL COORDINATES
                         GEOCENTRIC
                                                                                                                                                                                                                                                                                        DY -.14914270 02 DZ -.72176348 01
PTH .68466816 02 AZ .11972255 03
PTE .23706050 00 422 .26995361 03
DYS -.3543987 301 DZS -.15355094 01
DYH .-53276151 00 DZR -.13055575 00
DYT -.10891180 02 DZT -.26460171 01
RT .57798931 08 VT .12058464 02
AMM .12821159 03 LGW .10775079 03
DES -.23221815 02 DEM .19337558 02
      X -.41052372 08 Y -.38293904 08 Z -.13672441 08 R .57781104 08 DEC -.13687417 02 RA .22300891 03 R .57781103 08 L4T -.13687417 02 L6N .20254812 03 XS -.19592183 08 YS -.13388645 09 Z5 -.58056129 04 XH -.22548650 06 YM .284642300 06 Z7 -.13797500 06 XI -.41082576 08 YT -.38297208 09 Z7 -.13647793 08 RS .14724152 09 YS .30247584 02 RM .3863512 06 GEO -.13777151 02 ALT .57774727 08 L0S .24121388 03 OUT .35000000 02 0T .2400000 03 R.16474169 02 CCL .27338252 03 HCL .17999012 03 TCL .32423100 03
                                                                                                                                                                                                                  0X --74907017 01

V -18187518 02

VE -40880465 04

0XS -2999978 02

0XH --85825494 00

DXT --44351359 01

VM -10184061 01

RAS -25167468 03

SHA --35391055 08
                                                                                                                                                                                                                                                                                                                                           EQUATORIAL COORDINATES
                    HELIOCENTRIC
                                                                                                                                                                                                                                                                                                                                                               D2 -.56021254 O1
AZ .95825815 O2
DZE .15355094 O1
DZT -.11325077 O1
VST .35228410 O2
EEM .1032686 00
STP .70168220 O2
SPN .12301217 O3
      X -21459989 08

R .10755675 09

E .1952383 08

XT -21490192 08

LTE .23221815 02

LPS .12301450 03

HPS .12301450 03

HPS .12301490 03

EPT .11708922 03

SET .37791812 02

SAC .23815338-09

CCE .86617480 02

REP .57701104 08
                                                                     DX -3149080 02 07 -11370283 02

V 39588879 02 PTH -72124065 01

DXE -2999978 02 DYE 355439873 01

DXT -34435114 02 DYE -73471930 01

LOT -10267041 03 RST 1.0157002 09

EPH 37500648 00 EPP 8.070361 02

SEM 13664028 03 EKS 43256699 02

TFS 1.0981217 03 TSP 1.5539313-01

RPM 57842254 08 RPI .39123598 05
                                                                      GCT .23084848 03 SIP .10069395 03 CPT .12031818 03 SIR .11119995 03
                                                                                                                                                            18SYS-JPTRAJ-SPACE 022265
CASE 1
                                                                                                                                                                                                                                                                                                                                         EDUATORIAL COORDINATES
           APHRODICCENTRIC
                                                                                                                                             Z -.24548217 05 DX -.30555658 01 DY -.40230900 01 DZ -.45496177 01
RA .62437123 01 V .67985849 01 PTH .14131968 01 AZ .21265149 03
ALP .14175582 02 DR .16765977 00 DP .99533717-02 ASD .91182247 01
HVG .11394177 03 SIA .10797099 03
       X .30202689 05 Y .33043787 04

R .39123598 05 DEC -.39050781 02

ALT .32923599 05 SHA -.36803285 05

HDE .23698150 03 SVL -.33361545 02

SAL .23815338-09
                                                                                                                                                                           APHRODIOCENTRIC CONIC
EPOCH OF PERICENTER PASSAGE

SMA --10964909 05 ECC .45667502 01
VH .54423057 01 C3 .29618692 02
TA .17226236 01 MTA .10264878 03
ZAE .13776068 03 ZAP .39973169 02
                                                                                                                                                 235605635514202411466611 J.O.* 2438013-32428326 DEC. 14,1962 19 46 58.075 B. 48858736 05 SLR .21771055 06 APD .00000000 00 RCA .391010902 05 C1 .26590420 06 FFP .1728361 03 FF .10080794 03 FFF .10071252 03 EA .13790468 01 MA .49193264 01 FF .1008094 03 FFF .10080994 03 C2AC .71965352 02 DEF .252875750 02 IR .13206244 05 CP .30444630 02
                                                                                                                                                                                                                                                                                        REFERENCED IU EARTH FQUATOR PLANE
DY -.40230900 01 DZ -.45496177 01
HY -.59401770 00 HZ -.65386199 00
PY .10227852 00 PZ -.61006853 00
RY .71369419 00 RZ -.61339908 07
IY .42769483 00 TZ .00000000 00
RAI .22467816 03
RAI .22465815 03
                                                                                                                                                                                                                     ALL VECTORS

DX -.30555658 01

MX -.46862293 00

PX .78571959 00

RX .33778185 00

TX -.90792319 00

DAI -.52164309 02

DAO -.31506458 02
                                                                                                                                                 I -- 24648217 OS
APF - 22221299 O3
WI -- 41899072 OU
OI -- 677250511 OO
BI -- 44800140 OO
SZI -- 7897307 OO
SZI -- 52259466 OO
ETC - 25076850 O3
       X .30202689 05 Y .33043787 04
INC .11477088 03 L4V .20822792 03
WX -44920465 00 WY .80000593 00
QX -44520469 00 QY -59121028 00
BX .86413912 00 GY -59225603 00
SXI -26234762 00 SYI -55445566 00
SXI -60445213 00 SYI -5925835 00
ETE .30228987 03 FTS .29226401 02
         BTO -.33373633 05 BRQ .35684404 05
                                                                                                                                                    8 .48858736 OS THA .13308352 D3 T VECTOR IN EARTH FOUNTOR PLANE
                                                                                                                                                                                                                       ALL VECTORS REFERENCED TO ECLIPTIC PLANE
DX --30555658 01 DY --55009941 01 DZ --25735845 01
DX --46862293 00 PY --85501276 00 MZ --36357392 00
PX --85511969 00 PY --14886698 00 PZ --60040247 00
TX --195205093 00 TY --30374763 00 TZ --00000000 00
OAI --30265111 02 RAI --2521716 03
DAO --13948908 02 RAI --252132657 03
                                                                                                                                                 Z --23928315 05
APF -23754777 03
WZ --70267301 00
QZ --38179508 00
BZ --50222789 00
SZI --50400177 00
SZI --24105658 00
ETC -26213010 03
       X .30202659 05

INC .13464185 03

AX -42946164 00

QX -44520470 00

BX .86419313 00

SXI -26234761 00

SXI -26234761 00

SXI -504045214 00

SYI -82289483 00

SYI -82759980 00

SYI -82759980 00

SYI -8289085 02
                                                                                                                                                                                                                                                                                             T VECTOR IN ECLIPTIC PLANE
                                                                                                                                                    B .48858736 05 THA .14444512 03
         8TC -- 39749463 05 89C -28410495 05
                                                                                                                                                                                                                       ALL VECTORS REFERENCED TO ORBIT PLANE OF TANGET DX -.5065988 01 DY .15147077 01 DZ -.26679470 01 DX -.26679470 01 DX -.26679470 01 DZ -.26679470 01 DX -.26679470 01 DX -.26679470 01 DX -.40900556-01 PY -.83279862 00 PZ -.55205300 00 TX .3063483 00 PX -.18010689-01 RZ -.36721795 00 TX .39544721-01 TY .9993608 00 TZ .00000000 00 DAT -.3044602 0Z RAI .17795300 03 DAU -.15362793 0Z RAU .15579407 03
                                                                                                                                             X .55016583 03 Y -.32310610 05
INC .13722957 03 LAN .13851610 03
WX .44981135 00 WY .50811418 00
QX -.89218354 00 QY .218130321 00
QX .23527304 00 QY .218130321 00
QX 1.32527304 00 GY .30643794-01
SXI -.86157468 00 SYI .30643794-01
SXI -.87947010 00 SYI .30643794-01
SXI -.87947010 00 ETE .30926022 03 ETS .61081146 02
```

8TO --41602410 05 8R3 -25620611 05

8 .48858738 05 THA .14837339 03 T VECTOR IN ORBIT PLANE OF TARGET

```
CASE 1
                                                                                                            IBSYS-JPTRAJ-SPACE 022265
    217727454543 214522350406 617601552025 602611220703 603400724620 235605635567
                                                                                                                                                                                   603442741665
                                                                                                                                                                                                                                                                            VENUS
END
                                                                                                                                                             ITERATION NUMBER
                            U MATRIX FUR MAPPING FURWARD
                                                                                                                                                             ny
                                                                                                                                                                                              ΠZ
                                                                                                                                                                                                                               NV
                                                                                                                                                                                                                                                                AU
  X -39773183 01 -21087971 01 -10280934 01 -40090036-06 -12381001-05 -60017031-06 -00000000 00
Y -42917083 00 -3064667 00 -21520692 00 -14571548-06 -55214985-08 -10965075-06 -00000000 00
Z -15201560 00 -15553731 00 -13895444 00 -55276687-07 -73585203-70 -18114900-06 -00000000 00
DX -13987192 08 -68392037 07 -33544449 07 -16491137 01 -40547289 01 -19595172 01 -00000000 00
0007814130 07 -79537667 07 -19422097 07 -14248229 01 -25076387 01 -14931677 01 -00000000 00
621200700,00000000

x* -.36497311E+08 Y* -.29912202E+08 Z* -.9456733Rt+07 S

px* -.7364330E+01 DY* -.10826458E+02 DZ* -.54166483E+01 S

R01 24347426137221062600114578 24257631520520701436321278 S

2415664720732066222720179 21455747502416167211032078 S

21565661634016260352567478 21463565873516126531143378 S

53403671621510350007741/8 17757373651714452000000078 S

R02 2425763152052070143521278 24252163526520743061224278 S

2414354260122065520153338 21445734313616173105643578 S

54550253377651116421401278 2005070130201455600000078 S

GDPX*14130
                                                                                                                                                                      R03 241566472073206622272011/8
240714243507205560121535/8
21450177072161554051315/8
544550652467511063551222/8
804 214557475024161672110322/8
213443050767160413054706/8
167447242664134104655056/8
51755136273464644233410/8
805 215654616340162603525674/8
214501770732161554051315/8
QDPX+14130
                                                                                 241435426012206550201534/8 $
213443050767160413054706/8 6
212772213025571722636427/8 7
200406075476145440000000/8 8
214453020170614507145504/8 8
106420613604133125331104/8 8
166434734653133525462314/8 8
550774612640515700000000/8 3
21574360771162346376674/8 8
CONDITIONS AFTER FORWARD MAPPING
                                                                                                                                                                  62/09/05 002332.000 TD 62/12/07 000000.000
       STANDARD DEVIATIONS
        ITERATION NUMBER
                          COVARIANCE MATRIX AFTER MAPPING
                                                                                                                                                                                                                                                                 AU
              -21229665 11 .12831745 11 .62849716 10 .29409517 04 .56864511 04 .33107333 04 .294537555-09 .234780317-08 .63871396 00 .51895380 04 .25758895 04 .10125431 04 .226254072-08 .5187398 00 .29409537 04 .2942537 04 .10125431 04 .226254072-08 .5187398 00 .29409537 04 .2942537 04 .10125431 04 .26254072-08 .5187398 00 .29409537 04 .2942537 04 .10125431 04 .26758895 04 .10125431 04 .26254072-08 .5187398 00 .29409537 04 .2942537 04 .10125431 04 .25758895 04 .1265490-02 .24980159-02 .11891137-02 .32488109-14 .61674783-05 .33107333 04 .25555497 04 .10125431 04 .5486325-03 .1891137-02 .32488109-14 .61674783-06 .34810330 00 .35810307-08 .2555497 04 .294254072-08 .12541165-14 .33261903-14 .9999998-18 .00000000 00 .357103015 00 .63871396 00 .51195380 00 .59228753-07 .61674783-06 .40088574-06 .00000000 00 .25000000 02
```

```
CASE 1
                                                                                                                                                               IBSYS-JPTRAJ-SPACE 022265
DOUBLE PRECISION EPHENERIS TAPE - EPHENI
INJECTION CONDITIONS VENUS
                                                                                                                                              23560512344020200000000 J.D.= 243800>-50000000 DEC. 7.1962 00 00 00.000
GEDCENTRIC x0-.36497317 08 Y0-.29912202 08 20-.94567338 07 DXO-.73404306 01 DYO-.10826458 02 DZO-.54166483 01 CARTESIAN ID .00000000 00 GHA .75283983 02 GHO .7
           2 DAYS 14 HHS. 37 MIN. 51.431 SEC.
                                                                                                                                                       235605301517202667117775 J.D.= 2438008.10962304 DEC. 9,1962 14 37 51.431
                    HELIOCENIRIC
                                                                       Y .99263156 08 Z .46385103 08
LAT .25017435 02 L0M .92942495 02
YE .13172989 09 ZE .57121303 08
YT .97856441 03 ZI .44457140 03
L0E .75930985 02 L1T .24385356 02
ESP .15695705 02 SEP .35613715 02
ESP .15754194 02 SMP .35277588 02
ETP .41616356 02 TEP .18591082 01
STE .12978487 03 EST .16053214 02
                                                                                                                                                                                                                  DX -.36494255 02 DY -.58201981 01 V .37104844 02 PFH -.77073117 01 DXE -.29507275 02 DYE .60187012 01 DXT -.35091491 02 DYE .60187012 01 LOT .93387689 02 RST .10762945 09 EPH .79129366-01 EMP .1190247 02 SEM .15551576 03 EMS .24425467 02 TPS .34321412 02 JSP .75039911 00 RPM .51564819 08 RPT .25000001 07
           X -.51022687 07
R .10968493 09
XE .33012537 08
XI -.57926456 07
                                                                                                                                                                                                                                                                                                                                                                 DZ -33262720 DI
AZ -92101534 DZ
DZE -26093421 01
DZI -10045197 01
VST -352209096 02
MEP -16873040 03
ESM -57674939-01
STP -14492818 03
SPN -12888343 03
                                                                             GCT .32905257 03
VEP .14973608 02
                                                                                                                                                  SIP .34179318 02 CPT .72719297 02 SIN .72577202 02 CPE .80195375 02 CPS .10226691 03
                                                                                                                                                                          HELIOCENTRIC CONTC
 EPDCH OF PERICENTER PASSAGE
SMA -12722491 09 ECC -19144538 00
VM -26606642 02 C3 --10431457 04
TA --52176596 02 MTA -18000000 03
                                                                                                                                                      235607605023202607117775 J.D.= 2438037.01350758 JAN. 7,1963 12 19 27.056 0 .12437147 09 SLR .12256195 09 APD .15158153 09 RCA .10286829 09 C1 .40330757 10 IFP .22472956 07 IF .31513507 02 PER .2864619 03 EA -.43934640 02 MA -.36323937 02 IFI .26096230 01
                                                                                                                                                                                                                         ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE
DX --34649255 02 DY --58201981 01 DZ --33262720 01
RX --9880594 00 MY --35811855-01 MZ --3322998-01
PX --81740733 00 PY -52667545 00 PZ -23308269 00
RX --19593890 00 RY -12623585 00 RZ -97245697 00
TX -54159255 00 TY -84064133 00 TZ -00000000 00
       X -.51022667 07 Y .99263156 08

INC .25999881 02 LAN .35798326 03

XX -.14928002-01 WY -.4293464 00

0X -.57575317 00 WY -.73681227 00

BX .57575314 00 BY .73681223 00

DAP .13478611 02 RAP .14720788 03
                                                                                                                                                Z .46385103 08
APF .14656952 03
WZ .90556967 00
QZ -.35442357 00
BZ .35442355 00
         BTQ .11628278 09
                                                                          BRQ --455109/3 08
                                                                                                                                                                                                                                                                                                T VECTOR IN EARTH EQUATOR PLANE
                                                                                                                                                         B .124B7167 09
                                                                                                                                                                                                                    THA .33862552 03
                                                                                                                                                                                                                           ALL VECTORS REFERENCED TO ECLIPTIC PLANE

0X -.36494255 02 DY -.66630846 01 02 -.73627058 00

8X -.99880594 00 HY -.46075798-01 HZ -.16240136-01

PX -.81748730 00 PY .57593041 00 PZ .43166667-02

RX -.35288531-02 RY .24861228-02 RZ -.99999060 00

1X .57593579 00 IY .81794944 00 7Z .00000000 00
                                                                        Y -10952327 09
LAN -33250317 03
WY --28680308-01
QY --51699546 00
BY -81699547 00
                                                                                                                                                 2 -30665980 07
APF -17232766 03
HZ -99947717 00
QZ --32043277-01
BZ -32043278-01
         x -.51022667 07
INC .18528344 01
WX -.14928002-01
QX -.57575309 00
                                                                                                                                                                  IBSYS-JPTRAJ-SPACE 022265
         DAP .24732883 OD RAP .14483480 O3
                                                                                                                                                       8 .12487165 09 THA .35816372 03 T VECTOR IN ECLIPTIC PLANE
         BTC .12480753 09 8RC -.40013344 07
                                                                                                                                                                                                                      ALL VECTORS REFERENCED TO ORDIT PLANE OF TARGET 0X -.15093340 02 DV -.33785279 02 DZ -.27414236 01 PX -.28065739 00 NY ..95702552 00 AZ -.73031394-01 PX -.36657501 00 PY ..92900175 00 PZ -.50776340-01 RZ -.18637381-01 RY -.47232243-01 RZ -.99870991 00 TX -.9120180 00 TY -.36704654 00 TZ .0000000D 00
         X .10522014 09 Y .30951206 08

INC .42377882 01 LAN .20514101 03

UX -.31394614-01 WY .66895417-01

UX -.92985853 00 QY .36397883 00

DAP -.29105203 01 RAP .68466288 02
                                                                                                                                                   Z .12362214 07

APF .22340358 03

WZ .99726596 00

QZ -.53687747-01

BZ .53687754-01
                                                                                                                                                         B .12487163 09 THA .35691845 03 T VECTOR IN ORBIT PLANE OF TARGET
         810 -12469108 09 BRO --67127359 07
           2 DAYS 14 HRS. 37 MIY. 51.431 SEC.
CHANGE OF PHASE OCCURS AT THIS POINT
PROBE IS LEAVING VENUS. SHADDW
                                                                                                                                                 235605301517202667117775 J.D. 2438008-10962304 DEC. 9,1962 14 37 51.431
VENUS IS THE CENTRAL BODY FUR INTEGRATION CORELL EQUATIONS OF MOTION
             2 DAYS 14 HRS. 37 MIN. 51.431 SEC.
                                                                                                                                                              235605301517202667117775 J.D.= 2438008.10962304 OEC. 9,1962 14 37 51.431
                                                                                                                                                                                                                                                                                                                                              EQUATORIAL COORDINATES
                             GEOCENTRIC
                                                                                                                                                                                                                       DX -.69869809 01 DY -.11838899 02 DZ -.59356140 01 V .11973609 02 PTH .68707164 02 AZ .12448867 03 VE .39465785 04 PTE .21920980 00 AZE .26995161 03 VE .36957274 02 UNS -.60187012 01 DZ -.26097421 01 DXH .-85011119 00 UNM .59271622 00 DZM .29599127 00 DXH .295842176 01 UNM .59271622 00 DZM .29599127 00 VM .10777812 01 RT .53046872 08 VT .10472502 02 RS .25593098 03 RSM .-29672880 08 DES -.22812463 02 DEM .11927585 02
                                                                                                                                               Z --10736199 08
RA -22062483 03
LON -20310441 03
ZS --57121302 08
ZM -67759750 05
ZT --12684155 08
RN -35544014 06
LOS -31861057 03
IRR -13951460 02
GCL -61833078 02
                                                                        Y --32466733 08
DEC --12102737 02
LAI --12102737 02
YS --1317295 09
YH -27889500 06
YI --33873448 08
VS -30221679 08
AII -51199998 08
DI -3440000 04
MCL -17997891 03
                x -.38114804 08
         x -38114804 08
q -51206376 08
R -51206376 08
xS -33012737 08
xM -21915100 06
yT -38805183 08
xS -16732764 09
GED -12182759 02
UUT -35000000 02
CCL -27278050 03
                                                                                                                                                                                                                                                                                                                                                EQUATORIAL COORDINATES
                       HELIOCENTRIC
                                                                             Y .99263155 08
LAT .25017435 02
YE .13172989 09
YT .9785841 08
LOE .75930985 02
ESP .15695707 02
MSP .15754197 02
ESP .15754197 02
ESP .12978487 03
                                                                                                                                                  7 -46385103 08

LON -92942495 02

ZE -57121302 08

ZT -44437148 08

LIT -24385356 02

SEP -35413716 02

SMP -35277589 02

TEP -18581082 01

EST -16063214 02
                                                                                                                                                                                                                       DX -.36494254 02

V .37104043 02

DXE -.29507274 02

UXT -.35091491 02

LDT .93387689 02

EP4 .79129368-01

SE4 .15551569 03

TES .34321416 02

RPM .51564819 08
                                                                                                                                                                                                                                                                                             OY -.58201981 01
PTH -.7073120 01
DYE .60187012 01
UYI -.26941887 01
KST .10762945 09
EMP .11190135 02
EMS .24425541 02
TSP .75039911 00
RPT .25000001 07
                  x -.51022667 07
          x -.51022667 07

x -10968493 09

xE -33012537 08

xI -.57926456 07

LTE -22812463 02

EPS -12889057 03

EPI -13652552 03

EPI -13652552 03

SET -34151907 03

SAC -22900140-09

CCE 47219409 07
                                                                              .87219499 02
.51206376 08
```

COSSSO BIAGS-LARIGUEZZES

CASE 1

```
EQUATORIAL COORDINATES
             APHRODICCENTRIC
                                                                                                                                                                                                                                                                                                                                                                   x .69037884 06
R .25000001 07
ALI .24938001 67
HGE .23110942 03
SAC .22980140-09
                                                                                         Y .14067148 07 Z .19479556 07 DX -14027633 01

OEC .51185758 02 RA .63859402 02 V .55222701 01

SWA .14365071 07 ALP .50235550 02 BR -55208152 01

SVL .29805898 02 HYG .11862250 02 SIA .13638342 03
                                                                                                                                                                                                                         APHRODICCENTRIC CONIC
                                                                                                                                                                                       235605634233202077111775 J-D-= 2438013-27236092 (BC. 14-1902 19 01 00-473 8 5.57626458 05 SLR 30918769 06 APD -00000000 00 RCA 47679731 05 EA -25498070 03 MA -13134219 05 EA -25498070 03 MA -13134219 05 EA -25498070 07 DFF -21116094 02 IR -13180782 05 FP -22744401 02 DF -2114094 07 IR -13180782 05 FP -22744401 02
EPOCH OF PERICENTER PASSAGE
SMA -10741169 05 ECC -54575904 01
VM -44986957 01 C3 -30235654 02
TA -19239987 02 M7A -10055002 03
ZAE -13728728 03 ZAP -34431171 02
                                                                                                                                                                                                                                                                                 ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE OX -14427633 01 OY -31260145 01 OZ -43307917 01 NX -946107241 00 NY -154015756 00 NE -222934959 00 PX -90426010 00 PY -22422755 00 PZ -35148785 00 TX -32202514 00 NY -15593710 00 NZ -520944296 00 TX -91218214 00 TY -40933950 00 TZ -00000000 00 DAI -51651508 02 RAI -24593664 03 DAD -440953367 0Z RAD -221919069 03
                                                                                                                                                                                         I .19479556 07
APF .20564155 03
WZ -,58333038 00
QZ -,13224507 00
BZ -,21136701 00
SZI -,41136701 00
SZI -,55544458 00
ETC .25151734 03
          X .69037884 06 Y .14067148 07
INC .12568510 03 LAN .10063479 03
WX -89848358-02 WY .81218591 00
QX -42688758 00 QY -53064471 00
EX .96110951 00
EX .96110951 00
EX .96134882 00
EX .96134882 00
EX .30576850 03
EX .41834766 02
                                                                                                                                                                                                 B .57628458 05
                                                                                                                                                                                                                                                                                    THA _16008230 03 T VECTOR IN EARTH EQUATOR PLANE
           BTQ -.54181293 05 BRQ .19632282 05
                                                                                                                                                                                                                                                                                  ALL VECTURS REFERENCED TO ECLIPTIC PLANE
DX - 14027613 01 DY - 45907088 01 DZ - 27297030 01
RX - 94012233 00 PY - 232560024 00 PZ - 14912355 00
PX - 99426026 00 PY - 36224872 00 PZ - 22603090 00
RX - 14442024 00 PY - 36224872 00 PZ - 22603090 00
DX - 15442024 00 TY - 272767729 00 RZ - 36724288 00
TX - 95536871 00 TY - 27276749 00 TZ - 00000000 00
DAD - 26428634 0Z RAI - 25301253 03
DAD - 24428037 0Z RAI - 25304092 03
                                                                                            Y .20655583 07
LAN .18100463 03
WY .51308132 00
QY -.77615397 00
8Y -.21353375 00
SYI -.83135495 00
                                                                                                                                                                                         Z -12275367 07
APF -20613378 03
WZ -85029328 00
02 -46069070 00
8L -13778981 00
SZI -49431489 00
SZI -41148313 00
EIG -26231466 03
 x .69037884 06
- 1kC .14912548 03
wx -.89992301-02
ux -.42082763 00
BX .96716973 00
SXI -.25397181 00
                                                                                                                                                                                                   B .57628461 DS THA .17087959 D3 T VECTOR IN ECLIPTIC PLANE
           BTC -.56899888 05 BRC .91346706 04
                                                                                                                                                                                                                                                                                  All MECTURS REFERENCED TO URBIT PLANE OF TARGET DX -47921297 01 PY -11075317 PD DY -27414250 01 PX -95202354-03 PY -09590256 00 PX -9243392-01 PX -97591347 80 PY -12844413 DD PY -97591347 80 PY -12844413 DD PY -97591347 80 PY -12844413 DD PY -97591347 80 PY -12845808 DD PY -12845917-01                                                                                                                                                                                          Z .123622D2 07
APF .19959053 03
WZ -.86447056 00
0Z -.47157019 00
6Z -.76916122-01
SZI -.49643457 00
5ZO -.47467012 00
ETC .27321265 03
          X .21701734 07 Y -.11003277 06

JNC .14932230 03 LAN .99027389 02

WX .494.3508 00 WY .78871185-01

OX -.5559587 00 OY .20335195 00

SXI -.86781774 00 SYI .21093411-01

SXI -.88781774 00 SYI .21093411-01

SXI -.8170372 00 SYI .21072508 00

ETE .31229757 03 ETS .71005016 02
          6TO -.573898T3 05 8R0 .52390984 04 8 .57628455 05 THA .17478394 03 T VECTUR IN ORBIT PLAYE OF TARGEF
             7 DAYS 19 HRS. 46 MIM. 58.744 SEC. 235605635514202537267173 1.0.= 2436013.32429101 GEC. 14,1962 19 46 58.745
  CASE 1
                                                                                                                                                                                                            185YS-JPTRAJ-SPACE 022265
                                                                                                                                                                                                                                                                                                                                                                                                                                          EQUATURIAL COORDINATES
                                   GEOCENTRIC
                                                                                                                                                                                                                                                                                                                                                                          OY -.14909948 02 02 -.72400025 01
PTH .68392656.02 AZ .11983742 03
PTE .23682486 00 AZE .26995322 03
DYS -.35494921 01 025 -.15359235 0.
DYM -.52210920 00 02M -.13040295 00
UYT -.1080379 02 02T .-2667451 01
RT .57791036 08 VI .12057849 02
RAM .12813447 03 00M .1084436 03
DES -.23221698 02 0EM .19341637 02
          X -.41051080 08

R .57778186 08

R .57778186 08

X5 -.1957553 08

XM -.22533900 06

KT -.41001011 08

RS -.14724155 09

GEU -.13776580 02

DUI .35000000 02

CCL .27338213 03
                                                                                                                                                                                                                                                                             DX -_14625149 01
V .18177275 02
VE .40B78331 04
DXS .29999834 02
DXM -_85852918 00
DXT -_4356744 01
VM .10184388 01
SAA -_35388552 08
                                                                                            Y -.38291332 GB
DEC -.11686849 G2
LAT -.11686849 G2
YS -.13386824 G9
YH -.28651450 G6
YT -.34295332 GB
VS -30247574 G2
ALT -57771808 GB
DT -.24000000 G3
MCL .17990010 G3
                                                                                                                                                                                        Z -.13671194 08
RA .22300790 03
LON .203226706 03
ZS -.58055864 08
ZM .12794737 06
ZT -.13647333 08
RM .38631431 08
LOS .24193162 03
DR .16896480 02
ICL .32593707 03
                                                                                                                                                                                                                                                                                                                                                                                                                                          EQUATORIAL COORDINATES
                             HELIECENTRIC
                          -216326 08
-10755761 09
-1095755 08
-21484258 08
-21284258 08
-23221698 02
-12302091 03
-11479603 03
-11479603 03
-37776585 02
-23814957-04
-26617867 02
-57778186 06
                                                                                                                                                                                                                                                                                  UX -.37462349 D2
V .39561692 D2
DXE -.29999834 D2
LDT .1026686 D2
LDT .1026686 D3
EPM .37807118 D0
SEM .1366407 D3
TPS .10851672 D3
RPM .57839499 D8
                                                                                                                                                                                                                                                                                                                                                                           DY -.11355006 02 00 -.57040790 01
PTH -.72352603 01 AZ .95856866 02
DYE .35449421 01 DCE .15359235 01
NST ..77354057 01 DCF ..1315408 01
RST .310757004 09 VSJ .35228603 02
EMS .67232956 02 ESM .10279207 00
EMS .473282950 02 ESM .10279207 00
EMS .732840695-01 ESM .10279207 00
RPT .39912531 05 SPM .12301464 03
                                                                                               Y .95594911 08
LAT -24372011 02
YE .13388624 09
YT .75550911 08
LBE .81672483 02
ESP .19209272 02
MSP .19106513 02
ESP .6160999 02
SIE .17297786 03
                                                                                                                                                                                         Z -44384671 D8
LDN -10264881 03
ZE -58055864 08
ZT -44408531 08
LT -24430850 02
SEP -37769755 02
SEP -37769755 02
SEP -337494052 02
IEP -33542591-02
EST -19225546 02
                                                                                                  GCT -23205574 03
YEP -18177275 02
                                                                                                                                                                                          SIP .99394888 02 CPT .11992330 03 Sfn .11080247 03 CPE .80314970 02 CPS .1021893 03
                                                                                                                                                                                                                                                                                                                                                                                                                                      EQUATORIAL COORDINATES
               APHRODICENIES
                                                                                                                                                                                                                                                                                                                                                                            OY -.40195693 01 02 -.45723873 01
PTH .50229894-U6 AZ .21192407 03
OP .99597208-02 ASD .91208270 01
                                                                                                Y .40003297 04
DFC --37593313 02
SHA --37083647 05
SVL --32543115 02
                                                                                                                                                                                                                          APHRODIGCENTRIC CONIC
                                                                                                                                                                                         ALL VECTORS REFERENCED TO EARTH EQUATUR PLANE
DX -.0268405 01 DY -.40195693 01 DZ -.45723873 01
RX -.4451978 00 WY -.59120630 00 MZ -.57251580 00
PX .78573214 00 PY .10227745 00 PZ -.61005268 00
RX .39778178 00 WY .71394041 00 WZ -.6139481 00
IX -.90392327 00 TY .42759463 00 IZ .00000000 00
DAI -.52164327 02 MAI .24457865 03
DAI -.52156156 02 MAI .24457865 03
                                                                                               Y .40003297 04

L4N .20822719 03

WY .8000093 03

DY -597270628 00

RY .22924561 00

SYI -.55946547 03

SYB -.59925466 00

ETS .29230405 02
                                                                                                                                                                                         2 -.23860705 05
APF .22221180 03
MI -.41899652 00
02 -.6.7251547 00
82 -.44799561 00
S11 -.78977326 00
S20 -.52261993 00
ETC .25076851 03
           x .30731973 05
INC .11477124 03
MX -.42945022 00
OX -.44519376 00
BX .86414488 00
SXI -.26234739 00
SXD -.60643400 00
ETE .30229105 03
```

CASE 1

```
BTQ --33376521 05 88Q .35686536 05
                                                                                                                                                                            8 .48862267 05 THA .13308429 03
                                                                                                                                                                                                                                                                                                                                                      I VECTOR IN EARTH EQUATOR PLANE
                                                                                                                                                                                                                                                                  ALL VECTORS REFERENCED TO ECLIPTIC PLANE
DX -.3026405 01 DY -.55068224 01 DZ -.25
PX -.4451937 00 PY -.8099544 00 NZ -.38
PX .78571216 00 PY -.1488166 00 PZ -.06
PX .15308911 00 RY -.4801824 00 FZ -.08
TX -.95275260 00 TY .30374742 00 FZ .00
DAG -.3056128 02 RAG .253132775 03
DAG -.13950363 02 RAG .23132775 03
    X .30731973 05
INC .13464238 03
WX -.42945021 00
QX -.44519377 00
8X .86414488 00
SXI -.26234738 00
SXO -.60643401 00
EFE .31365265 03
                                                                                      Y --58223563 04
LAN .21712663 03
WY .56728648 00
QY -800955446 00
BY .32007629-01
SYI --82289474 00
                                                                                                                                                                      L -.23482676 05
APF .23754634 03
WZ -.70267956 00
QZ -.38180655 00
BZ -.50221846 00
SZO -.24108124 00
ETC .26213010 03
                                                                                                                                                                                                                                                                                                                                                                                                                                             C PLANE
DZ -.25958754 01
MZ -.38180656 00
PZ -.60038753 00
RZ -.86370243 00
TZ .00000000 00
                                                                                         SYO -.75770547 00
ETS .40592005 02
      BTC -.39752716 05
                                                                                         BRC .28412022 05
                                                                                                                                                                            8 .48862270 05
                                                                                                                                                                                                                                                                    THA .14444588 03 I VECTOR IN ECLIPTIC PLANE
                                                                                                                                                                                                                                                                   ALL VECTORS REFERENCED TO DRBIT PLANE OF FARCET DX --60658784 01 DY -14841501 01 02 --26884647 NX --809218155 00 MY -21829178 00 MZ -399824647 PX -40908683-01 PY --83280857 00 PZ --55204733 RX -50638509 00 RY -18010522-01 RZ --8621733 DX -35544653-01 TY -99936809 00 FZ -00000000 DX -3084619 02 RX -17579524 03 UAO --15364189 02 RAD -15579524 03
                                                                                    PLANE OF TARGET

DZ -.26884647 01

HZ -.39542477 00

PZ -.55204743 00

RZ -.86211930 00

TZ .00000000 00
     x .16000420 04
INC .13723007 03
WX .44982056 00
0X -89218155 00
       BX .23526740 00

SXI -.86157454 00

SXO -.87948921 00

EYE .30926123 03
      BTG --41605758 05 BRD -25621909 05 B -48862269 05 THA -14837416 03 T VECTOR IN ORBIT PLANE OF TANGET
         7 DAYS 19 HRS. 49 HIM. 51.431 SEC. 235605635567202667117775 J.D. 2438013.32628971 DEC. 14,1967 19 49 51.431
                               GEOCENTRIC
      X --41052371 08

R -57781104 08

A -57781103 08

XS --19592373 08

XH --22546700 06

XI --41082577 08

RS -14724152 09

GEO --13177152 02

RO --335000000 02
                                                                                                                                                                                                                                                                 DX -.74905246 01 DY -.14914115 02 DZ -.72176794 01
V .18183384 02 PTH .68464354 02 AZ .11972300 03
DX .40880464 04 PTE .23705679 00 AZE .26995361 03
DX .2999978 02 UXS -.35459652 01 DZS -.15355085 01
DXT -.44851348 01 DYT -.10991182 02 DZT -.26806179 01
VM .10184061 01 RT .57798935 08 VT .12058465 02
DXH -.35391056 08 UES -.23221815 02 DEM .19337545 02
                                                                                        Y -,38293907 08
DEC -,13687418 02
LAT -,13687418 02
YS -,1338685 09
YM -,28642250 06
YI -,38297212 08
VS -,30247584 02
ALT -,57774727 08
DI -,24000000 03
MCL -,17998012 03
                                                                                                                                                                          Z -.13672442 08
RA .22300892 03
LON .20254657 03
ZS -.58056130 08
ZM .12792487 06
ZI -.13647744 08
RM .38612500 06
LOS .24121233 03
DR .169133991 02
ICL .32423443 03
         0U1 .35000000 02
CCL .27338252 03
                      HEL 10CENTRIC
                                                                                                                                                                                                                                                                                                                                                                                                                   EQUATORIAL COURDINATES
                                                                                                                                                                                                                                                                                                                                                   DY --11370190 02
PTH --72123937 01
DYE -355439852 01
DYT --7371948 01
RST -10757002 09
EMP -80703716 02
EMS -43256800 02
TSP -18504685-01
RPT -39126984 05
       X -21459998 08

R .10755675 09

XE .19592273 03

XI -21490205 08

LIE .23221815 02

EPS .12301850 03

HPS .12339618 03

EPT .11709311 03

SET .37797815 02

SAC .23815338-09

CC+ .86617470 DZ
                                                                                                                                                                         1 .44383687 08
LDM -10265276 03
2E .58056130 08
LTT .44408336 03
LTT .24302956 02
SEP .3777065 02
SMP .37495777 02
LEP .34970568-01
EST .19226759 02
                                                                                                                                                                                                                                                                 DX -.37490503 02
v .39585691 02
DXE -.2999978 02
DXI -.34435113 02
LDI .10267042 03
EPM .37807118 00
SEM .13654018 03
JPS .10980964 03
RPM .57842254 08
                                                                                        Y .95592947 08
LAT .24371663 02
YE .13388085 09
YT .95589642 08
LOE .81676042 02
ESP .19210845 02
MSP .19108037 02
ETP .52872362 02
STE .12297542 03
                                                                                                                                                                                                                                                                                                                                                                                                                                            --56821710 01
42 -95825938 02
02E -15355085 01
02T --11325095 01
VST -35228410
                                                                                            GCT .23085190 03 SIP .10069221 03 CPT .12031658 03 VEP .18183384 02 CPE .80314978 02 CPS .10271903 03
                                                                                                                                                                                                                                                                                                                                                   SIN .11119915 DJ
CASE L
                                                                                                                                                                                                 185YS-JPIRAJ-SPACE 022265
                                                                                                                                                                                                                                                                                                                                                                                                                     EQUATORIAL COBRDINATES
             APHRODI OCENTRIC
          X .3020808 05 Y .33059017 04
R .39126984 05 DFC -.39046991 02
ALT .32926984 05 SMA -.36807055 05
HGE .23698150 03 SVL -.33359017 02
SAC .23815338-09
                                                                                                                                                                               Z -.24648340 05 UX -.30553899 01 UY -.40229929 01 02 -.45496615 01 RA .62457221 01 V .6796777 01 PTH .14106330 01 AZ .21265003 03 ALP .1417916 02 DR .16736299 00 UP .99523646-02 450 .91174290 01 HNG .11393792 03 SIA .10797568 03
                                                                                                                                                                                                                APHRUDIOCENTRIC CUNIC
                                                                                                                                                                               235605635514207537325475 J.D.= 2438013.32429103 DEC. 14,1967 19 46 58.745 8 .44862266 05 SLR .21774185 06 APU .00000000 00 RCA .39112531 05 C1 .26552331 06 IFP .17268546 03 IF .78242908 01 IFF .7788874 01 EA .13765490 01 MA .49108403 01 IF .78262896 01 IR .1320628 05 GP .30444012 02 GP
 EPOCH OF PERICENTER PASSAGE
SMA -.10964917 05 ECC .45670611 01
VH .54423037 01 C3 .29618670 02
TA .17194770 01 PTA .10264791 03
Z4E .13776067 03 Z4P .39973176 02
                                                                                                                                                                                                                                                                      ALL VECTORS REFERENCED TO GAATH POWATOR PLANE
DX - 30553899 01 DY - 40729029 01 DZ - 45496615 01
PX - 46857002 00 HY - 59409080 00 MZ - 66390717 00
PX - 78573211 00 PY - 10227740 00 PZ - 61005274 00
PX - 33578181 00 PX - 7138943 00 PZ - 61339499 00
PX - 90392326 00 TY - 4276947 00 TZ - 00000000 00
DAU - 52164319 02 RAU - 242647864 03
DAU - 31508153 02 RAU - 22465883 03
          X .30206H0H 05
INC .11477125 03
MX --52945024 00
MX -445193H1 00
HX .864144H4 00
SXI -26234747 00
SXD -66643400 00
ETE .30228987 03
                                                                                       Y .33059017 U4
LAN .20822719 03
WY .80000898 00
QY -59120634 00
BY .22922661 03
SYI -55446555 00
SYO -59925471 00
LTS .29226412 02
                                                                                                                                                                              L -.24648340 05
APF -2222180 03
AZ -.4189966 00
QZ -.67251579 00
BZ -.44739571 00
SZI -.787977318 00
SZI -.52261987 00
ETC .25076851 03
                                                                                                                                                                                                                                                                      THA _13308428 03 I VECTOR IN EARTH EGJATOR PLANE
           RTO -. 33376518 05
                                                                                            B90 .35686538 05
                                                                                                                                                                                       # .48862266 G>
                                                                                                                                                                                                                                                                    Z -.23929034 05
APF .23154635 03
WZ -.31800645 00
UZ -.38180645 00
UZ -.50221853 00
SZI -.50400192 00
SZI -.24108117 00
ETC .26213010 03
                                                                                            Y -.61728103 04
LAW .21712864 03
MY .66728643 03
QY -.80995450 00
BY .32097534-01
SYI -.82289480 00
SYD -.75770550 00
LTS .405#8012 02
                                                                                                                                                                                                                                                                                                                                                                                                                                                   C PLANE
DZ -.25736633 01
MZ -.36361931 00
MZ -.60038754 00
RZ -.86370252 00
12 .00000000 00
          X .30206808 05
INC .13464238 03
NX -.42745024 00
0X -.44519382 00
BX .86414484 00
SXI -.26234749 00
SXD -.60643402 00
E1E .31365147 03
                                                                                                                                                                                      B .48862266 US THA .14444588 03 T VECTUR IN ECLIPTIC PLANE
           BIC -. 39752710 05 BRC -28412022 05
                                                                                                                                                                                                                                                                     ALL VECTURS REFERENCED TU ORBIT PLANE OF TARGET BX -.60668877 01 DY .15165495 01 DZ -.26680164 01 HX -.889407156 00 HY .261818280 00 HZ -.318848185 00 PX .4098606-01 PY -.83280366 00 PZ -.55204755 00 TX .50084949 00 HY -.1010642-01 HZ -.862111938 00 TX .50084940 01 TY -.1010642-01 TZ .3554620-01 TY .99918408 00 TZ .00000000 00 01 -.3044610 02 HAI .17945101 03 000 -.15364182 02 RAU .15579524 03
                                                                                                                                                                                  Z -.22054476 05
APF .23438645 03
HZ -.73408640 00
0Z -.31542464 00
HZ -.45206972 00
5ZI -.50670516 00
5ZO -.26475338 00
          X .55244946 03 Y -.32314327 05

INC .13723007 03 LAN .13851527 03

MX .4492048 00 MY .50870296 00

0X -89218161 00 YY .71829185 00

BX .235267310 00 GY -.80939558 00

SXI -.86157462 00 SYI .30643707-01

SXI -.8494925 00 SYI .30643707-01

SXI -.30926021 03 EIS .41081140 02
                                                                                                                                                                                   ETC .27339348 03
```

BID -.41605749 05 880 .25621916 05

8 .48862266 05 THA .14837414 0) I VECTOR IN DRUIT PLAYE OF TARGET

CASE	i 1			185YS-JPTRAJ-S	PACE 022265			7
63	12427135666	631706661666	630437201056	603730124337	604531616254	603532066024		EARTH
21	7727515530	21174200096	617601553043	17448304640	407400704475	0000000000000		INITIAL
••		21-022711174	417001553043	0020[1203236	603400721472 235605635567	603442743230 202667117775		ENO AEUN2
	υ,	MATRIX FOR MAPPIN	G FORWARD		ITERATION	REMUM	o	
	x	Y	Z	ÐX	DY	DŁ	NY	AU
X Z Z X Y Y Z V V V V V	.10965738 .26546167 .1489051 .75527639 .16940996 .90707747 -11663469 .10846571	00 .10270700 0 00 .18058750 0 06 .15126123 0 06 .67661268 0	1 .15926470 00 0 .92539933 00 6 .61127321 05 6 .87107397 05 6 .62498159 06 111436336 T1	-38408640-04 46092930-07 -27719612 02 -25112386 02 10337408 01 11459470 07	.29859219-04 - -12657212-04 - 31993660-05 - .20412752 02 - -75215715 01 - .21220986 01 - -46688873 06 - .53912774-05 -	40772528-05 22687172-04 94570649 01 23777418 01 14140570 02 27653503 06	.00000000 00 .00000000 00 .00000000 00 .00000000	.00000000 00 .00000000 00 .00000000 00 .00000000
	COVA	RIANCE PATRIX	AF IMPACT		ETERATION	NUMBER	۵	
	x	Y	Z	DX	DY	DZ	WA	AU
X Y	.47175286 .38625947	11 -34565694 11	1 -17722058 11	-20495190 07	-73161593 06 - -54127742 06 -	88647863 06 - 74744181 04 -	15561391-07	-34610615 D1 -31844926 D1
Z DX	.20113070	11 .17722058 11 07 .20495190 01	1 -10110164 12 7 -10488527 07	-10488527 07 -12505391 03	.28992393 06 -	41058659 06 - 45555512 02 -	17080953-07 13852334-11	-22685311 01 -55924926-03
NA DT DA	08647863	06 .54127742 06 06 - 74744181 06 -07 - 18174269-01	5 -28992393 06 541058659 06	.35761035 02 45555512 02	_13487717 A2 _	- 1370Ennt na .	- 48COS47E-17	-13441655-03 19567239-03
ĀÜ	-34610615	01 431844926 01	22655311 01	-55924926-03	-13441655-03 -	•43232970-12 •19567239-03	.00000000 00	.00000000 00 .25000000 02
	U PR	GDUCT MATRIX DF S	STEP HAPPING		ITERATION	чинаєя	۵	
	×	¥	Ł	ОX	OY	DZ	RV	ΑU
X	-56437986			.27753481-03	-95502075-04	.10171062-03	-000000000 00	-00000000 00
Y Z DX	.70343361 .22739587 .19764677	00 .21658658 QC	19479862 00	+32273659-04 +15866207-04	.12672869-04 - .21757187-05	,14935375-04 ,23402609-05	-000000000 00	-00000000 00
70 30	.80540421 .40302329	D7 .11420462 D6	-46543167 D7	.549DB4B3 03	.34895230 03 - .27889194 02 - .48287167 02 -	.16720412 03	-00000000 00	.00000000000
HV AU	-+15561391 -13844246	11 -418174269 11	17080953 11	13852334 07	48508476 06 -53766621-05 -	.43232971 06	.00000000 00 .10000000 01	.00000000 00 .00000000 00 .10000000 01
	ENCO	UNIER PARAMETERS	AND STATISTICS	62/12/14 194	658.745			•
	6	.48R63381 05	SHAA .1	13027722 06				
	8.40	£25623516 O5	SHIA .4	46289093 05				
	8-10	41606075 05	•	31 379607 05				
	8.80	.28412022 05		(4581931 06				
	6. FC	39752716 05		17077879 06				
	THEI	.10077253 03 20 37570843 02		91950071 05   }} 7444 06				
			VEC 51 11	11717444 08				
	N HA	TR1X						
		6-RD	B. TO	TL	C3	S.T	s s	iRS
	8.KU	.84548154 10	82051087 10	.26191344	1059891937	05 .225383	12 06 -1164	6524 06
	6.10	82051090 10	-12608495 [1	30171756	.647 <del>4</del> 3844	05 284562	52 061459	10043 06
	ŤL	.26191345 10	30171755 10	.98467973 (	7920924726	05 .805063	90 05 -4176	2426 05
	C3	59891386 05	-64743775 05					1404 00
	5.15	-22578313 06	28456251 06					0356 01
	S.AS	.11646524 06	14590043 06	-41782426 (	1590371227	00 .363403	58 OL .1875	0305 01
	NORHA	LIZED H MAIREX						
		B.RD	B-10	ŤL	<b>C</b> 3	5.1	, ,	.RS
	8.20	-10000000 01	78846768 00					9722 00
	B. 10	78846771 00	•9999999 00					6186 00
	71	.90773323 QQ	84958051 00	-10000000 C	198661894	00 .9666778	32 00 .9721	8390 00
	C3	96348663 00	-84621952 00	98662093 0	.10000000	019721094	69 009760	3794 00
	S_1S	-92356665 00	94738875 00	.96667784 0	097211093	00 .1000000	10 Ol -9997	4760 00

S.KS .97479726 00 -.94126187 00 .91218370 00 -.97603603 00 .99974765 00 .9999999 00

```
XISTAN GOGVAG
                  9.R0
                                 B_ to
                                               TL
                                                                             5.15
               -19432538 01 -_30687803 01 .71810839 00 -_15502912-04 .67876250-04
                                                                                         -34794829-04
               .12661832 00 -.37872949 00 .99066247-01 -.15491778-05
                                                                          .79940872-05
                                                                                         .41622495-05
        Z
               .34317390 00 -.17255082 00 -.66687256-02
                                                          .86614341-06
                                                                          .37648641-05
                                                                                         -17358850-05
        οx
                -64180771 07 --11189817 08
                                            -23965677 07 -.51176269 02
                                                                          .23588366 03
                                                                                         .12053721 03
        ۵¥
               -65818714 07 --15729188 07 -16748565 07 --41520920 02 -11365567 03
                                                                                        460119785 D2
               -.18816308 06 -.63123703 05 .11391386 07 -.15014267 02 .43341703 02 .24794701 02
        DZ
               -.56560455-01 -.96096551-01 -.26623533-01 .14224668-03 .50777522-06 .40448822-06
        DM/DD MATRIX
                  B.RO
                                 8-10
                                               TL
                                                              Ç3
                                                                             5.15
                                                                                            S.RS
                                            .55341266-01
                -21496986 00 --71465895 00
                                                          .33640446-03
                                                                          -10118067-04
                                                                                         -62313123-05
                -60157931 00
                              .64188397 00
                                             .14796535 00
                                                           --11025433-03
                                                                          -20658079-05
                                                                                         -12722483-05
        2
               --91472661 00 --98744307-01
                                            .97671857-01 +.23437026-03 -.56003876+05 -.51622475-06
        òχ
               -.11462816 04 -.42096133 04 -.67860414 03 -.71121789 01 -.20354937-01 -.12535793-01
                .42021899 04 -.25239673 04 .46801974 03 -.10266800 02
        D.F
               .82683492 04 -.68900916 03 .67421384 03 -.53769287 01 -.57946112-01
                                                                                        -42589657-01
               -.33843363-01 -.14002395 00 -17583676-D2 -91201498-04 -18249393-05
        ИU
                                                                                         .11239068-05
        N MATRIX FOR MANEUVER CALCULATIONS
                   B_RC
                              B.TC
                                                TL
        B.AC
             .95964167 IO -.84255963 IO .32634384 OS
        8.70 -.84255986 10 -11666853 11 -.32763030 05
TL
0000 141631
0000 14163
9000 14163
              .32634384 05 -.32763030 05 .13190690 00 OFFELINE CONTROL TERMINATE JOU END DATA
                                                                             127
                                                                              10
```

```
PACE HEADING
(HOON CHECK CASE - PREDICTIONS)
EPOCH
640702910,2758000
GEOCHNIC PUSITION AND VELOCITY AT EPDCM
X=.15667453E6 Y=.6304161565 Z=.8077720464
DX=.14342616E1 DY=.97258996E0 DZ=.28116199E0
OTHER OPTIONS AND CONSTANTS
SYMPRO=2030516.
IPREQ=7900300.
POINTING TIMES, SAMPLE RATE, CJUNT TIMES
JETGLZ=640702911,0,640703011,0,1800-,60.
IRANSHITTER ID TABLE
Z,2036516.
OFFLINE CONTROL
KEY18) KEY113)
END DATA
  0000 141751
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
0000 14175
                                                                                                                                                                                                                                                                                                                                                                                                               101
                                                                                                                                                                                                                                                                                                                                                                                                               102
                                                                                                                                                                                                                                                                                                                                                                                                              126
                                                                                                                                                                                                                                                                                                                                                                                                               127
                                                                                                                                                                                                                                                                                                                                                                                                                    60
                                                                                                                                                                                                                   1BSYS-JPTRAJ-SPACE 022265
 CASE 1
 DOUBLE PRECISION EPHEMERIS TAPE - EPHEMI
         GME .39860063 06 J .16234500-02 G .66709998-19 A .88781796 29 GMH .49026293 04 GWS .13271411 12 GEGM .39860320 08 MGM .49027779 04
                                                                                                                                                                                          H --5749999-05 D -7874999-05 RE -63781650 04 REM -63783112 04
B .88800194 29 C -88836978 29 DME -41780741-02 AU .14959850 09
GMY .32476627 06 GMA -42977368 05 GMC .37918700 08 GMJ .12670935 09
JA .2920000-02 MA .00000000 00 DA .00000000 00 RA .34170000 04
  INJECTION CONDITIONS MOON
                                                                                                                                                                                                           235666506353202400000000 J.D.= 2438605.93608795 JULY 29,1954 10 Z7 58.000
 GEUCENTRIC X0 .15667453 06 V0 .63041615 05 Z0 .80777203 04 DX0 .14342616 01 DY0 .97256996 00 DZ0 .28116199 00 CARTESIAN ID .37678000 05 GHA .10409373 03 GHO .30667227 03 DATE OF RUN 022265A 14185 EARIH IS THE CENTRAL BODY FOR INTEGRATION COWELL EQUATIONS OF MOTION
                                                                                                                                                                                                         235666506353202400000000 J.D.= 2438605.93608796 JULY 29,1964 10 27 58.000
               O DAYS O HRS. O MIN. 0.000 SEC.
                                   GEOCENTRIC
                                                                                                                                                                                                                                                                                                                                                                                                                                                          EQUATORIAL COORDINATES
                                                                                                                                                                                                                                                                                       DX .14342615 01 DY .97256992 00 02 .28116198 00 V .17555769 01 PTH .76231923 02 AZ .61612202 02 VE .12070910 02 PTE .81207508 01 AZ .227095862 03 DXS -2.52515999 02 DYS -1.0077681 02 DZS -6.6971998 01 DXT -834399888-01 DYT .93230192 00 DZT .40985490 00 VM .10218268 01 PTY .93230192 00 DZT .40985490 00 VM .10218268 01 PTY .93230192 00 DZT .40985490 00 VM .10218268 01 PTY .93230192 00 DZT .40985490 DZT .4098549
            X .15667452 05
R .16907513 06
R .16907512 06
X5 -18949372 08
XM .38246410 06
XI .38246410 06
GED .27570333 01
DUI .3500000 02
CCL .25840724 03
                                                                                               Y .63041612 05
DEC .27384004 01
LAT .27384004 01
YS .11227336 09
YM .27455507 05
YT .27455507 05
YS .29327501 02
ALT .16269697 06
DT .12000000 03
MCL .11052996 00
                                                                                                                                                                                           Z -80777202 04
RA -21918529 02
LON -27782479 03
ZS -48886598 08
ZM -26012551 05
ZT -26012551 05
RM -38432968 06
LOS -24606715 02
DR -17051340 01
TCL -11052996 00
                                                                                                                                                                                                                                               GEOCENTRIC CONIC
                                                                                                                                                                                                       235666450062202621300000 J.D.= 2438605-21642518 JULY 28,1964 17 11 39-136
B .55279828 05 SLR .12519505 05 APO .48183367 06 RCA .63421468 04
C1 .70641933 05 IFP .62178864 05 IF -17271907 02 PER .20002258 05
EA .71607988 02 MA .18651553 02 C3J -20370820 01 IFI .00000000 00
 EPUCH OF PERICENTER PASSAGE
SMA .24408791 06 ECC .97401695 00
VH .14661061 00 C3 -.16330208 01
TA .16192599 03 MTA .18000000 03
                                                                                                                                                                                                                                                                                           ALL VECTORS REFERENCED TO EARTH EQUATOR PLAME

DX .14342615 01 DY .97255992 00 DZ .28116198 00

AX .34398640 00 MY .80607909 00 MZ .47795833 00

PX .-77265498 00 PY .60455107 00 PZ .-19370658 00

RX .15255788 00 RY .11936638 00 RZ .98105948 00

IX .61622264 0D IY .78757200 00 TZ .00900000 00
         X .15667452 06 Y .63041612 05 INC .28707658 02 LAV .16908121 02 OX .161926389 00 OX .61926389 00 OX .65062083 00 OX .61926466 00 OX .65062083 00 OX .61926466 00 OX .65062083 00 OX .61926466 00 OX .61926469 03 OX .61926469 
                                                                                                                                                                                              1 .80777202 04

APF .20378273 03

WZ .87708195 00

QZ -.43955031 00

BZ .43955042 00
         8TQ .49421011 05 8RQ -.24767379 05
                                                                                                                                                                                                        8 .55279828 05 THA .33338222 03 T VECTOR IN EARTH EQUATOR PLANE
CASE 1
                                                                                                                                                                                                                   IBSYS-JPTRAJ-SPACE 022265
                         HEL10CENTRIC
                                                                                                                                                                                                                                                                                                                                                                                                                                                           EQUATORIAL COORDINATES
         DX -24950250 02 DY -17050251 02 DZ -72531608 01 V -31077876 02 PFH --21988085 00 AZ -75813410 02 DXE -23515989 02 DYE -16077681 02 DZE -69719988 01 DXT -23452549 02 DYT -1700998 02 DZT -73818558 01 DXT -1700988 02 DXT -73818558 03 EPH -16723361 03 EMP -13173981 02 EMP -18992403 02 SEM -12393574 03 EMS -55944139 02 ESM -12012787 00
             1 DAYS O HRS. 32 MIN. 2.000 SEC.
                                                                                                                                                                                                        235666561454202000000000 J.D.= 2438606.95833333 JULY 30,1964 11 00 00.000
                                   GEOCENTRIC
                                                                                                                                                                                                                                                                                                                                                                                                                                                        EQUATORIAL COORDINATES
        X .25084050 06
R .28601301 06
R .28601300 06
XS -92013117 08
XM .36449540 06
XI .36449540 06
RS .15185954 09
GED .61270553 01
DUI .3500000 02
CCL .25920166 03
                                                                                                                                                                                                                                                                                             OX -81614400 00 DY .67319618 00 DZ .22454298 00 V .10819020 01 PTH .77274393 02 AZ .61615284 02 DX .20556600 02 PTE .29427204 01 AZE .27031620 03 DXS -.25214413 02 DYS -.16451548 02 DZ -.71333353 01 DXH -.32284243 00 DYN .88717703 00 DXH .41420339 00 DXH .91420339 00 DYN .88717703 00 DXI .41420339 00 DXH .10309582 01 RT .38035037 06 VI .10309582 01 RS .12969836 03 RAH .16532524 02 DXH .2542290 03 SHA .28251917 06 DES .18451435 02 DEM .15891664 01
                                                                                             Y .13402614 0b
DEC .60858957 01
LAT .60858957 01
YS .11083678 09
YM .10819357 06
YT .10819357 06
YT .10819357 06
OS .27963504 06
OS .480000000 03
MCL .11192516 01
                                                                                                                                                                                         2 .30322890 05
RA .28115954 02
LON .27500634 03
ZS .48063660 08
ZH .10548403 05
ZT .10548403 05
ZT .36036037 06
LOS .16588743 02
DR .10552863 01
TCL .11192516 01
                                                                                                                                                                                                                                  GEOCENTRIC CONIC
                                                                                                                                                                                                       235666450333202544400000 J.D.= 2438605.22424520 JULY 28,1964 17 22 54,785
B .53607772 05 S.R .11656695 05 APQ .48718128 06 RCA .58988163 04
C1 .68163673 05 IFP .14982521 06 TF .17084226 02 PER .22304432 05
SEA .99441039 02 NA .44273647 02 C3J -20471674 01 TFI .24533889 02
 EPOCH OF PERICENTER PASSAGE
SHA .24654005 05 ECC .97607359 00
VH .13991439 00 C3 -.16167784 01
TA .16934513 03 HA .18000000 03
                                                                                                                                                                                                                                                                                           ALL VECTORS REFERENCED TO EARTH EQUATOR PLANE

0X .81614400 00 DY .67379618 00 DL .22458298 00

MX .445905374 00 MY .75220646 00 ML .47271026 00

PX -.77702823 00 PY -.59959958 00 PL -.19159199 00

RX .15168230 00 RY .1704677 00 RL -.98147467 00

TX .61091702 00 TY .79169463 00 TL .00000000 00
          2 .30322890 05
APF .20329595 03
WZ .87481701 0D
QZ -.44495799 00
62 .44495801 00
          BTQ .47782174 05 BRQ -.24303436 05
                                                                                                                                                                                                           B .53607772 05 THA .33304076 03 T VECTOR IN EARTH EQUATOR PLANE
```

CASE 1 18SYS-JPIRAJ-SPACE 022265 3 RELICCENTRIC EQUATURIAL COORDINATES Z -.48033337 08 DX .24030557 02 DY .17125344 02
LDN .30980913 03 V .30412018 02 PTH -.39832041 00
ZE -.48063860 08 DXC .22214413 02 DTE .16451548 02
ZT -.48053111 08 DXT .222891571 02 DYT .17339725 02
LTT -.18429768 02 LDT .30983725 03 RST .15199850 09
SEP .38964786 02 EPM .13630553 03 EMP .31269390 02
SMP .37216690 02 SEM .11136139 03 EMS .68505077 02 DZ .73583793 01 AZ .75360793 02 DZE .71338363 01 DZT .75480396 01 VSI .29692226 02 MEP .12398535 02 ESH .13343921 00 1 DAYS 0 HRS. 36 HIM. 7.000 SEC. 23566656155020200000000 J.D.= 2438606.95111111 JULY 30:1964 11 04 00:000 GEOCENTRIC EQUATORIAL COORDINATES X .25103626 06 R .28626615 06 R .28626615 06 X -92018688 08 X -936441784 06 X -36441784 06 RS .15185950 09 EU .61325336 01 UT .35000000 02 CL .25920271 03 Y .13418778 06
DEC .60913377 01
LAT .50913377 01
YS .11083283 09
YM .10840647 06
YI .10840647 06
VS .29333513 02
ALT .27988819 06
DT .48000000 03
MCL .11199342 01 Z .30376764 05 DX .81519112 00 DY .67322571 00 DZ .22440404 00 RA .28126092 02 V .10807990 01 PTH .77275751 02 42 .61616519 02 LON .27401373 01 VE .20574902 02 PTE .29371153 01 AZE .27031557 03 ZS .48061947 08 DXS -.23213586 02 DYS -.16452559 02 DZS -.71342735 01 ZH .10647809 05 DXM -.32348484 00 DYT .88698269 00 DZM .41418345 00 ZT .10647809 05 DXM -.32348484 00 DYT .88698269 00 DZM .41418345 00 DXS -.33334944 05 VM .10309833 01 RT .3833494 05 VM .10309833 01 RT .3833494 05 VM .10309833 01 DS .15588715 02 RAS .12970107 03 RAM .1658608 02 LOM .262454525 03 DR .10842562 01 SHA .28277598 06 DES .18450759 02 DEH .16041929 01 TCL .11199342 01 GEOCENTRIC CONIC EPDCH OF PERICENTER PASSAGE
SMA .24655219 05 ECC .97608645 00
VM .1397292 00 C3 ~.16166988 01
TA .16935879 03 MTA .18000000 03 235666450335202006000000 J.D.= 2438605.22430609 JULY 28,1964 17 23 00.047
8 .53596202 05 SLR .11650884 05 APO .48720843 06 RLA .58959386 04
C1 .68147266 05 IFP .15005995 06 IF -17082765 02 PER .20305931 05
EA .99498608 02 RA .44339739 02 C3J -20471971 01 IFF .24600555 02 ALL VECTORS REFERENCED TO EARTH EQUATUR PLANE
DX .81519112 00 DY .67322571 00 DZ .2244040 00
RX .45922550 00 RY .75208422 00 RZ .47271713 100
PX -77705276 00 PY -59957162 00 PZ -19157992 00
RX .15167722 00 RY .7170370 00 RZ -98147708 00
TX -51088707 00 FY .7917175 00 FZ .00000000 00 X .25103626 06
INC .28978429 02
WX .14177917 00
QX .61325999 00
BX -61325984 00
OAP -.11045001 02 Y -13418778 06
LAY -17016156 02
HY --46327146 03
GY --65260538 00
BY -55260523 00
RAP -21765367 03 Z .30376764 05
APF .20329300 03
HZ .87480215 00
QZ -44499313 00
BZ .44499303 00 BTQ .47770919 05 BRQ -.24300045 05 B .53596202 05 THA .33303853 03 T VECTOR IN EARTH EQUATOR PLANE

CASE 1 18SYS-JPTRAJ-SPACE 022265

PAGE 1

64/07/29

STATION HUMBER 2

TIME	ĸ	DR	EL	AZ	DEC	HA	Cì	CC3	C3	RU
110000	-16755796 06	-15050596 01	47.636	136.455	1.705	332.326	1010789-13	1023069.11	1023069.11	001144252115
113000	-17028671 06	.15274765 OL	51.590	146.367	1.846	339.850	1010960.92	1023412.66	1023412.66	001156233643
128000	+27305866 06	_155290B2 D1	54.595	157.866	1.982	347.384	1011155.81	1023802.44	1023802.44	001170336233
123000	LE7587845 G6	.15804852 DI	56.389	170.809	2.113	354.924	2011367-16	1024225.10	1024225.10	001202572225
130000	·17874913 00	.16093108 OI	56.767	104.497	2.236	2-466	1011588-06	1024666.91	1024666,91	001215145375
133000	.18167214 06	.16384802 01	55.686	197.936	2.350	10.006	F611911-91	1025114.00	1025114.00	001227722501
140000	.15464729 06	.16670989 01	53.281	210.238	2.473	17.540	1012030-93	1025552.64	1025552.64	001242621237
143000	.18767281 06	.16943014 01	49.800	220.912	2-584	25-046	1012239.39	1025949.58	1023969.38	001255656163
150000	.19074539 06	.17192679 01	45.509	230-131	2.690 2.791	10.070	1015430-15	1020332.23	1020372.23	001271043153
153000	.19386032 06 .19701158 06	.17595327 01	36 253	244.64R	2.890	47.561	1012739.23	1026060.79	1026969.29	001317757503
160000 163000	.20019198 06	.17735484 OL	20.790	250.562	2.985	55.024	1012856.56	1027183.99	1027183.99	001333456279
170000	.20339335 06	.17827840 01	24-362	255.830	3.078					001347224263
173000	-20660669 06	. I 7868442 Ol	18_179	260.685	3-171	69.884	1012948-01	1027386.95	1027386.95	001363020561
180000	.20982235 OG	.17854687 01	12-248	265-250	3-261	77-261	1012936.66	1027364.27	1027364.27	001363020561
183000	-21303024 06	.17767053 01	6.368	269.662	3.392	84.596				001412403511
190000	.ZI622002 06	.17707772 01	.803	273.967	3.697	91.643	1012782.75	1027056.52	1027056.52	001426123757
				64/07/	30					
070000	.27016468 D6	.74415302 DC	1.327	84.522	5.232	267.906	1004988.50	1011455-10	1011468.10	001733233605
073000	-27150337 06	.73977832 00	7.101	88.762	5.109	275.090	1004928.05	1011347-15	1011347-15	75444100
000000	.27283360 06	.73864767 00	13-138	93-055	5.125	282.487	1004916-70	1011324.41	1011324.41	001745035411
00000	.27416693 06	.74363618 00	19.237	97.503	5.181	289.942	1004954.21	L011399.38	1011399.38	001751734177
090000	.27551468 06	.75475478 DO	25-283	102.226	5.234	297.453	1005039-15	1011569.20	1011569.20	001756665755
093000	-27688772 06	.77173927 00	31.236	107-375	5.289	304.964	1005169-18	1011829.23	1011829.23	001763676773
100000	.27829530 <b>0</b> 6	.79419018 00	37.030	113-131	5-344	312.493	1005341-17	1012173.16	1012173.16	001771012373
103000	40 \$8645615.	.02159878 00	42.574	119.735	5.398	320.037	1005551-19	1012593-16	1012593.16	001776252055
530000	.28125670 04	.85336002 OC	47.745	127.49B	5.450	327.594	1005794-57		1013079.91	002003655402
0000 1	42041 PDIY	TING TIMES LV=640702911.	A ((CTO						.,	
0000 t	4204 JETG	:CV=640/02411. :SMITTER 10 TA	0+64U+U.	3071.0.19	10. +00.				26	
0000 1		38516.	CE					• •		
0000 1	4204 [#26 4304 BEFI	THE CONTROL						13	27	
0000 1										
0000 1	4204 END	DATA						1	lo	
	STATEON HUM	JER 1		64/07/	29					PAGE 1
11HE	STATEON HUM	DR DR	٤٤	64/07/ Al	29 DEC	на	C1	CC3	£3	PAGE 1
	R	DR		AŽ	DEC 1.709	332.381	1010789.32	1023069.48	1023069.48	RU 001144227211
110000			47.174	AŽ 136.409	DEC 1.709	332.381	1010789.32	1023069.48	1023069.48	RU 001144227211 001154211137
110000	R -16754786 06 -17027675 06 -17304902 06	DR -15050839 01 -15276002 01	47.774	AZ 136.409 146.322	DEC 1.709 1.851	332.381 339.905	1010789.32 1010961.87	1023069.48 1023414.55 1023805.83	1023069.48 1023414.55 1023805.83	RU 901144227211 901154211137 901170314205
110000 113000 120000 123000	R -16754786 06 -17027675 04 -17304902 06 -17586929 06	DR -15050839 01 -15276002 01 -15531298 01 -15808013 01	47.774 51.738 54.750 56.542	Až 136.409 146.322 157.878 170.873	DEC 1.709 1.051 1.986 2.117	332.381 339.905 347.439 354.980	1010789.32 1010961-87 1011157-52 1011369-58	1023069.48 1023414.55 1023805.83 1024229.95	1023069.48 1023414.55 1023805.83 1024229.95	RU 001144227211 001154211137 001170314205 001202551115
110000 113000 120000 123000 130000	R -16754786 06 -17027675 04 -17304902 06 -17586929 06	DR -15050839 01 -15276002 01 -15531298 01 -15608013 01	47.774 51.738 54.750 56.542 56.911	AZ 136.409 146.322 157.878 170.873 184.619	DEC 1.709 1.051 1.986 2.117 2.242	332.381 339.905 347.439 354.980 2.522	1010789.32 1010961.87 1011157.52 1011369.58	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13	RU 001144227211 001154211137 001170314205 001202551115 001215145445
110000 113000 120000 123000 130000	R -16754786 06 -17027675 06 -17304902 06 -17586929 06 -17874062 06 -18166443 06	DR -15050839 01 -15276002 01 -15531298 01 -15808013 01 -16097161 01	. 47.774 51.738 54.750 56.542 56.911 55.813	AL 136.409 146.322 157.678 170.873 184.619 198.099	DEC 1.709 1.051 1.986 2.117 2.242 2.362	332.381 339.905 347.439 354.980 2.522	1010789.32 1010961.87 1011157.52 1011369.58 1011591.17	1023069.48 1023414.56 1023805.83 1024229.95 1024673.13	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13	RU 001144227211 001154211137 001170314205 001202551115 001215145445
110000 113000 120000 123000 130000 130000	R -16754786 06 -17027675 06 -17308902 06 -17386929 06 -17874062 06 -18166443 D6 -18464053 D6	DR -15050839 01 -15276002 01 -15531298 01 -15808013 01 -16097161 01 -16389676 01	47.774 51.738 54.750 56.542 56.911 55.813	AZ 136.409 146.322 157.878 170.873 184.619 198.099 210.424	DEC 1.709 1.051 1.986 2.117 2.242 2.362 2.477	332.381 339.905 347.439 354.980 2.522	1010789.32 1010961.87 1011157.52 1011369.58 1011591.17	1023069.48 1023414.56 1023805.83 1024229.95 1024673.13	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13	RU 001144227211 001154211137 001170314205 001202551115 001215145445
110000 113000 120000 123000 130000 140000 140000	R -10754786 06 -17027675 06 -17304902 06 -17586929 06 -17874062 06 -18166443 06 -18464053 06 -18464053 06	DR -15050839 01 -15276002 01 -15531298 01 -15608013 01 -16097161 01 -16389676 01 -16676600 01 -16949262 01	47.774 51.738 54.750 56.542 56.911 55.813 53.386	AL 136.409 146.322 157.878 170.873 184.619 198.099 210.424 221.162	DEC 1.709 1.051 1.986 2.117 2.242 2.362 2.477 2.587	332.381 339.905 347.439 354.980 2.522 10.062 17.597	1010789.32 1010961.87 1011157.52 1011369.58 1011591.17 1011615.34 1012035.23	1023069.48 1023414.5 1023805.83 1024229.95 1024673.13 1025121.47 1025561.24	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13 1025121.67 1025561.24 1025979.16	RU 001144227211 001156211137 001170314205 001202551115 001227704147 001242404517 00125643447
110000 113000 120000 123000 130000 133000 140000 143000	R -16754786 06 -11027675 06 -117304902 06 -17364902 06 -17874062 06 -18166443 D6 -18464053 D6 -18766711 D6 -19074087 06	DR -15050839 01 -15276002 01 -15531298 01 -156097161 01 -16389676 01 -16676600 01 -16949252 01	. 47.774 51.738 54.750 56.542 56.911 55.813 53.388 53.388	AZ 136.409 146.322 157.878 170.873 184.619 198.099 210.424 221-162 230.310	DEC 1.709 1.051 1.986 2.117 2.247 2.362 2.477 2.587 2.693	332.381 339.905 347.439 354.980 2.522 10.062 17.597 25.123 32.637	1010789.32 1010961.87 1011157.52 1011369.58 1011591.17 1011815.34 1012235.23 1012246.18 1012435.91	1023069.48 1023414.56 1023805.83 1024229.95 1024673.13 1025121.47 1025561.24 1025979.16	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13 1025121.47 1025561.24 1025979.16	RU 0011442272L1 00115421137 001170314205 001202551115 001215145445 001227704147 001242404517 001255643469 001271032575
11000 113000 12000 12000 13000 13000 14000 14000 15000 15300	R -16754786 06 -17027675 06 -17304902 06 -17364929 06 -17874062 06 -18166443 06 -1846453 06 -1846673 06 -18766711 06 -19074087 06	DR -15050839 01 -15276002 01 -15531298 01 -15808013 01 -1607161 01 -16389676 01 -16676600 01 -16049262 01 -177199454 01	47,174 51,138 54,150 56,542 56,911 55,813 53,386 49,981 49,588	AL 136.409 146.322 157.678 170.873 184.619 198.099 210.424 221.162 230.310 236.093	DEC 1.709 1.051 1.986 2.117 2.242 2.362 2.477 2.587 2.693 2.795	332.381 339.905 347.439 354.980 2.522 10.062 17.597 25.123 32.637 40.136	1910789.32 1910961-87 1911157-52 19111591-58 1911591-17 1911615-34 1912244-18 1912435-91 1912435-91	1023069.48 1023414-55 1023805-83 1024229-95 1024673-13 1025121-47 1025561-24 1025979-16 1026362-62	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13 1025121.47 1025979.16 1026362.62	RU 0011442272L1 00115421.137 00170314205 00120255115145445 001227704167 001227404517 001255643449 0012710255053469
11000 11300 12000 12000 13000 13000 13000 14000 14300 15000 15300	R -16754786 06 -17027675 06 -17304902 06 -17358929 06 -17874062 06 -18166443 06 -18464053 06 -78766711 06 -19074087 06 -19074087 06 -1910993 06	DR -15050839 01 -15276002 01 -15531298 01 -155808013 01 -16097[61 01 -16389676 01 -16476600 01 -16476600 01 -17199454 01 -177419581 01	47.174 51.738 54.750 56.542 56.911 55.813 53.386 49.891 45.568 40.674	AL 136.409 146.322 157.878 170.873 184.619 198.099 210.424 221.162 230.310 238.093 244.791	DEC 1.709 1.951 1.986 2.117 2.242 2.362 2.477 2.587 2.693 2.795 2.893	332.381 339.905 347.439 354.980 2.522 10.062 17.597 25.123 32.637 40.136	1010789.32 1010051.87 1011157.52 1011369.58 1011591.17 1011615.34 1012035.23 1012244.18 1012435.91 1012604.59	1023069.48 1023414-55 1023805-83 1024229-95 1024673-13 1025121-47 1025561-24 1025979-16 1026362-62 1026699-73	1023069.48 1023414.55 1023805.83 1024229.95 1024673.13 1025121.47 1025541.24 1025979.16 1026162.62 1026699.73	RU 0011442272L1 0011542L1.137 001170314250 001202551115 001215145445 001227704147 001242404517 00125564344) 001271032575 001304341251
110000 113000 120000 123000 130000 133000 140000 143000 153000 160000	R -16754786 06 -17027675 04 -17304902 06 -17364929 06 -17874062 08 -18166643 08 -1846673 06 -18766711 06 -19074087 06 -19174087 06 -191740910 06 -20019140 06	DR -15050839 01 -15276002 01 -15531298 01 -15508013 01 -16097161 01 -16389676 01 -16676600 01 -16749762 01 -17149744 01 -177437581 01	47.174 51.138 54.150 56.542 56.911 55.813 53.366 49.991 45.568 40.674 35.374	AL 136.409 146.322 157.878 170.873 184.619 198.099 210.424 221.162 230.310 236.093 244.791	DEC 1.709 1.051 1.986 2.117 2.242 2.362 2.367 2.587 2.693 2.795 2.988	332.381 339.905 347.439 354.980 2.522 10.062 25.123 32.637 40.136 47.618	1010789-32 1010961-87 1011157-52 10111591-58 1011591-17 1011615-34 1012235-23 1012245-91 1012604-59 1012744-59 1012745-24	1023069.48 1023414-55 1023805-83 1024229-95 1024673-13 1025121-47 1025561-24 10255979-16 1026699-78 1026980-73 1027195-68	1023069.48 1023414.55 1023405.83 1024229.95 1024673.13 1025121.47 1025541.24 10255979.16 10265979.8 1026599.98 10269697.73	RU 001144227211 001154211137 001170314205 001202551115 001215145445 00122770415 001255643449 001271032575 001304341251 001317753775
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